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CORONA J

PERFORMANCE EVALUATION REPORT

MISSION 1044-1 and 1044-2

FTV 1639, J-41

Approved

[REDACTED]

Manager

Advanced Projects

Approved

[REDACTED]

Manager

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In Accordance with E. O. 12958

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## INTRODUCTION

This report presents the final performance evaluation of Missions 1044-1 and 1044-2 of the Corona Program. The purpose of this report is to define the performance characteristics of the J-41 payload system and to identify the source of in-flight anomalies.

The performance evaluation was jointly conducted by representatives of Lockheed Missiles and Space Company (LMSC) and ITEK at the facilities of NPIC and AFSPPF. The off-line evaluation using Corona engineering photography acquired over the United States was performed at the individual contractors plants.

The quantitative data used for this report is obtained from government organizations. The diffuse density data, and MTF/AIM resolution are produced by AFSPPF. The vehicle attitude error values, frame correlation times are made at NPIC who also supply the Processing Summary reports published by [REDACTED]

Computer programs developed by A/P are utilized to calculate and plot the frequency distribution of the various contributors to image smear to permit analysis and correlation of the conditions of photography to the information content and quality of the acquired pictures. Computer analysis of the exposure, processing and illumination data provides the necessary data to analyze the exposure criteria selected for the mission.

This report contains certain data summarized from [REDACTED] Processing Summary, [REDACTED] and from AFSPPF TERO Report [REDACTED]

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SECTION 1  
SYSTEM PERFORMANCE

A. MISSION OBJECTIVES

The payload section of Mission 1044, placed into orbit by Flight Test Vehicle #1639 and THORAD Booster #513, consisted of two panoramic cameras, two Stellar-Index cameras, two Mark 5A recovery capsules and a space structure to enclose the cameras and provide mounting surfaces for all equipment. Figure 1-1 presents an inboard profile of the J-41 payload system. This Corona "J" system is designed to acquire search and reconnaissance photography of selected areas of the earth from orbital altitudes. A seven day -1 mission and a seven day -2 mission was planned.

B. MISSION DESCRIPTION

The payload was launched from Vandenberg Air Force Base (VAFB) at 2131:19Z (1331:19 PST) on 2 November 1967. Ascent and injection were normal and the achieved orbit was within nominal tolerances. Tracking and command support was effected by the Air Force Satellite Control Facility consisting of tracking and command stations at [REDACTED] [REDACTED] under central control of the Satellite Test Center at Sunnyvale, California. Mission 1044-1 consisted of a 6 day operation and was completed by air recovery on 8 November 1967. Mission 1044-2 was completed with an air recovery on 11 November 1967 following a 3 day photographic operation. The very short -2 mission was precipitated by a potential failure in the lifeboat system.

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The comparison of the planned and actual orbit parameters is tabulated as follows:

ORBITAL PARAMETERS		
<u>Parameter</u>	<u>Predicted</u>	<u>Orbit 110 Actuals</u>
Period (Min.)	90.47	90.333
Perigee (N.M.)	99.876	99.531
Apogee (N.M.)	223.750	219.940
Inclination (Deg.)	81.5	81.539
Perigee Latitude (Deg. N.)	19.24	33.816
Eccentricity	0.017191	0.01673

A single OAS rocket was fired on Rev 4, Rev 17, and Rev 113. These rocket firings produced the following results:

OAS ROCKET PERFORMANCE

OAS Rocket	Velocity Gained
#1	+ 16.2 FPS
#2	+ 15.7 FPS
#3	+ 17.5 FPS

C. PANORAMIC CAMERAS

Both instruments operated satisfactorily throughout both missions, and produced good image quality except for minor bands of image smearing near the takeup and on many frames. The imagery was very sharp, and verified the validity of the new focus settings.

SCHEMATIC INBOARD PROFILE - CORONA J-41 SYSTEM

MISSION 10/14

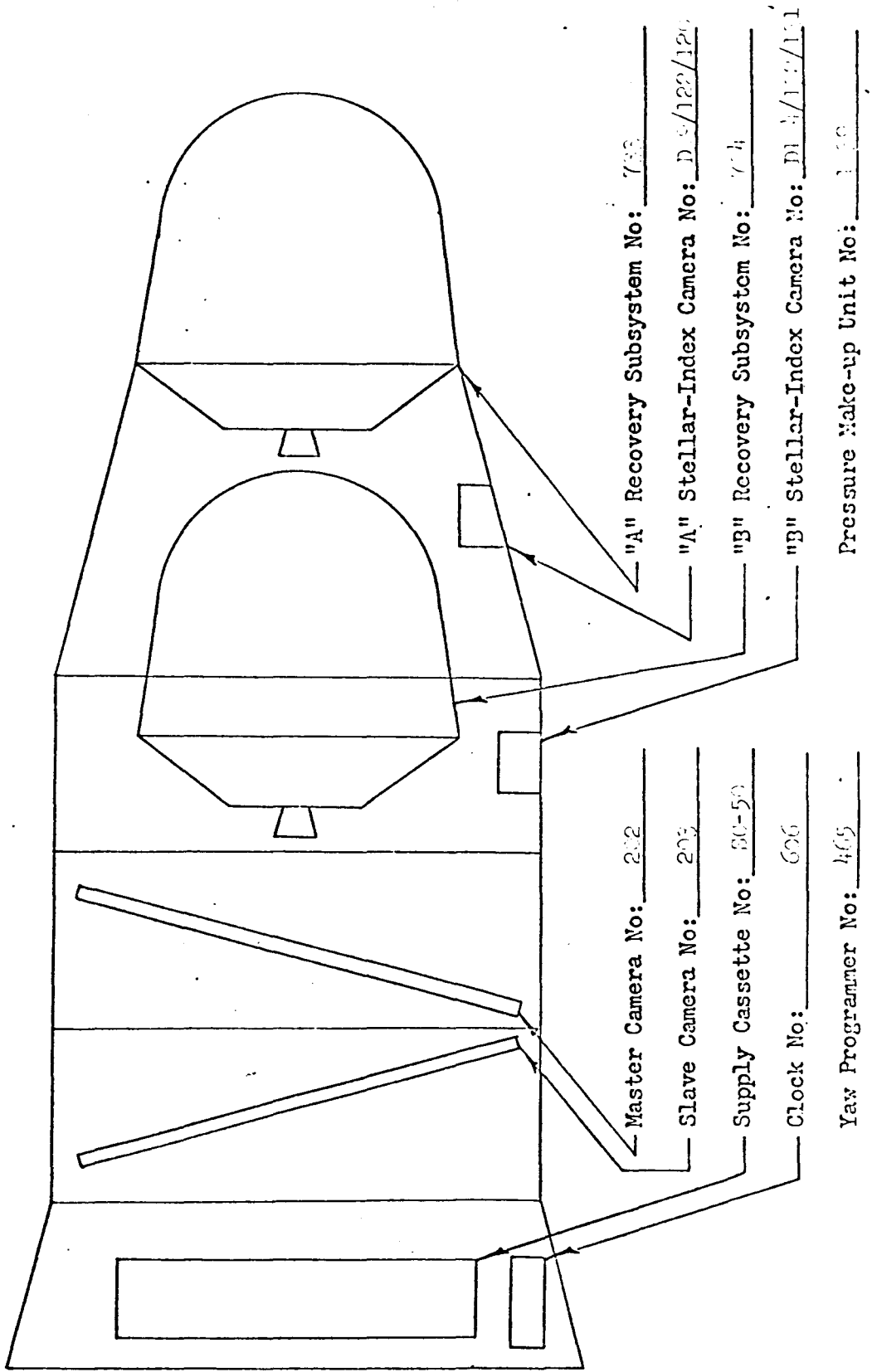
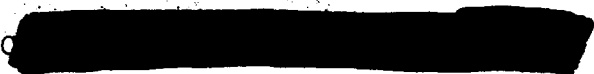


FIGURE 1-1





D. STELLAR-INDEX CAMERAS

Both the "A" and "B" S/I's operated satisfactorily and most Stellar images appear as points rather than the usual odd shaped stars.

E. OTHER SUB-SYSTEMS

The clock, instrumentation, pressure make-up, command and thermal control subsystems performed satisfactorily.

F. COMPONENT IDENTIFICATIONS AND SETTINGS

1. MASTER PANORAMIC CAMERA

a. COMPONENT ASSIGNMENT

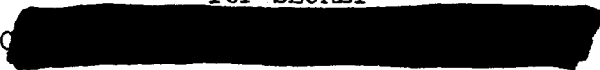
<u>Component</u>	<u>Serial Number</u>
Main Camera	202
Main Camera Lens	2042435
Supply Horizon Camera	308-G6
Supply Horizon Camera Lens	12889
Take-up Horizon Camera	318-G5
Take-up Horizon Camera Lens	12886
Supply Cassette	SC-50

b. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24"f/3.5
Slit Width	0.225"
Filter Type	Wratten 23A
Film Type	Eastman Type 3404





Supply (Port) Horizon Camera:

Lens	55 mm f/6.3
Aperture Setting	f/6.3
Exposure Time	1/100 second
Filter Type	Wratten 25

Take-up (Starboard) Horizon Camera:

Lens	55 mm f/6.3
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

2. SLAVE PANORAMIC CAMERA

a. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Main Camera	203
Main Camera Lens	2162435
Supply Horizon Camera	299-G6
Supply Horizon Camera Lens	12903
Take-up Horizon Camera	297-G5
Take-up Horizon Camera Lens	12883
Supply Cassette	SC-50

b. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5
Slit Width	0.175"
Filter Type	Wratten 21
Film Type	Eastman Type 3404





Supply (Starboard) Horizon Camera:

Lens	55 mm f/6.3
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

Take-up (Port) Horizon Camera:

Lens	55 mm f/6.3
Aperture Setting	f/6.3
Exposure Time	1/100 second
Filter Type	Wratten 25

3. MISSION 1044-1 STELLAR-INDEX CAMERA

a. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D-99
Index Reseau	122
Stellar Reseau	120

b. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85 mm f/1.8
Exposure Time	1 second
Filter Type	None
Film Type	Eastman Type 3401

Index Camera:

Lens	38 mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 3400





4. MISSION 1044-2 STELLAR-INDEX CAMERA

a. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D-104
Index Reseau	132
Stellar Reseau	131

b. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85 mm f/1.8
Exposure Time	1 second
Filter Type	None
Film Type	Eastman Type 3401

Index Camera:

Lens	38 mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 3400

SECTION 2

PRE-FLIGHT SYSTEMS TESTS

As a standard procedure, the J payload systems are subjected to a series of tests which demonstrates a satisfactory level of confidence that the systems will indeed perform as required in their respective missions. The tests include and operational-type exposure to simulate thermal/altitude environment, a light-leak evaluation, and a dynamic measure of the photographic performance capabilities. Significant baseline levels and anomalies experienced with this system during the pre-flight testing are as follows:

A. ENVIRONMENTAL TEST

The J-41 payload system was subjected to an environmental HIVOS Chamber test from August 15 through August 19, 1966, and from August 27 through September 1, 1966. The interruption was caused by the actuation of a camera failsafe control during the cut and wrap sequence.

Except for some minor acceptable corona marking, the panoramic instruments performed satisfactorily. The Master camera failsafe was activated at the cut and wrap sequence when the film jammed in the felt seal preventing take-up into the "B" SRV. The failure was attributed to a combination of a misaligned "B" take-up cassette, and the vertical attitude of the camera system during the sequence which permitted the cut film to fall back and jam in the felt door opening.

The clock accuracy was satisfactory, except for one correlation that was outside of the accepted tolerance range.

The pressure make-up system operated normally. During PMU operate, internal pressure increased to 37-39 microns. Gas consumption was as high as 7.45 lbs/min. during -2 portion of the test.

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The command system functioned properly for both bucket tests with no evidence of any equipment malfunctions.

B. RESOLUTION TEST

Initial resolution and theodolite tests were performed on 20 September 1966. Results of the thru-focus resolution tests of pan instruments 202 and 203 show the following characteristics:

Master Pan Instrument No. 202

Maximum high contrast resolution 175 lines/mm at -0.002 focal position.

Maximum low contrast resolution 115 lines/mm at -0.002 focal position

Slave Instrument No. 203

Maximum high contrast resolution 176 lines/mm at +0.001 focal position.

Maximum low contrast resolution 112 lines/mm at +0.001 focal position.

Additional Boston investigations indicated that optimum focus position would be attained by adding 0.002" shim to the scan head of the Slave instrument, and 0.001" shim to the Master instrument. The modified instruments were retested 22 October 1967, with the following results:

Master Pan Instrument No. 202

Maximum high contrast resolution 183 lines/mm at -0.0025 focal position.

Maximum low contrast resolution 120 lines/mm at -0.0020 focal position.

Slave Pan Instrument No. 203

Maximum high contrast resolution 185 lines/mm at -0.0015 focal position.

Maximum low contrast resolution 118 lines/mm at -0.0020 focal position.

The final test data for both instruments is shown in Figures 2-1 and 2-2. Both instruments met the system requirements specification.

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Camera No: 202

Payload No: J-41

Resolution (l/mm) 183

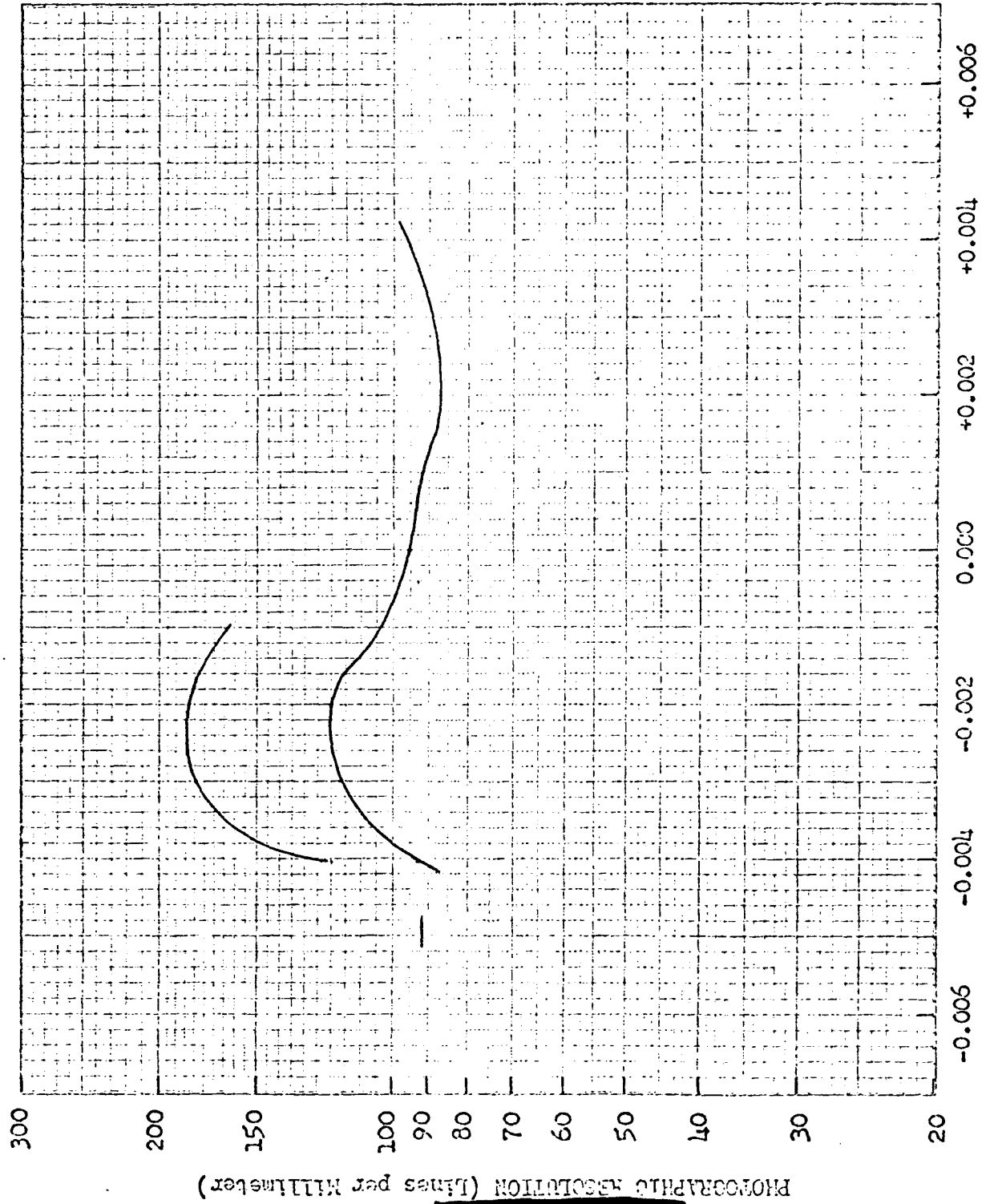
High Contrast: 183

Low Contrast: 120

Film Type: 3404

Test Date: 22 Oct. 1967

PRE-FLIGHT DYNAMIC RESOLUTION



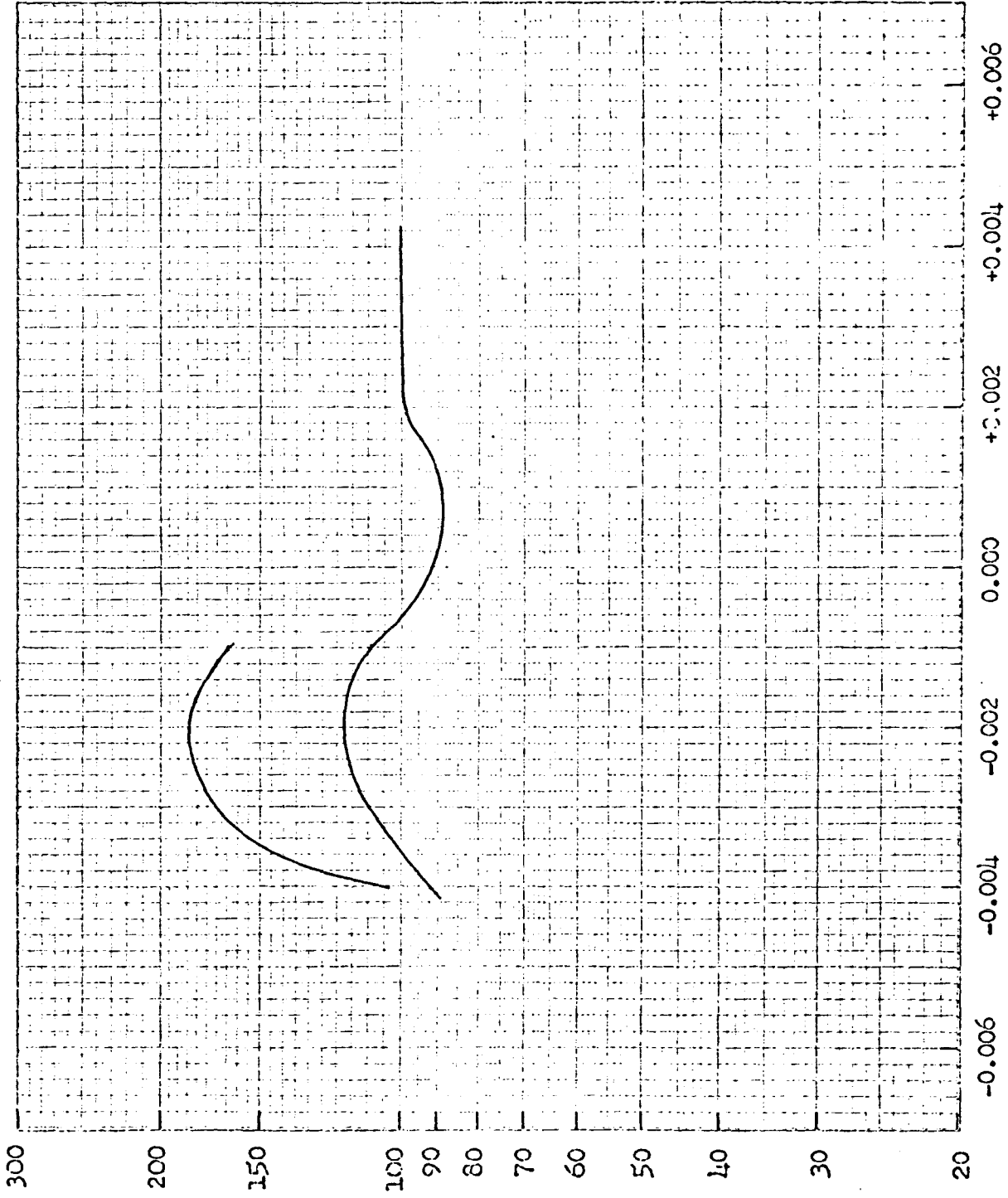
THROUGH FOCUS INCREMENTS (Inches)

FIGURE 2-1

PHOTOGRAPHIC RESOLUTION (Lines per Millimeter)

[REDACTED]

PRE-FLIGHT DYNAMIC RESOLUTION



THROUGH FOCUS INCREMENTS (Inches)

FIGURE 2-2

Camera No: 203  
Payload No: J-41  
Resolution (1/mm) 185  
High Contrast: 118  
Low Contrast: 118  
Film Type: 3404  
Test Date: 22 Oct. 1967





C. LIGHT LEAK TEST

The J-41 system was tested for light leaks on 12 August 1966, revealing major leaks at three of the four H.O. boot installations and at the Agena interface cover. Photomultiplier sensing techniques were used to verify the validity of repairs made.

D. FLIGHT LOADING AND CERTIFICATION

Loading of flight film was accomplished on 24 October 1967, and final pre-flight acceptance tests performed 25 October 1967. All functions were nominal, with no indications of light leaks or other sources of performance degradation.





SECTION 3

FLIGHT OPERATIONS

A. SUMMARY

Ascent through Agena ignition was nominal. The Agena engine "coughed" at approximately 128 seconds after Agena ignition and combustion chamber pressure was reduced during the last period of engine burn. The engine burned approximately 5 seconds longer than nominal. A hard engine shut-down was confirmed.

The achieved orbit parameters were low, but were within the three (3) sigma dispersions.

Both panoramic cameras operated satisfactorily throughout the flight.

Both Stellar/Index cameras operated satisfactorily throughout the flight.

The instrumentation system, clock system, and the yaw function generator performed normally for the duration of the flight.

An intermittent anomaly in the Lifeboat system developed in the -2 mission, with the possible initiation of an unplanned recovery sequence. As a result, the mission was intentionally terminated as early as possible.

Several commanding problems were encountered during this flight while commanding in the repetitive mode.

B. PANORAMIC CAMERA PERFORMANCE

Both panoramic cameras operated normally throughout the flight. Camera system dynamic operation, 99/101 clutch operation, start-up, shut-down, and film transport functions were normal on all monitored passes.

The cut and wrap operation and transfer to the -2 system occurred as programmed utilizing the KIK-ZORRO 38 command (early A to B switchover) on Rev 88.





The panoramic film was exhausted on Rev 140 frame No. 25 and frame No. 60 on the Master and Slave cameras respectively.

Panoramic Film Consumption

	<u>Actual Frames</u>	
	<u>Master</u>	<u>Slave</u>
Pre-Launch	137	137
-1 Mission	2898	2880
-2 Mission	3011	3030
Total	6046	6047

FMC Match

The V/H ramp to orbit match was acceptable throughout the flight. The following settings of RTC 6, 8, and 10 were utilized to obtain the optimum FMC match during the flight.

	<u>RTC Commands</u>			<u>REMARKS</u>
	<u>6</u>	<u>8</u>	<u>10</u>	
RTC Positions	7	4	6	Launch thru Rev 3
	7	2	9	Rev 4 thru Rev 12
	6	3	9	Rev 13 thru Rev 16
	7	3	9	Rev 17 thru Rev 44
	8	2	8	Rev 45 thru Rev 65
	7	3	9	Rev 66 thru Rev 75
	8	2	9	Rev 76 thru Rev 91
	7	3	9	Rev 92 thru Rev 114
	7	3	10	Rev 115 thru the end of the mission

However, the design of the 1000-series ramp programmer limits this optimum FMC match to a nominal band of latitude defining areas of primary interest. The extensive operations over a wide range of latitude experienced in this mission (Ref. Figs. 5-5 to 5-10) increases the statistical deviation, as is evident in Figures 3-1 through 3-4.



Y V/H RATIO ERROR - PERCENT (Y) VERSUS FREQUENCY - PERCENT (X)

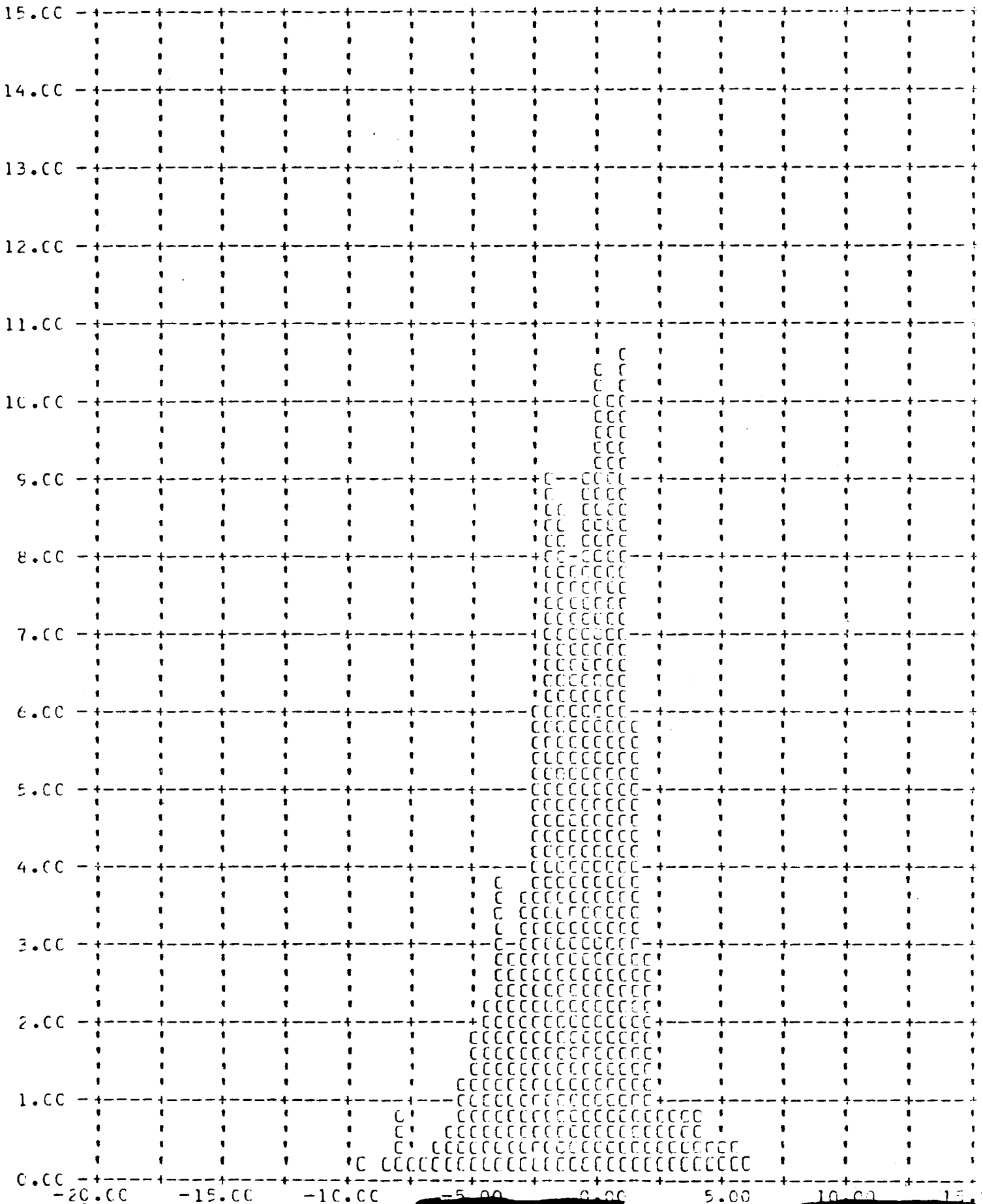
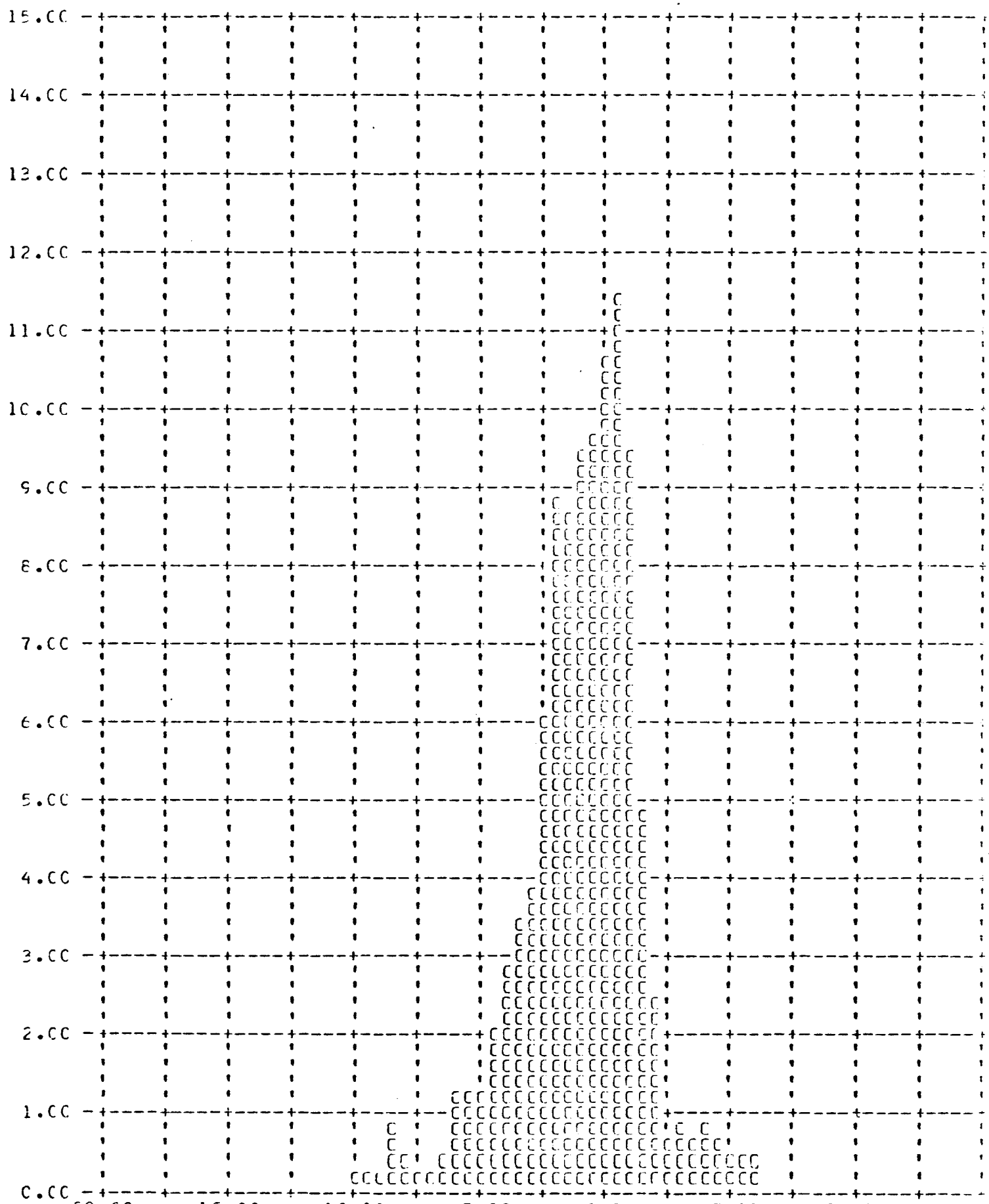
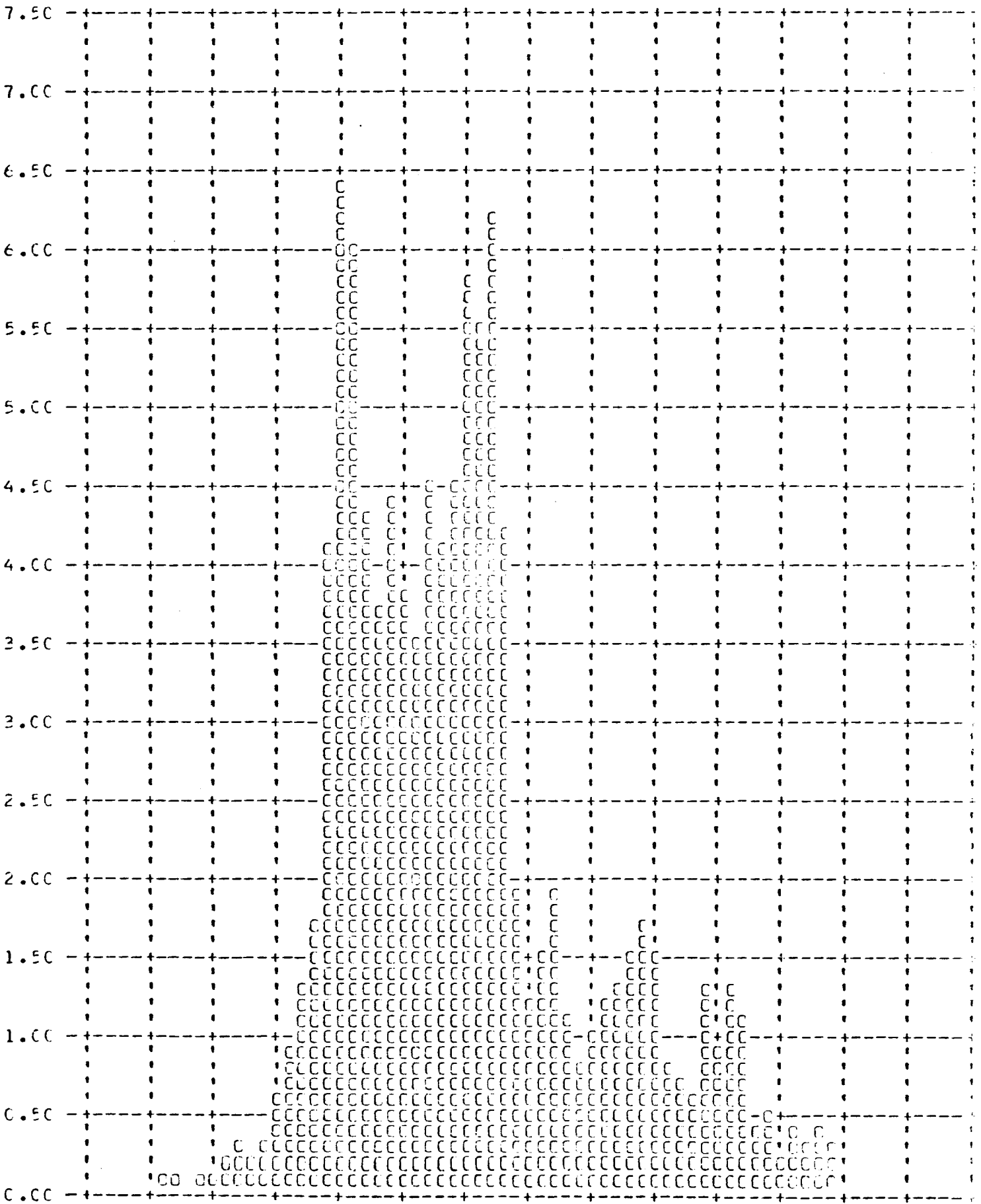


Figure 3-1

Y V/F RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



Y V/F RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

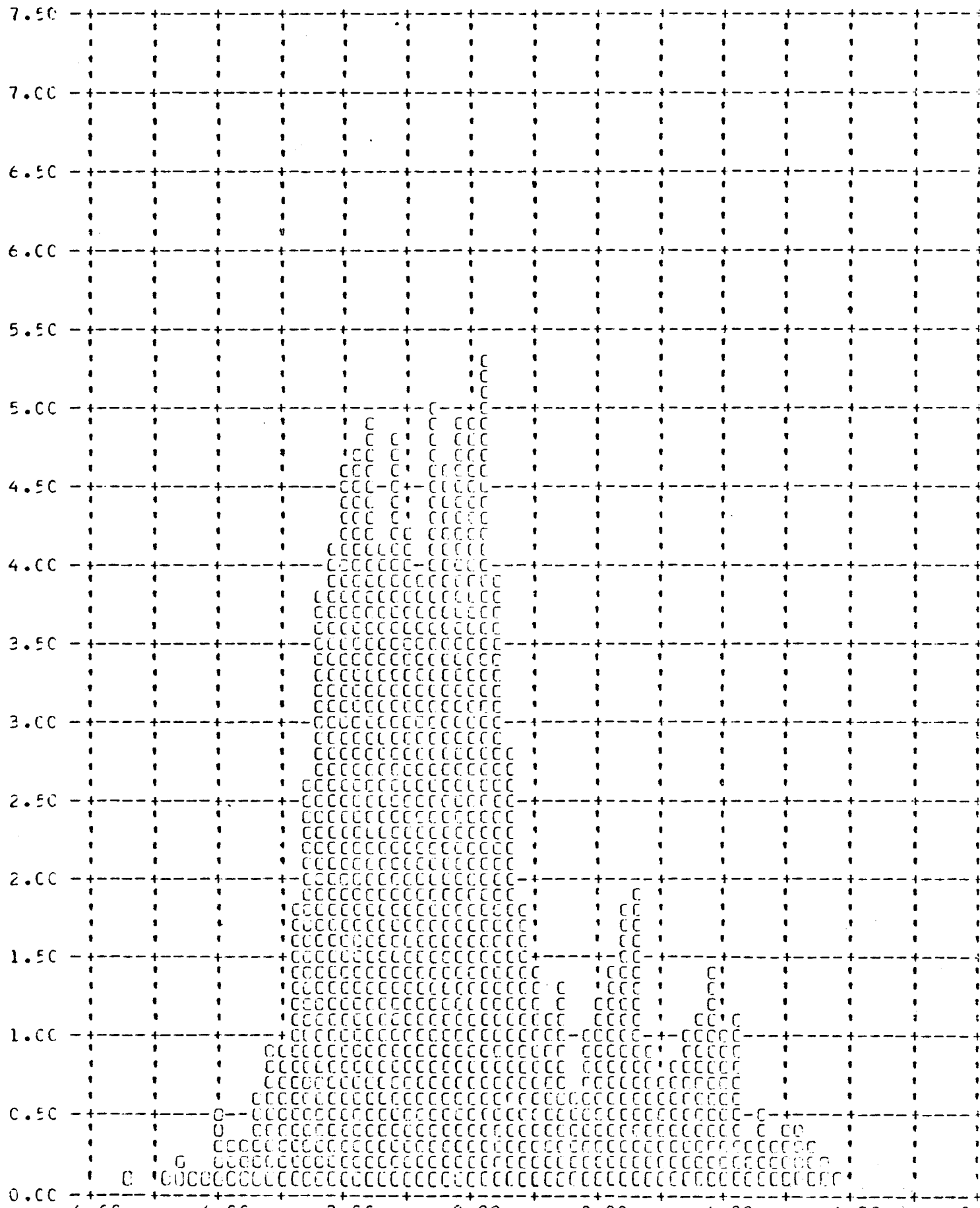


Figure 3-4

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C. STELLAR/INDEX CAMERA PERFORMANCE

Both the -1 and -2 Stellar/Index cameras operated satisfactorily on all monitored engineering passes. Telemetry data indicated the programmer, metering functions, and shutter monitors performed satisfactorily.

D. INSTRUMENTATION AND COMMAND SYSTEM PERFORMANCE

The instrumentation system performed normally throughout the total mission.

The command system performance was satisfactory for both missions. However, numerous command anomalies were encountered during the mission when real time commanding (RTC) was performed in the repetitive mode.

Analysis indicated that the RTC commands issued at the [REDACTED] [REDACTED] showed erratic command duration times. It was recommended that the command generation equipment be verified for proper operation.

One RTC 9 was missed on Rev 56 [REDACTED] while transmitting in the repetitive mode. The A/P stepper also failed to advance on two correctional commands and a third command was required to place the A/P stepper in the proper position. Analysis indicates that the stepper failed to respond to the issued command. A specific cause of this anomaly could not be determined from the available data. However, this command box was checked prior to shipment for proper command response time to command durations of 65 milliseconds and the system functioned normally.

E. CLOCK SYSTEM PERFORMANCE

The clock system operation was normal for the entire mission. Satisfactory time correlation between the flight clock and the [REDACTED] was obtained. The ratio of clock time to system time was 1:00000026563.

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F. PRESSURE MAKE-UP SYSTEM PERFORMANCE

The pressure make-up system performance was normal for the duration of the mission. Average gas consumption was approximately 8.4 Δ PSI/min for the 240 minutes of total operate time. The system had a reserve of 20 PSI at the end of the flight.

G. THERMAL ENVIRONMENT

The thermal control pattern on this payload system was modified prior to launch to produce a thermal environment of  $75 \pm 10^{\circ}\text{F}$ .

Temperature data from the [REDACTED] acquisitions are included in Tables 3-1 and 3-2. The average instrument temperatures ranged from a high of 85<sup>o</sup>F. and 86<sup>o</sup>F. to low of 68<sup>o</sup>F. and 68<sup>o</sup>F. on the Master and Slave instruments respectively.

H. YAW PROGRAMMER

The vehicle Yaw Programming functioned properly throughout the mission. However, because of pre-flight programming error which placed the function start pulse approximately 800 seconds late, the Yaw attitude achieved was approximately 55 Degrees out of phase with the desired profile. A more complete description of this function and its effect on mission performance is presented in Sections 4, 7 and 8.

CHRYSLER CREDIT CORPORATION  
MEMPHIS - PARK

CREDITS ACQUIRED

STATUS

Account/Serial # (27)	5	16	24	32	40	47	56	63	72	79	88	95	103	111	120	127
1	59	32	36	39	29	32	49	25	25	25	49	25	24	24	25	24
2	16	-4	6	-4	0	0	16	-4	0	-4	16	-4	16	18	40	40
3	2	2	-2	-5	-12	2	6	2	-7	-1	2	2	75	62	12	72
4	54	40	43	41	41	45	54	44	33	41	54	48	51	51	25	54
5	77	64	64	55	55	55	74	59	51	51	71	51	40	46	23	57
6	79	75	57	60	50	69	72	60	47	57	69	54	--	--	--	--

Serial No. 2

1	55	77	52	74	46	60	52	69	43	59	55	59	40	43	23	33
2	72	95	52	31	43	47	62	78	43	76	68	75	55	45	33	57
3	67	65	61	67	59	55	73	79	55	62	76	65	70	50	34	76
4	64	43	54	45	48	51	67	45	51	43	61	51	37	42	45	42
5	61	57	54	51	45	54	61	57	48	51	57	54	33	26	22	28

Credit Acquired

1	75	70	55	65	57	63	70	60	50	54	66	57	50	37	54	31
2	142	95	55	95	50	80	91	62	79	76	65	79	75	47	43	42
3	160	60	54	50	65	65	102	74	63	60	69	64	34	72	60	45

Serial

1	83	81	71	77	75	81	85	78	72	80	81	79	77	63	63	61
2	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--

PLEASE GO TO "A" TO "B"

1 (Serial)	87	89	81	86	81	84	84	87	80	87	81	87	81	68	68	61
2 (Serial)	70	68	68	57	62	60	68	66	64	67	68	71	60	67	63	61

PLEASE GO TO "A" TO "B"

1	57	64	65	64	65	66	69	69	61	66	66	66	66	66	66	66
2	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--

PLEASE GO TO "A" TO "B"

1	74	61	67	60	69	70	65	69	64	64	64	64	64	64	64	64
2	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--

REAL TIME DATA  
TEMPERATURE SUMMARY  
DORRNEY - PAPER

ORBITS ACQUIRED

SENSOR

1	8	16	24	32	40	47	56	63	72	77	80	85	102	111	115	120	125	130
2	76	75	75	72	72	73	74	72	71	71	74	72	67	63	62	61	62	63
3	74	75	75	75	76	76	76	74	74	75	77	74	72	68	66	64	63	64
4	71	74	77	75	74	73	76	72	72	71	74	72	70	72	75	71	71	72
5	77	75	74	74	75	74	76	72	72	71	75	71	71	72	71	71	72	71
6	77	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
7	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
8	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
9	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
10	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
11	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
12	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
13	76	75	74	75	72	72	72	71	72	71	75	71	71	72	71	71	72	71
AVG. TEMP. TEMP.	75	72	74	71	72	71	72	70	71	70	71	70	70	69	69	69	69	69

ORBITS

1	81	86	89	89	87	84	83	82	83	81	84	81	77	77	84	81	79	83
2	80	82	83	85	85	82	81	81	81	82	83	83	83	83	82	80	83	81
3	81	81	85	81	83	81	84	83	81	80	83	80	75	75	75	75	80	81
4	81	81	80	79	77	79	77	77	75	77	77	77	72	72	72	72	72	72
5	80	83	83	85	84	83	84	83	82	82	83	81	77	77	77	74	74	77
6	82	83	85	82	81	81	84	81	80	80	84	81	77	77	75	75	75	75
7	82	79	82	83	81	80	82	77	81	79	82	79	75	75	75	75	75	75
8	84	77	79	76	74	77	82	77	76	76	82	80	70	81	82	82	82	81
9	82	84	85	83	81	83	82	82	83	81	83	81	75	77	77	77	77	77
10	82	78	77	76	77	75	77	77	76	76	77	77	70	77	77	77	77	77
11	83	83	84	81	79	82	81	81	81	81	85	83	76	82	82	82	82	82
AVG. TEMP. TEMP.	79	74	73	73	75	74	77	76	74	73	73	76	76	74	74	74	74	74

Summary Data

1	72	74	73	73	75	74	77	76	74	73	73	76	76	75	75	75	75	75
2	76	75	76	77	73	76	72	71	73	73	73	77	75	74	74	74	74	74

I. RECOVERY SYSTEM

An early switchover from the A to the B Recovery systems was performed on Pass 88, with all functions appearing normal. The 1044-1 recovery capsule was successfully recovered by air-catch on Rev 97 at 1608 PST on 9 November 1967. Capsule impact was approximately 50 N.M. south of the predicted impact. All available data has been analyzed and all functions appeared to have occurred normally. All re-entry events appeared normal and close to the predictions except for deceleration chute deployment which occurred 0.12 seconds late.

	<u>Latitude</u>	<u>Longitude</u>
Predicted	25° 56.4' N	165° 51.08' W
Actual	25° 06' N	165° 42' W

The intermittent failure of a relay in the "Lifeboat" system electronics during the 1044-2 mission resulted in several cases of inadvertent lifeboat timer starts as well as one case of power being applied to the primary recovery system. Because of the possibility of an uncontrolled recovery, the 1044-2 mission was terminated early. The 1044-2 recovery capsule was successfully recovered by air-catch on Rev 144 at 1509 PST on 11 November 1967. All re-entry events appeared normal and close to the predictions except for parachute cover off which occurred 1.97 seconds early. Capsule impact was close to nominal.

	<u>Latitude</u>	<u>Longitude</u>
Predicted	21° 0.56' N	154° 28.1' W
Actual	21° 5.0' N	154° 33.0' W

J. RADIATION DOSAGE

Each recovery system flown on a Corona mission contains a sealed packet of Eastman Type 3401 and Royal X Pan emulsions to determine the total radiation received at the take-up cassette. Both film types have been irradiated by IMSC at various levels and the base plus fog densities recorded after controlled processing.

Following recovery the film dosimeter packets are removed at A/P and processed with a pre-flight sample of the same film type and sensitometric control film. The resulting base plus fog density measurement of the dosimeter strips is used to ascertain the total radiation level. The table below presents the base plus fog readings for the dosimeter strips and the radiation level equivalents.

<u>Emulsion</u>	<u>Mission 1044-1</u>		<u>Mission 1044-2</u>	
	<u>B + F Density</u>	<u>Radiation</u>	<u>B + F Density</u>	<u>Radiation</u>
Type 3401	0.22	0.9 R	0.25	1.3 R
Royal X Pan	0.27	0.5 R	0.30	0.6 R

These levels are below that which will degrade the photography.

SECTION 4

PHOTOGRAPHIC PERFORMANCE

A minimum of payload system photographic anomalies occurred during Missions 1044-1 and 1044-2, thus providing one of the most trouble-free flights to date. The image sharpness attained was considered equal to any previous Corona J-1 photography, permitting most imagery to be viewed at 60 x magnification. The overall image quality was judged to be generally good where not degraded by atmospheric attenuation; however, there was a predominance of cloud cover over the highest priority targets.

A. PANORAMIC INSTRUMENTS

The Master Camera produced 2898 frames (8049 feet) of photography during Mission 1044-1, and 3011 frames (7951 feet) during Mission 1044-2. The Slave camera produced 2880 frames (8009 feet) during Mission 1044-1, and 3030 frames (7963 feet) during Mission 1044-2. The quality of the photography produced by the two cameras was very similar, and was rated comparable to Mission 1035. The MIP Frames were rated 85.

The array of fixed resolution targets at Holloman AFB, New Mexico, were recorded during Mission 1044-2. The average system resolution of these targets was judged to be approximately eight feet for both instruments.

C. 

Both instruments exhibited characteristic anomalies, most objectionable of which was an appreciable build-up of emulsion particles. This condition was apparently accentuated by the very long operate times commanded during Mission 1044-2 so as to facilitate an early recovery. However, there appeared to be no significant reduction in information content because of this condition.

All auxiliary data recording functions operated normally throughout the flight, with the exception of missing binary data blocks on six occasions randomly over the missions. (Four occasions were on the forward camera, two on the aft). In each instance all of the other auxiliary data was present. This behavior was observed in pre-flight altitude testing, but was not considered detrimental so corrective action was waived.

The quality of the photography was adequate to readily identify intermittent bands of smearing near the takeup end of format caused by film flutter as the scan head enters the photographic format area. This anomaly is characteristic of instrument operation, and should be reduced considerably with the CR concepts.

It must be noted that this system was the first of the 1000-series to have the revised focus settings for a more precise compensation of the vacuum focal shift characteristics of the lenses used. Although there are many factors influencing the photographic quality achieved, it is reasonable to assume that the desirable performance of Mission 1044 verifies the validity of the new peak focus positions.

C

B. STELLAR/INDEX CAMERAS

The Stellar/Index film recovered consisted of 449 frames of photography from each film path of S/I D99/122/120 (Mission 1044-1), and 464 frames from each path of S/I D104/132/131 (Mission 1044-2). The cameras operated normally throughout the respective mission. There were 15 to 30 or more stellar images detectable on most frames despite a level of flare which affected approximately 50 percent of each frame. Most of the stellar images were good, and were point-type images. There was an appearance of Corona static marking occurring intermittently throughout the Mission 1044-2 stellar record.

The index cameras produced good quality imagery through each of the respective missions. The reseaus were sharp and well defined in both instruments. Several instances of dendritic static were recorded on the preflight, postflight and the last eight frames of the Mission 1044-2 index film.

C. OBSERVED DATA

Detailed evaluation of the engineering materials available at A/P indicated that the smearing effects from the V/h and yaw steering errors (see Section 8) did indeed create a detectable limitation to system performance in many instances in the mission. As predicted in the smear analysis, frames obtained with a large yaw steering error show a distinct disparity in quality between the forward and aft photography directly related to the difference in their exposure times.



C

When the ground smear contributions drop below some apparent threshold value (estimated to be approximately five feet for this system) on both instruments, the resulting forward and aft photography becomes very comparable and very good in quality. The Holloman AFB targets photographed during Pass 126 indicated approximately eight feet ground resolution for both instruments with a calculated theoretical smear of about  $4\frac{1}{2}$  feet (which corresponds to a theoretical ground resolution limitation of  $6\frac{1}{2}$  feet). In comparison, Pass 63 photography had noticeable disparity between forward and aft performance corresponding to relatively high smear values for the forward looking imagery (approximately  $8\frac{1}{2}$  feet theoretical smear induced ground resolution limitation).

The mission processing summary indicated a major disparity in original negative development in several instances throughout the mission. Evaluation of engineering pass 125 indicated that the aft-looking record, which was processed at the primary development level, had a significant loss of detail and image quality when compared with the corresponding forward-looking photography which was processed at the full level. There was excellent cloud highlight definition in the aft photography, but the important ground and culture imagery was suppressed to the extent of a distinct loss in information content.

#### D. PERFORMANCE MEASUREMENTS

A summary of MTF/AIM resolution values measured by SPPF is tabulated below. The microdensitometer slit used was 1 micron by 80 microns.



<u>Mission</u>	<u>Camera</u>	<u>Cycles/mm</u>	<u>Avg</u>	<u>Ground Resolution</u>
1044-1	Fwd	78		
1044-2	Fwd	61	70	15'
1044-1	Aft	71		
1044-2	Aft*	84	78	13½'

\*Samples from portion processed by dual gamma method

The details of the measurement and computing techniques, targets measured and target locations are fully reported in the evaluation report published by AFSPPF and are not included in this report. These values were determined by using the "Interim MTF/AM Program" technique.

It should be noted that the value shown for 1044-2 Fwd camera includes one reading of only 32 cycles/mm. The reading may be accurate, but does not represent the nominal level of system performance. In comparison, visual resolution targets recorded ten passes after this reading location indicated an effective ground resolution of approximately eight feet, which corresponds to an MTF reading on the order of 130 cycles/mm.



SECTION 5

PANORAMIC CAMERA EXPOSURE

The Master camera contained a 0.200 inch slit and a Wratten 23A filter. The Slave camera had a 0.150 inch slit and a Wratten 21 filter. These conditions placed the nominal exposure between the full and the intermediate processing curve.

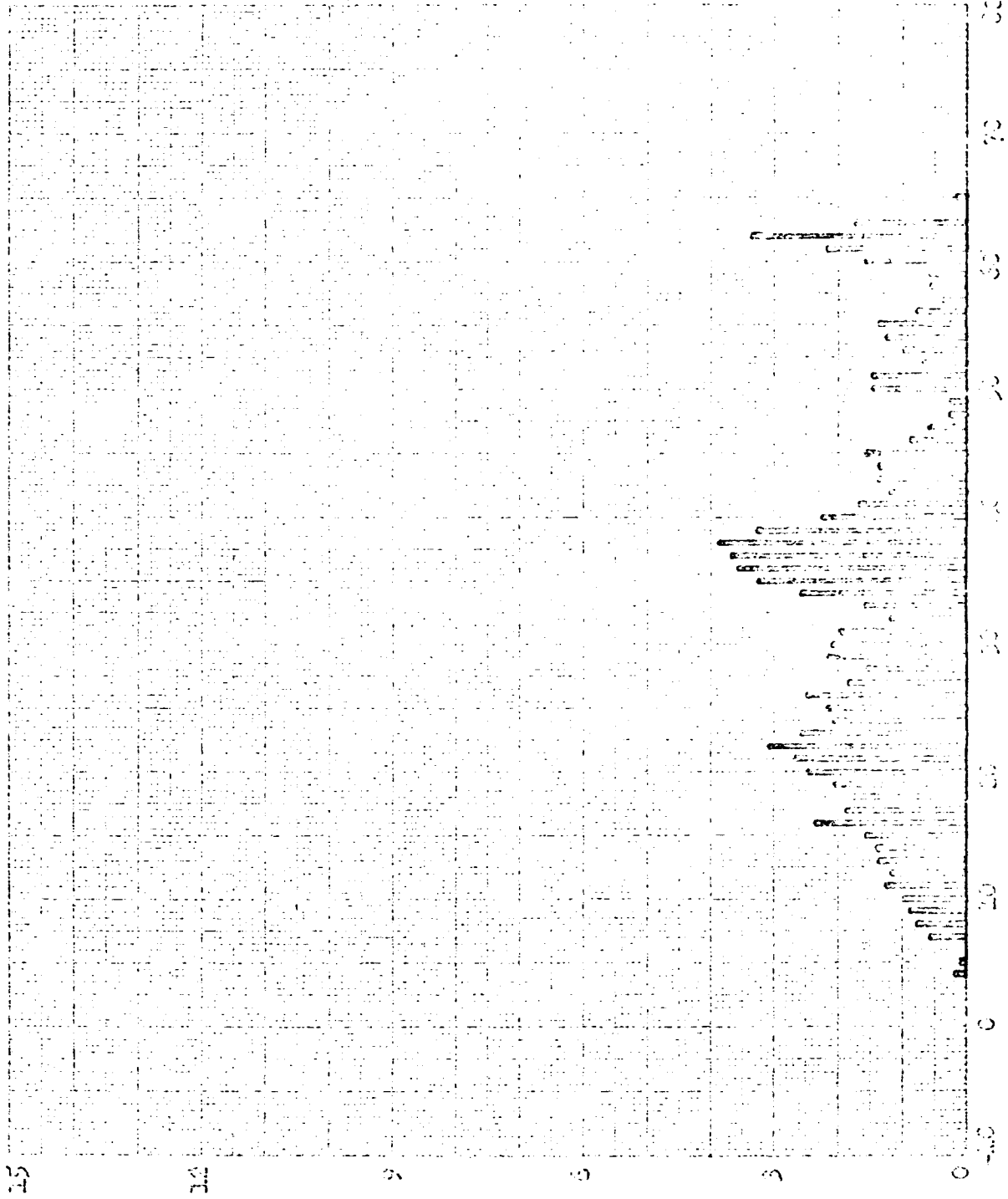
The frequency distributions of the solar elevations and solar azimuths encountered during the photographic operations are shown in Figures 5-1 to 5-4.

The nominal exposure times of the Master and Slave cameras are shown as a function of latitude for passes D-25, D-70, and D-116 in Figures 5-5 to 5-10. Superimposed on these plots are relative distributions of camera operations for the portion of the mission represented by each plot. These distributions became very uniform with latitude as the mission progressed because of the extended operations programmed in order to reduce mission duration.

~~TOP SECRET~~

C

2000 2000 2000 2000 2000 2000



Mission No: 1044-1  
Reference No: 1-41  
Contract No: 292  
Location: 11/2/57  
Station: 2131 Z  
Altitude: 81.50

1 2000 2000 2000 2000 2000

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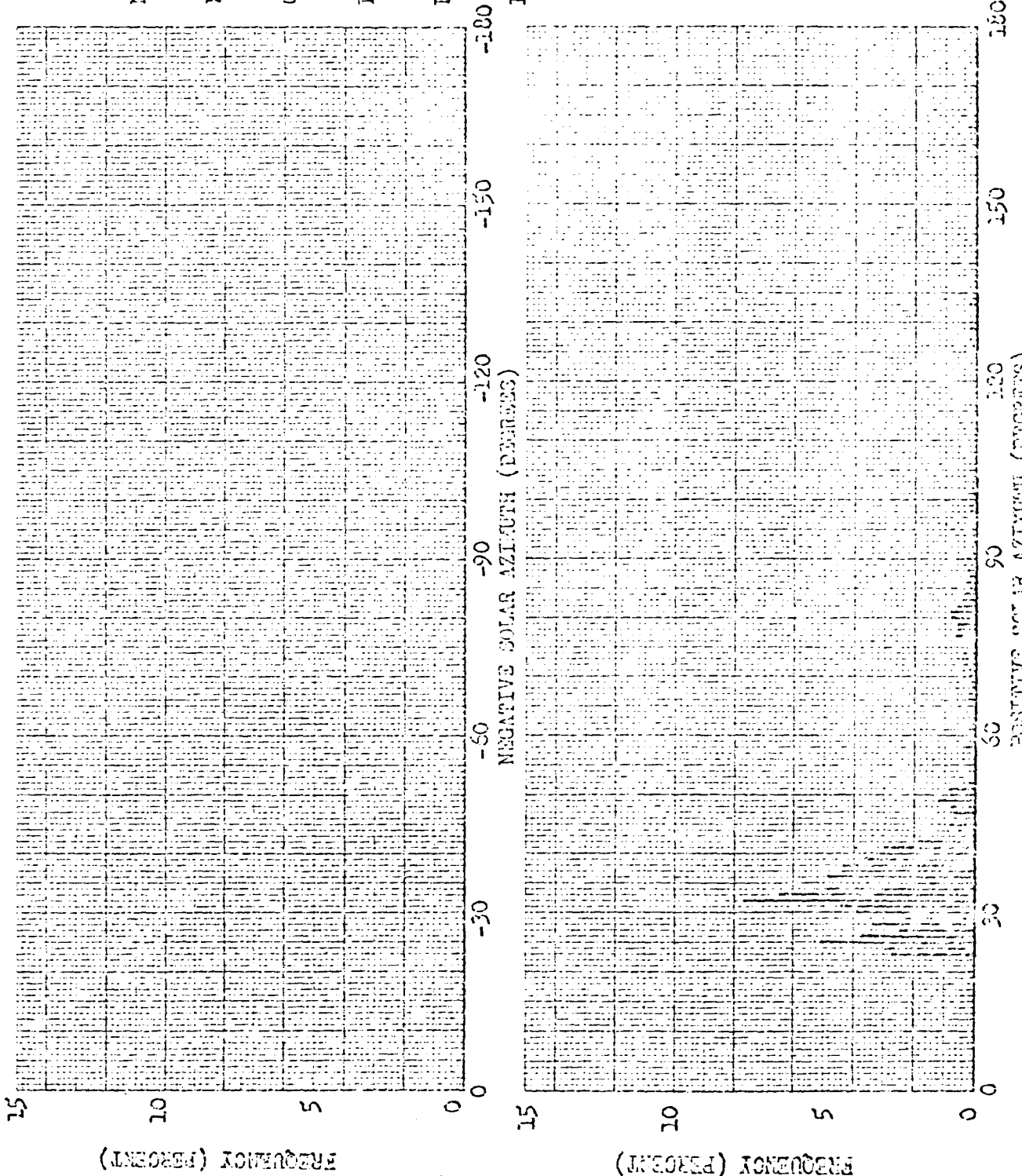
(UNCLASSIFIED)

FIGURE 5-1

TOP SECRET

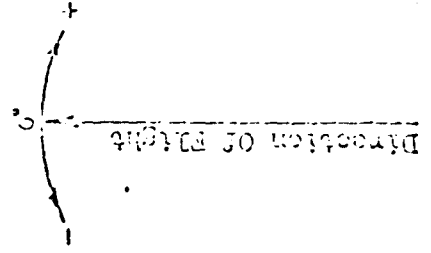
C/

SOLAR AZIMUTH FREQUENCY DISTRIBUTION



Mission No: 1044-1  
 Payload No: J-41  
 Camera No: 202  
 Launch Date: 11/2/67  
 Launch Time: 2131 Z  
 Inclination: 82.5°

SIGN NOTATION



TOP SECRET

[REDACTED]

Mission No: 1044-2

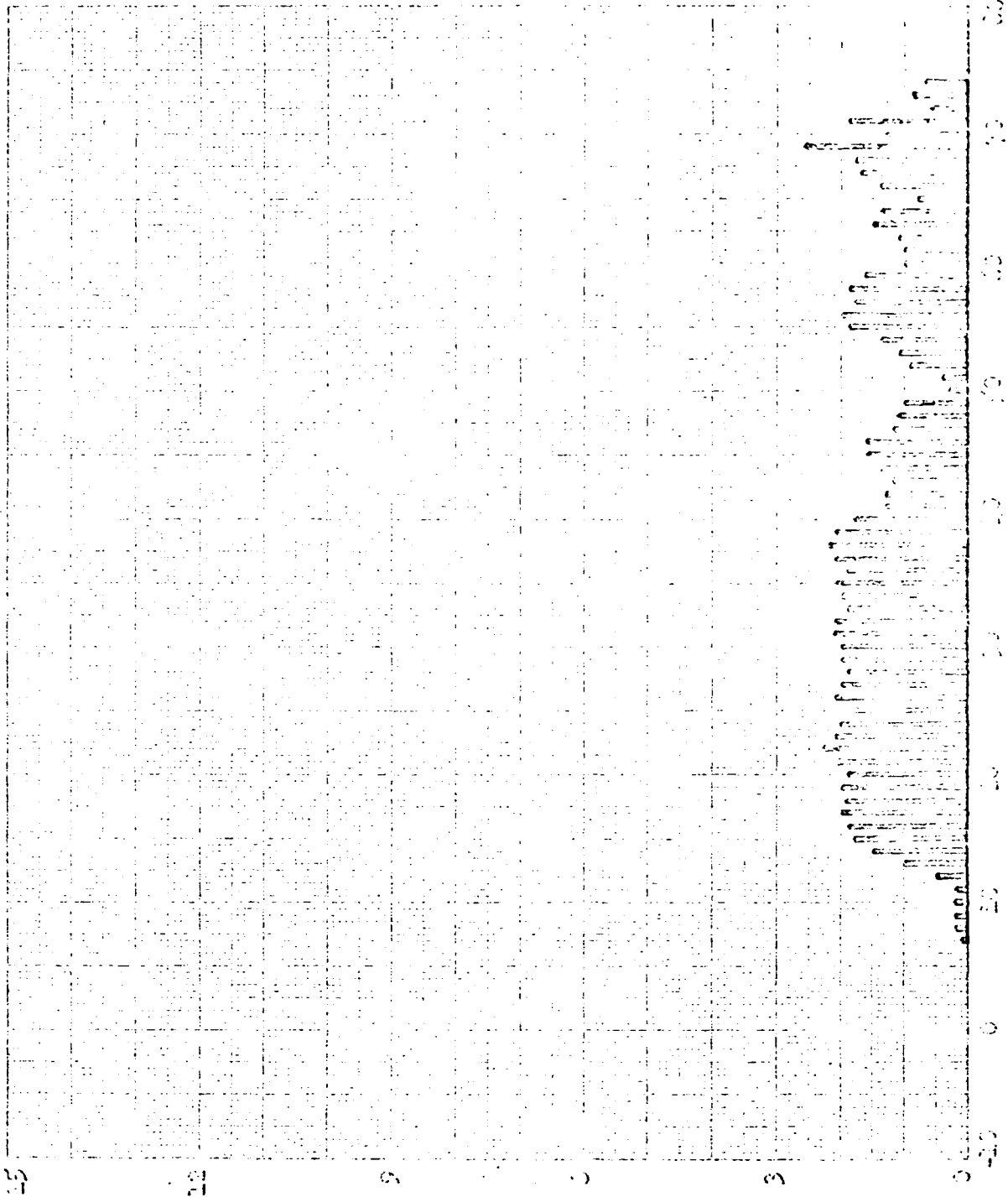
Project No: J-41

Contract No: 202

Launch Date: 11/2/67

Launch Time: 2131 Z

Latitude: 81.5°



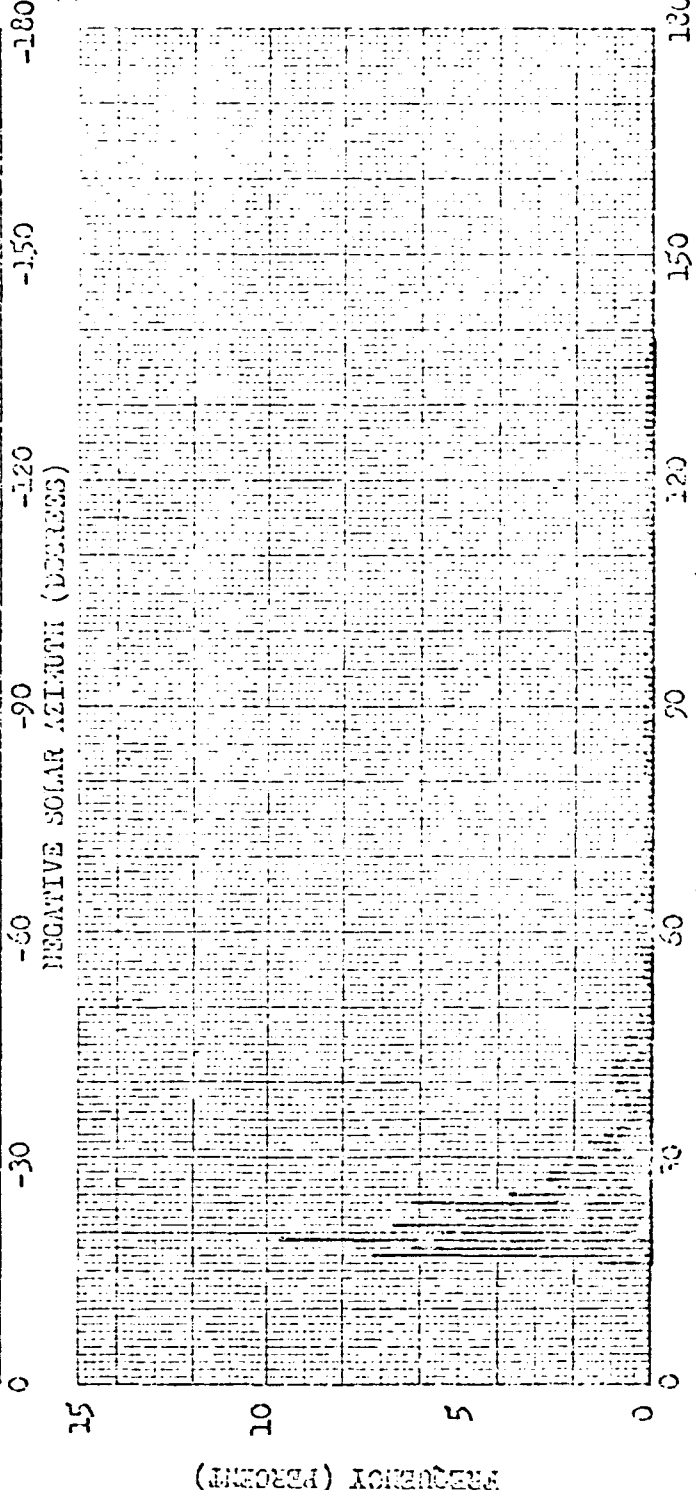
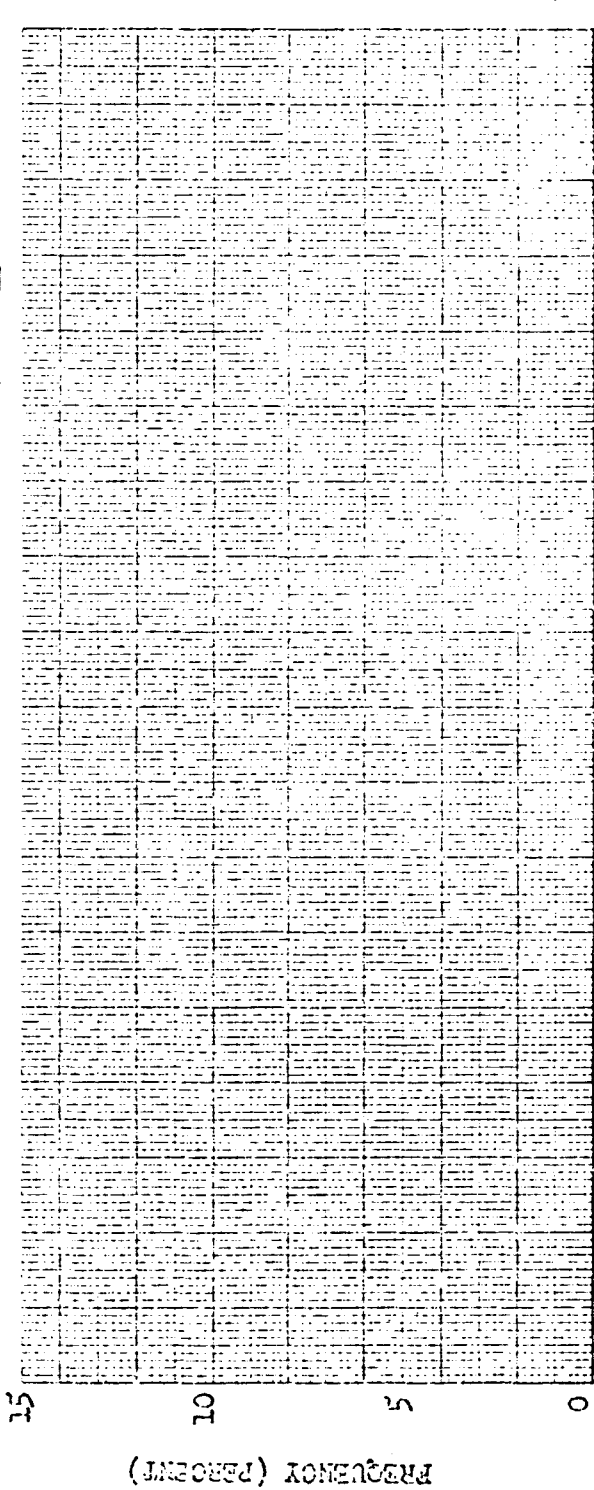
(UNCLASSIFIED)

FIGURE 5-3

TOP SECRET

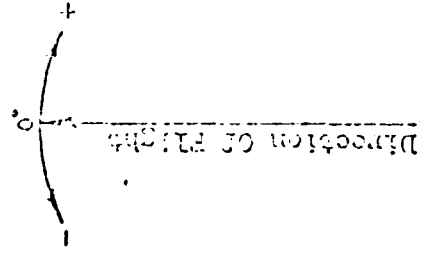
[REDACTED]

SOLAR AZIMUTH FREQUENCY DISTRIBUTION



Mission No: 10641-2  
 Payload No: J-11  
 Camera No: 202  
 Launch Date: 11/2/57  
 Launch Time: 2152 Z  
 Inclination: 81.5°

STON NOTATION



C [REDACTED]

EXPOSURE POINTS

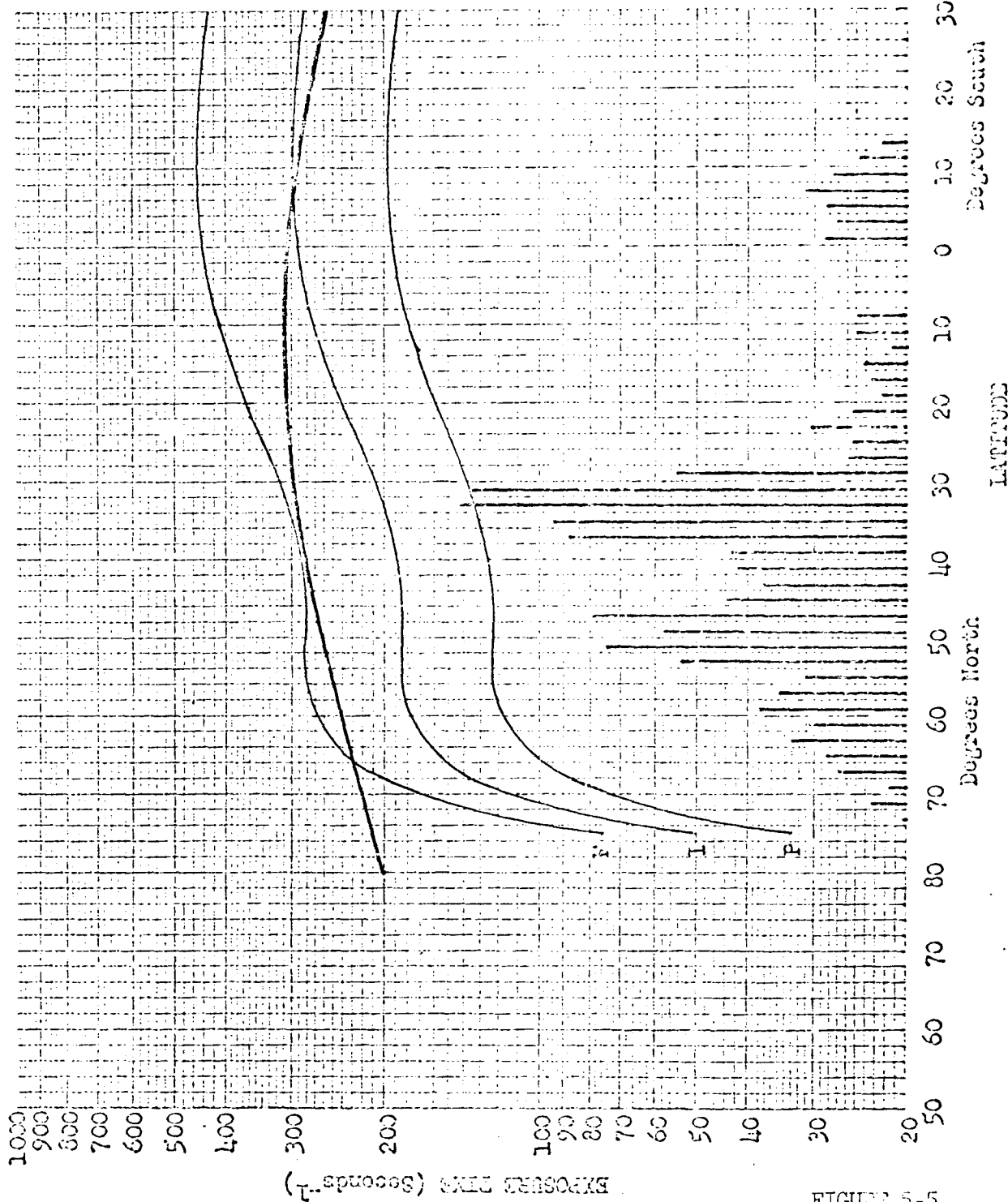
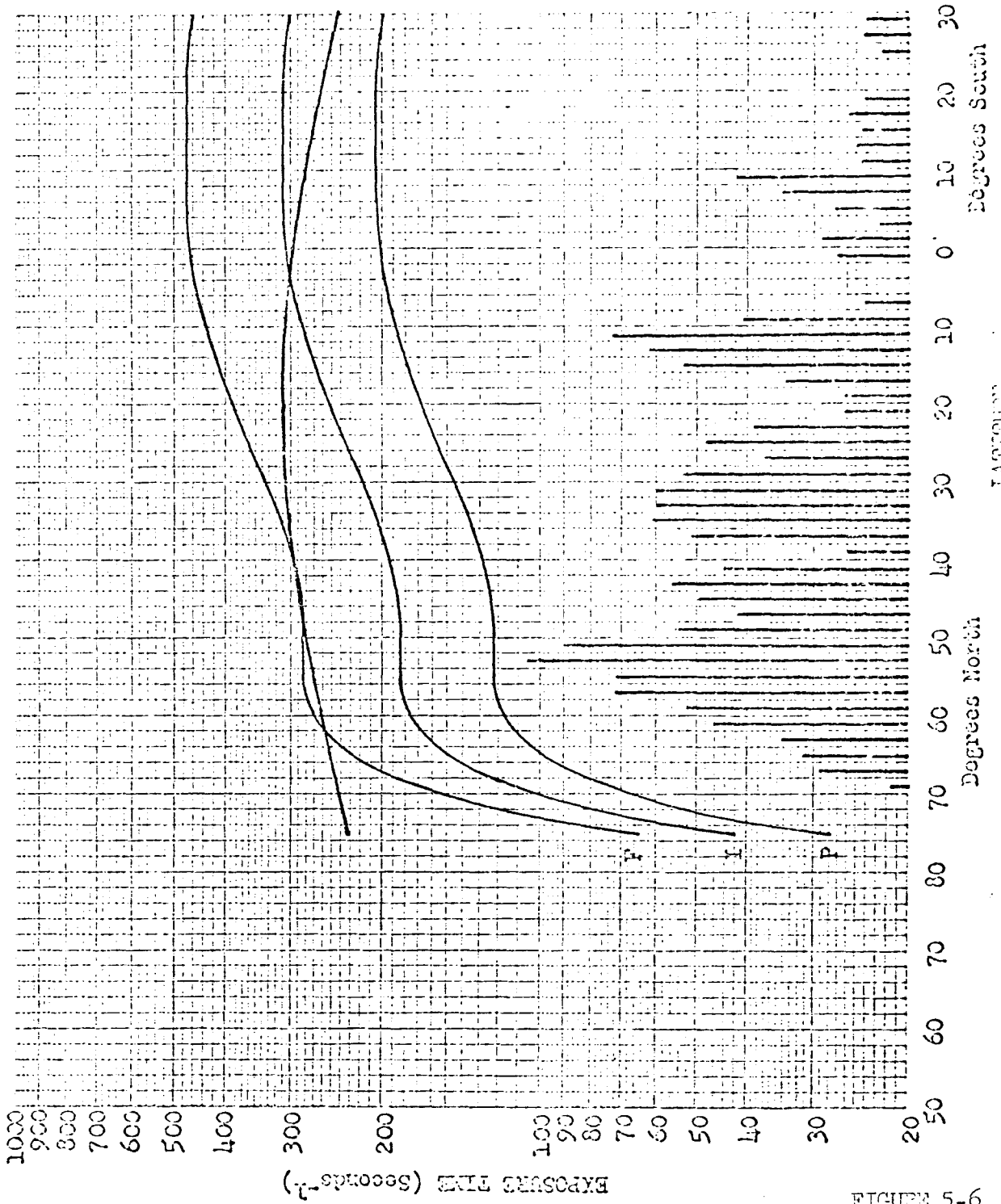


FIGURE 5-5

Mission No: 1044  
 Payload No: J-11  
 Camera No: 202  
 Pass No: 25  
 Launch Date: 11/2/67  
 Launch Time: 0151 Z  
 Slit Width: .225  
 Filter Type: Wratten 25  
 Film Type: 3404



EXPOSURE POINTS

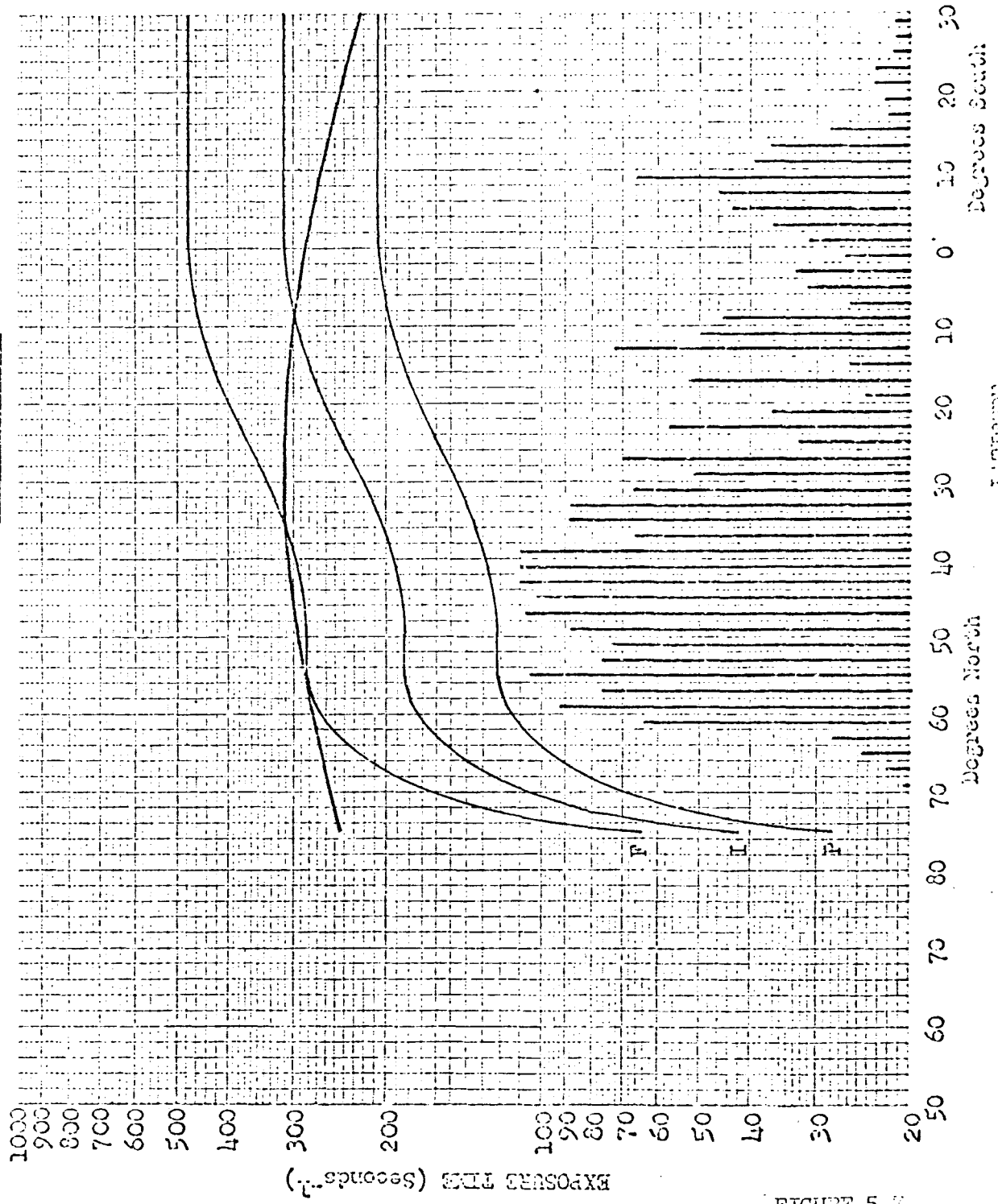


Mission No: 10Mh  
 Payload No: J-41  
 Camera No: 202  
 Pass No: 70  
 Launch Date: 11/2/67  
 Launch Time: 2131 Z  
 Slit Width: .225  
 Filter Type: Wratten 23  
 Film Type: 3404

FIGURE 5-6

~~TOP SECRET~~

EXPOSURE POINTS



Mission No: 1044  
Payload No: J-41  
Camera No: 202  
Pass No: 116  
Launch Date: 11/2/67  
Launch Time: 2131 Z  
Slit Width: .225  
Filter Type: Wratten 23  
Film Type: 3404

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FIGURE 5-7

TOP SECRET



EXPOSURE POINTS

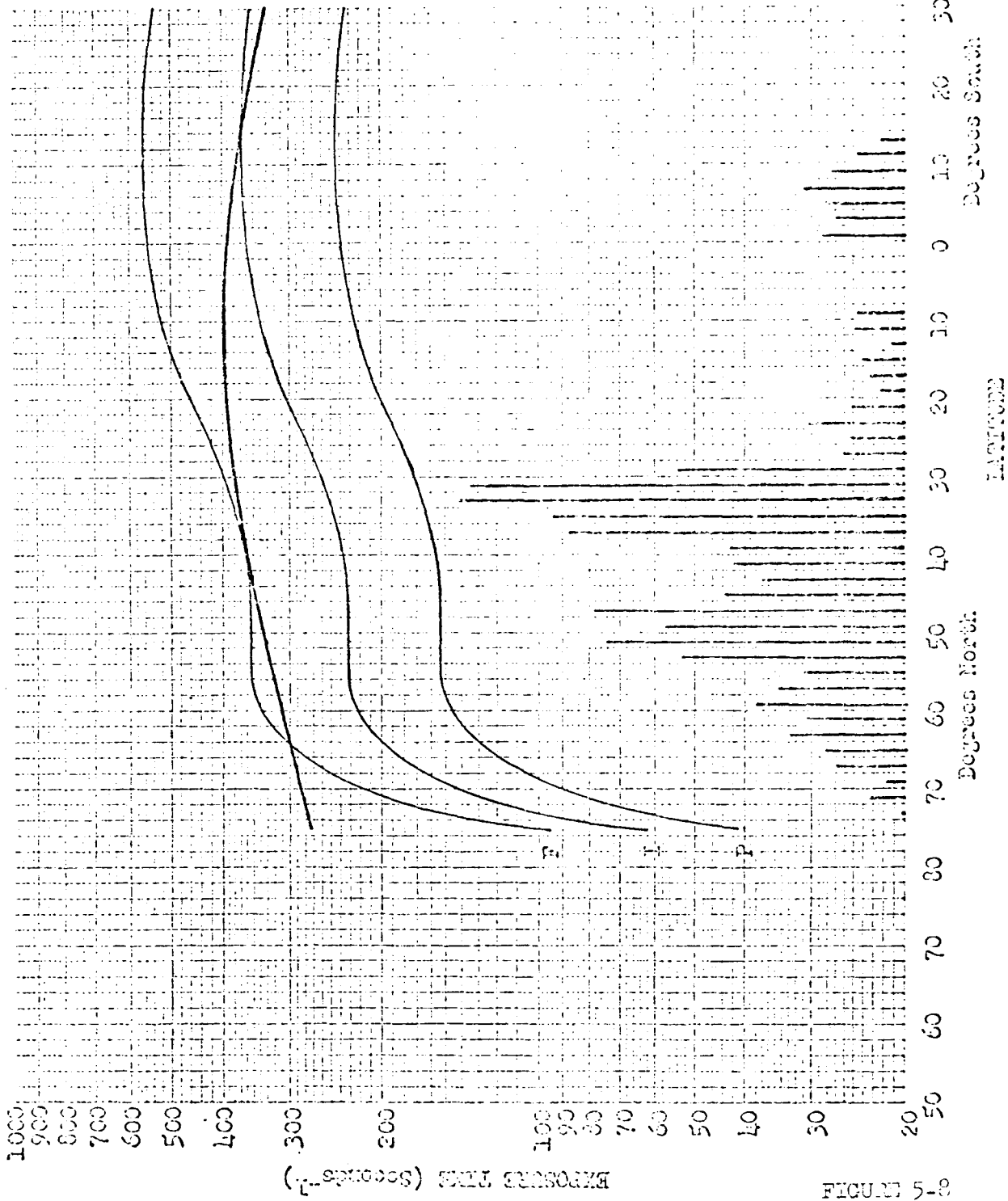


FIGURE 5-18

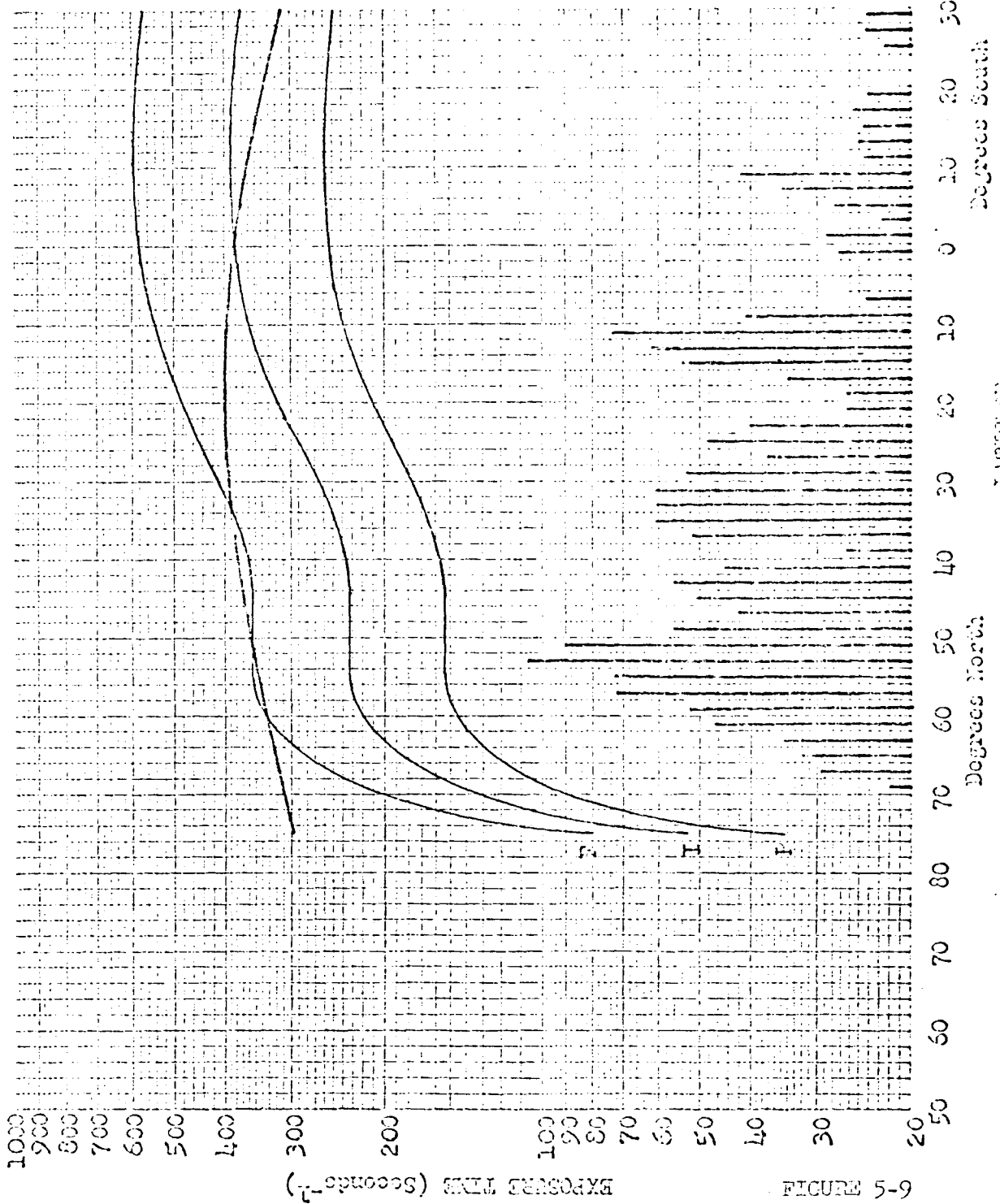
Mission No: 1044  
 Payload No: J-11  
 Camera No: 203  
 Pass No: 25  
 Launch Date: 11/2/67  
 Launch Time: 2132 Z  
 Slat Width: .175  
 Filter Type: Wetted 21  
 Film Type: 3104



TOP SECRET

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EXPOSURE POINTS



Mission No: 1044  
Payload No: J-42  
Camera No: 263  
Pass No: 70  
Launch Date: 11/2/57  
Launch Time: 2131 Z  
Slit Width: .175  
Filter Type: Wratten 21  
Film Type: 3404

FIGURE 5-9

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[REDACTED]

EXPOSURE POINTS

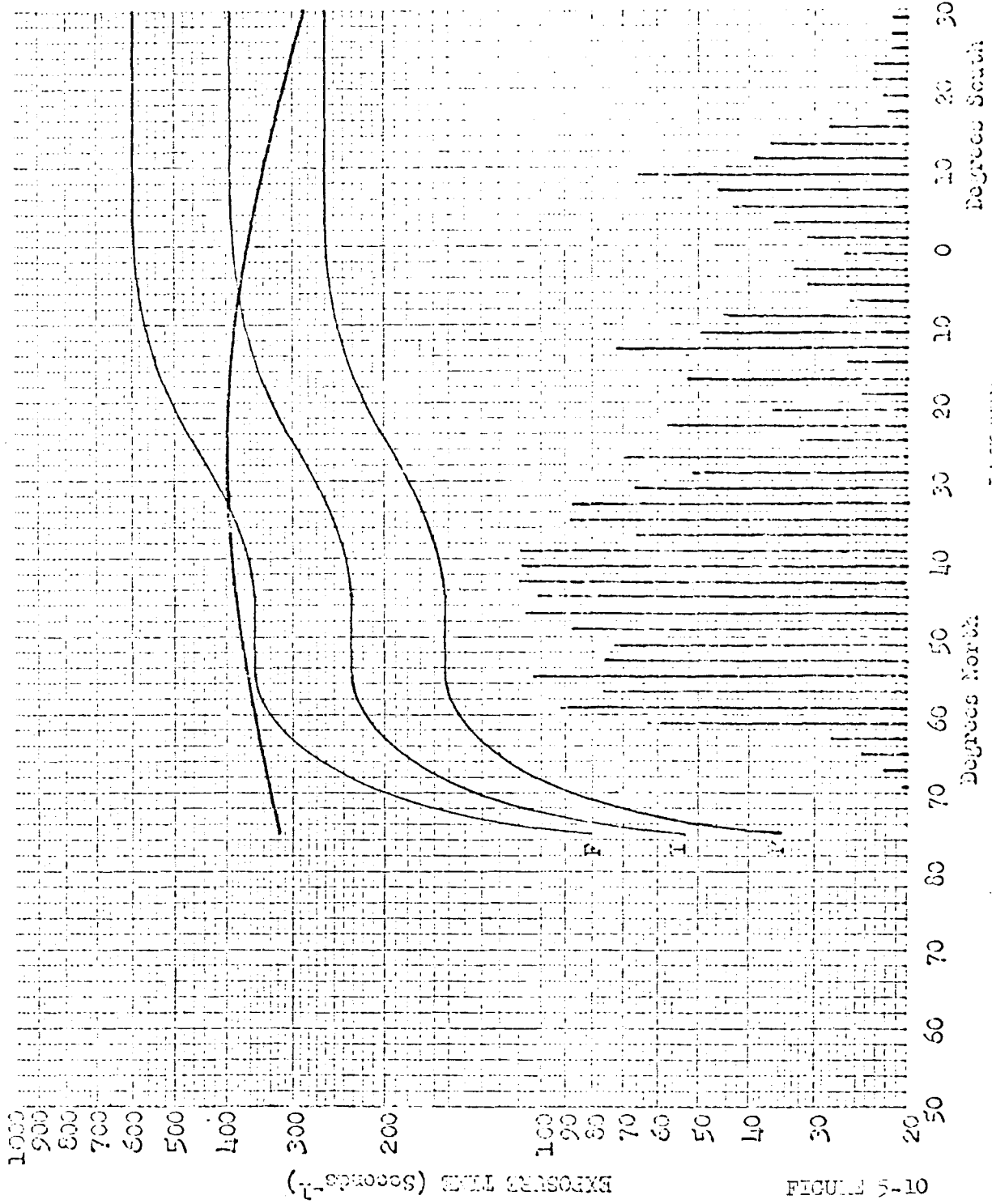


FIGURE 5-10

Mission No: 1014  
 Payload No: J-41  
 Camera No: 203  
 Pass No: 176  
 Launch Date: 11/2/57  
 Launch Time: 2131 Z  
 Slit Width: .175  
 Filter Type: Water 21  
 Film Type: 3104

[REDACTED]



SECTION 6

DIFFUSE DENSITY MEASUREMENTS

The diffuse density measurements made by AFOPPF were computer sorted at A/P to permit analysis of the density ranges resulting from the three levels of conventional processing and from the dual gamma process experiment. The sorting technique utilizes the base plus fog density values for the conventionally processed materials where measurements up to 0.09 density are considered as having received Primary processing, 0.10 to 0.17 as Intermediate, and above 0.17 density as Full. The percentage of this material that was processed at each level, based on the computer sort, is tabulated below with the predicted and reported processing percentages.

<u>Mission</u>	<u>Camera</u>		<u>Primary</u>	<u>Intermediate</u>	<u>Full</u>	<u>Transition</u>
1044-1	Fwd	Predicted	0	13	87	-
		Reported	0	6	88	6
		Computed	0	8	92	-
1044-1	Aft	Predicted	0	19	81	-
		Reported	2	12	71	15
		Computed	0	23	77	-
1044-2	Fwd	Predicted	4	16	80	-
		Reported	0	4	92	4
		Computed	0	7	93	-
1044-2	Aft	Predicted	8	24	68	-
		Reported	5	17	63	15
		Computed	0	28	72	-



C [REDACTED]

Approximately 30 percent of the total mission original negative was subjected to a "dual-gamma" processing experiment. The results indicate a very effective reduction in the maximum cloud and snow densities with only minor influence in the normal range of terrain densities.

Graphical computer plots of the sampled density distributions are presented in Appendix A, Pages A-1 through A-48. Note the variation between the conventional processing plots and those for the dual gamma process. The differences in the cloud  $D_{max}$  are very distinct. There is, however, a more subtle distinction that should be emphasized; namely, the incidence of lower terrain  $D_{min}$  densities with the dual gamma process than with conventional processing. The reasons for these variations are obvious upon comparison of the corresponding sensitometric curves, Figures 6-2 through 6-13.

The sensitometric curves also illustrate the distinct deviations of the actual flight material processing from the standards and from the R-2 day samples. Obviously, there is need to maintain exposure control based on actual effective processed film speeds rather than on the standards. This will be especially true for the dual gamma process if future deviations continue to be as significant as was experienced in this case. As the dual gamma process becomes operational, it is anticipated that reliable processing controls will be attained. Likewise, it is anticipated that as additional progress in target density analysis is made a corresponding reliable exposure criteria will become a reality.

C

A summary of the processing and exposure analysis for the conventionally processed material is shown in Table 6-1. The terrain D-Min criteria, (range) for proper exposure and processing is 0.40 to 0.90 density units. The area measured for D-Min is selected subjectively and is not necessarily the absolute D-Min in the photography.

The terrain D-Min criteria has been found to be an inadequate indicator of optimum target exposure. Maximum intelligence is derived from specific target densities meeting this criteria; which, in general, results in overall terrain D-Min values repeatedly below the 0.40 density level. It is therefore apparent that the more desirable missions will, most likely, be reported as significantly underexposed by the present terrain D-Min criteria.

A density range chart, Figure 6-1, is included in this report. This type of chart for Missions 1004 to 1031 is included in the A/P final report for Mission 1031.

These charts are produced from the same density measurements previously mentioned in this section. The computer produced the mean, median and range figures for the various processing levels used. The chart includes the number of frames (samples) in which the density measurements were made. These measurements are made on approximately every tenth frame throughout the mission. It should be noted that the density figures shown for Missions 1044-1 and 1044-2 include both dual-gamma and conventionally processed materials, thus tending to artificially enlarge the apparent range of densities, especially for the cloud D-Max values.



MISSION 1044-1 INSTR - FWD 1/16/68 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXPOSED	OVER PROCESSED	OVER EXPOSED
PRIMARY	C	0 PC	0 PC	0 PC	7 PC	7 PC
INTERMEDIATE	13	0 PC	0 PC	46 PC	46 PC	8 PC
FULL	147	35 PC	0 PC	54 PC	10 PC	1 PC
ALL LEVELS	140	32 PC	0 PC	53 PC	13 PC	1 PC

MISSION 1044-1 INSTR - AFT 1/16/68 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXPOSED	OVER PROCESSED	OVER EXPOSED
PRIMARY	C	0 PC	0 PC	0 PC	19 PC	19 PC
INTERMEDIATE	37	0 PC	11 PC	65 PC	19 PC	5 PC
FULL	126	28 PC	0 PC	65 PC	7 PC	0 PC
ALL LEVELS	163	21 PC	2 PC	65 PC	10 PC	1 PC

MISSION 1044-2 INSTR - FWD 1/16/68 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXPOSED	OVER PROCESSED	OVER EXPOSED
PRIMARY	C	0 PC	0 PC	0 PC	7 PC	7 PC
INTERMEDIATE	12	0 PC	0 PC	42 PC	58 PC	0 PC
FULL	168	38 PC	0 PC	58 PC	4 PC	0 PC
ALL LEVELS	180	36 PC	0 PC	57 PC	7 PC	0 PC

MISSION 1044-2 INSTR - AFT 1/16/68 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXPOSED	OVER PROCESSED	OVER EXPOSED
PRIMARY	C	0 PC	0 PC	0 PC	24 PC	24 PC
INTERMEDIATE	46	0 PC	25 PC	59 PC	15 PC	0 PC
FULL	116	31 PC	0 PC	66 PC	3 PC	0 PC
ALL LEVELS	162	22 PC	7 PC	64 PC	6 PC	0 PC

MISSION 1044-2 INSTR - AFT 1/16/68 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	BASE & FUG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXPOSED	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.60	-----	0.91 AND UP
INTERMEDIATE	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.60	0.91-1.26	1.35 AND UP
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.60	0.91-1.69	1.70 AND UP

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TABLE 6-1

# J MISSION DENSITY RANGES

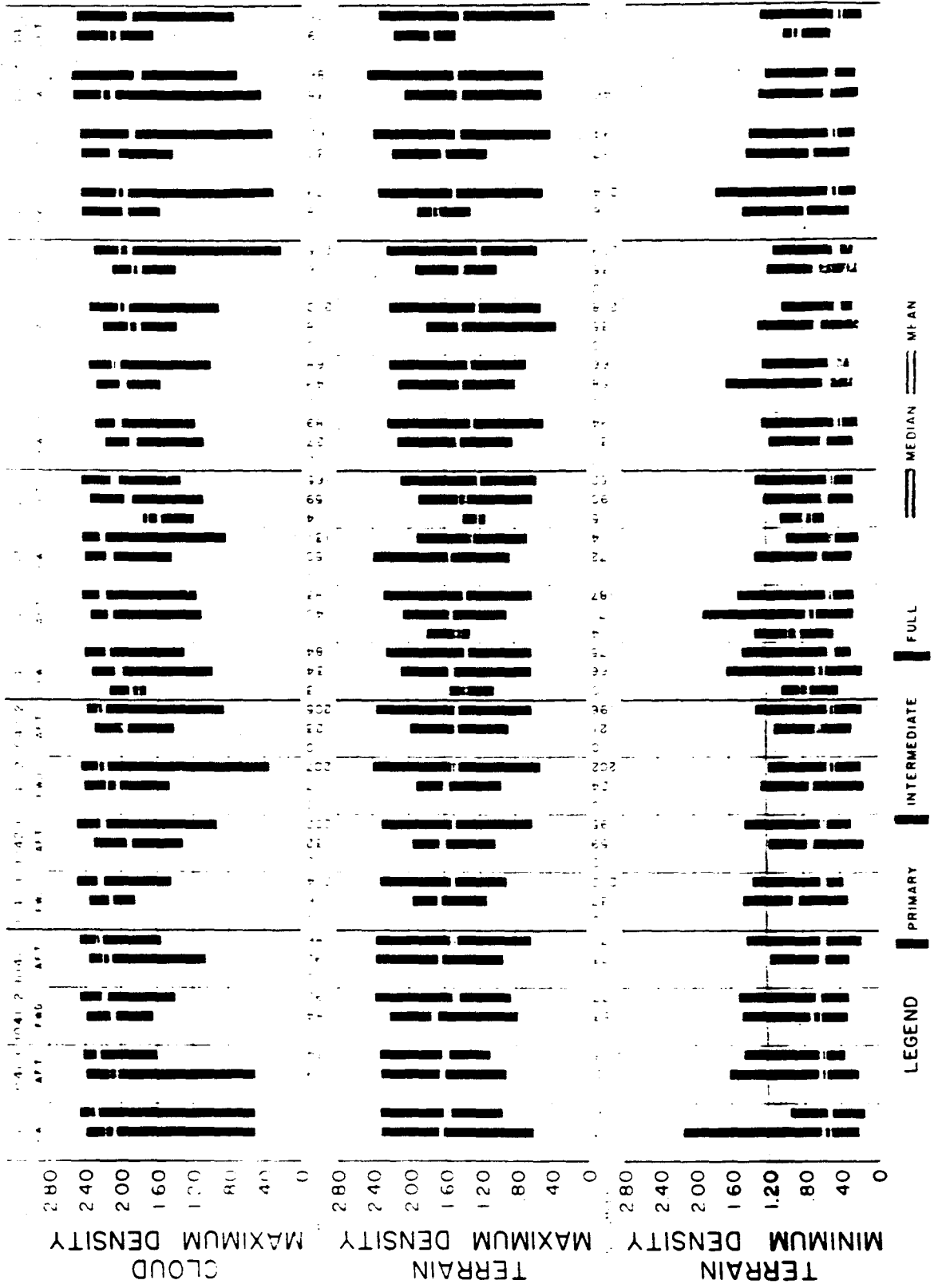
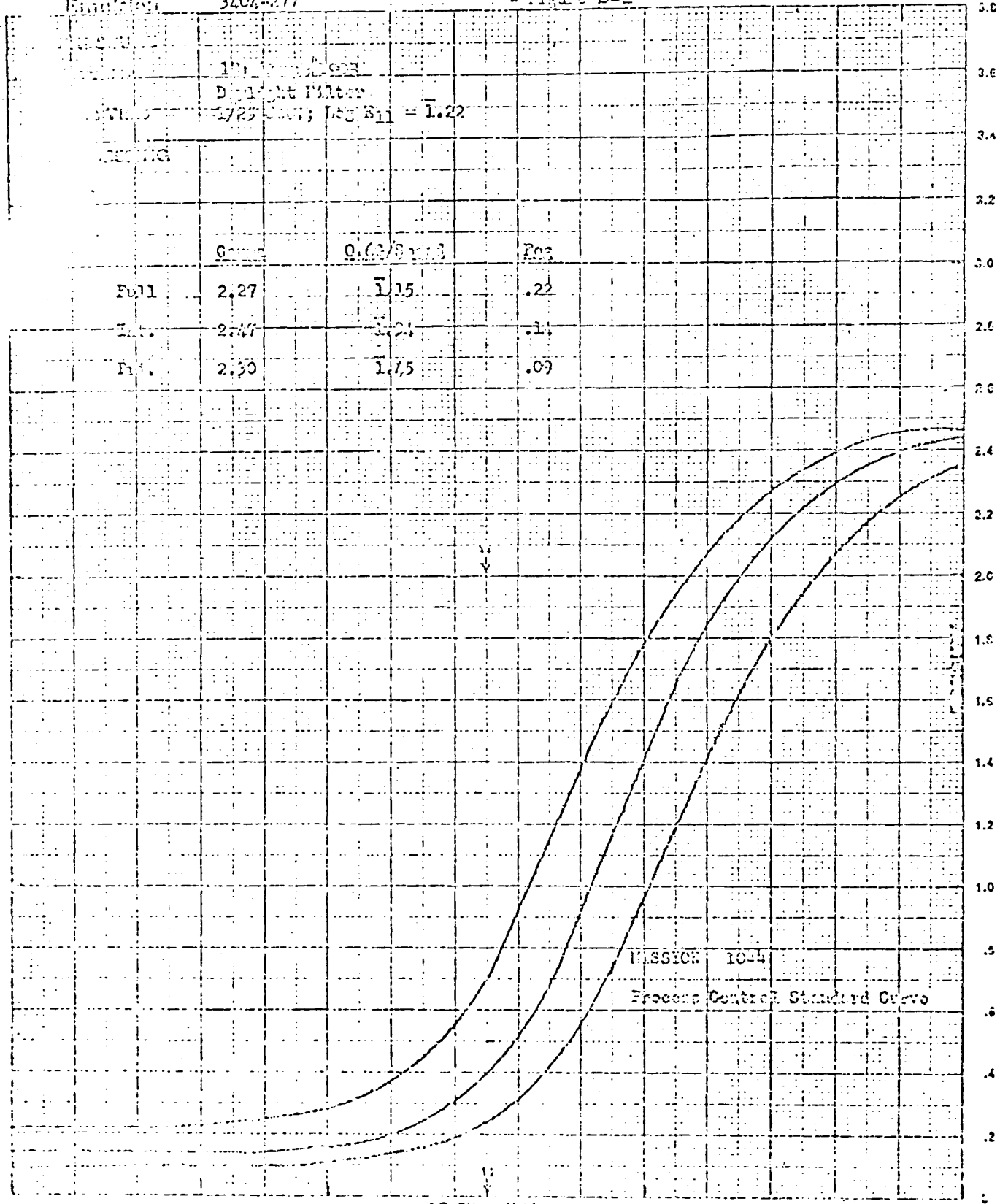


FIGURE 6-1

Emulsion 3404-277

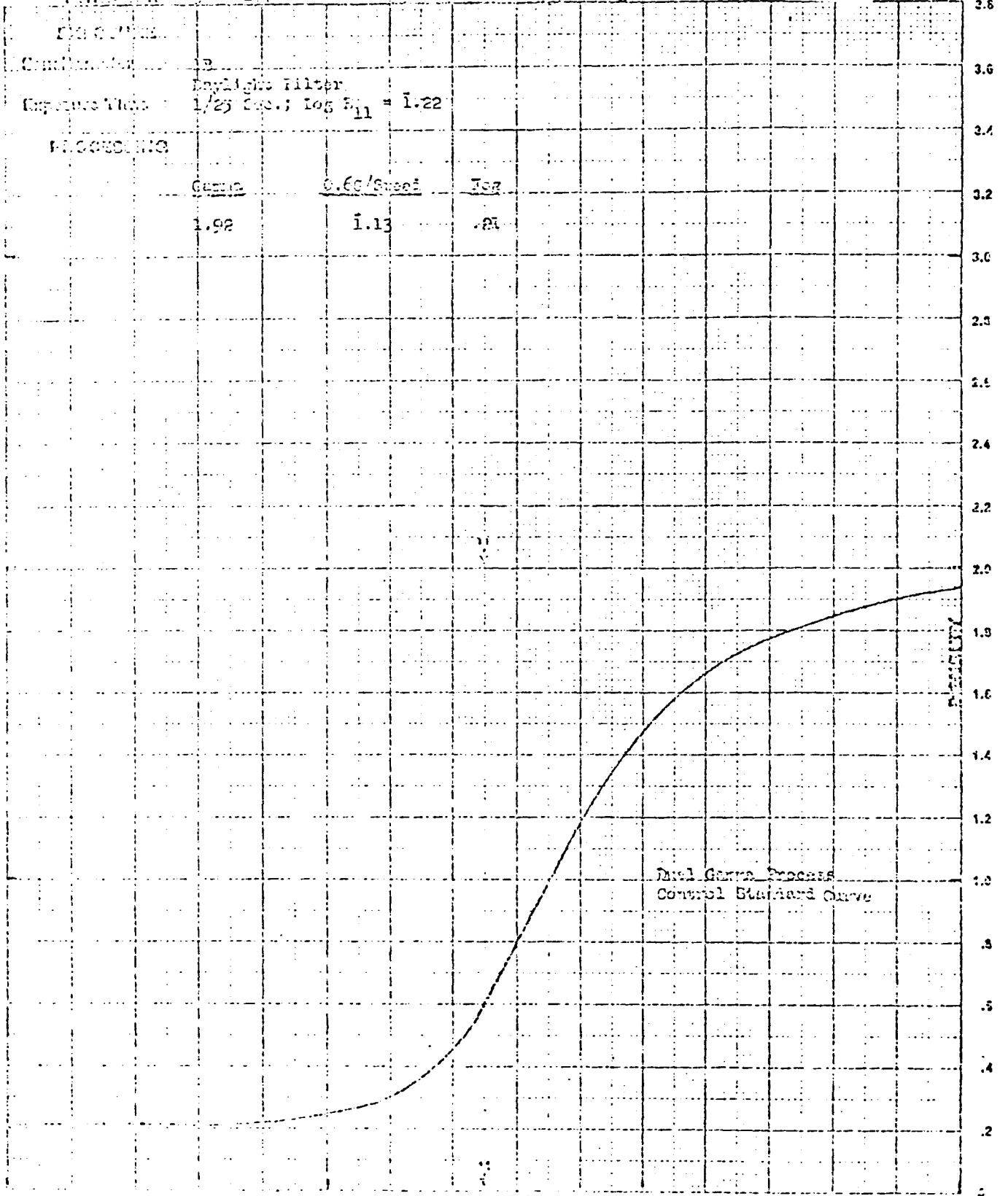
Figure 6-2



[REDACTED]  
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[REDACTED]

Emulsion 3401-277

6-3



Condition: B  
Exposure Time: 1/25 Sec.; Log E<sub>11</sub> = 1.22

RECORDING

Gamma 1.92      0.60/Speed 1.13      ISO 20

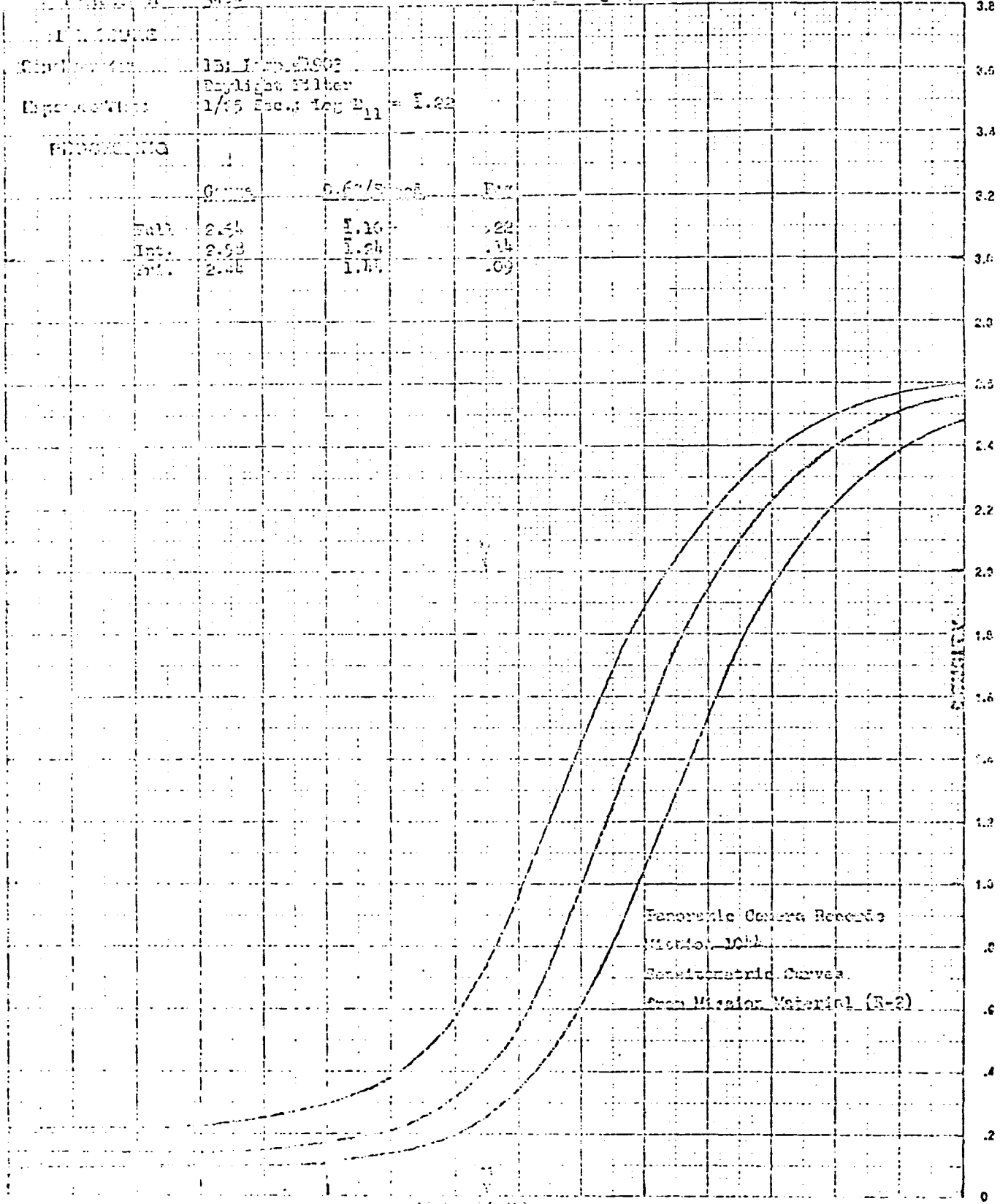
[REDACTED]  
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C/ [REDACTED]

Emulsion

3404

Fig. 6-4



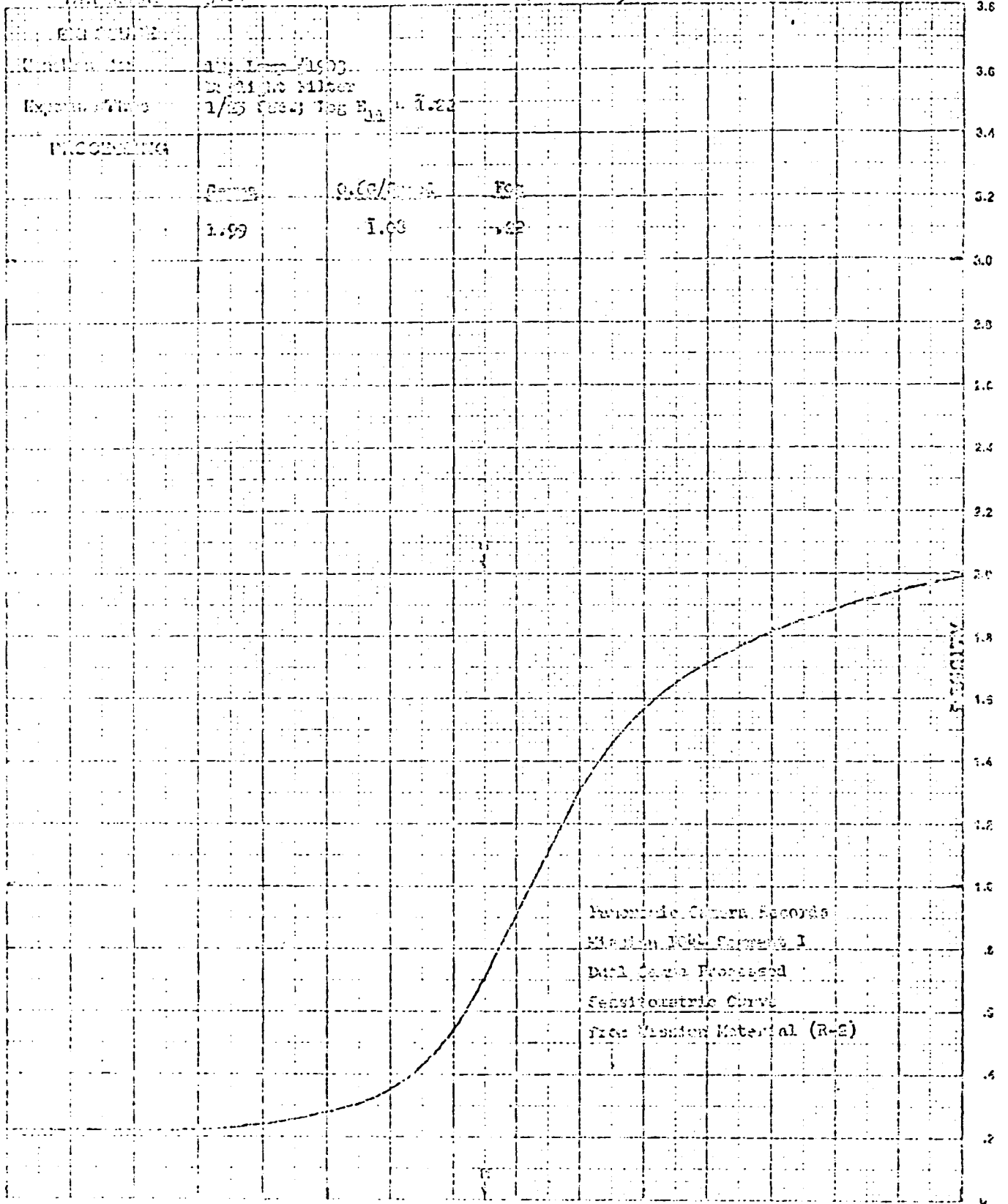
LOG TIME

[REDACTED]

C [REDACTED]  
 [REDACTED]

Emulsion 3494

Figure 6-5



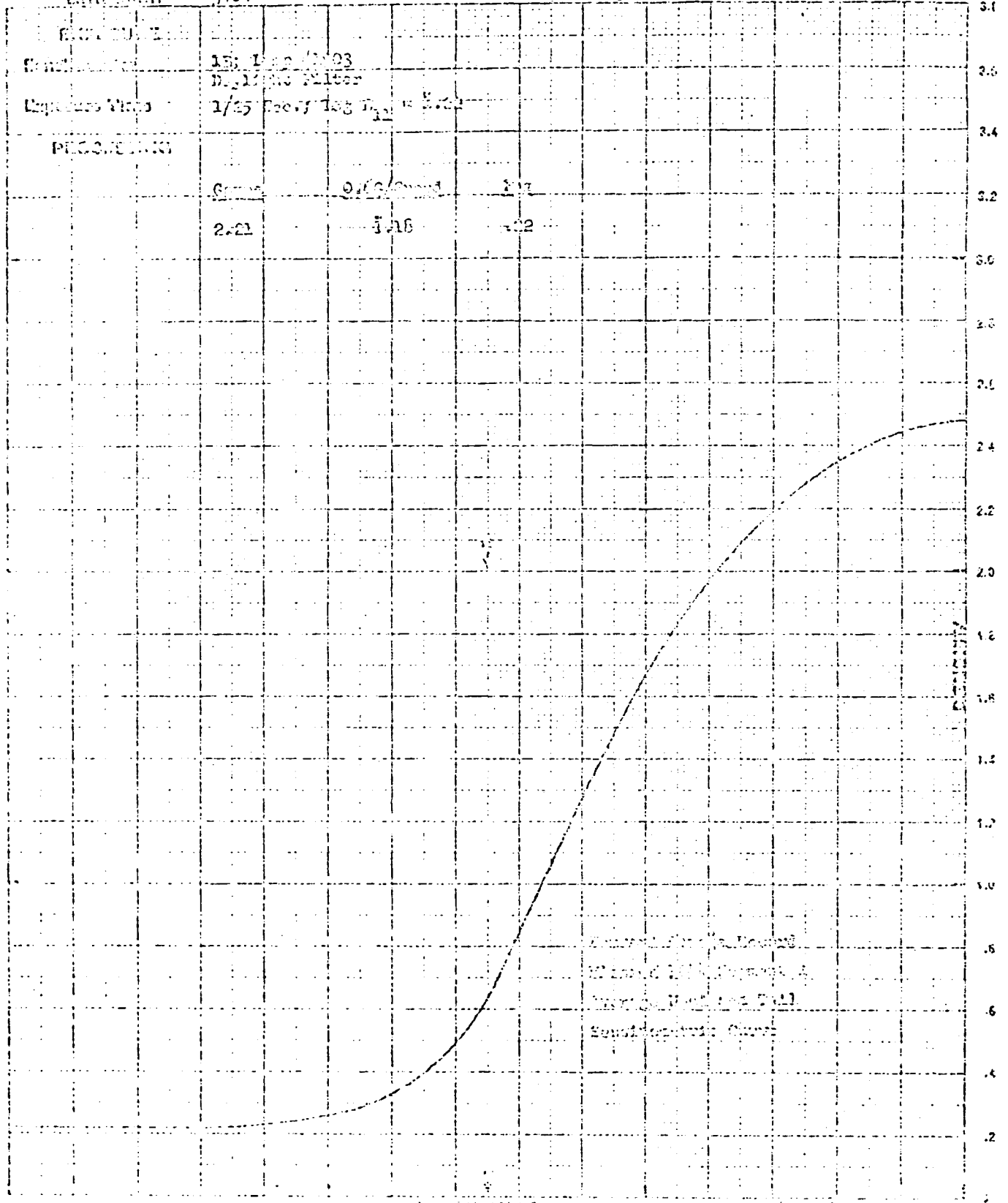
[REDACTED]  
 [REDACTED]

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Equipment 3404

Figure 6-6

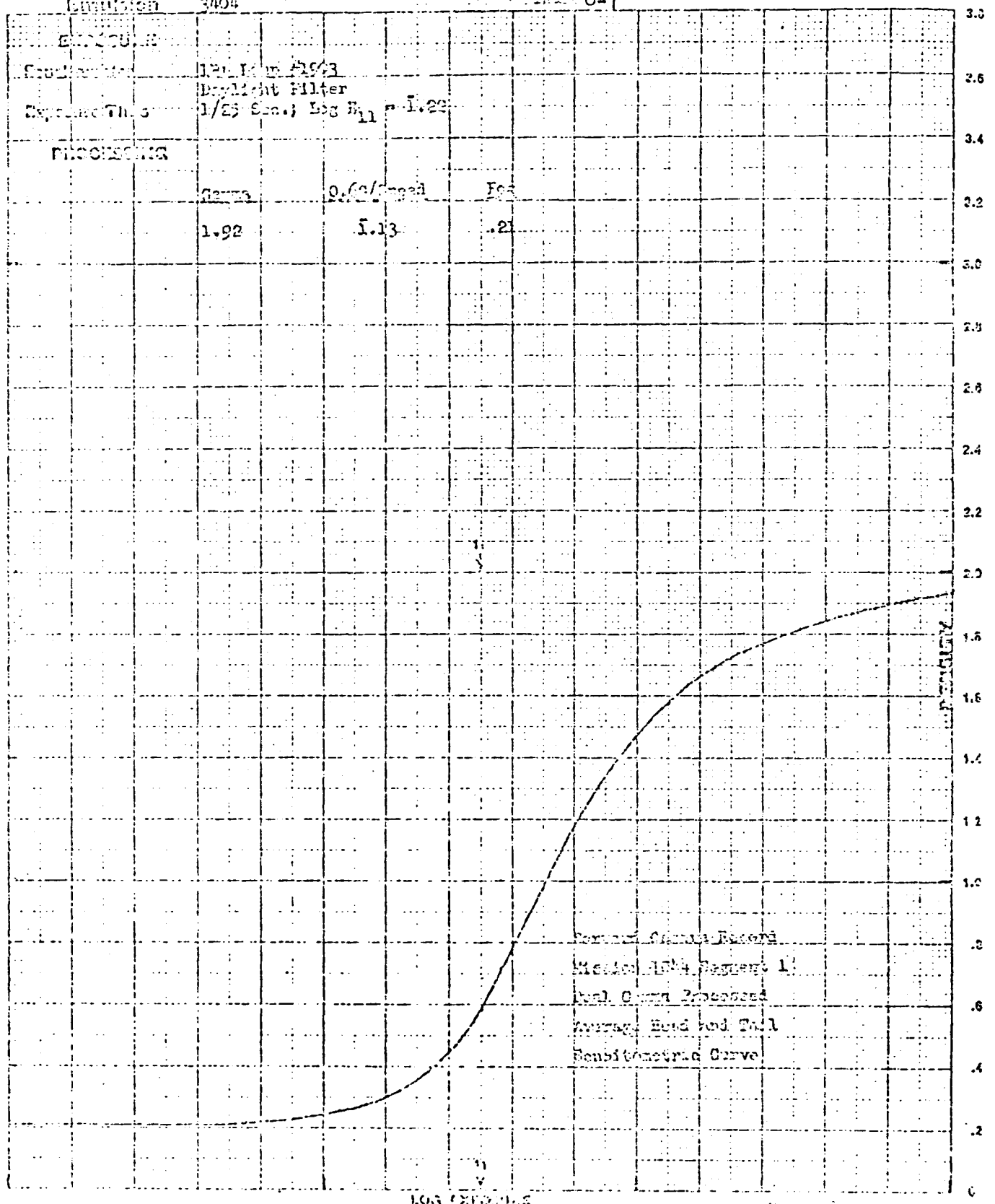


Filter 153 1/2" 1/2" 23  
 D. 1/2" 23 Filter  
 Impedance Wind 1/25 1/2" 1/2" 23  
 PRESSURE 2.21 0.16 1.16 1.12

General Electric  
 Model 100  
 Series 100  
 Semi-conductor

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Emulsion 3404 Figure 6-7



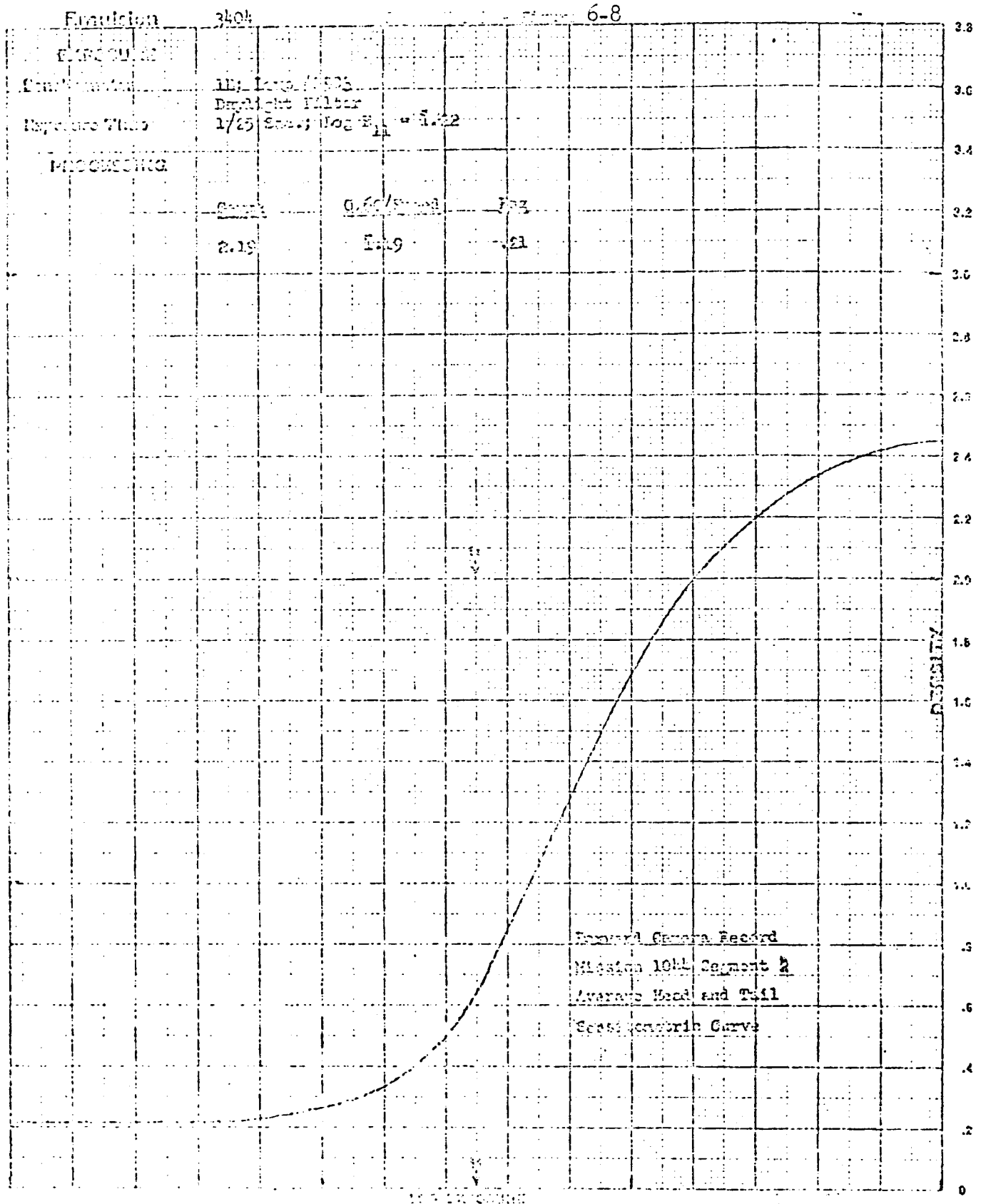
EMULSION  
 3404  
 120 Micron 1943  
 Daylight Filter  
 Exposure Time 1/25 Sec.; Log E<sub>11</sub> = 1.22  
 PROCESSING  
 Gamma 1.92  
 0.60/Speed  
 Fog 0.21  
 I.13

Kodak Research Laboratories  
 Kodak 1634, Negative 1  
 Final Curve Processed  
 Average Head and Tail  
 Sensitometric Curve

LOG EXPOSURE



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 [REDACTED]

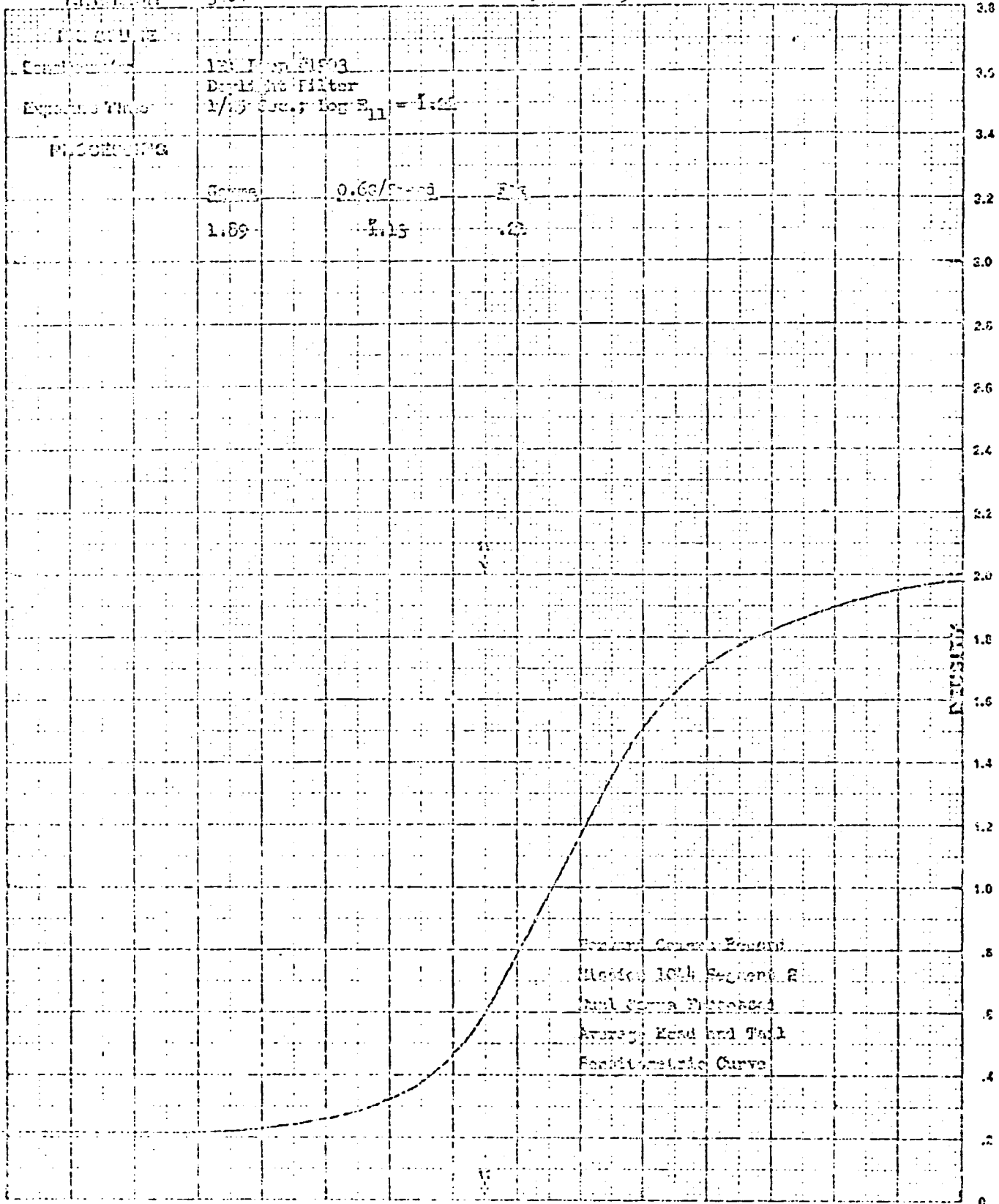


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Rawlston 3404

Page 6-9



RAWLSTON

Count Rate

Exposure Time

PHOTOMETER

1201 on 1503  
 Double Filter  
 1/15 Sec.; log  $R_{11} = 1.42$

Source 0.66/5-rod  $R_{11}$   
 1.89 7.15 1.2

Fourier Coeff. Found  
 Modified 10th Segment 2  
 and Gamma Plot  
 Average Head and Tail  
 Facilitate Curve

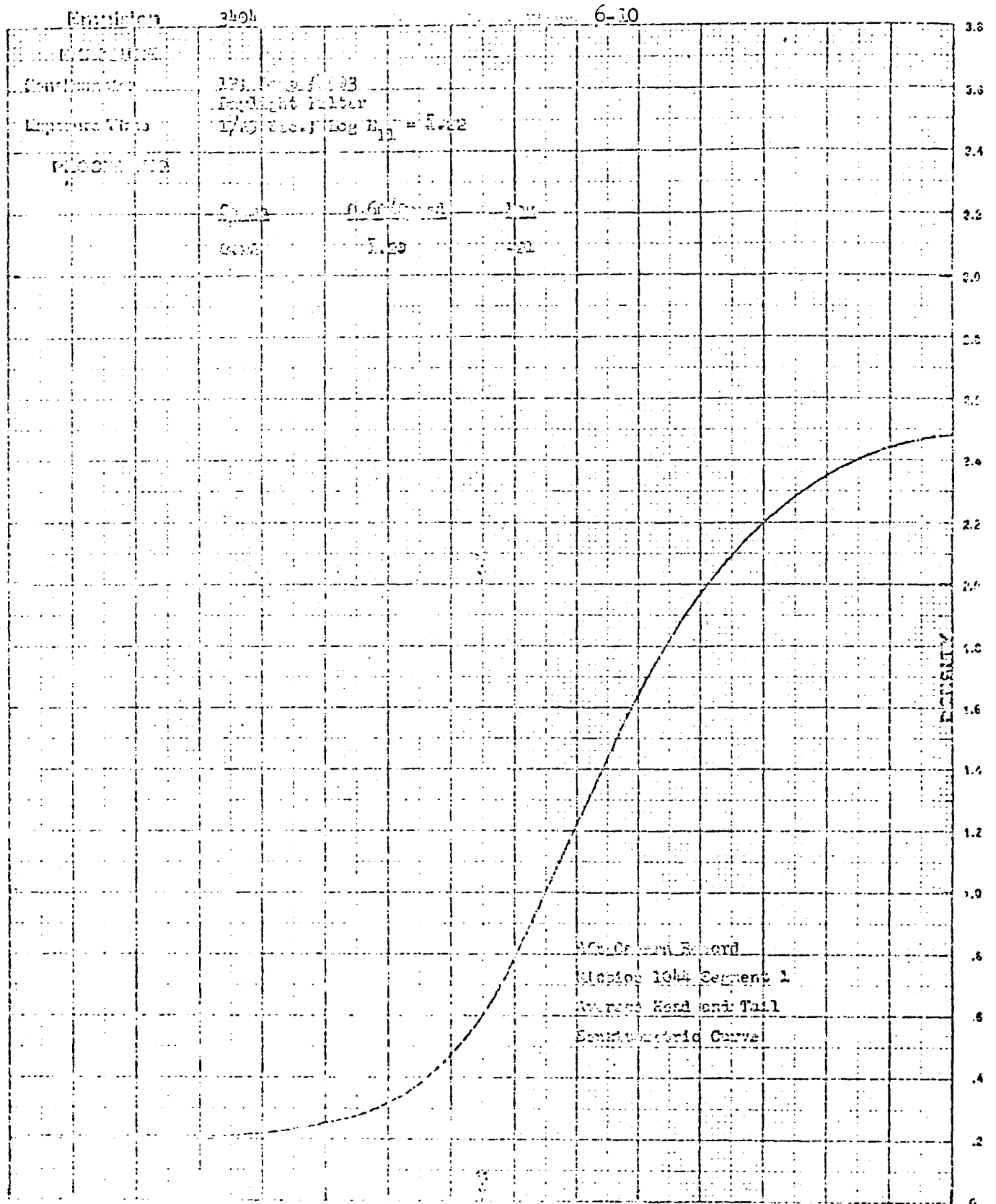
NO. 1000000

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C/ [REDACTED]

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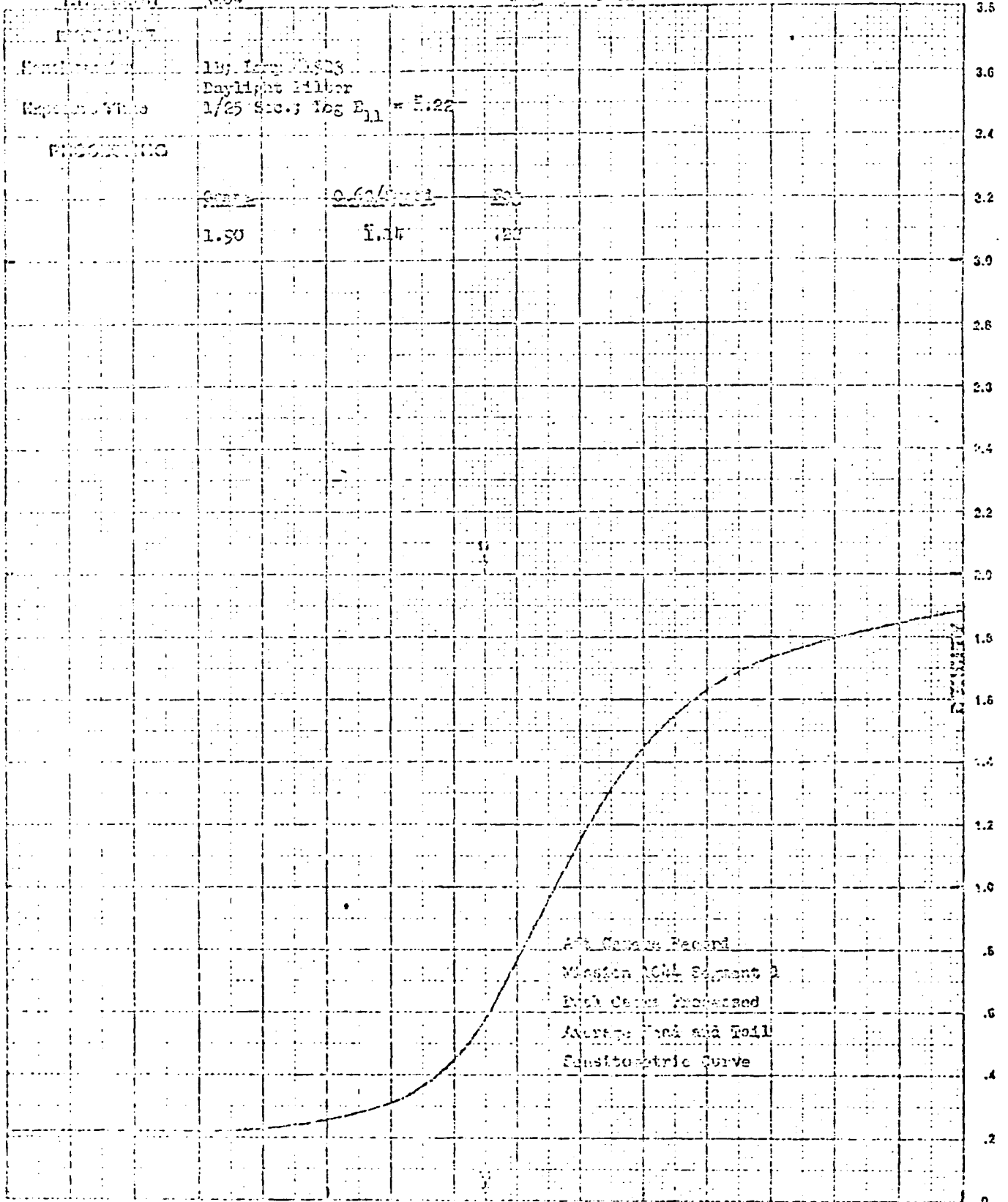
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[REDACTED]

Figure 3-04

6-11



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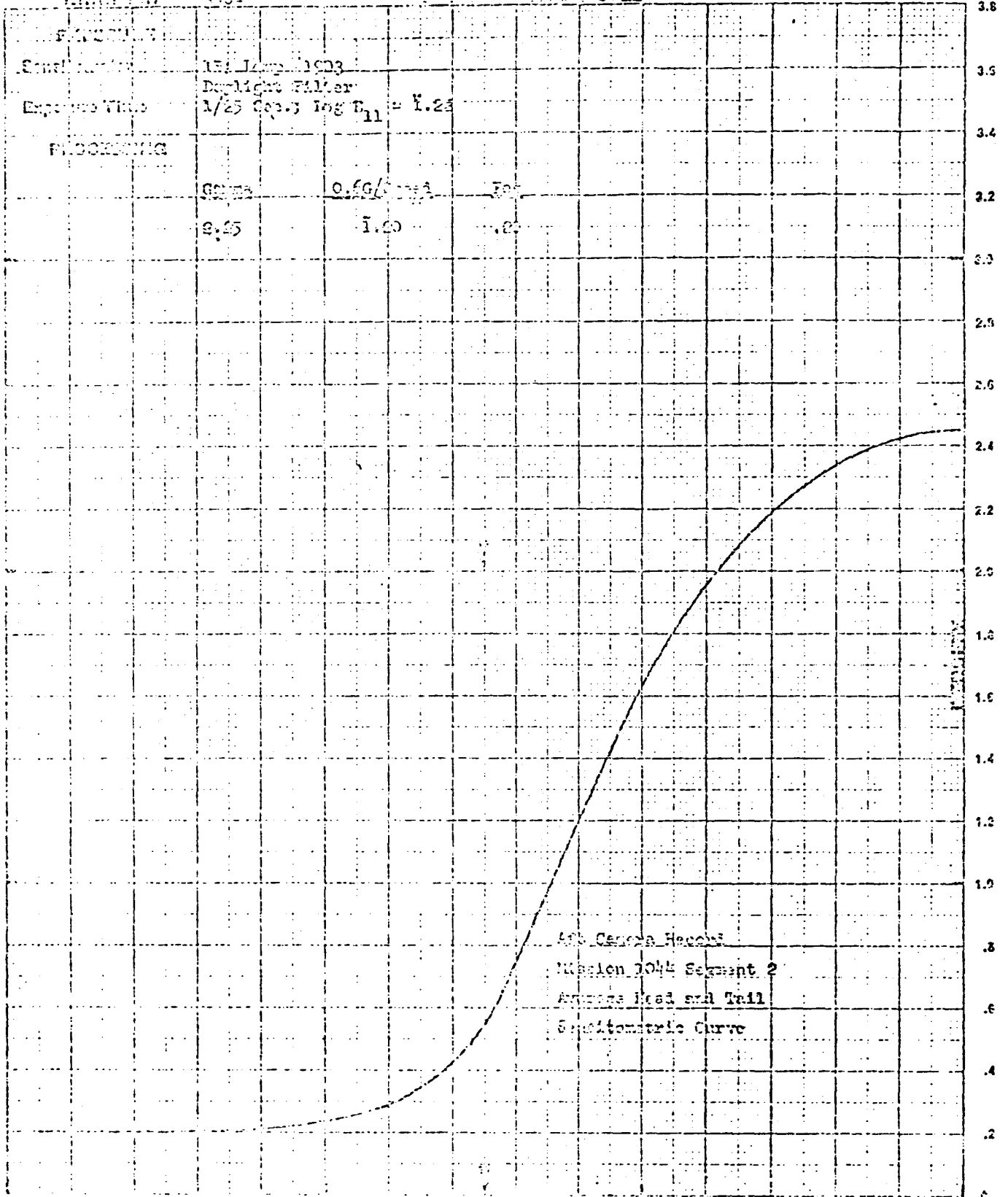
C/ [REDACTED]

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[REDACTED]

Enrollment 2:04

Figure 6-12

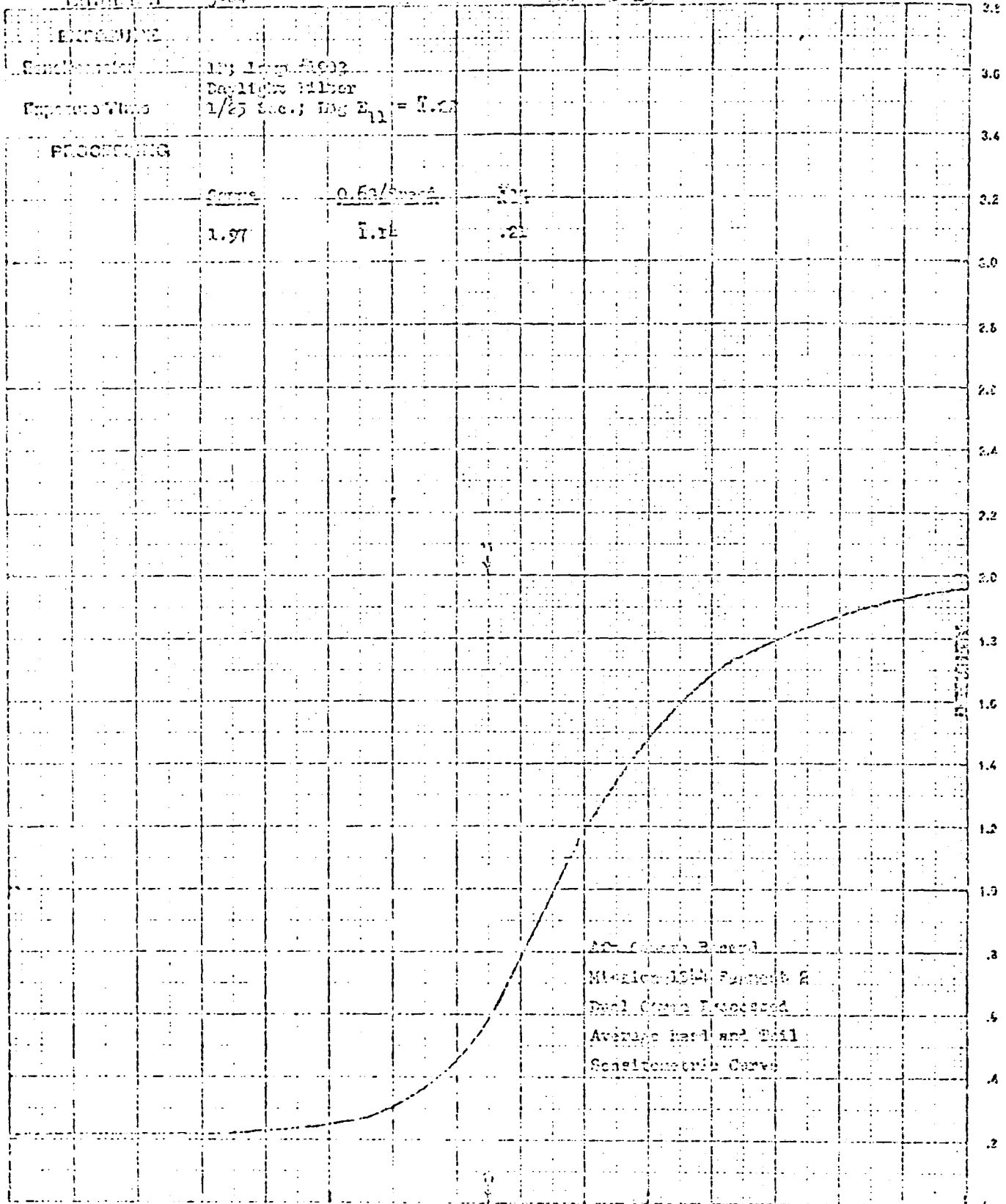


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[REDACTED]

Drainage 3504

Figure 6-13



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SECTION 7

VEHICLE ATTITUDE

The vehicle attitude errors for both Mission 1044-1 and 1044-2 were derived from the reduction of the Stellar camera photography. This attitude data is supplied to A/P by NPIC.

The attitude errors for each frame and the attitude control rates are calculated at the A/P computer facility. The computer also plots the frequency distribution of the rates and errors. Figures 7-2 through 7-7 show these distributions for Mission 1044-1 and Figures 7-8 through 7-13 for Mission 1044-2.

The summary table below lists the maximum attitude errors and rates that were experienced during 90 percent of the FWD camera photographic operations, excluding the first six frames of each operation, and the total range of the errors and rates.

<u>Value</u>	<u>Mission 1044-1</u>		<u>Mission 1044-2</u>	
	<u>90%</u>	<u>Range</u>	<u>90%</u>	<u>Range</u>
Pitch Error (°)	0.30	-0.35 to + 0.02	0.37	-0.62 to + 0.10
Roll Error (°)	0.15	-0.28 to + 0.46	0.37	-0.57 to + 0.06
Yaw Error (°)	3.42	-1.20 to + 3.80	3.31	-0.40 to + 3.60
Pitch Rate (°/hr.)	14.53	-85 to +85	23.64	-65 to +75
Roll Rate (°/hr.)	26.23	-58 to +76	30.62	-85 to +50
Yaw Rate (°/hr.)	51.28	-98 to +24	29.78	-80 to +10

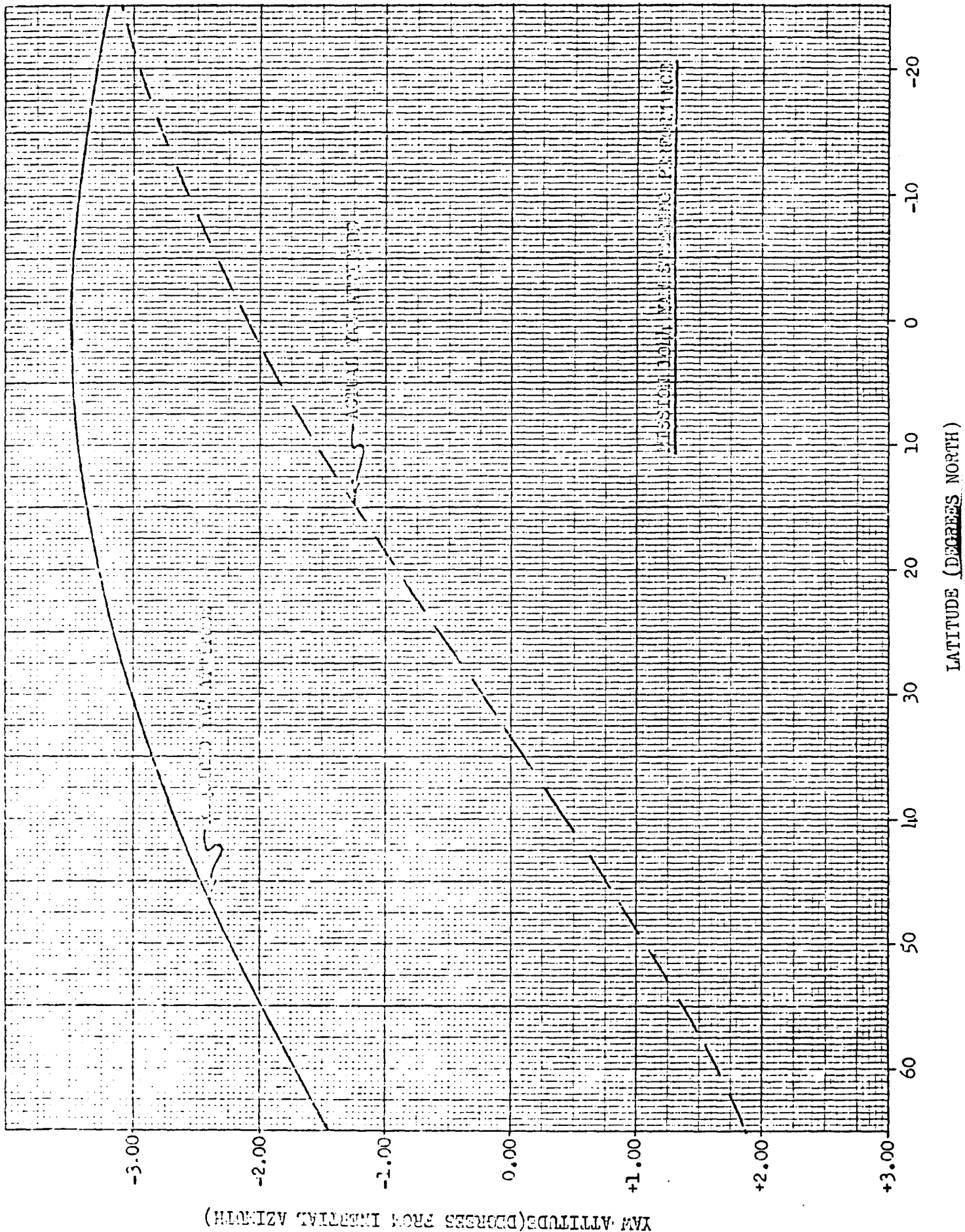
The yaw angle error represents the difference between the actual vehicle yaw attitude and the ideal yaw angle that would provide correct ground image motion. Because of a pre-flight programming error in the

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placement of the function start position, the yaw programmer was approximately 800 seconds out of phase with the desired performance. The large yaw angle error indicated reflects this condition. Figure 7-1 graphically depicts these relationships. The effects on image quality are discussed in Sections 4 and 8.



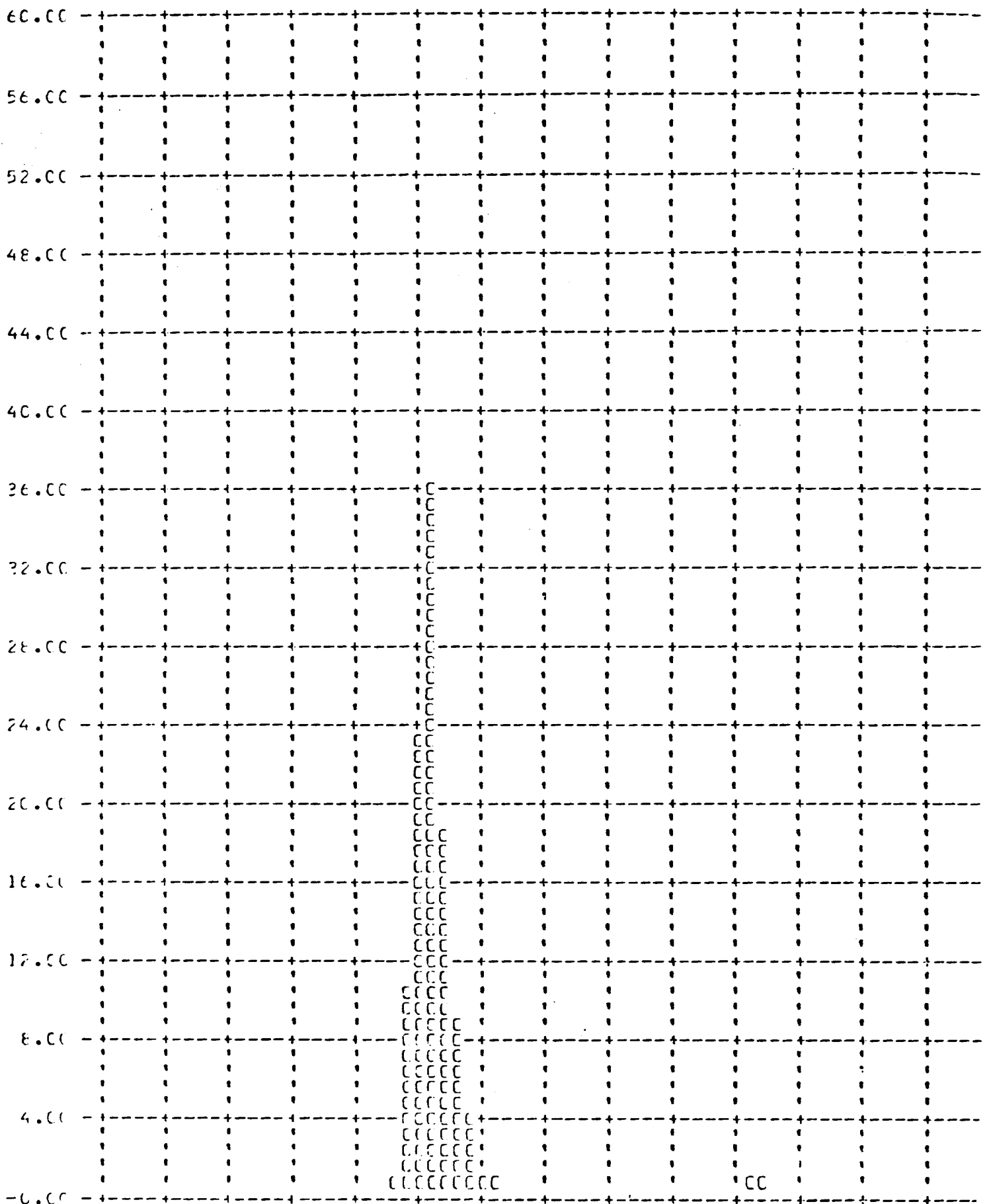
SEC 57



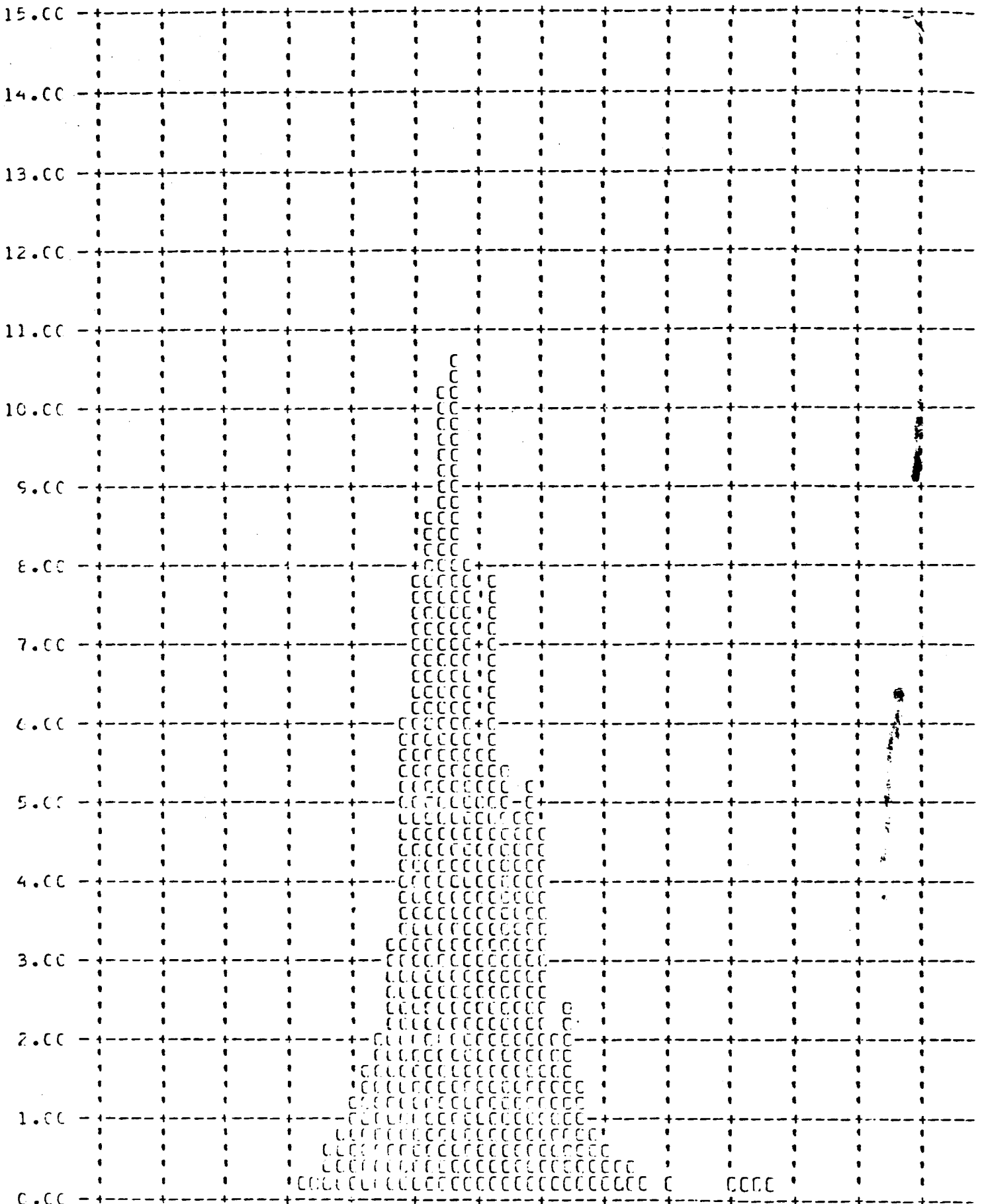
YAW ATTITUDE (DEGREES FROM INERTIAL AZIMUTH)

FIGURE 7-1

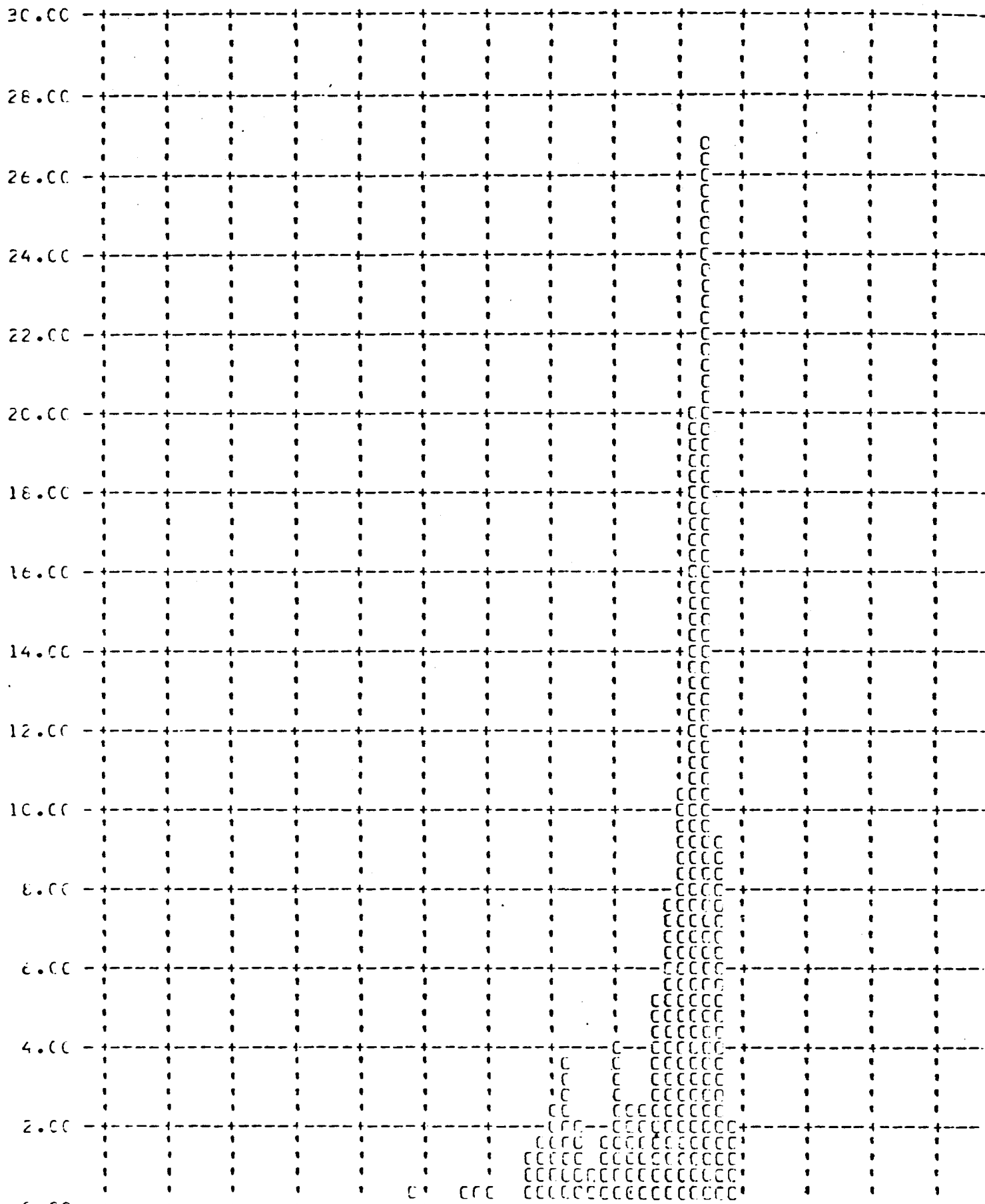
Y PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



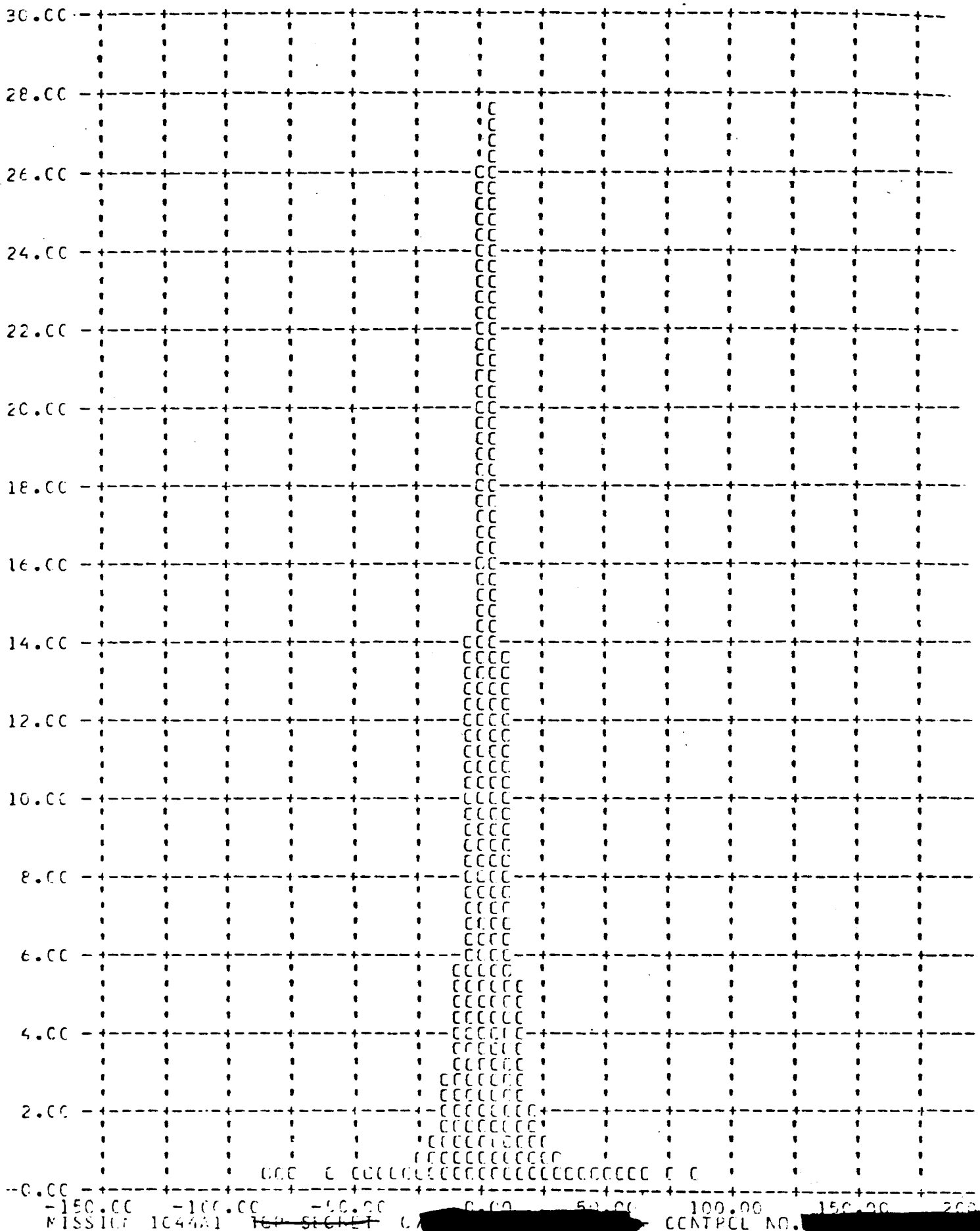
Y ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



Y YAW ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



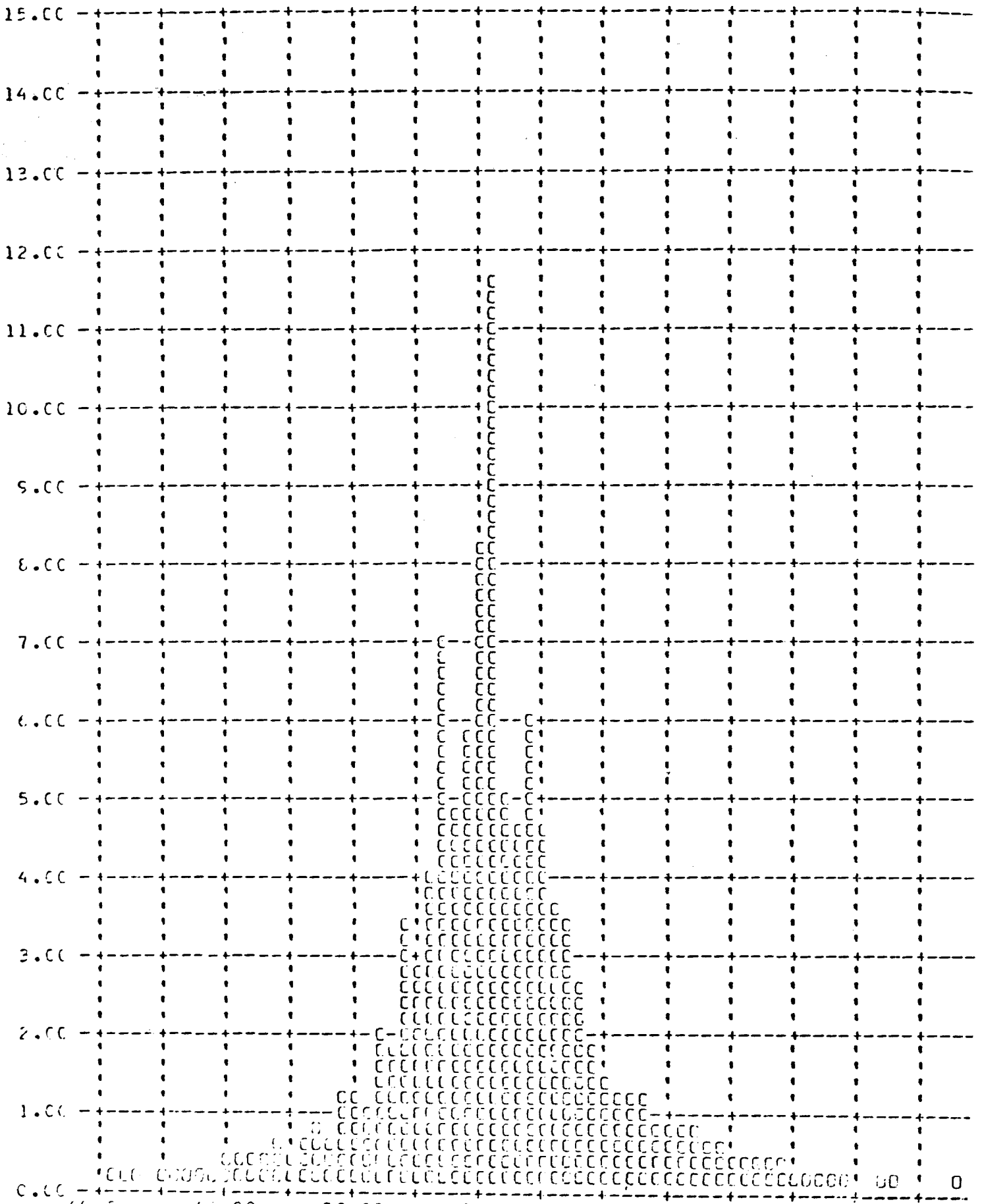
Y PITCH RATE ERROR - DEG/HCLR (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1044A1 169 SECRET [REDACTED] CONTROL NO. [REDACTED]

Figure 7-5

Y ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



YAW RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)

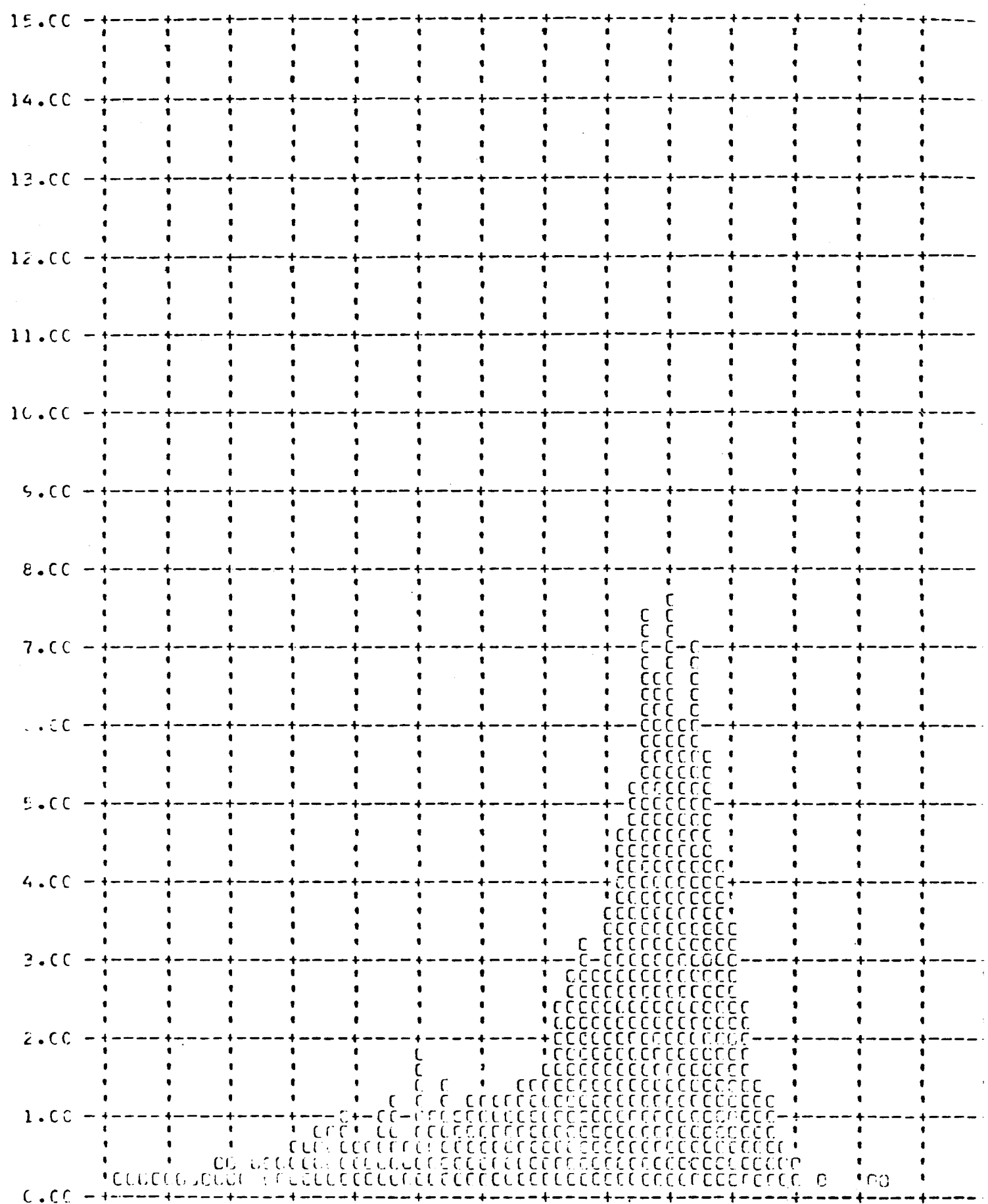
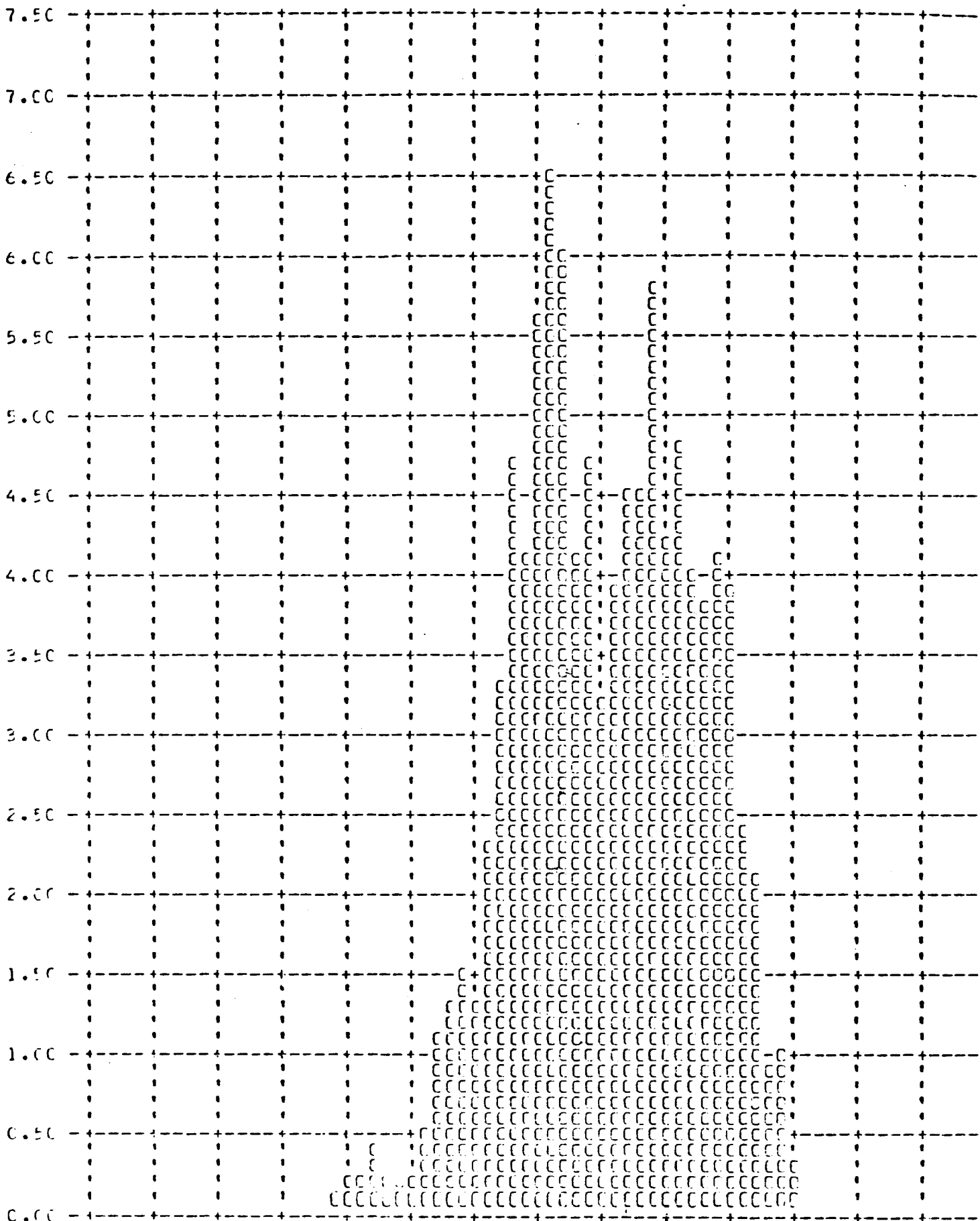


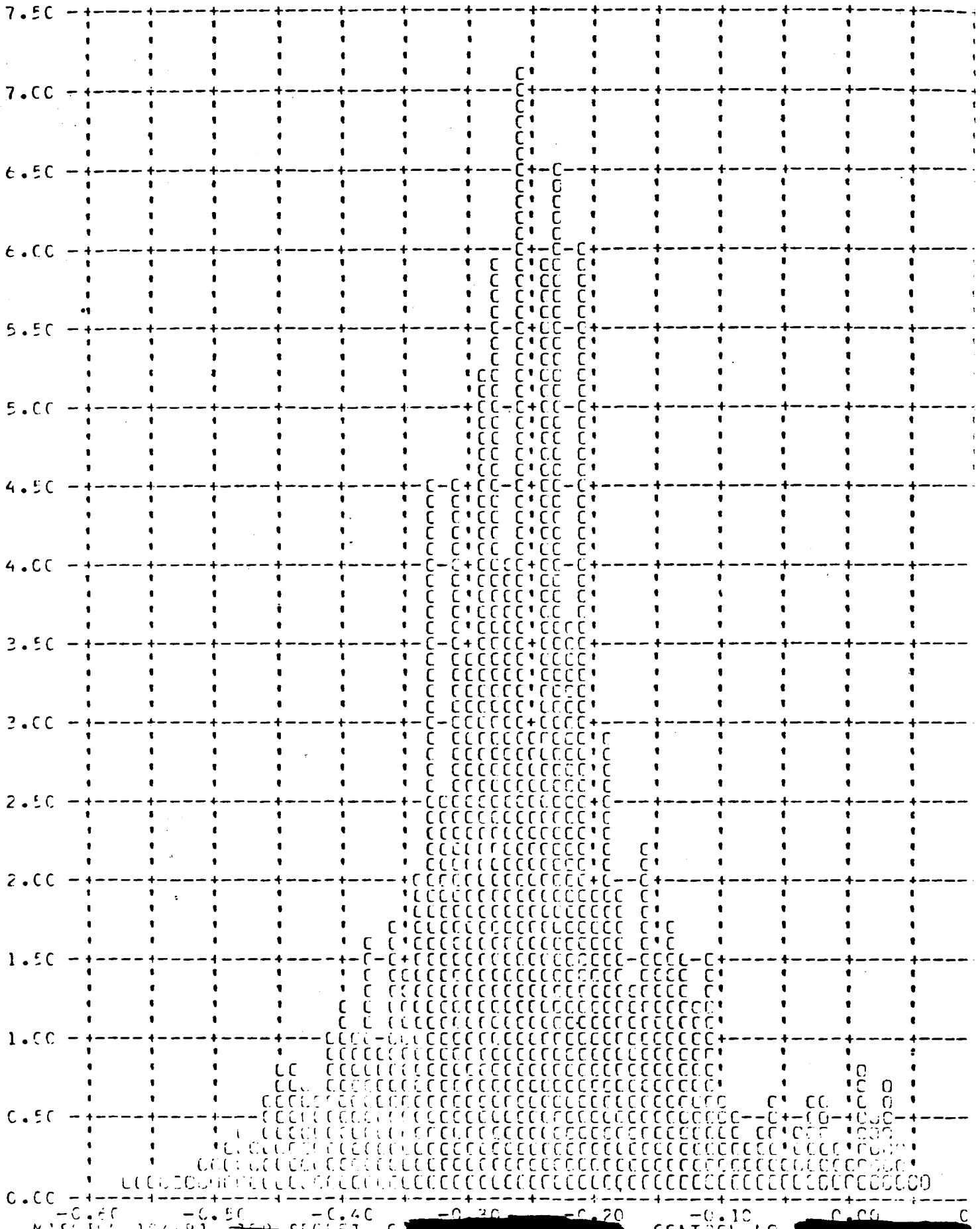
Figure 7-7

Y PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)

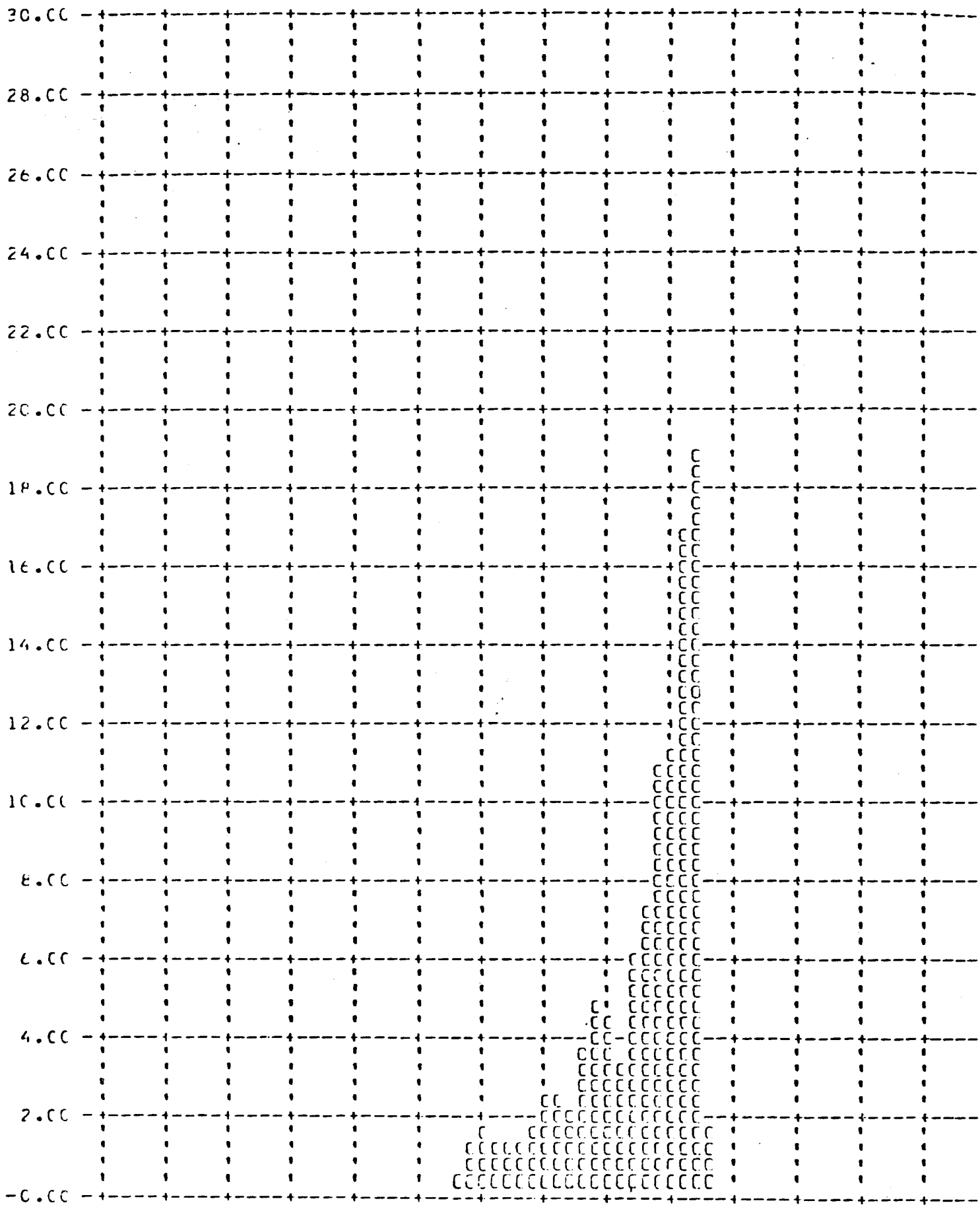




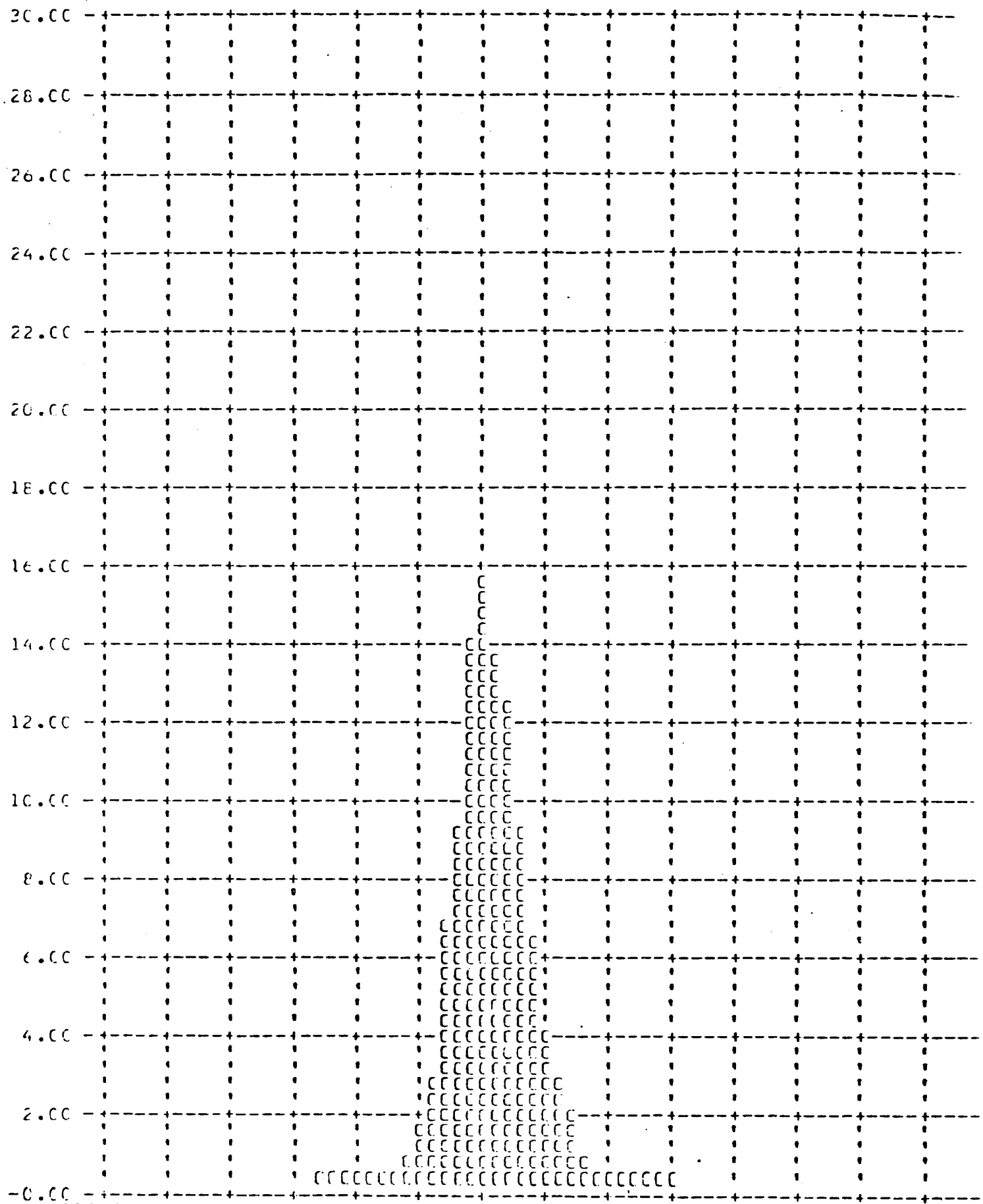
Y ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



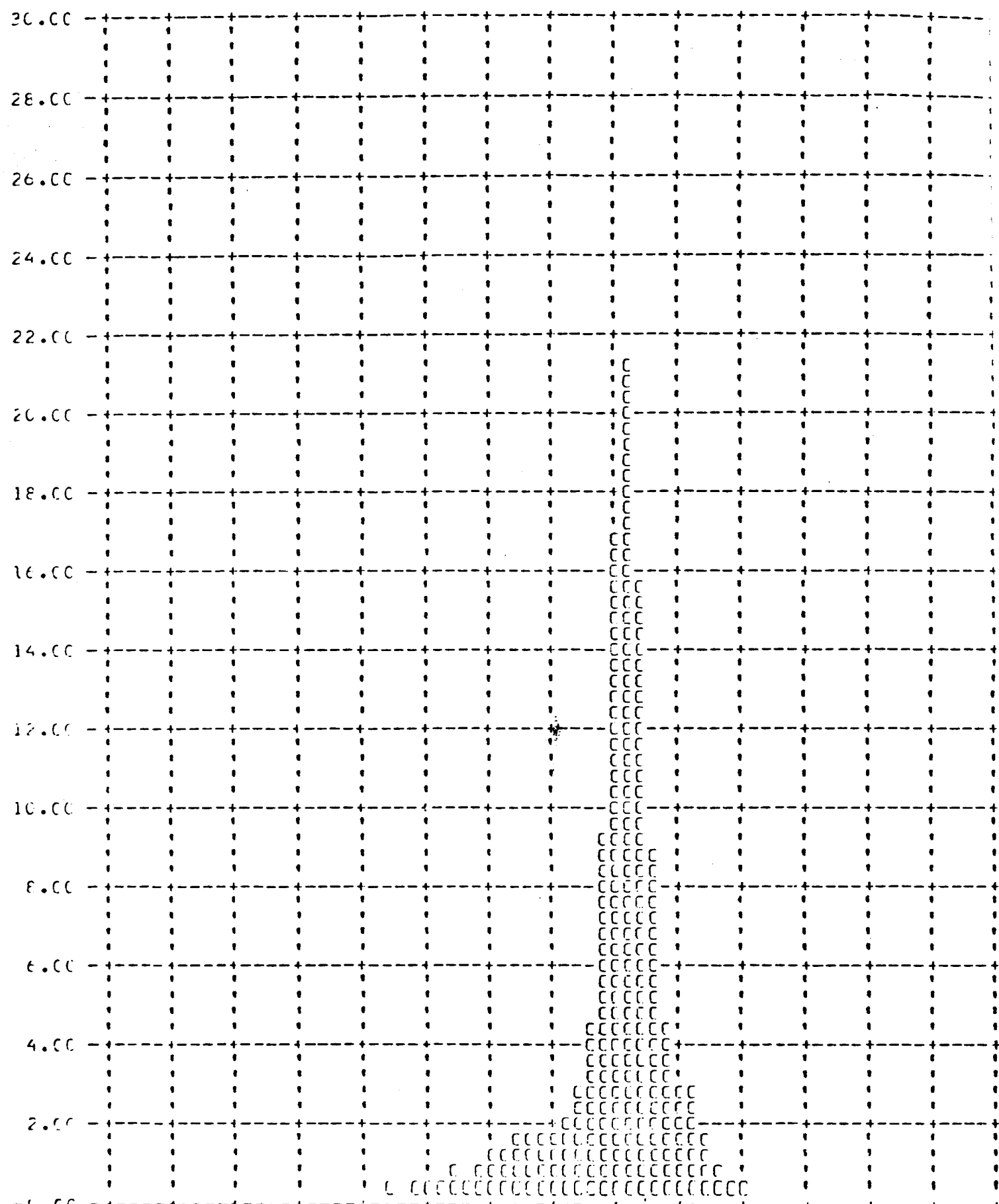
Y YAW ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)



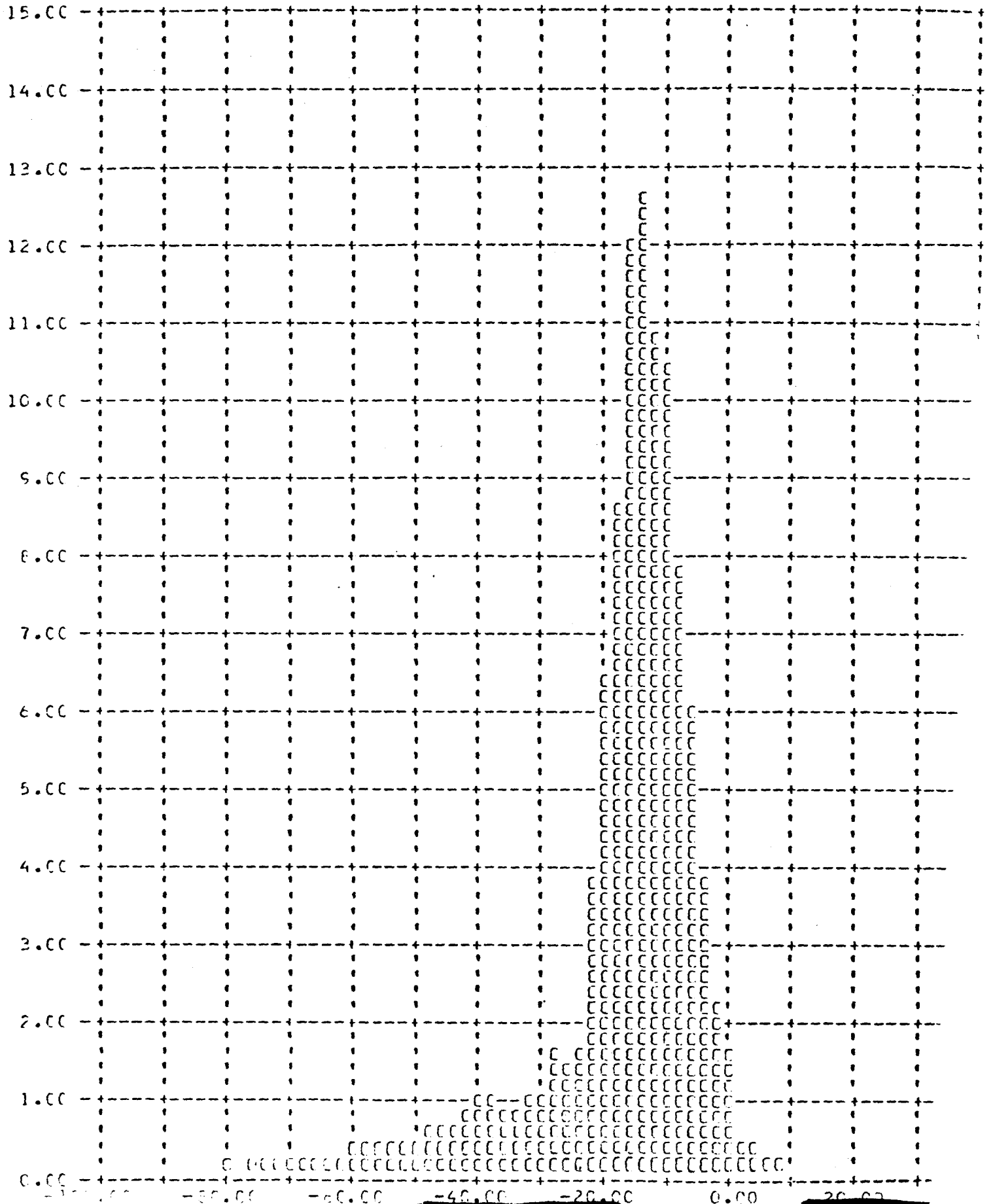
Y PITCH RATE ERROR - DEG/HCLR (X) VERSUS FREQUENCY - PERCENT (Y)



Y ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



Y YAW RATE ERROR - DEG/HCLR (X) VERSUS FREQUENCY - PERCENT (Y)





SECTION 8

IMAGE SMEAR ANALYSIS

The frame correlation tape supplied to A/P by NPIC contains the binary time word of each frame of photography. A computer program has been assembled at A/P which calculates the exposure time of each frame and compares the camera cycle rate with the ephemeris to calculate the V/h mismatch (Section 3), which is then combined with the vehicle attitude error and rate values of each frame and the crab error caused by earth rotation at the latitude of each frame. The program outputs the net IMC error and the total along track and cross track limit of ground resolution that can be acquired by a camera regardless of focal length and system capabilities.

The computer rejects the first six frames of all operations as the large V/h error induced by camera start-up is not representative of the overall system operations. The frequency distribution of the IMC errors and resolution limits are computer plotted and are shown in Figures 8-1 through 8-12.

The summary table 8-1 presents the maximum IMC errors and resolution limits that existed during 90% of the photographic operations and the total range of values during all operations that were computed.



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The relatively high values obtained in Mission 1044 reflect the combined effects of imperfect V/h and yaw steering matching as discussed in Sections 3 and 7. The apparent discrepancy in resolution limit values between the forward and aft-locking instruments is, in reality, a dramatic illustration of the relative influence of the difference in exposure time when coupled with smear contributing V/h and attitude errors.

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MISSION 1044

IMC RATIO AND RESOLUTION LIMITS

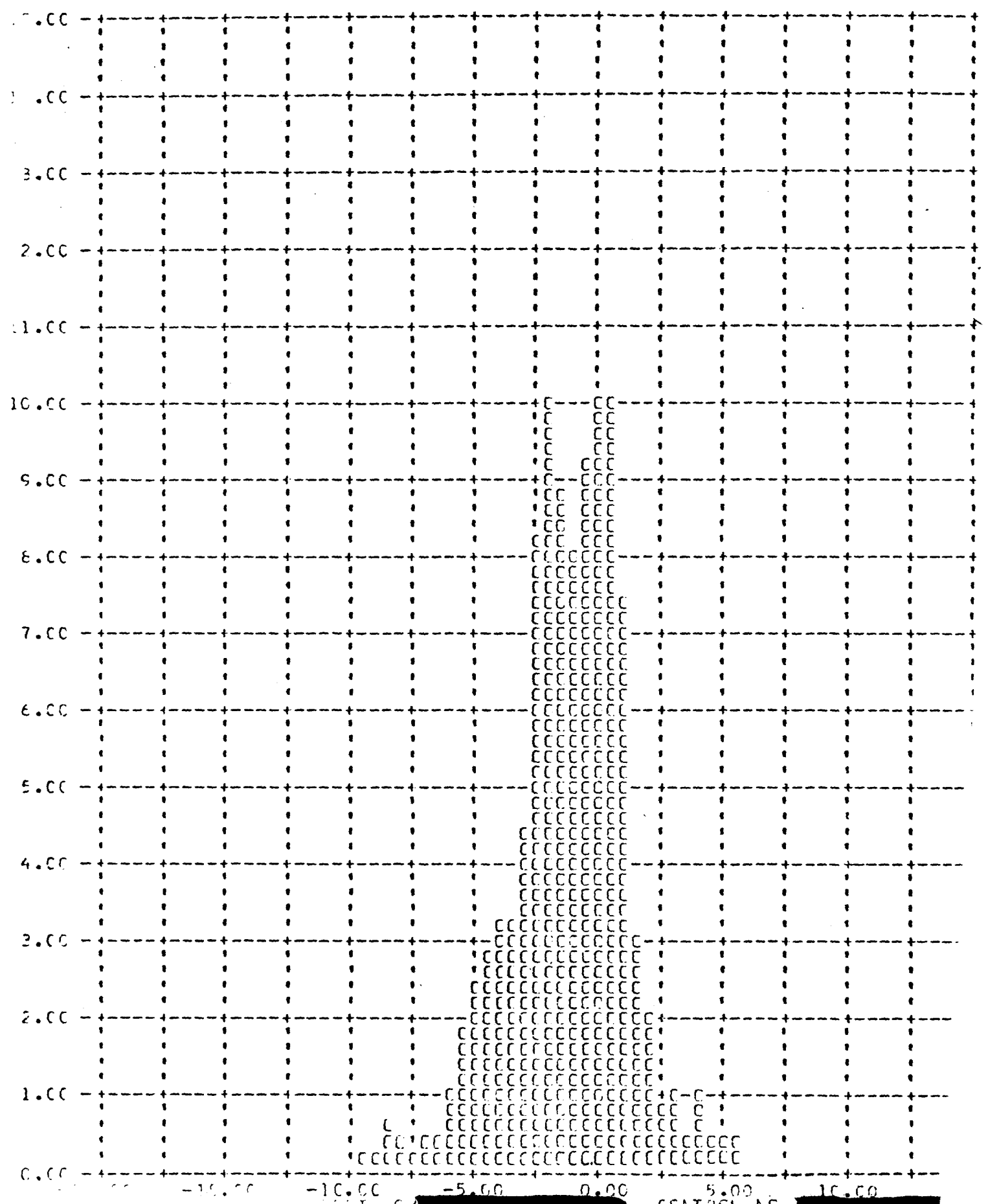
<u>VALUE</u>	<u>UNITS</u>	<u>CAMERA</u>	<u>MISSION 1044-1</u>		<u>MISSION 1044-2</u>	
			<u>90%</u>	<u>RANGE</u>	<u>90%</u>	<u>RANGE</u>
IMC Ratio Error	%	FWD	4.55	-9.5 to +5.5	3.21	-5.4 to +6.0
		AFT	4.06	-10.0 to +6.5	3.26	-5.6 to +6.2
Along Track Resolution Limit	Feet	FWD	6.98	0.2 to 13.6	4.38	0.2 to 8.8
		AFT	4.36	0.2 to 9.8	3.25	0.2 to 6.6
Cross Track Resolution Limit	Feet	FWD	9.75	0.2 to 11.0	8.39	0.2 to 10.4
		AFT	6.19	0.2 to 7.6	5.30	0.2 to 6.0

TABLE 8-1

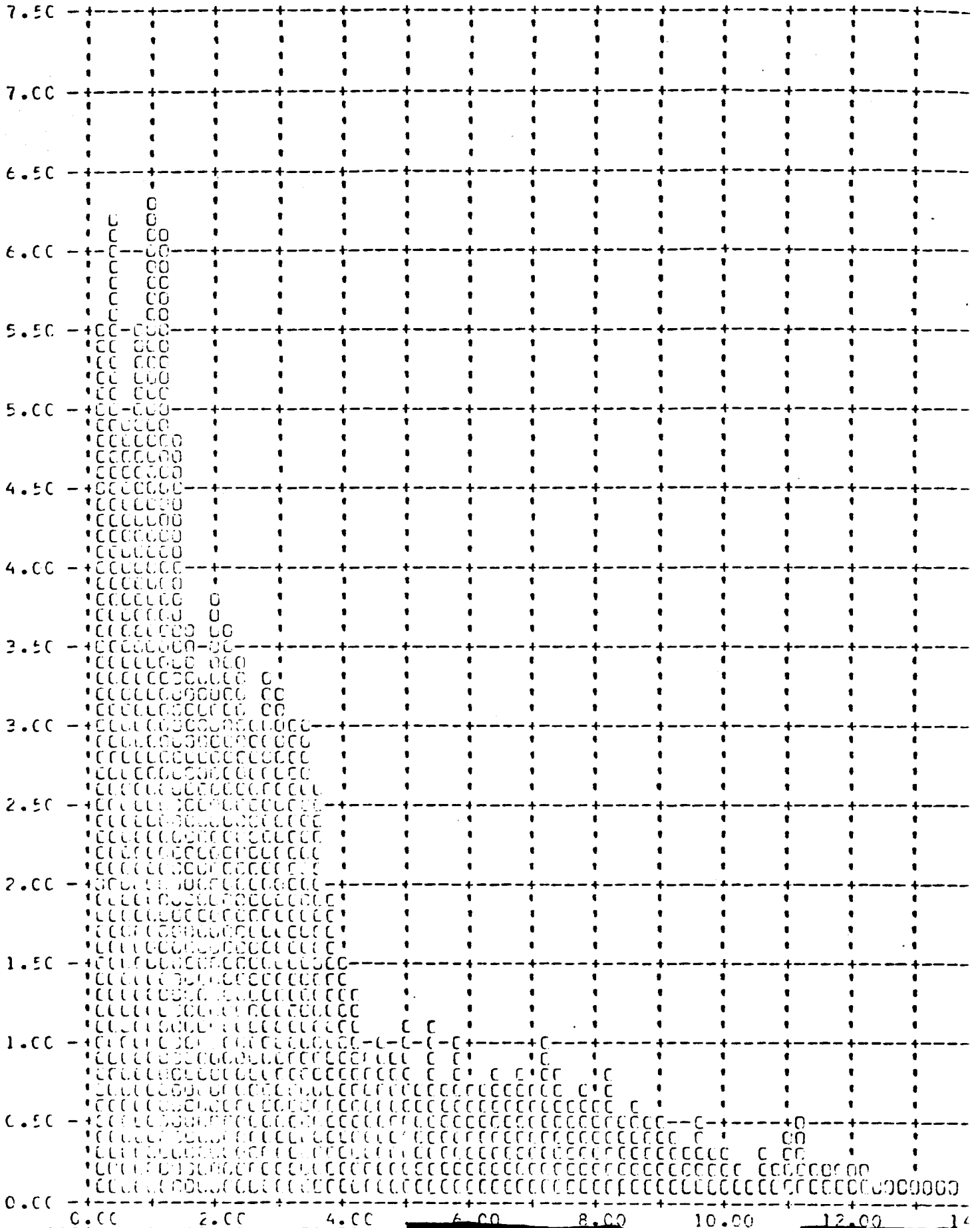
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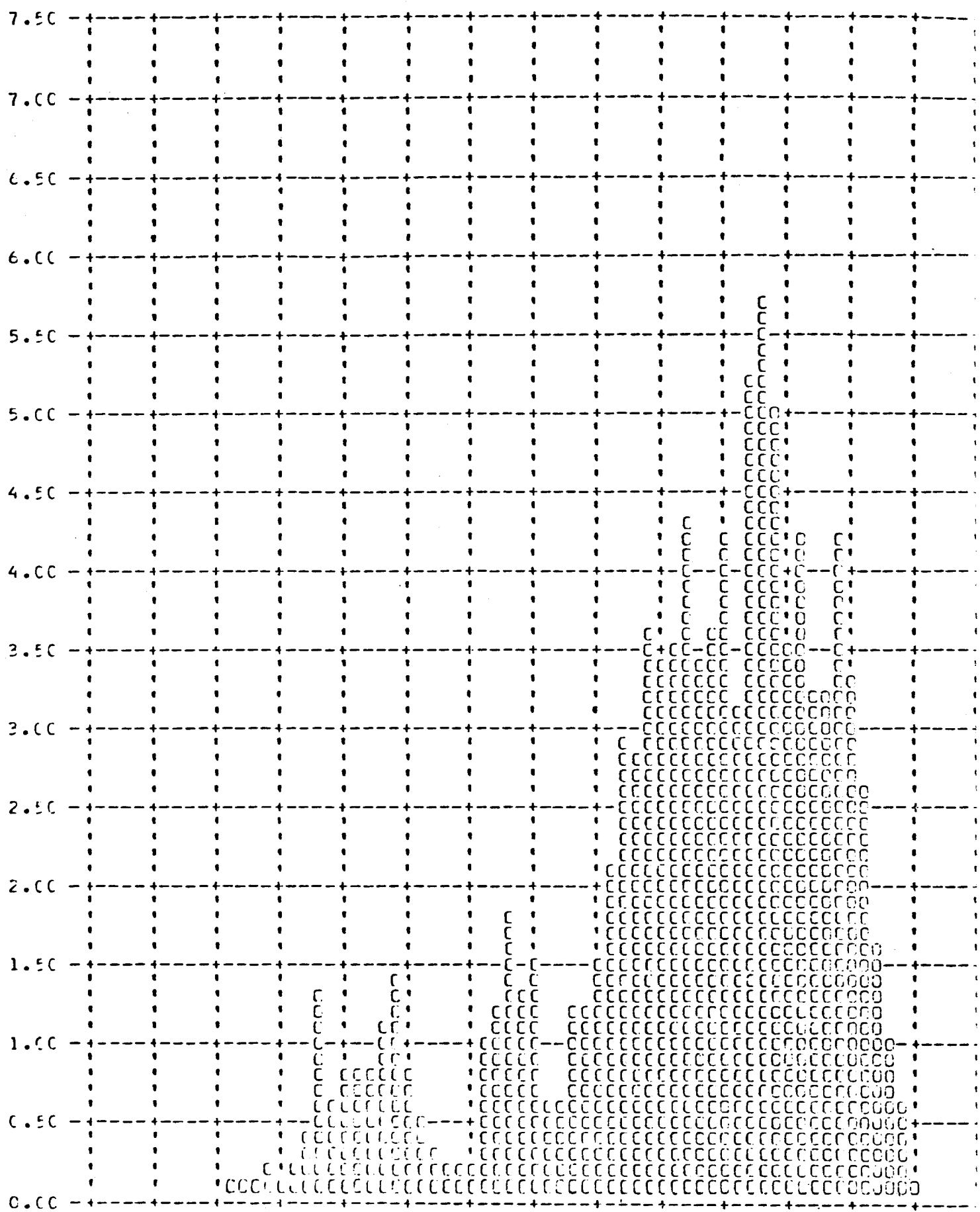
Y IMC ERROR -- PERCENT (X) VERSUS FREQUENCY -- PERCENT (Y)



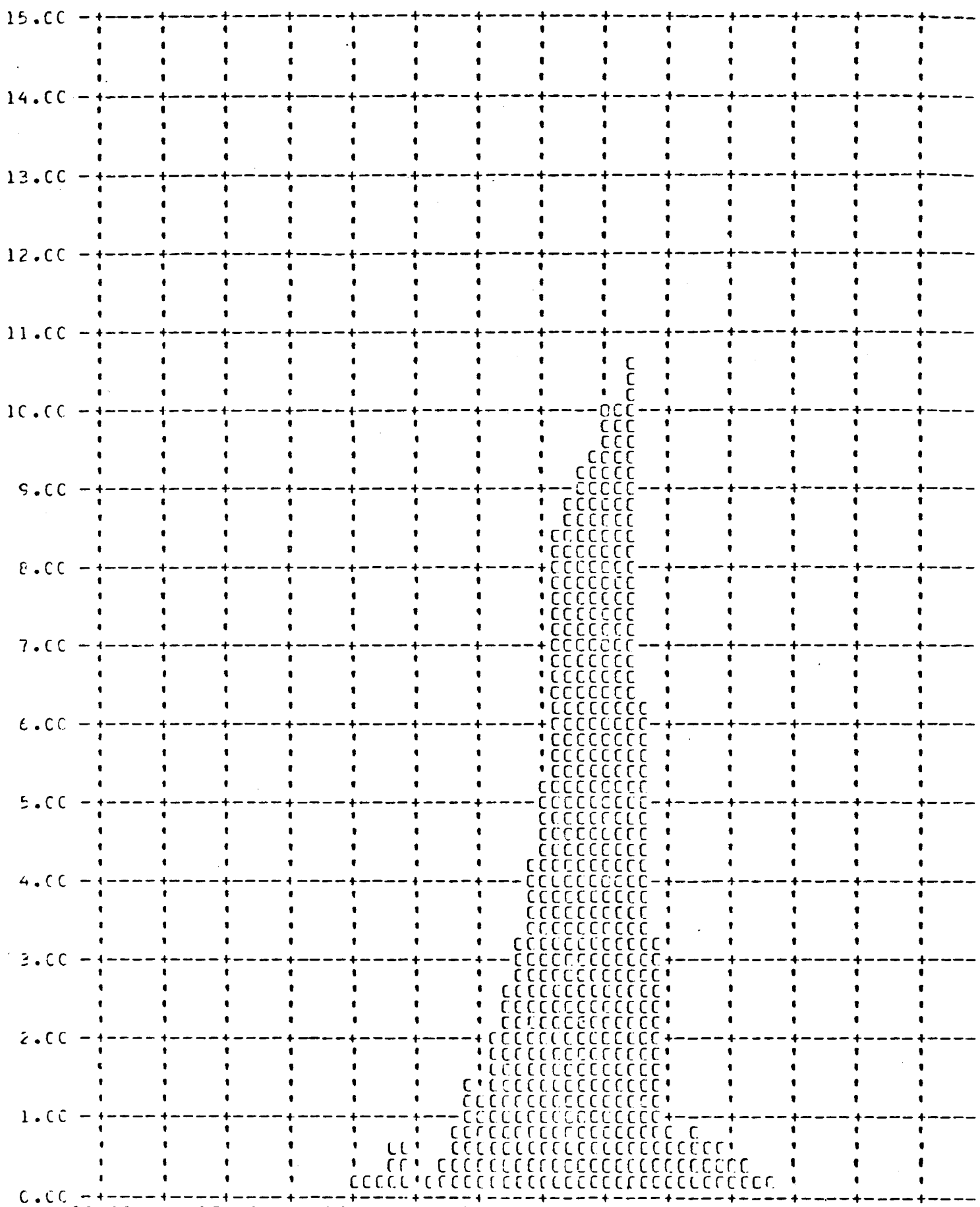
Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (



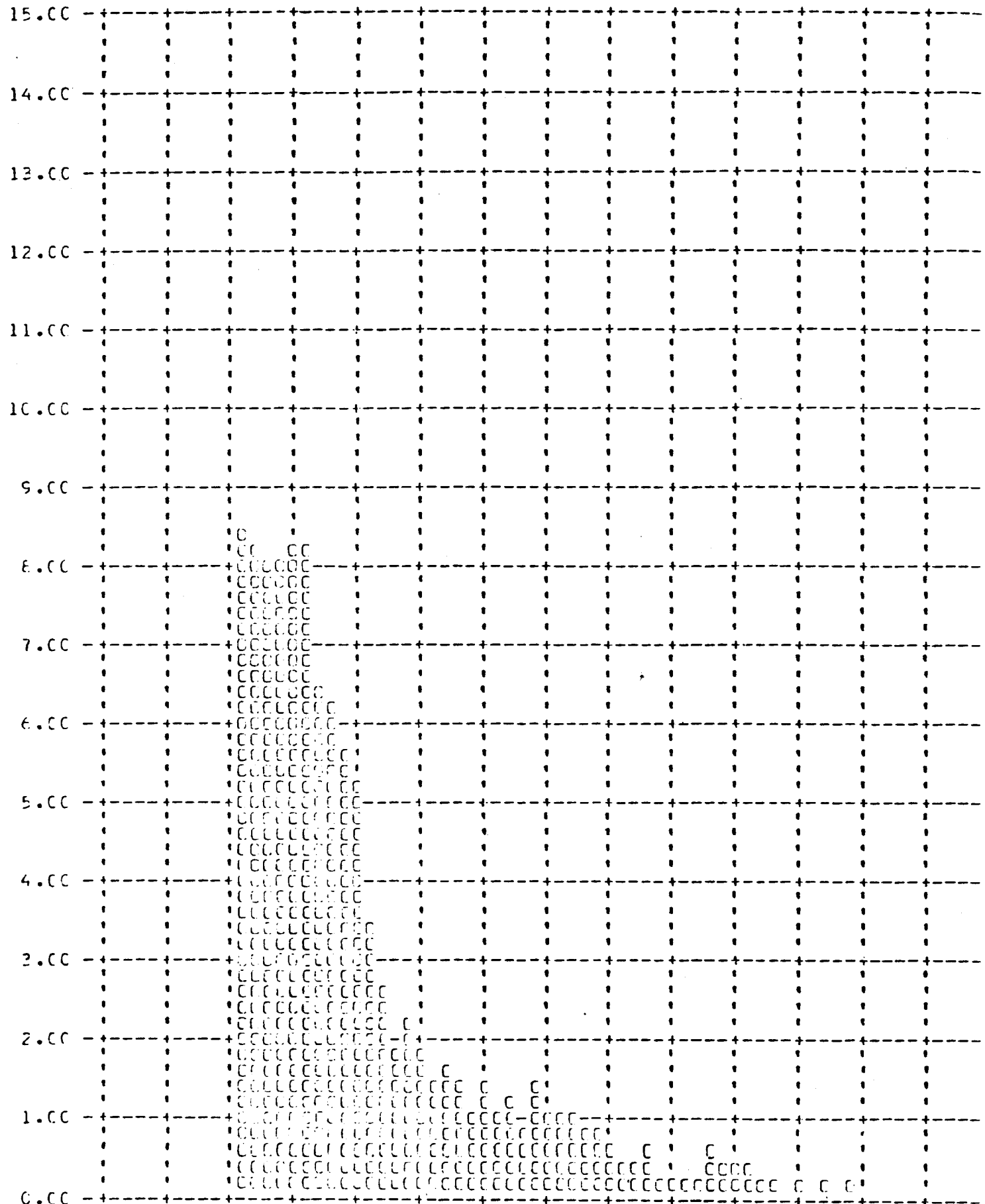
Y CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (X)



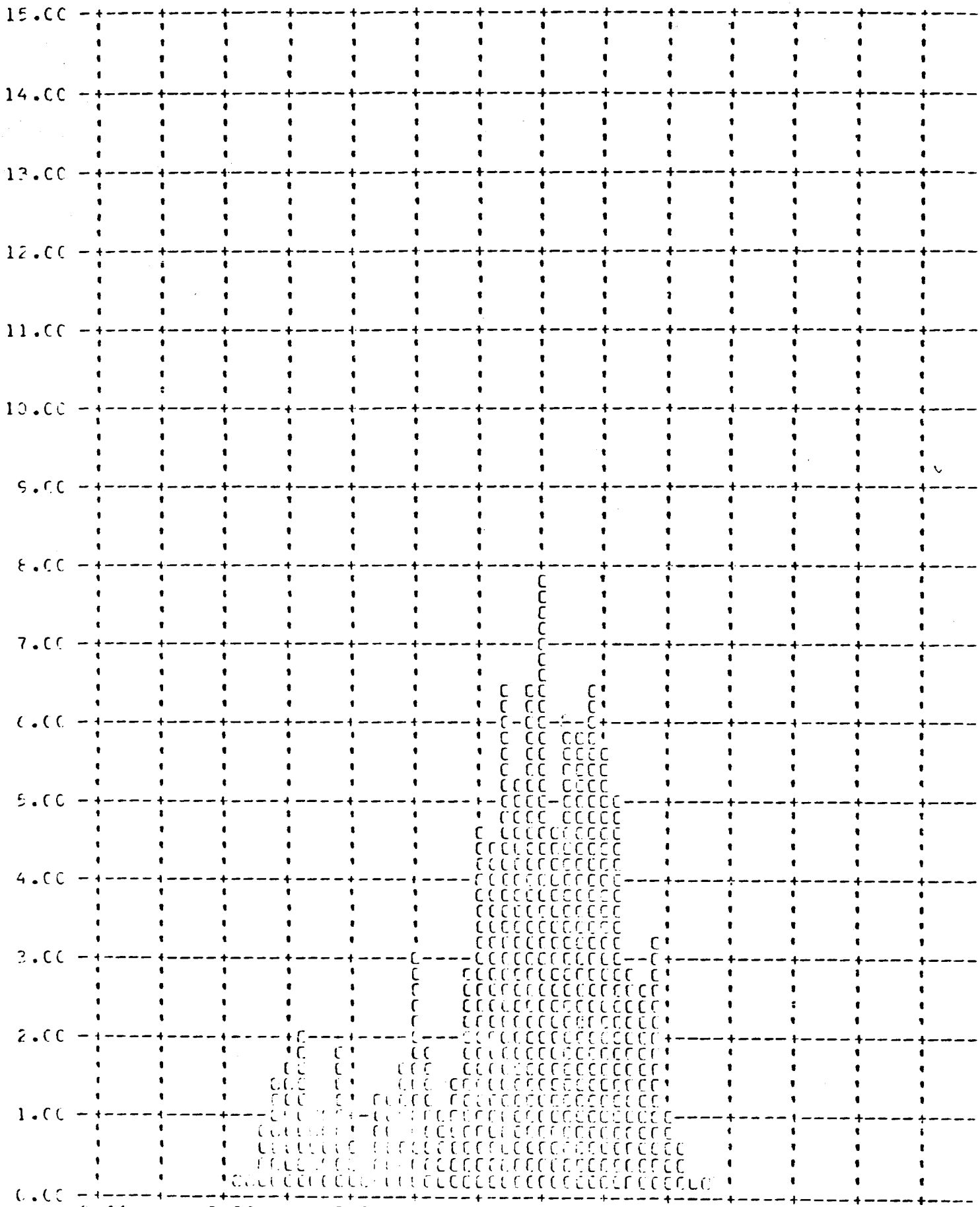
Y IMC ERROR -- PERCENT (X) VERSUS FREQUENCY -- PERCENT (Y)



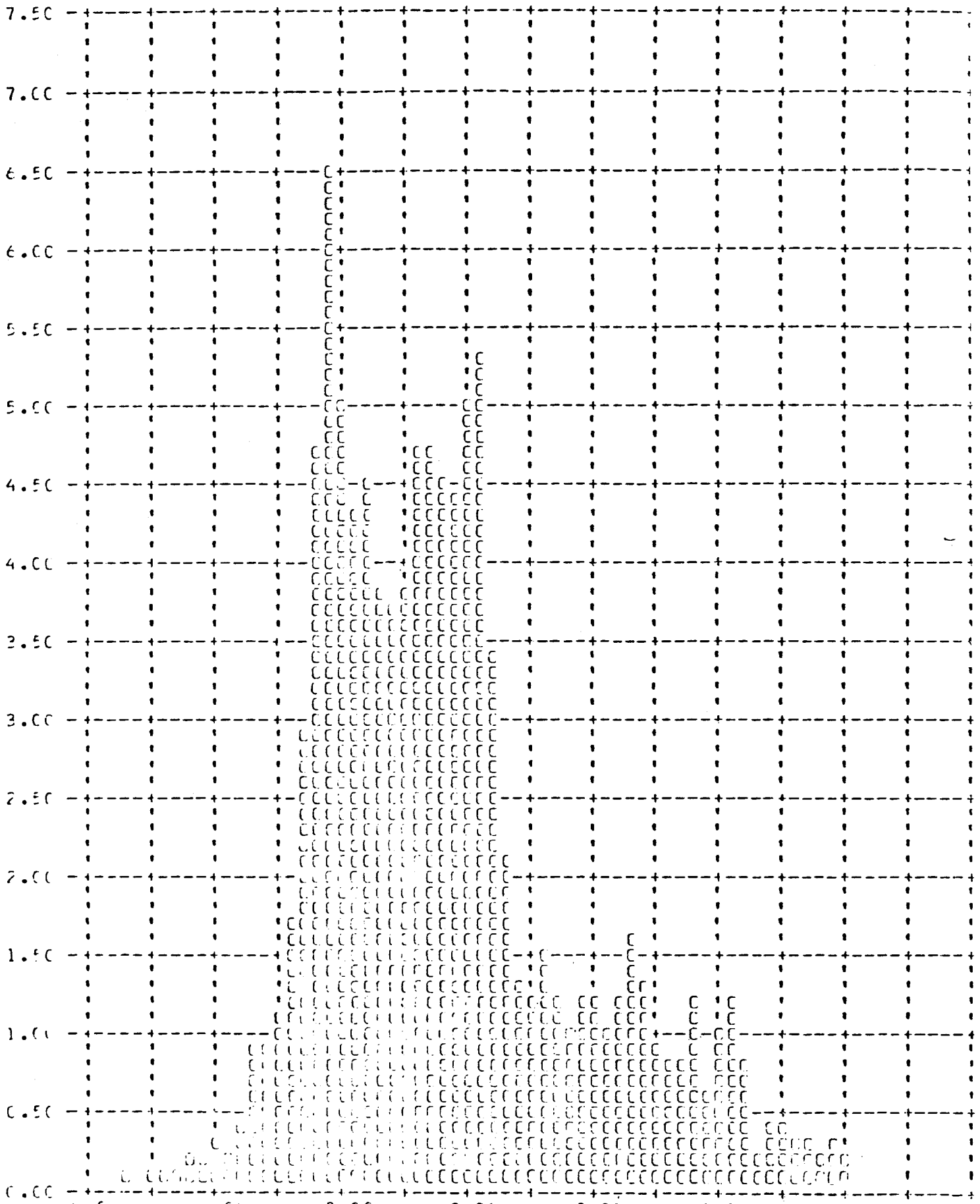
Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (



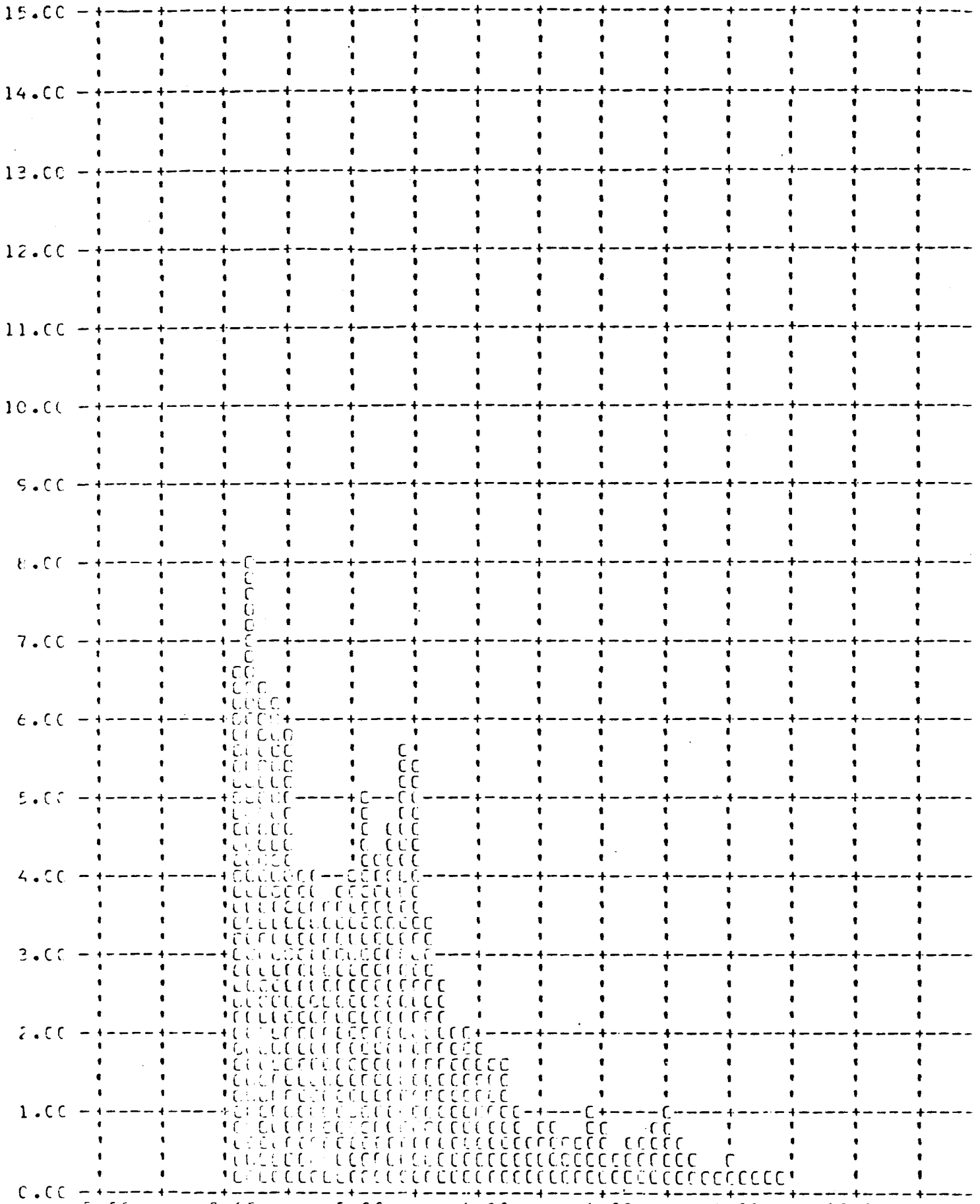
Y CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (



Y IFC ERROR -- PERCENT (X) VERSUS FREQUENCY -- PERCENT (Y)

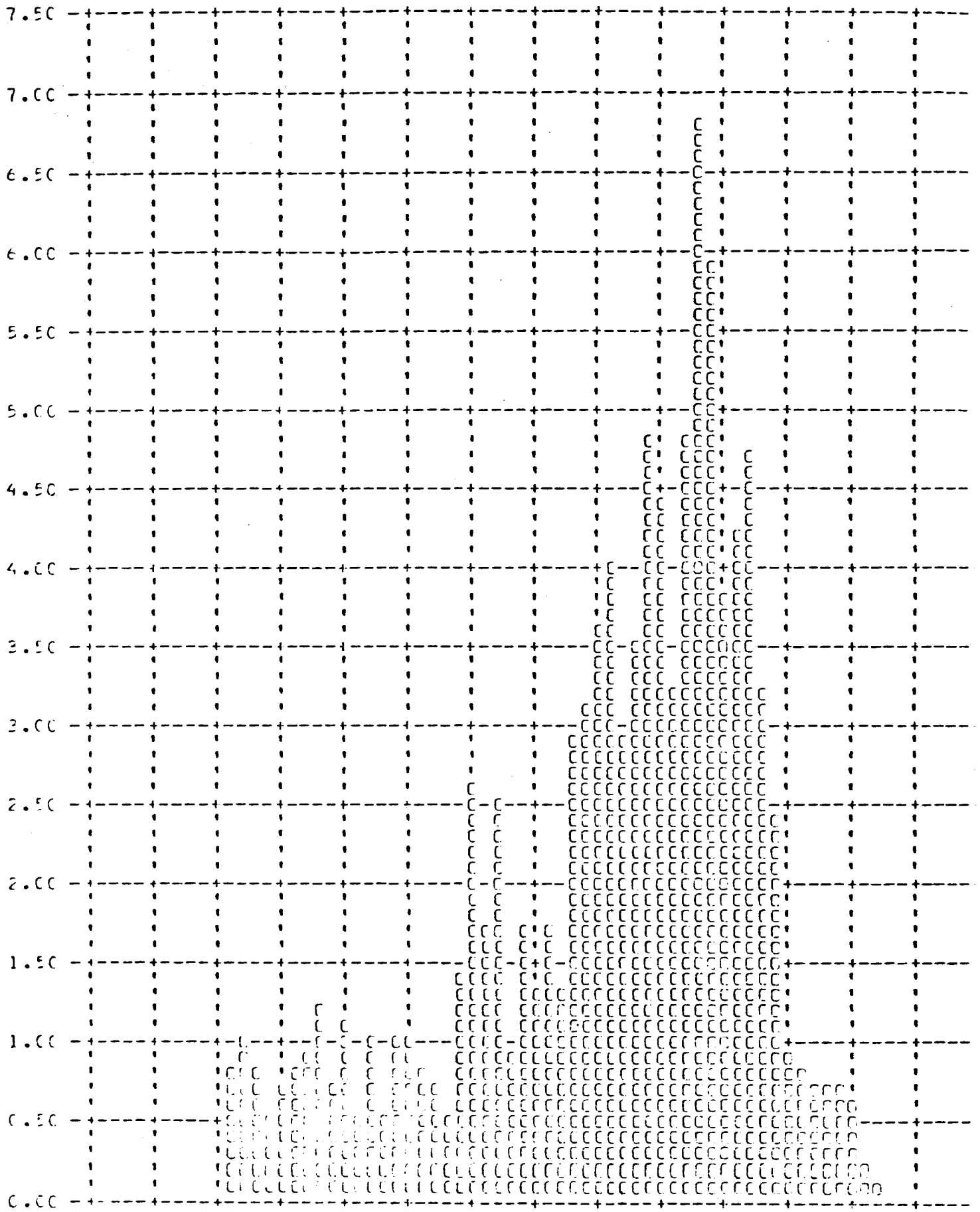


Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (

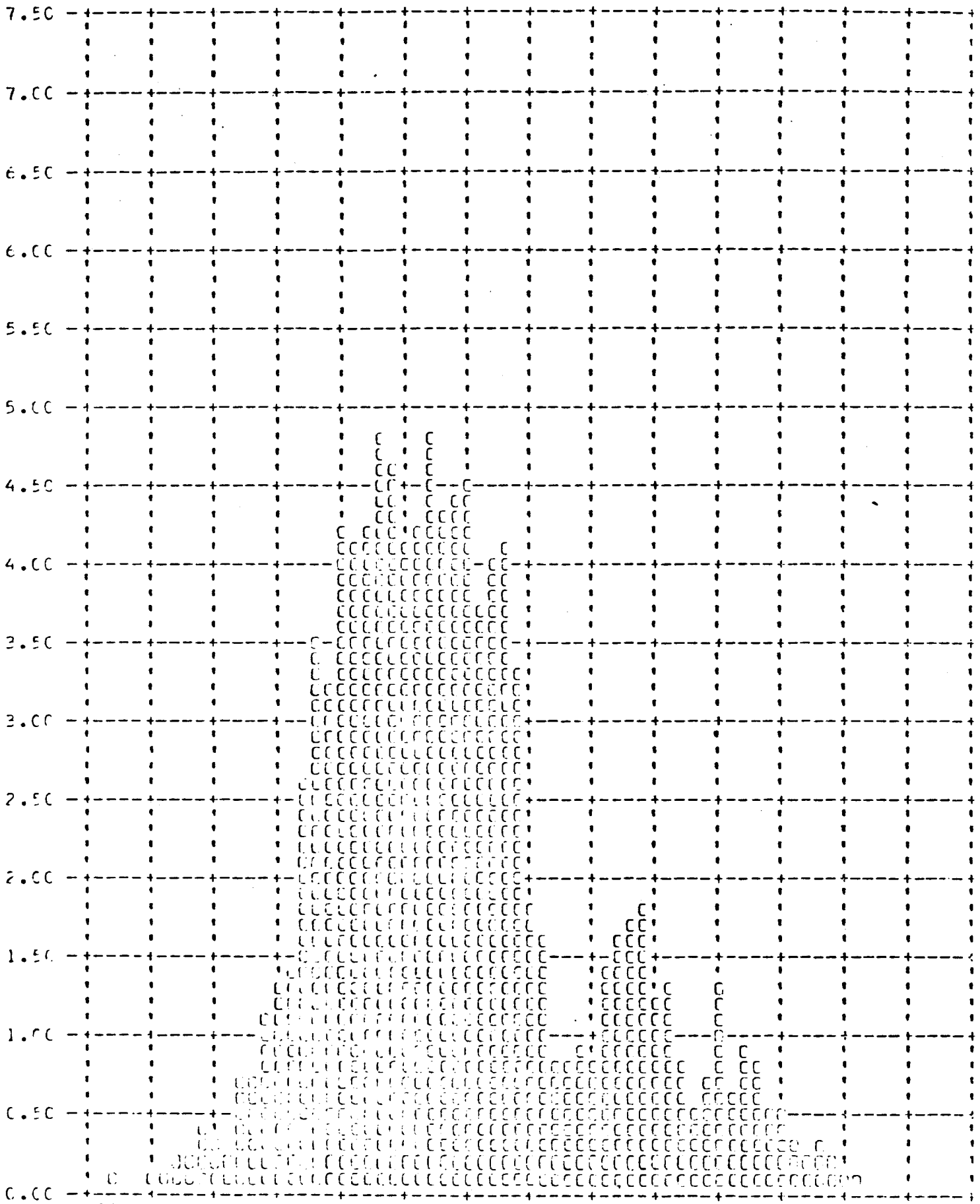




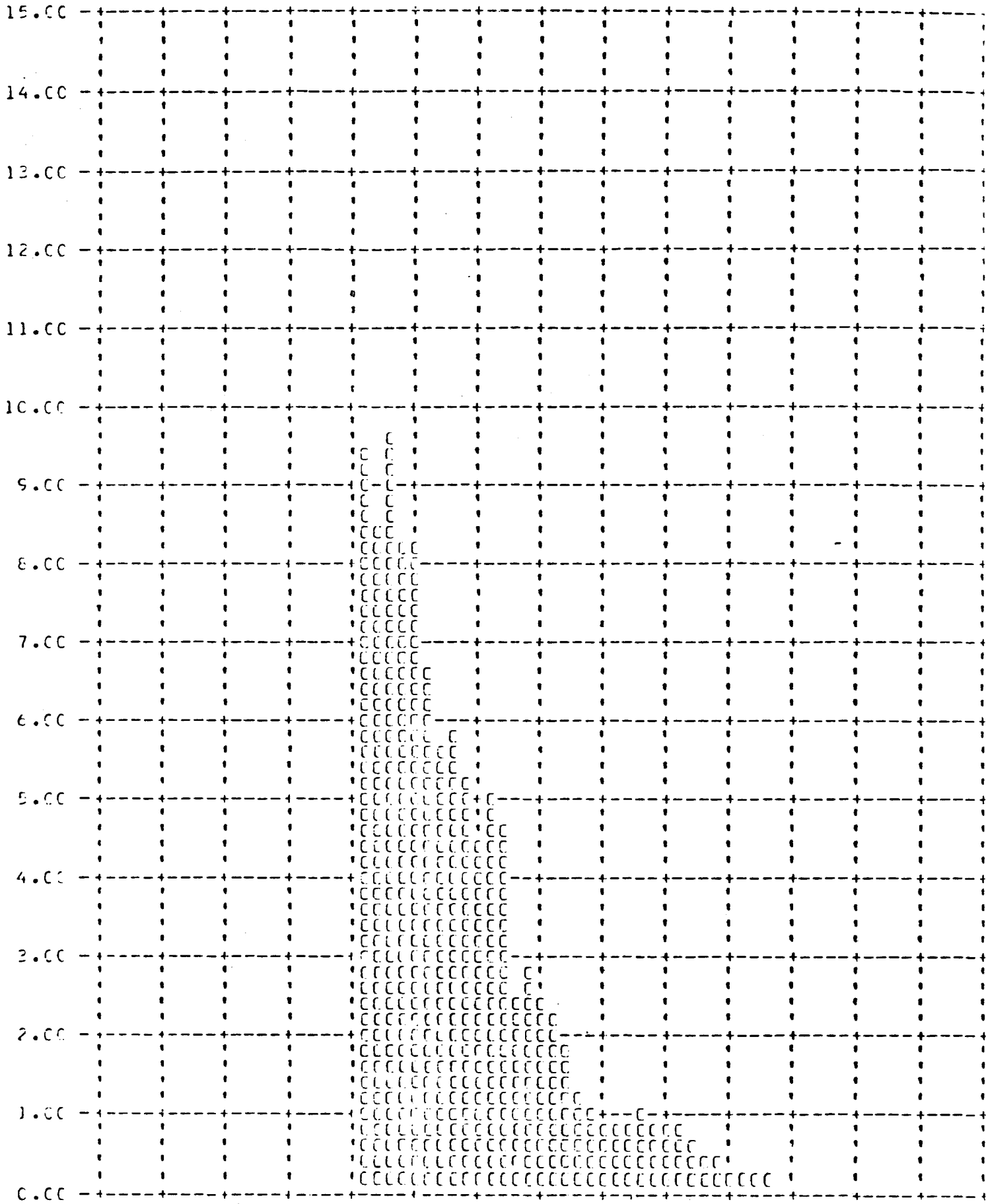
Y CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (



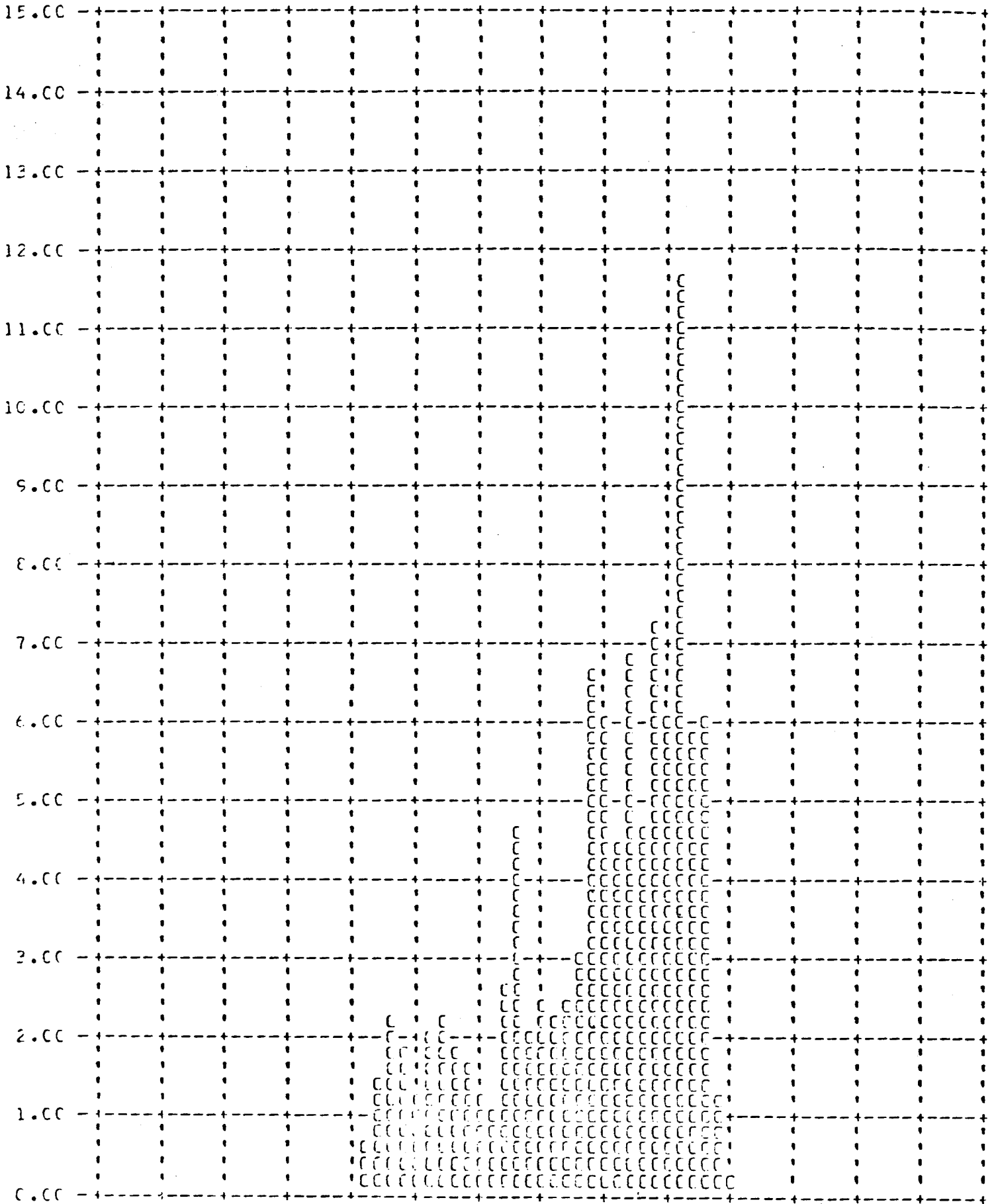
Y IMC ERROR -- PERCENT (X) VERSUS FREQUENCY -- PERCENT (Y)



Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)



Y CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (Y)





SECTION 9

SYSTEM RELIABILITY

Reliability calculations for the payload are based on a sample beginning with M-7. Hence both the major part of the Mural program and the "J" program are covered in the calculation. For certain auxiliaries, i.e., the stellar-index camera and the horizon cameras, the sample size is changed to recognize incorporation of modified equipment or new designs where reliability was one of the principal reasons for the modification. However, for primary mission function, the sample size is consistent with reliability reporting for the vehicle.

The reliability estimates of this section deal exclusively with the payload. Failures to achieve orbit or vehicle induced failures are thereby excluded. Recoveries before a complete mission has been completed are considered as full missions providing that early termination was caused by reasons not connected with payload operation. Film quality is not considered in the reliability estimate calculation. Hence, only electrical and mechanical functioning are considered.

The reliability estimate is also divided into primary and secondary functions. The primary functions are operation of the panoramic cameras, main camera door operation, operation of the payload clock, and recovery operations. The secondary mission functions are horizon camera operation excluding catastrophic open shutter failure mode, auxiliary data recording, and stellar-index camera operation. A summary of estimated reliability is shown in Table 9-1.



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Panoramic Camera Reliability

Sample Size - 195 opportunities to operate.

Two failures - S/I Programmer on System J-19

Film Transport on System J-42

Assume - 3000 cycles per camera per mission.

Estimated Reliability = 98.6 at 50% confidence level

Main Camera Door Reliability

Sample Size - 62 vehicles x 3 doors = 124 opportunities to operate

Estimated Reliability = 99.5 at 50% confidence level.

Payload Command and Control

Sample Size - 11,424 hours operation in sample

Two failures

Estimated Reliability = 96.1% at 50% confidence level

Payload Clock Reliability

Sample Size - 11,424 hours operation in sample

No failures

Estimated Reliability - 99.0% at 50% confidence level

Estimated Reliability of Payload Functioning on orbit = 96.5% at

50% confidence level

Recovery System Reliability

29 opportunities to recover

1 failure - improper separation due to water seal - cutter failure

Estimated Reliability - 98.2% at 50% confidence level



Stellar-Index Camera Reliability

Sample begins with J5 (Does not include DISIC units in 1100 series systems)

Sample size = 28,480 cycles

Four failures

Estimated Reliability = 93.3% at 50% confidence level.

Horizon Camera Reliability

Sample begins with J5 - 115,000

Estimated Reliability of Single Camera - 99.1% at 50% confidence level

Estimated Reliability of Four Horizon Cameras at a Parallel Redundant System = 99.9% at 50% confidence level.



# ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS						ON-OBIT FUNCTIONS			RECOVERY SYSTEM			SECONDARY FUNCTIONS		
	PANORAMIC CAMERA FAILURES	PANORAMIC CAMERA DOORS FAILURES	PANORAMIC CAMERA COMMAND & CONTROL SYSTEM FAILURES	PAYLOAD CLOCK FAILURES	ON-OBIT FUNCTIONS RELIABILITY	RECOVERY SYSTEM SAMPLE CAPSULES	STELLAR - INDEX CAMERAS SAMPLE	HORIZON CAMERAS SAMPLE	PANORAMIC CAMERA FAILURES	PANORAMIC CAMERA DOORS FAILURES	PANORAMIC CAMERA COMMAND & CONTROL SYSTEM FAILURES	PAYLOAD CLOCK FAILURES	ON-OBIT FUNCTIONS RELIABILITY	RECOVERY SYSTEM FAILURES	STELLAR - INDEX CAMERAS FAILURES
9038 to 1008	60	52	3124	3124	96.1	10	3400	12,000	0	0	0	96.1	1	3	0
1009	64	54	3216	3216	96.2	20	4250	15,000	0	0	0	96.2	1	3	0
1010	60	56	3432	3432	96.4	22	5100	10,000	0	0	0	96.4	1	3	0
1011	72	58	3600	3600	96.8	24	5525	21,000	0	0	0	96.8	1	0*	0
1012	76	60	3720	3720	96.9	26	5525	24,000	0	0	0	96.9	1	0	0
1013	78	62	3940	3940	96.0	20	5950	25,500	0	0	0	96.0	1	0	0
1014	82	64	4056	4056	96.1	30	6375	28,500	0	0	0	96.1	1	1	0
1015	86	66	4320	4320	96.1	32	7225	31,500	0	0	0	96.1	1	1	0
1016	90	68	4560	4560	96.4	34	7650	34,500	0	0	0	96.4	1	1	0
1017	94	70	4760	4760	97.6	36	10925	37,500	0	0	0	97.6	1	1	0
1018	98	72	4920	4920	96.7	30	8900	40,500	0	0	0	96.7	1	1	0
1019	102	74	5136	5136	96.8	39	8075**	43,500	0	0	0	96.8	1	1	0

\* DESIGN FIX NEGATED PREVIOUS FAILURE CONSIDERATIONS

\*\* 1019 SAMPLE OUT OF SEQUENCE

TABLE 9-1



# ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS						SECONDARY FUNCTIONS														
	PANORAMIC CAMERA		PANORAMIC CAMERA DOORS		COMMAND & CONTROL SYSTEM		PAYLOAD CLOCK		ON-ORBIT FUNCTIONS		RECOVERY SYSTEM		STELLAR-INDEX CAMERAS		HORIZON CAMERAS						
	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY	SAMPLE FAILURES	RELIABILITY					
1020	108	1	98.5	78	0	99.1	5544	1	97.1	5544	0	98.9	43	1	96.1	10,580	2	69.9	48,000	0	97.9
1021	104	1	98.5	76	0	99.1	5376	1	97.0	5376	0	98.8	41	1	96.0	9830	2	89.1	46,500	0	97.8
1022	112	1	99.5	80	0	99.2	5784	1	97.3	5784	0	98.9	45	1	96.3	11,550	2	90.7	51,000	0	99.0
1023	114	1	90.6	82	0	99.2	6000	2	95.8	6000	0	98.9	47	1	96.5	12,190	2	91.1	54,000	0	99.1
1024	118	1	98.6	84	0	99.2	6240	2	96.0	6240	0	98.9	49	1	96.6	13,040	2	91.6	57,000	0	98.2
1025	122	1	98.6	86	0	99.2	6480	2	96.1	6480	0	99.0	51	1	96.7	13,890	2	92.1	60,000	0	98.3
1026	126	1	98.7	88	0	99.2	6720	2	96.3	6720	0	99.0	53	1	96.8	14,740	2	92.6	63,000	0	98.4
1027	128	1	98.7	90	0	99.2	6744	2	96.3	6744	0	99.0	55	1	97.0	15,165	3	90.0	64,500	0	99.4
1028	132	1	98.7	92	0	99.2	6960	2	96.4	6960	0	99.0	57	1	97.1	16,015	3	30.7	67,500	0	96.5
1029	136	1	98.8	94	0	99.3	7200	2	96.5	7200	0	99.1	59	1	97.1	16,580	4	58.7	70,500	0	98.5
1030	140	1	98.9	96	0	99.3	7440	2	96.6	7440	0	99.1	61	1	97.2	17,430	4	89.3	73,500	0	98.6
1031	143	1	98.8	98	0	99.3	7704	2	96.7	7704	0	99.1	63	1	97.3	18,280	4	89.7	76,500	0	98.6

# ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS				ON-DUTY FUNCTIONS RELIABILITY	SECONDARY FUNCTIONS			
	PARADEING CAMERA SAMPLE FAILURES RELIABILITY	PARADEING CAMERA DOWNS SAMPLE FAILURES RELIABILITY	COMMAND & CONTROL SYSTEM SAMPLE FAILURES RELIABILITY	PLAYBACK SUCCESS SAMPLE FAILURES RELIABILITY		RECOVERY SYSTEM SAMPLE FAILURES RELIABILITY	STELLAR INDEX CAMERAS SAMPLE FAILURES RELIABILITY	MISSION CAMERAS SAMPLE FAILURES RELIABILITY	
1033	47	100	7968	7968	57.1	65	19,150	19,150	79,500
1034	151	55.9	10208	560	99.2	67	19,080	19,080	82,500
1035	155	98.9	870	90.9	99.2	71	21,060	21,060	88,500
1036	159	59.0	8500	97.1	99.2	69	20,830	20,830	85,500
1037	163	166	9043	5048	97.4	73	22,530	22,530	91,500
1038	167	110	9336	9336	97.5	75	23,360	23,360	94,500
1039	171	112	9600	9600	97.5	77	24,230	24,230	97,500
1040	175	114	9840	9840	97.5	79	25,040	25,040	100,500
1041	179	116	10,176	10,176	97.6	81	25,930	25,930	103,500
1042	183	118	10,536	10,536	97.7	83	26,780	26,780	106,500
1043	187	120	10,896	10,896	97.2	85	27,530	27,530	109,000

TABLE 9-1

RELIABILITY CRITERIA UPDATED FOR SUBSEQUENT MISSIONS

# ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

FUNCTIONS	PRIMARY FUNCTIONS				SECONDARY FUNCTIONS			
	FUNCTIONS		FUNCTIONS		FUNCTIONS		FUNCTIONS	
	RELIABILITY	FAILURE RATE	RELIABILITY	FAILURE RATE	RELIABILITY	FAILURE RATE	RELIABILITY	FAILURE RATE
1.01	99.0	0	99.0	0	99.0	0	99.0	0
10.04	99.0	0	99.0	0	99.0	0	99.0	0

TABLE 9-1

\*\* CALCULATIONS ADJUSTED TO NOMINAL 14-DAY MISSION STANDARD  
 \*\* DISC REPLACES S/I CAMERAS ON 1100 SERIES SYSTEMS

SECTION 10

SUMMARY DATA

The comparison of the operating parameters and the performance achieved by previous missions has been difficult due to the large volume of data that results from each mission. Some of the pertinent characteristics from prior missions have been summarized in Tables 10-1 through 10-3.

The summary data was started with Mission 1004 as the J-05 camera system was the first to incorporate the major modifications of the titanium drum and scan arm, four roller scan head and Corona J capabilities. Only those missions that culminated in the recovery of some photography have been listed, therefore Missions 1003, 1005 and 1032 are deleted.

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# MISSION SUMMARY

MISSION NUMBER	PAVLOD NUMBER	VEHICLE NUMBER	LAUNCH DATE	LAUNCH TIME	ORBIT INCLINATION (°)	PERIGEE ALTITUDE (NM)	PERIGEE LOCATION (°E)	RECOVERY PASS	MASTER CAMERA CAMERA NUMBER	MASTER CAMERA SLIT (°)	MASTER CAMERA FILTER TYPE	SLAVE CAMERA CAMERA NUMBER	SLAVE CAMERA SLIT (°)	SLAVE CAMERA FILTER TYPE	STELLAR-INDUX CAMERA NUMBER
1005	J-05	1174	2/15/64	2138 Z	74.9	99.9	29.0	49 112	124	0.250	W-21	125	0.250	W-21	D29/29/29
1006	J-06	1175	6/4/64	2259 Z	79.9	84.0	63.2	65 128	148	0.200	W-21	149	0.200	W-21	D42/42/37
1007	J-07	1609	6/19/64	2318 Z	85.0	99.2	41.5	65 128	144	0.250	W-25	145	0.200	W-21	D45/47/45
1008	J-08	1177	7/10/64	2314 Z	85.0	99.4	40.8	49 112	150	0.200	W-21	151	0.200	W-21	D49/53/42
1009	J-09	1605	8/5/64	2316 Z	80.1	99.6	39.5	49 128	154	0.200	W-21	155	0.200	W-21	D54/56/51
1010	J-10	1178	9/14/64	2254 Z	84.9	97.4	42.5	65 144	152	0.175	W-21	153	0.175	W-21	D48/45/40
1011	J-11	1170	10/5/64	2150 Z	79.9	99.3	20.9	65 —	160	0.175	W-21	161	0.175	W-21	D56/54/56
1012	J-12	1179	10/17/64	2202 Z	75.0	96.2	32.4	49 81	156	0.200	W-21	157	0.200	W-21	D38/30/34
1013	J-13	1173	11/2/64	2130 Z	80.0	100.0	25.0	65 81	158	0.225	W-21	159	0.225	W-21	D41/41/41
1014	J-14	1180	11/9/64	2036 Z	70.0	103.2	65.6	31 145	162	0.250	W-25	139	0.175	W-21	D30/36/30
1015	J-15	1607	12/19/64	2110 Z	74.9	96.7	21.5	81 175	138	0.250	W-25	141	0.175	W-21	D51/51/47
1016	J-16	1608	1/15/65	2101 Z	74.9	99.4	30.2	81 159	132	0.250	W-25	133	0.175	W-21	D52/49/55
1017	J-17	1611	2/25/65	2144 Z	75.0	97.2	25.9	81 145	140	0.250	W-25	165	0.175	W-21	D53/59/49
1018	J-18	1612	3/25/65	2111 Z	96.0	100.2	40.3	66 99	122	0.250	W-25	123	0.175	W-21	D61/61/61
1019	J-19	1614	4/29/65	2144 Z	85.0	99.1	27.1	80 —	118	0.250	W-25	119	0.175	W-21	D55/55/50
1020	J-20	1613	6/9/65	2158 Z	75.1	97.1	40.6	97 113	136	0.250	W-25	137	0.175	W-21	D21/21/21
1021	J-21	1615	5/18/65	1803 Z	75.0	109.2	24.3	81 161	166	0.175	W-21	167	0.250	W-25	D20/20/20
1022	J-22	1617	7/19/65	2201 Z	85.0	99.7	30.3	65 144	168	0.250	W-25	169	0.175	W-21	D59/50/59
1023	J-23	1618	8/17/65	2100 Z	70.0	97.8	29.0	81 144	170	0.225	W-25	171	0.150	W-21	D21/21/21
1024	J-24	1619	9/22/65	2131 Z	80.0	95.9	10.4	81 161	172	0.225	W-25	173	0.150	W-21	D59/39/35
1025	JX-28	1616	10/9/65	1746 Z	75.0	112.9	44.3	81 161	142	0.175	W-21	127	0.175	W-21	D67/85/80
1026	J-25	1620	10/28/65	2117 Z	75.0	93.0	17.0	81 160	174	0.225	W-25	175	0.150	W-21	D63/69/69
1027	JX-27	1621	12/9/65	2110 Z	80.0	97.4	17.3	17 33	164	0.230	W-25	163	0.175	W-21	D65/77/70
1028	J-26	1610	12/24/65	2106 Z	80.0	97.6	28.4	81 144	176	0.250	W-25	177	0.175	W-21	D17/19/82
															D56/75/72
															D64/82/66
															D70/89/81
															D75/92/93
															D71/87/87
															D77/81/97
															D74/76/95

REC-434 4/68

TOP SECRET

# MISSION SUMMARY

MISSION NUMBER	MISSION DATE	VEHICLE NUMBER	LAUNCH TIME	CONC. INCLINATION (°)	RECOVERY ALTITUDE (M)	RECOVERY LOCATION (°S)	RECOVERY PAGES	COMM. SLIT NUMBER	COMM. SLIT TYPE	CARRIER NUMBER	SLAVE SLIT NUMBER	SLAVE SLIT TYPE	STELLARANCE CARERA NUMBER
1029	3-27	1625	2132 Z	75.1	99.5	22.5	81	178	W-25	179	0.175	W-21	079/04/94
1030	3-29	1622	2202 Z	75.0	97.5	18.7	81	182	W-25	183	0.175	W-21	079/100/07
1031	3-30	1627	2202 Z	75.1	104.5	23.3	113	164	W-23A	185	0.150	W-21	083/101/89
1032	3-30	1625	1935 Z					180	W-21	181	0.150	W-21	081/07/01
1033	3-33	1619	2213 Z	60.1	102.0	60.7	82	194	W-21	195	0.200	W-21	081/05/109
1034	3-31	1620	2131 Z	60.1	105.4	18.2	81	185	W-23A	187	0.150	W-21	085/09/76
1035	3-29	1621	2146 Z	65.0	99.5	29.1	81	186	W-23A	189	0.175	W-21	089/12/10
1036	3-30	1620	2046 Z	60.0	102.4	22.9	115	190	W-23A	191	0.150	W-21	089/10/11
1037	3-30	1619	1917 Z	60.0	91.6	14.5	85	197	W-23A	199	0.175	W-21	087/06/28
1038	3-31	1626	2056 Z	60.1	96.5	29.2	81	193	W-23A	193	0.175	W-21	083/06/11
1039	3-30	1625	2102 Z	60.0	97.0	30.2	81	206	W-23A	207	0.175	W-21	083/11/02
1040	3-30	1620	1856 Z	85.1	99.7	20.3	81	145	W-21	97	0.225	W-23A	078/09/10
1041	3-30	1634	2153 Z	60.1	100.1	33.0	91	215	W-23A	209	0.175	W-21	085/10/10
1042	3-37	1633	2130 Z	60.0	96.5	29.1	97	240	W-23A	205	0.150	W-21	087/02/117
1043	3-12	1627	2144 Z	60.0	102.1	16.3	113	240	W-23A	201	0.150	W-21	087/05/135
1101	CR-1	1641	1941 Z	80.0	84.8	5.7	97	238	W-21	303	*	W-23A	DISIC NO 3
1044	3-41	1639	2131 Z	81.5	93.9	16.4	97	144	W-23A	203	0.175	W-21	099/12/120

\* 300 SERIES INSTRUMENTS USE VARIABLE SLIT EXPOSURE CONTROL. REFER TO FINAL REPORT, SECTION 2

# PERFORMANCE SUMMARY

MISSION NUMBER	CATEGORY	SERIES NUMBER	M.T.P. VALUE	VISUAL RES.	ALT. AVERAGE (ft)	EFFECTIVE		M.T.P. AVERAGE		SUIT AVERAGE		SUIT AVERAGE		90% ATTITUDE ERROR (°)		90% ATTITUDE RATES (°/HR)		90% V/MH ERROR (%)	90% RESOLUTION LIMIT (FEET)	
						ALT. AVERAGE (ft)	SUIT AVERAGE (ft)	ALT. AVERAGE (ft)	SUIT AVERAGE (ft)	PITCH	ROLL	YAW	PITCH	ROLL	YAW	FLOW TRACK	CROSS TRACK			
1004-1	FWD	124	85	78	350	97	109	115	127	0.45	0.42	1.08	30.0	25.0	21.0	5.1	7.7	6.1		
1004-2	FWD	125	85	76	350	88	113	117	124	0.74	0.50	0.91	44.0	30.0	29.0	4.9	6.8	6.5		
1006-1	FWD	140	90	73	350	55	88	84	97	0.41	0.42	1.14	26.8	28.5	27.8	15.4	13.8	6.7		
1006-2	FWD	149	90	35	350	64	90	84	90	0.49	0.40	1.08	31.1	27.9	30.0	11.6	10.1	7.0		
1007-1	FWD	144	85	80	350	60	87	82	91	0.58	0.46	1.43	37.6	23.9	29.9	3.6	3.1	9.4		
1007-2	FWD	145	85	79	350	72	81	68	74	0.64	0.47	—	43.0	25.8	—	4.6	2.1	7.6		
1008-1	FWD	150	85	80	350	80	95	81	81	0.39	0.39	0.94	43.8	23.9	29.6	2.9	4.9	5.9		
1008-2	FWD	151	85	79	350	84	96	82	83	0.63	0.36	0.71	42.9	24.0	32.5	2.8	4.2	5.4		
1009-1	FWD	154	85	92	350	80	—	75	98	0.65	0.65	0.71	29.2	22.7	27.6	3.3	5.3	5.8		
1009-2	FWD	155	85	94	350	85	—	76	83	0.48	0.65	0.59	33.6	23.9	27.2	2.6	4.9	5.9		
1010-1	FWD	152	85	90	350	90	83	87	96	0.93	0.30	0.87	39.1	23.6	30.8	4.5	2.3	4.4		
1010-2	FWD	153	85	92	350	86	82	92	93	0.59	0.70	1.21	45.4	23.6	30.7	4.6	7.5	3.8		
1011-1	FWD	160	90	84	350	76	86	78	87	0.77	0.39	0.97	43.1	28.9	31.1	2.3	5.3	5.6		
1012-1	FWD	156	85	92	350	—	91	84	98	0.65	0.51	—	47.1	33.2	—	1.5	4.8	—		
1012-2	FWD	157	85	91	350	—	89	84	100	0.97	0.77	0.51	45.2	30.7	20.4	5.9	3.3	5.9		
1013-1	FWD	158	85	89	350	—	94	85	98	0.64	0.32	1.34	36.9	29.0	32.3	3.7	7.8	8.3		
1014-1	FWD	162	80	87	350	—	78	74	86	0.62	0.41	1.46	35.0	36.1	33.5	2.2	6.2	8.8		
1014-2	FWD	139	90	83	350	—	80	95	107	1.06	0.55	1.44	34.8	36.0	38.3	3.3	2.8	6.3		
1015-1	FWD	138	85	87	350	—	75	70	77	0.50	0.38	0.53	47.0	29.4	38.2	5.0	5.3	7.8		
1015-2	FWD	141	85	83	350	—	72	89	89	0.64	0.61	0.64	46.9	29.2	39.2	6.3	6.0	5.5		
1016-1	FWD	132	85	85	350	—	80	90	90	0.50	0.61	0.64	39.1	27.1	36.2	3.2	4.6	7.5		
1016-2	FWD	133	85	91	350	—	55	91	91	0.72	0.83	2.01	48.9	30.2	40.4	2.0	5.5	10.5		
1017-1	FWD	140	85	72	350	—	57	78	86	0.49	0.76	2.50	35.5	32.2	38.4	3.3	3.3	7.1		
1017-2	FWD	165	85	85	350	—	70	94	107	0.49	0.76	2.49	35.3	32.0	38.5	4.3	9.0	11.6		
1018-1	FWD	122	65	79	350	—	70	82	92	0.91	0.48	—	47.4	36.7	—	3.4	5.6	—		
1018-2	FWD	123	85	88	350	—	75	77	89	0.64	0.63	—	46.2	36.2	—	3.2	3.7	—		





# PERFORMANCE SUMMARY

REG. NO.	TYPE	CLASS.	CLASS.	WIND	WIND	WIND	WIND	90° SPEED (MPH)			90° SPEED (MPH)			90°/270°		90°/270° (Error %)	90°/270° (Error %)	90°/270° (Error %)	90°/270° (Error %)	90°/270° (Error %)	90°/270° (Error %)	90°/270° (Error %)	
								WIND	WIND	WIND	WIND	WIND	WIND	WIND	WIND								
1041-1	FWD	100	100	60	51	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-2	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-3	FWD	100	100	60	60	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-4	FWD	100	100	60	51	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-5	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-6	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-7	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-8	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-9	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-10	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-11	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-12	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-13	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-14	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-15	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-16	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-17	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-18	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-19	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8
1041-20	FWD	100	100	60	53	0.30	0.30	0.30	19.2	20.7	20.2	19.0	17.5	17.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8	5.8



# EXPOSURE PROCESSING SUMMARY

INSUFFICIENT DATA

CAMERA	EXPOSURE NO.	EXPOSURE DATE	EXPOSURE TIME	EXPOSURE DATA		EXPOSURE DATA		EXPOSURE DATA		EXPOSURE DATA		UNDER EMPLOYED (%)	UNDER PRODUCED EXP. (%)	OVER PRODUCED EXP. (%)	OVER EMPLOYED (%)	CLOUD COVER (%)	
				APERTURE	SHUTTER	APERTURE	SHUTTER	APERTURE	SHUTTER	APERTURE	SHUTTER						
1020-1	FWD	04	70	24	100	0	21	79	100	0	21	79	100	0	21	79	
1020-1	AFT	23	70	21	100	0	22	60	25	19	3	67	10	13	70	20	
1020-1	FWD	20	75	19	100	0	19	81	100	0	19	81	100	0	19	81	
1020-1	AFT	29	75	17	100	0	24	36	100	0	24	36	100	0	24	36	
1020-2	FWD	47	69	15	100	0	47	69	100	0	47	69	100	0	47	69	
1020-2	AFT	48	65	17	100	0	48	65	100	0	48	65	100	0	48	65	
1021-1	FWD	15	55	163	100	0	60	32	14	39	47	1	52	47	50	0	
1021-1	AFT	14	55	157	100	0	57	43	17	70	0	57	43	17	70	0	
1021-2	FWD	12	55	153	100	0	57	43	17	70	0	57	43	17	70	0	
1021-2	AFT	13	55	154	100	0	50	56	0	33	38	0	50	56	0	33	38
1022-1	FWD	19	67	35	100	0	42	54	0	16	49	0	42	54	0	16	49
1022-1	AFT	27	67	24	100	0	53	47	0	24	49	0	53	47	0	24	49
1022-2	FWD	20	75	21	100	0	43	57	0	26	50	0	43	57	0	26	50
1022-2	AFT	26	74	19	100	0	53	47	0	21	40	0	53	47	0	21	40
1023-1	FWD	22	82	35	100	0	72	26	0	15	26	0	72	26	0	15	26
1023-1	AFT	29	81	35	100	0	42	54	0	16	49	0	42	54	0	16	49
1023-2	FWD	29	81	13	100	0	18	62	0	22	38	0	18	62	0	22	38
1023-2	AFT	28	80	13	100	0	20	72	0	22	16	0	20	72	0	22	16
1024-1	FWD	10	51	24	100	0	72	26	0	17	24	0	72	26	0	17	24
1024-1	AFT	13	51	21	100	0	32	18	0	20	12	0	32	18	0	20	12
1025-1	FWD	7	79	11	100	0	25	75	0	25	17	0	25	75	0	25	17
1025-1	AFT	11	79	9	100	0	66	34	0	13	39	0	66	34	0	13	39
1025-2	FWD	0	70	10	100	0	64	32	0	10	41	0	64	32	0	10	41
1025-2	AFT	0	70	10	100	0	72	28	0	18	45	0	72	28	0	18	45
1026-1	FWD	0	57	23	100	0	51	49	0	21	32	0	51	49	0	21	32
1026-1	AFT	0	57	21	100	0	38	32	0	16	33	0	38	32	0	16	33
1026-2	FWD	1	72	18	100	0	5	95	0	15	10	0	5	95	0	15	10
1026-2	AFT	0	57	13	100	0	15	85	0	21	13	0	15	85	0	21	13
1027-1	FWD	2	63	26	100	0	3	57	0	25	19	0	3	57	0	25	19
1027-1	AFT	2	63	26	100	0	20	80	0	16	34	0	20	80	0	16	34
1028-1	FWD	3	73	15	100	0	11	89	0	22	10	0	11	89	0	22	10
1028-1	AFT	1	73	14	100	0	11	89	0	10	10	0	11	89	0	10	10
1028-2	FWD	2	81	5	100	0	15	85	0	23	9	0	15	85	0	23	9
1028-2	AFT	1	80	5	100	0	6	94	0	22	4	0	6	94	0	22	4
1029-1	FWD	3	63	18	100	0	14	86	0	25	18	0	14	86	0	25	18
1029-1	AFT	2	63	16	100	0	25	75	0	16	17	0	25	75	0	16	17
1029-2	FWD	0	80	4	100	0	20	80	0	20	70	0	20	80	0	20	70
1029-2	AFT	0	80	3	100	0	25	75	0	22	74	0	25	75	0	22	74
1030-1	FWD	5	63	27	100	0	52	48	0	18	55	0	52	48	0	18	55
1030-1	AFT	4	63	22	100	0	71	29	0	15	44	0	71	29	0	15	44
1030-2	FWD	3	78	10	100	0	30	70	0	30	70	0	30	70	0	30	70
1030-2	AFT	2	78	7	100	0	30	70	0	34	66	0	30	70	0	34	66
1031-1	FWD	17	70	20	100	0	51	49	0	25	16	0	51	49	0	25	16
1031-1	AFT	17	70	18	100	0	47	53	0	24	15	0	47	53	0	24	15
1031-2	FWD	19	82	6	100	0	60	40	0	26	13	0	60	40	0	26	13
1031-2	AFT	19	82	6	100	0	60	40	0	26	13	0	60	40	0	26	13
1033-1	FWD	10	47	45	100	0	2	98	0	28	10	0	2	98	0	28	10
1033-1	AFT	11	47	45	100	0	3	97	0	30	10	0	3	97	0	30	10
1033-2	FWD	0	54	34	100	0	10	90	0	30	10	0	10	90	0	30	10
1033-2	AFT	0	54	34	100	0	1	99	0	26	14	0	1	99	0	26	14





SECTION A

APPENDIX



[REDACTED]

MISSION # 1044-1 \* INSTR # 1/16/68 PLOT OF D MIN \* TRAIN # PROCESSING \* INTERMEDIATE  
ARITH MEAN # 0.89 \* MEDIAN # 0.94 \* STD DEV # 0.29 \* RANGE # 0.48 TO 1.48 WITH 13 SAMPLES

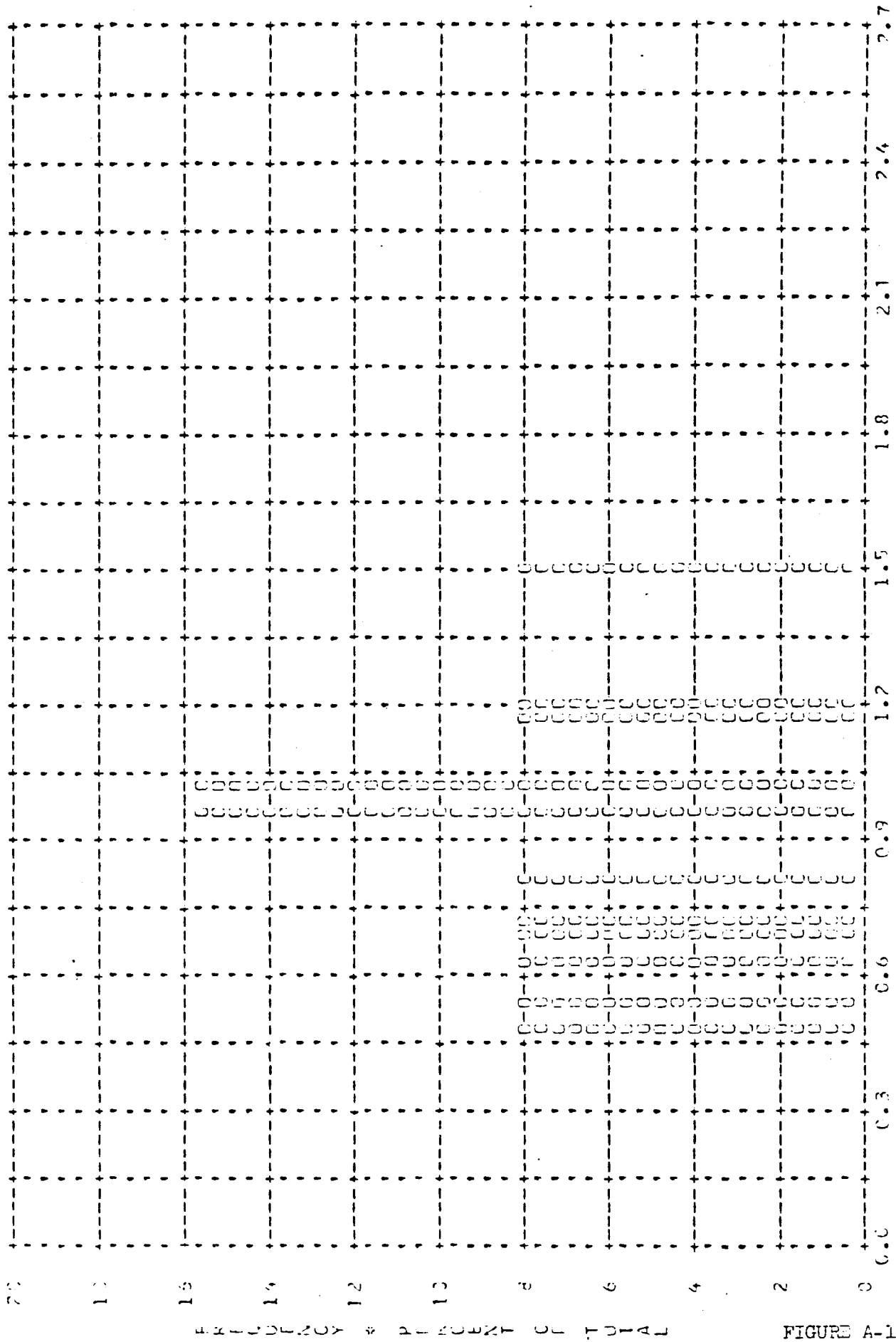


FIGURE A-1

[REDACTED]

~~TOP SECRET~~

MISSILE # 1044-1 \* INSTID # 6WD \* 1/16/68 PLCT CF 0 MAX \* TERRAIN \* PROCESSING \* INTERMEDIATE  
GRITF BEAR \* 1.00 \* MEDIAN # 1.71 \* STD DEV # 0.20 \* RANGE # 1.32 TO 1.92 WITH 13 SAMPLES

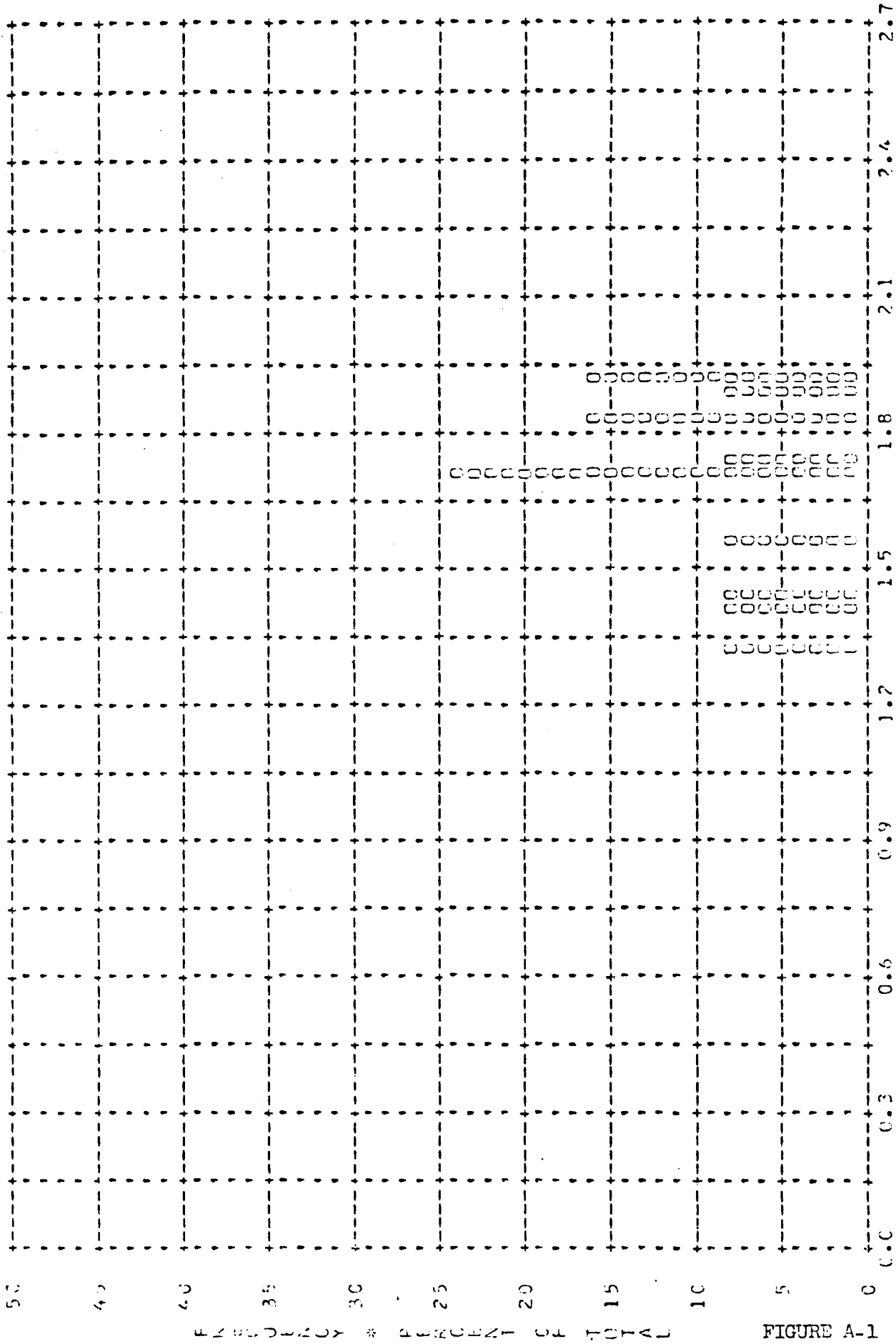


FIGURE A-1

\* DENSITY \*

~~TOP SECRET~~

[REDACTED]

MISSION \* 1044-1 \* INSTR \* F40 \* 1/16/58 PUCT OF D MAX \* CLOUD \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 2.07 \* MEDIAN \* 2.00 \* STD DEV \* 0.25 \* RANGE \* 1.58 TO 2.44 WITH 13 SAMPLES

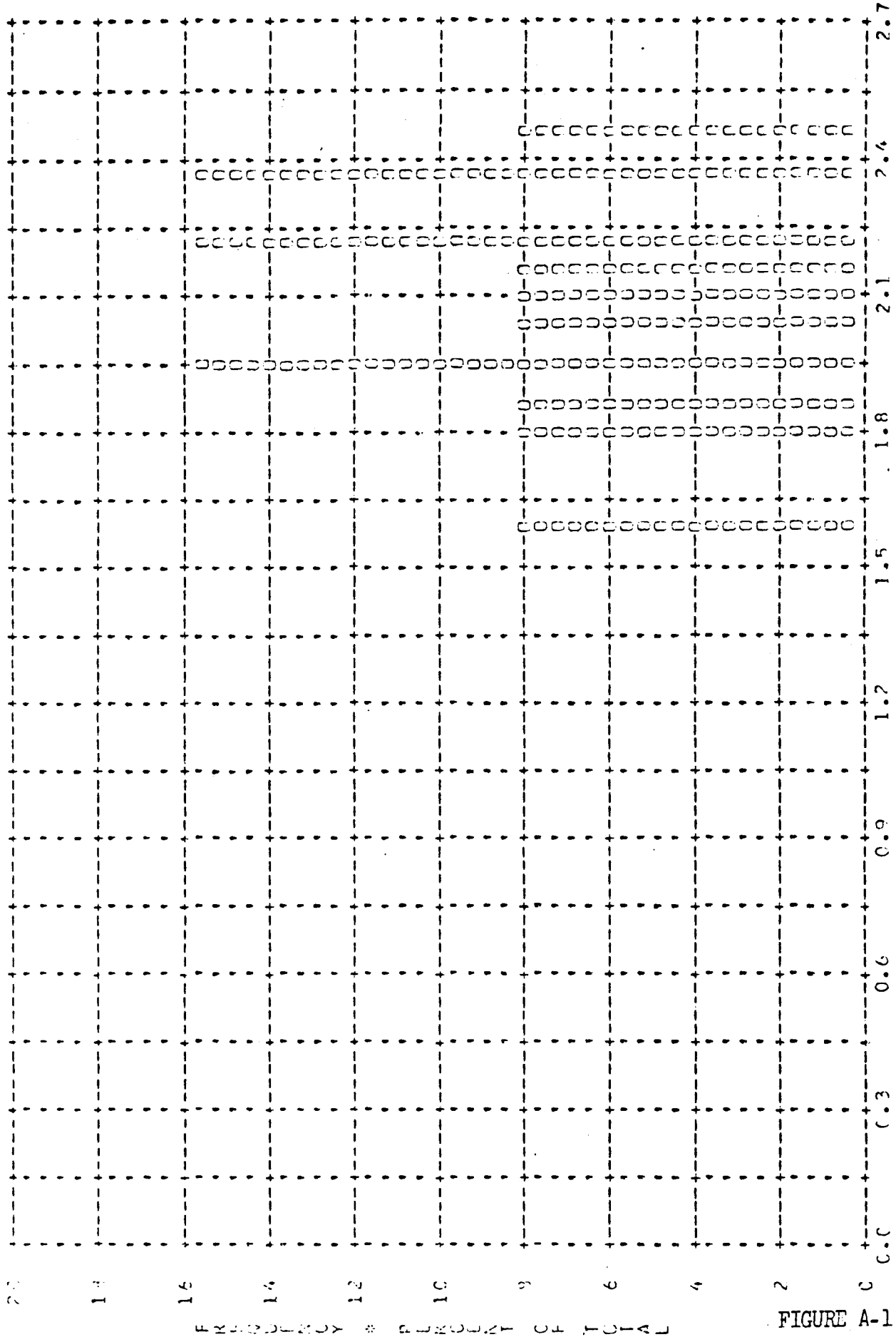


FIGURE A-1

\* DENSITY #

[REDACTED]



TOP SECRET C

MISSION # 1044-1 \* INSTR \* FWD \* 1/16/68 PLCT CF D MIN \* TERRAIN \* PROCESSING \* FULL  
AKITH PLAN \* 0.55 \* MEDIAN \* 0.67 \* STD DEV \* 0.24 \* RANGE \* 0.23 TO 1.78 WITH 147 SAMPLES

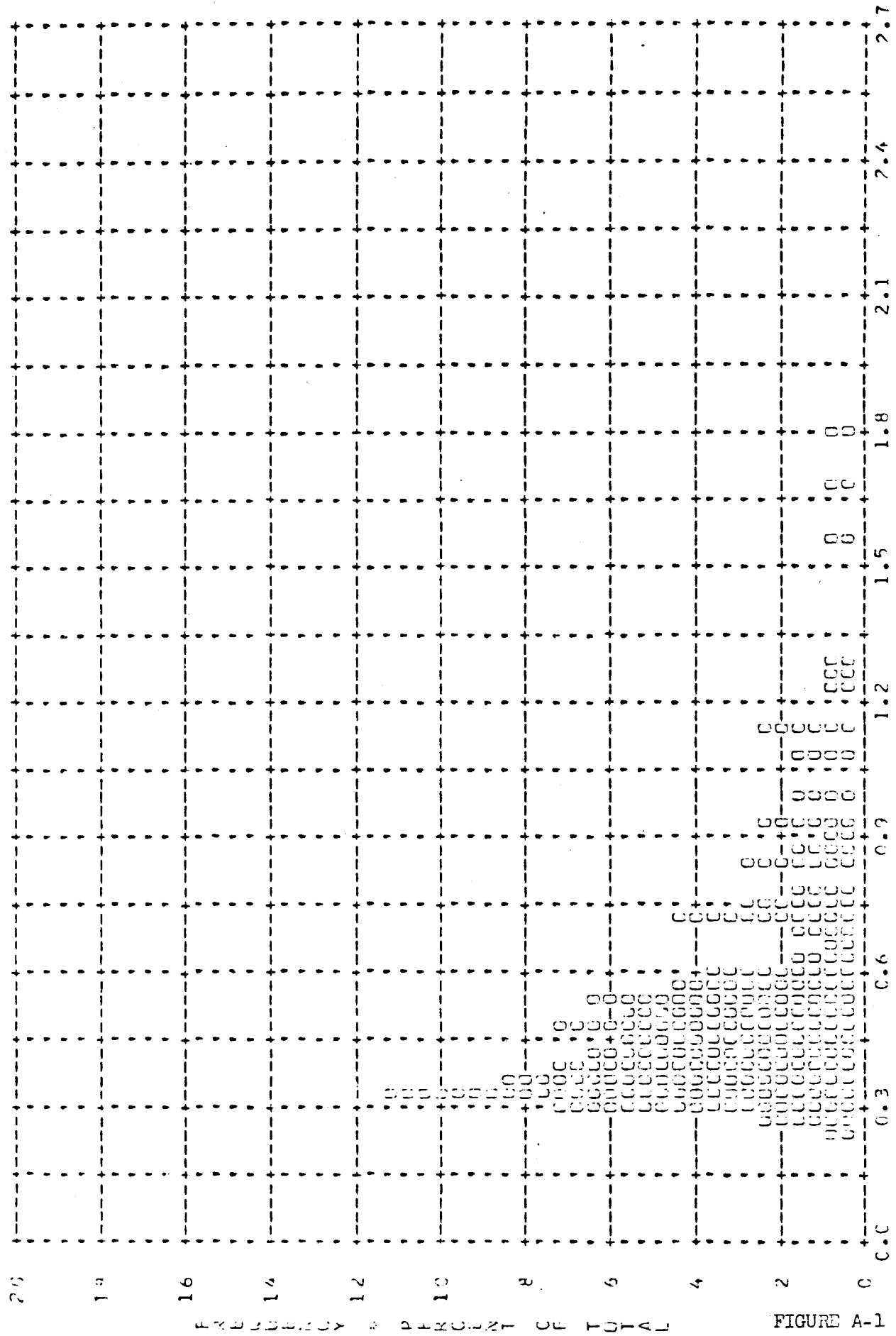


FIGURE A-1

\* DENSITY #

TOP SECRET C

~~TOP SECRET~~ C

MISSILE \* 1044-1 \* INSTP \* FWD \* 1/16/68 PLOT OF D MAY \* TERRAIN \* PROCESSING \* FULL  
 ORTHO \* 1.57 \* MEDIAN \* 1.57 \* STD DEV \* 0.40 \* RANGE \* 0.67 TO 2.36 WITH 147 SAMPLES

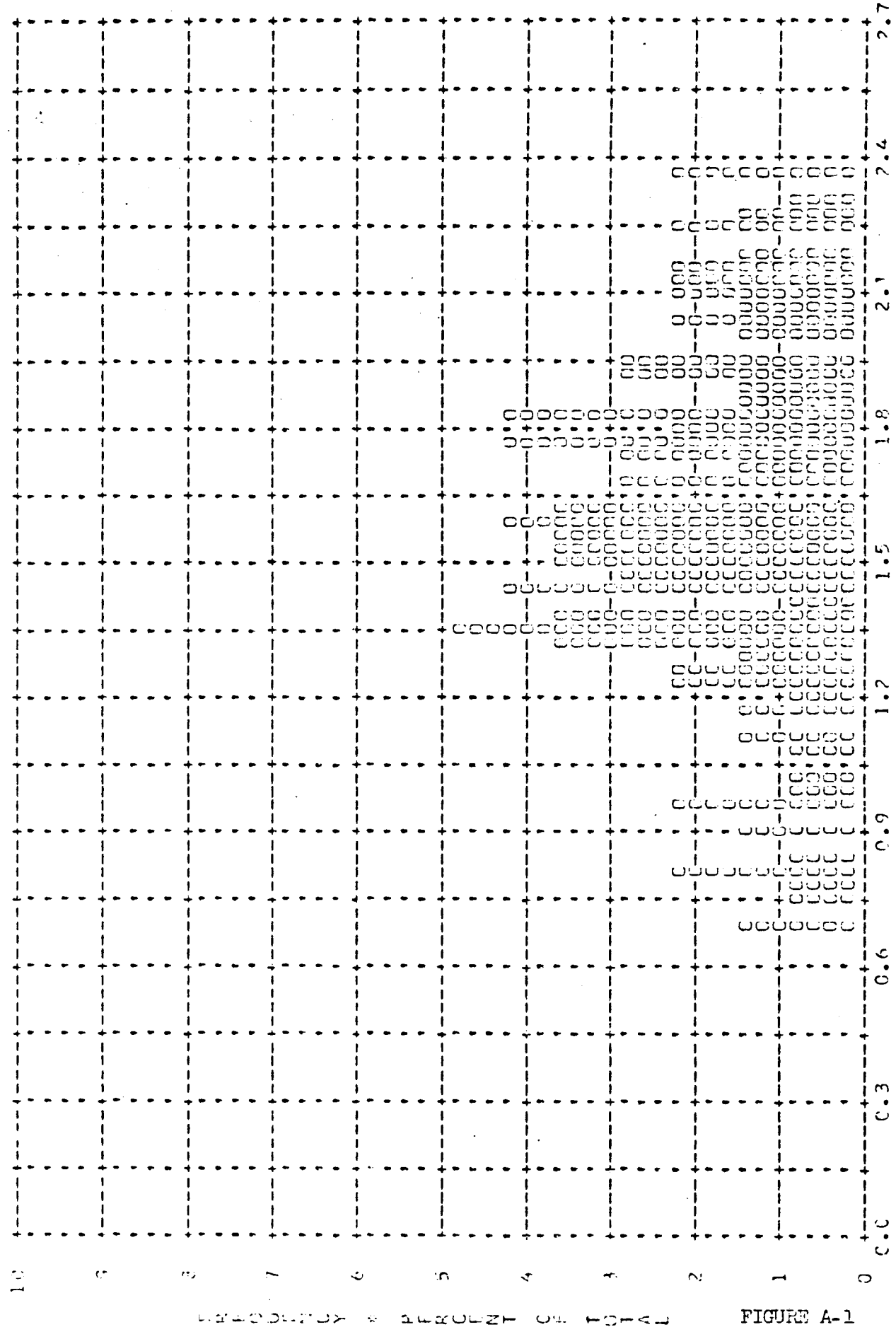


FIGURE A-1

~~TOP SECRET~~ C

~~TOP SECRET - C~~

MISSILE # 1040-1 \* INST # FWD \* 1/16/68 PLCT CF D MAX \* CLCUD \* PROCESSING \* FULL  
ARITH MEAN \* 2.09 \* MEDIAN \* 2.18 \* STD DEV \* 0.33 \* RANGE \* 0.31 TO 2.45 WITH 128 SAMPLES

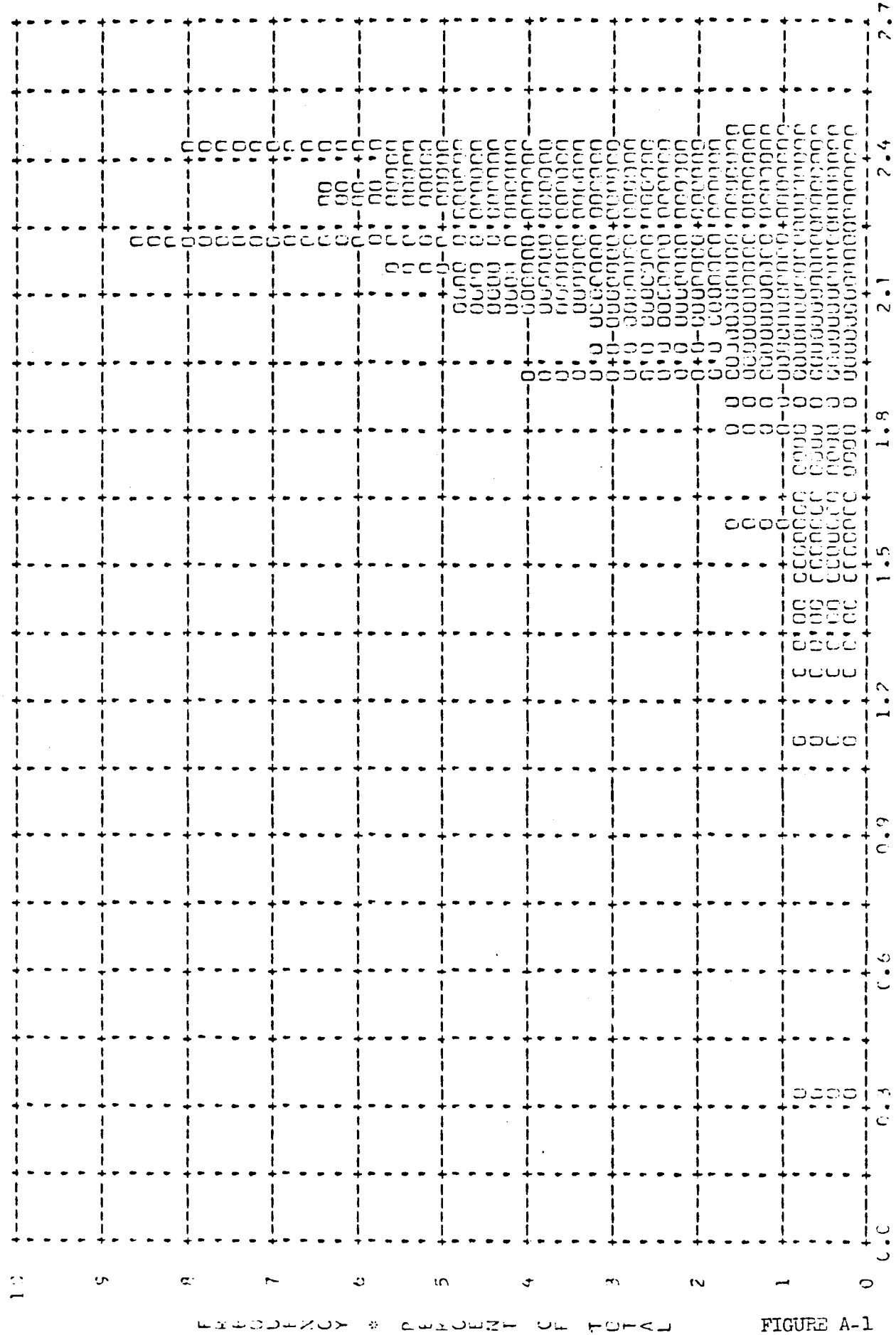


FIGURE A-1

\* DENSITY \*

~~TOP SECRET - C~~

[REDACTED]

MISSION \* 1044-V \* INSTR \* FLD \* 1/16/68 PICT CF D MIN \* TERRAIN \* PROCESSING \* ALL LEVELS  
 ARITH MEAN \* 0.50 \* MEDIAN \* 0.40 \* STD DEV \* 0.30 \* RANGE \* 0.23 TO 1.78 WITH 160 SAMPLES

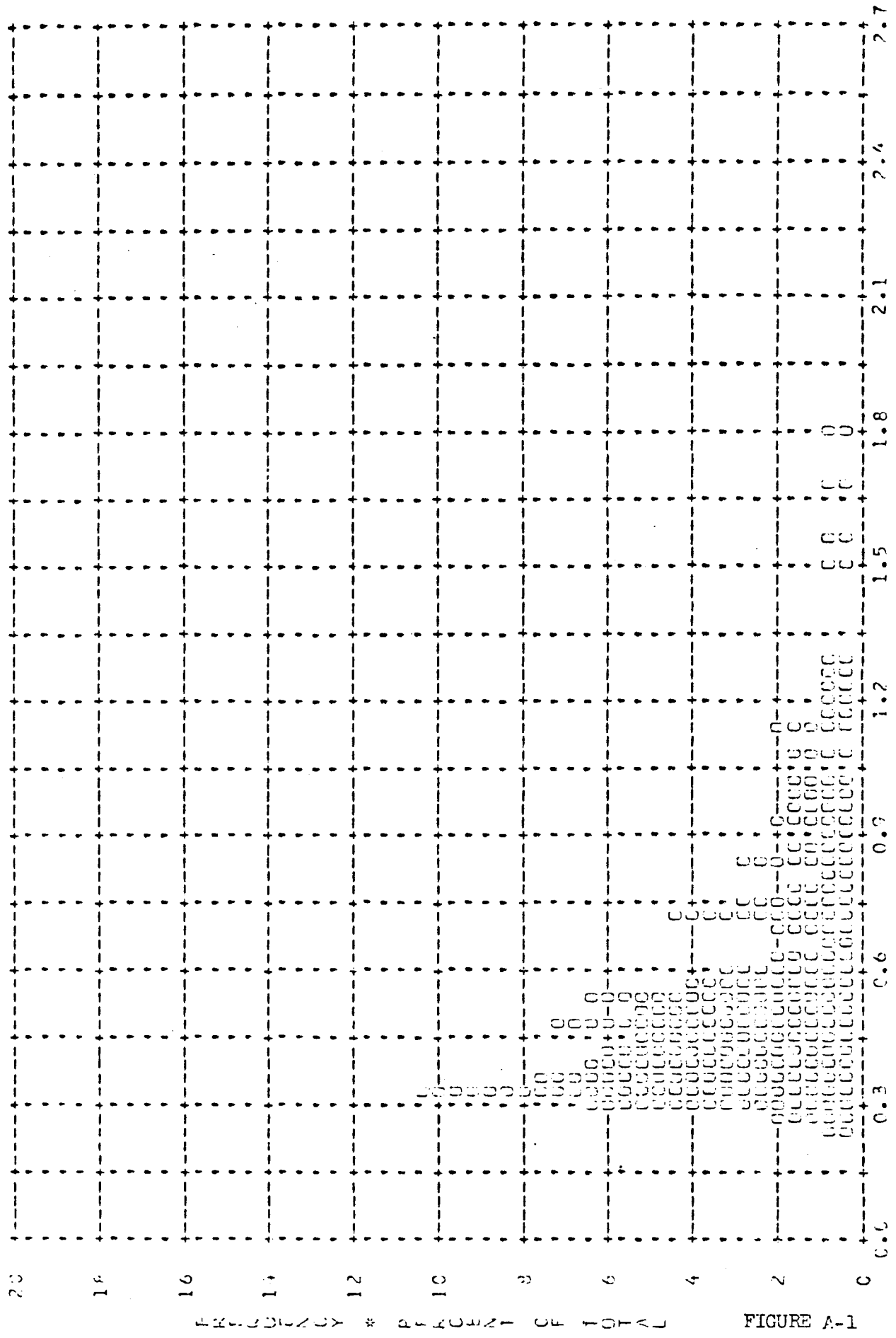


FIGURE A-1  
 \* DENSITY \*  
 [REDACTED]

~~TOP SECRET~~ CA

MISSION \* 1044-1 \* INSTR \* F&D \* 1/16/68 \* PLOT OF D MAX \* TERRAIN \* PROCESSING \* ALL LEVELS  
ARITH MEAN \* 1.59 \* MEDIAN \* 0.39 \* RANGE \* 0.67 TO 2.36 WITH 160 SAMPLES

	0	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1.0	1.1	1.2	1.3	1.4	1.5	1.6	1.7	1.8	1.9	2.0	2.1	2.2	2.3	2.4	2.5	2.6	2.7		
10																														
9																														
8																														
7																														
6																														
5																														
4																														
3																														
2																														
1																														
0																														

FREQUENCY \* PERCENT OF TOTAL

FIGURE A-1

\* DENSITY \*

~~TOP SECRET~~ CA

~~TOP SECRET~~

MISSION \* 1044-1 \* INSTR \* FMD \* 1/16/68 PLCT CF D MAX \* CLOUD \* PROCESSING \* ALL LEVELS  
ORITH BEAR \* 2.00 \* MEDIAN \* 2.15 \* STD DEV \* 0.32 \* RANGE \* 0.31 TO 2.45 WITH 141 SAMPLES

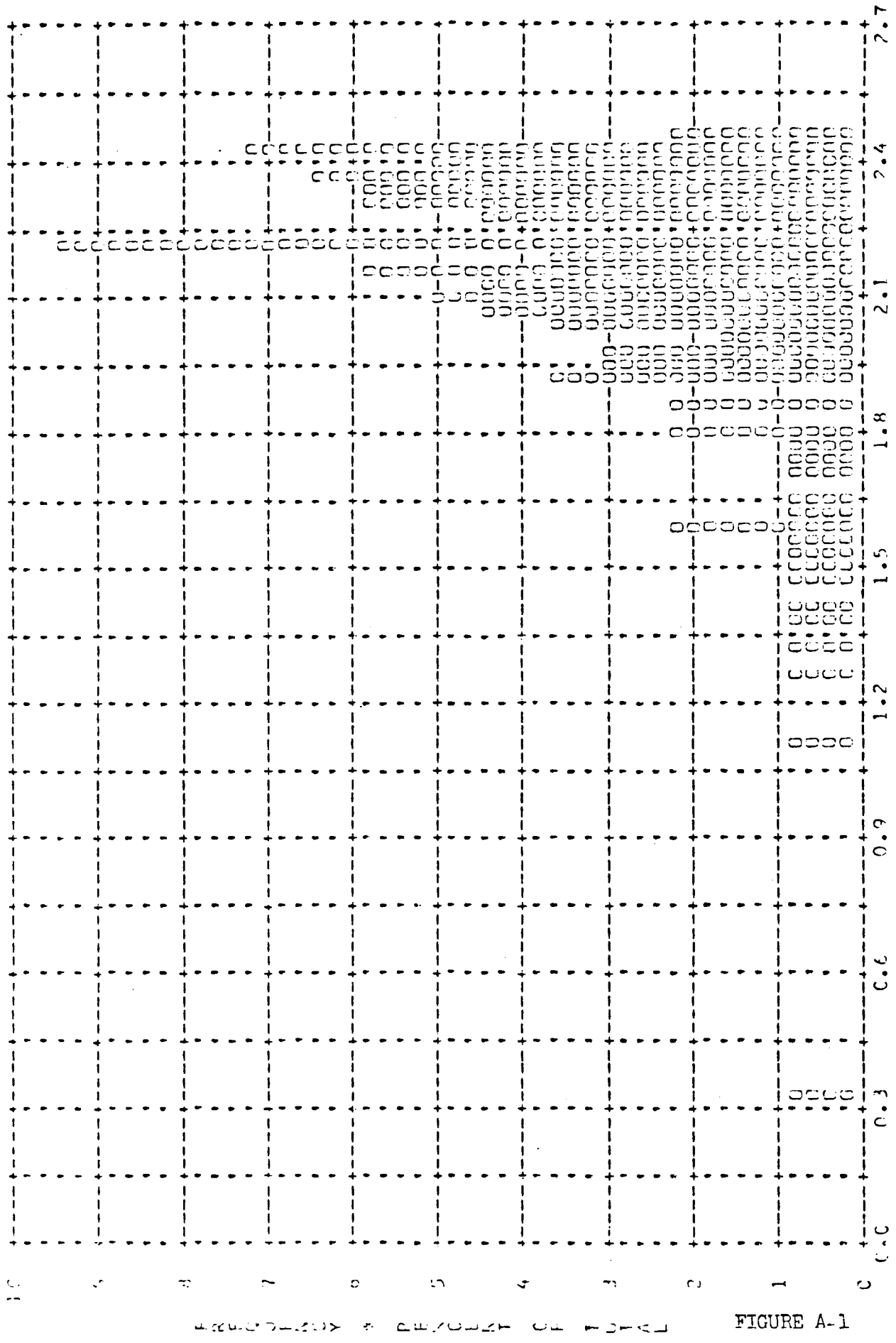


FIGURE A-1

\* DENSITY #

~~TOP SECRET~~

[REDACTED]

MISSION # 1044-1 \* INSTR # FWD \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* DUAL GAMMA  
ARTIF # 174 \* 0.40 \* SLP DRV \* 0.17 \* RANGE \* 0.24 TO 0.92 WITH 67 SAMPLES

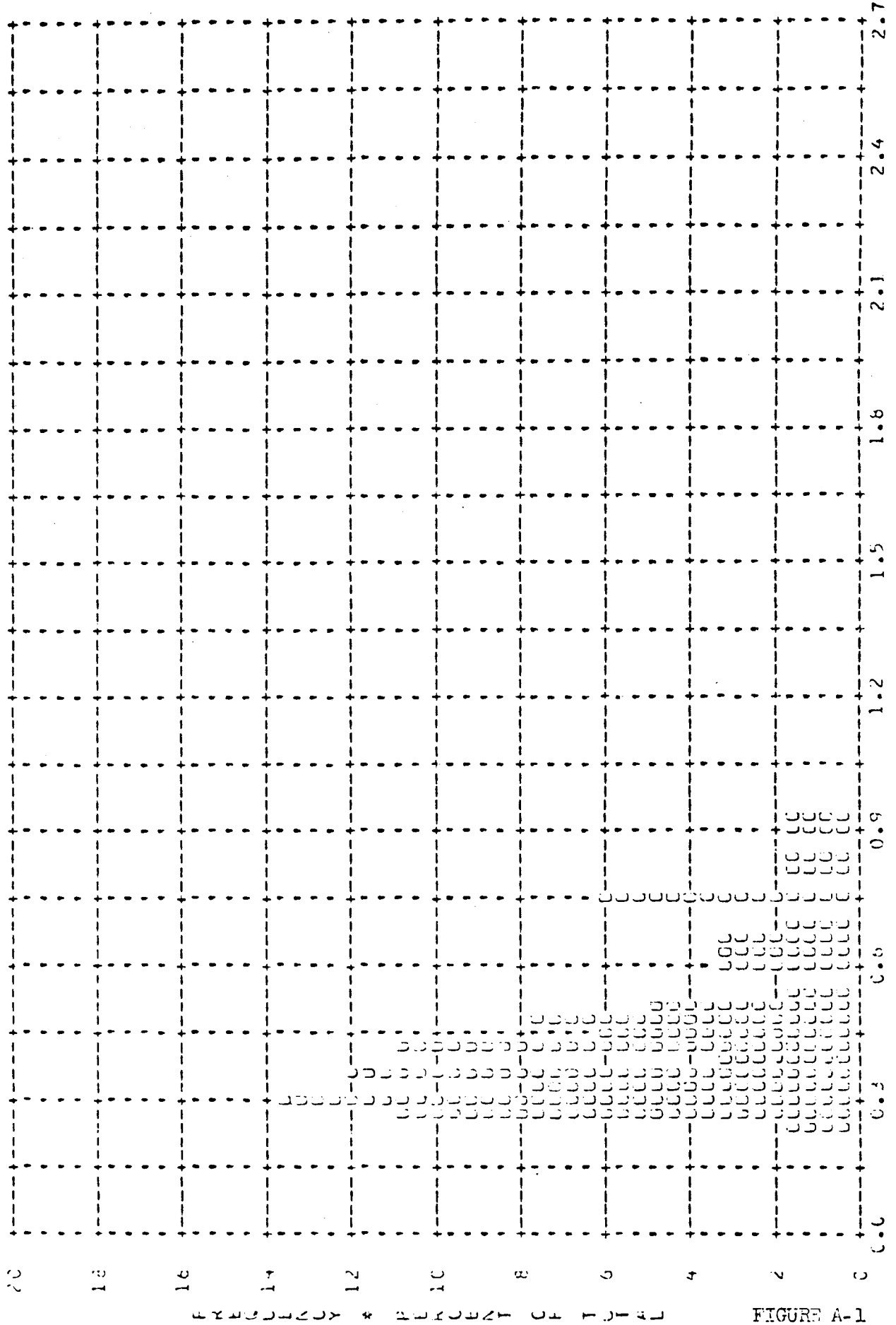


FIGURE A-1

[REDACTED] \* DENSITY \* [REDACTED]

~~TOP SECRET~~

MISSION \* 1044-1 \* INSTR \* FAD \* 1/16/68 PLOT CF D MAX \* TERRAIN \* PROCESSING \* DUAL GAMMA  
ARITH MEAN \* 1.30 \* MEDIAN \* 1.41 \* STD DEV \* 0.37 \* RANGE \* 0.49 TO 1.82 WITH 67 SAMPLES

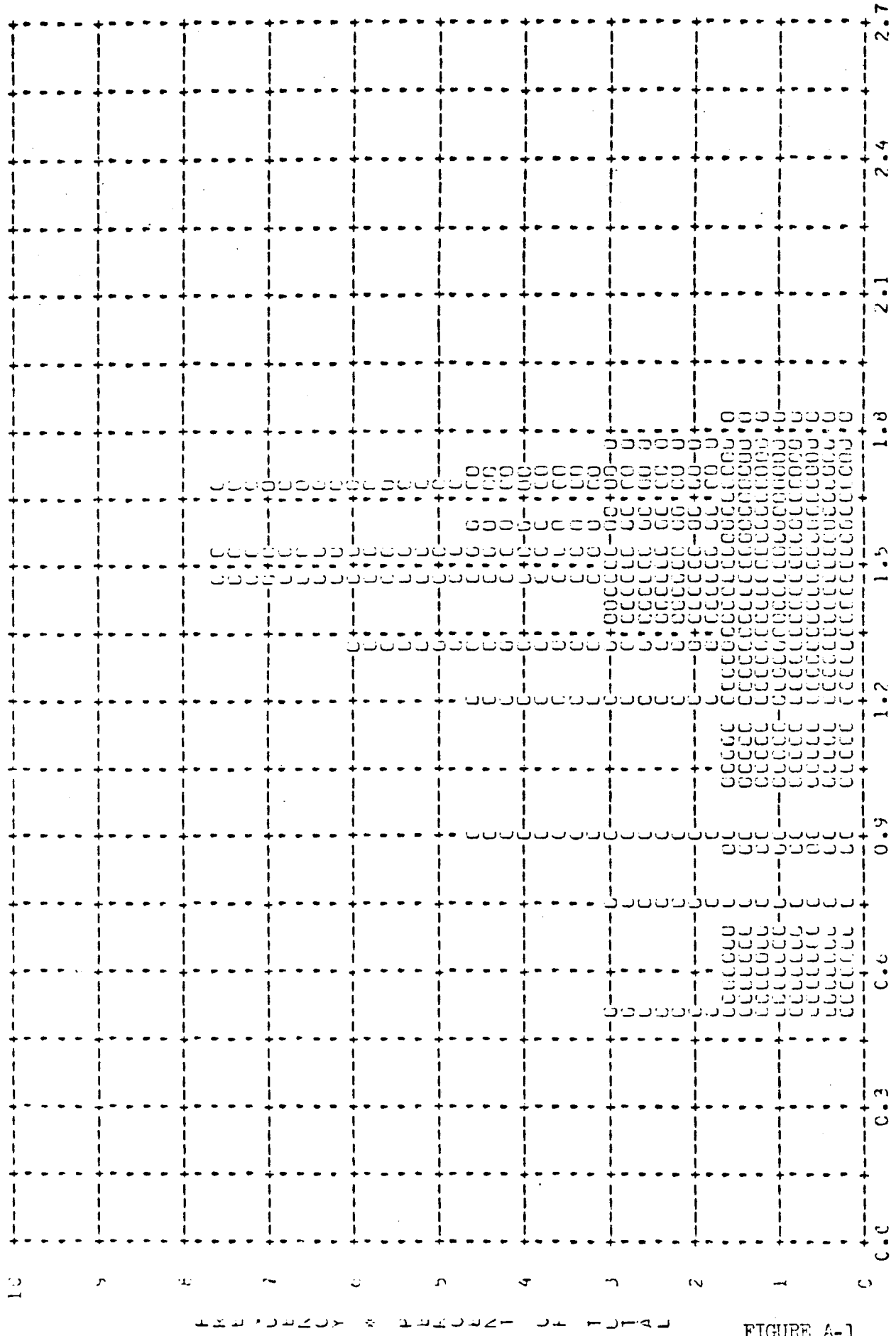


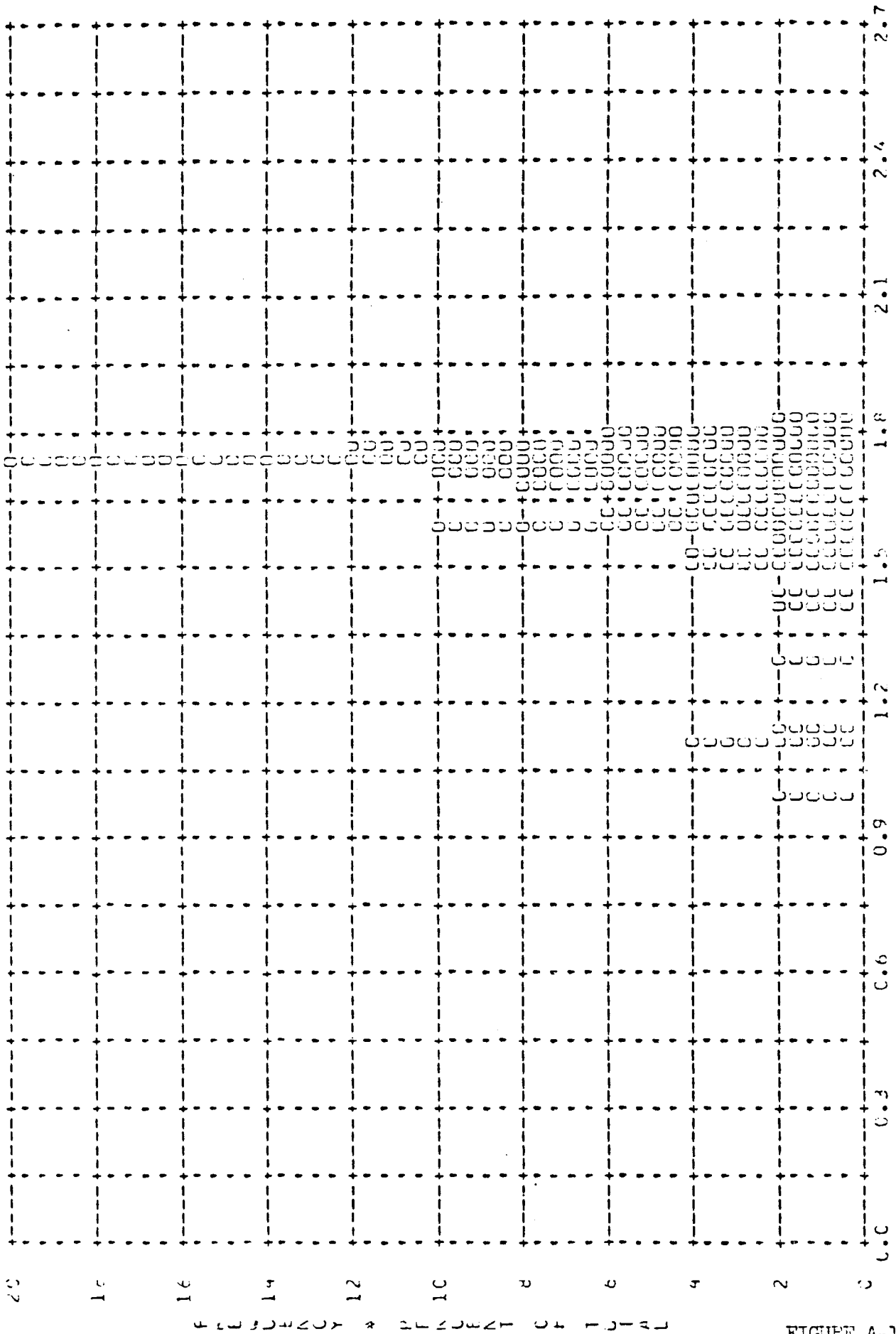
FIGURE A-1

~~TOP SECRET~~



~~TOP SECRET C/~~

MISSILE # 1C44-1 \* INSTR # FWD \* 1/16/68 PLCT CF D MAX \* CLCUD \* PROCESSING \* DUAL GAMMA  
WITH BEAN # 1.01 \* FIDIAN # 1.03 \* STD DEV # 0.19 \* RANGE # 0.97 TO 1.83 WITH 51 SAMPLES



~~TOP SECRET C/~~

FIGURE A-1

~~TOP SECRET~~ I.C.

MISSION # 1044-1 \* INSTR # 1/16/60 PLCT CF D MIN # TERRAIN # PROCESSING # INTPTFDIATE  
ARITH MEAN # 0.72 \* MEDIAN # 0.74 \* STD DEV # 0.30 \* RANGE # 0.30 TO 1.45 WITH 37 SAMPLES

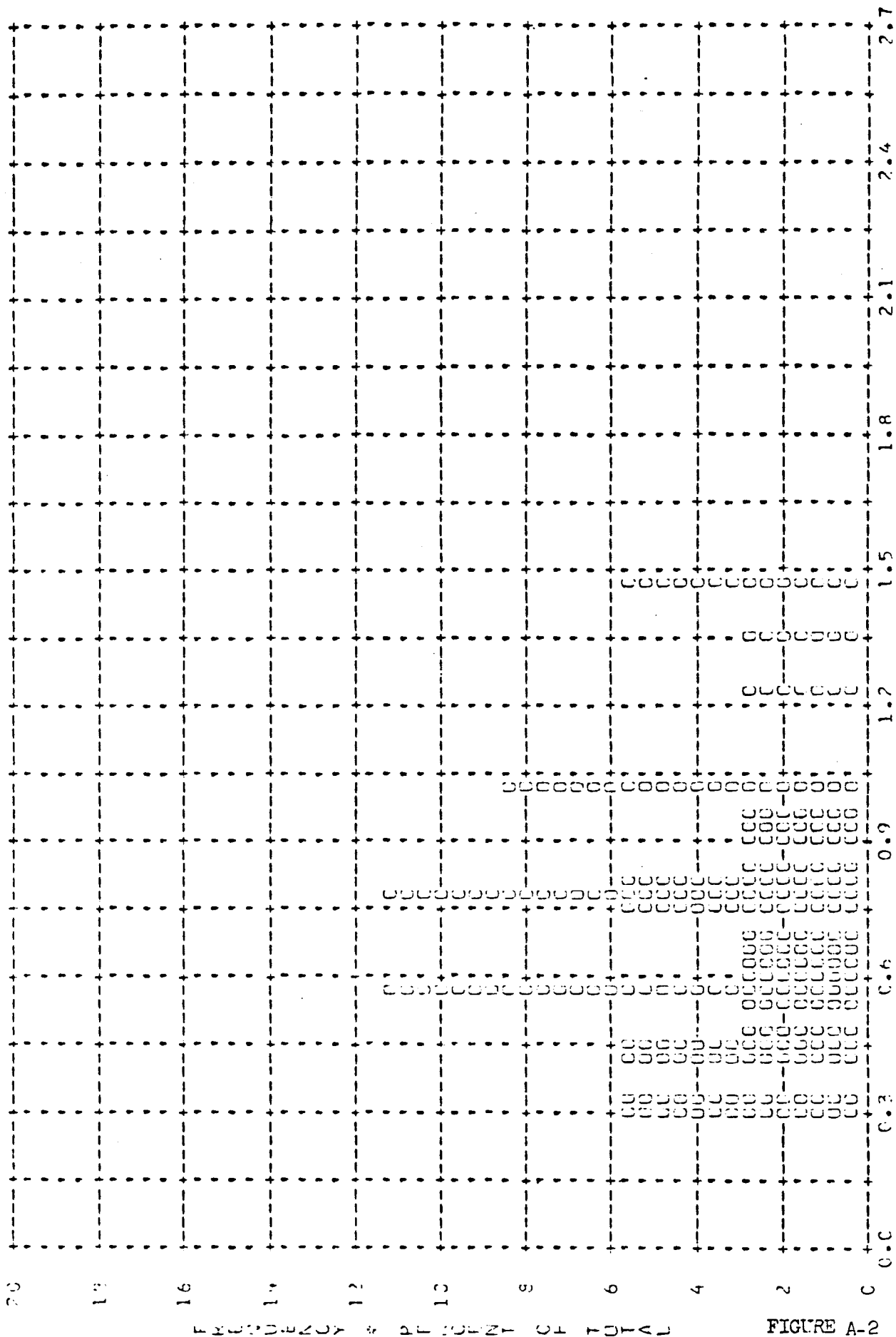


FIGURE A-2

~~TOP SECRET~~ I.C.

~~TOP SECRET~~ C

MISSION \* 1044-J \* INSTR \* AFT \* 1/16/68 PLCT OF D MAX \* TERRAIN \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 1.63 \* MEDIAN \* 1.63 \* STL DEV \* 0.26 \* RANGE \* 1.13 TO 2.20 WITH 37 SAMPLES

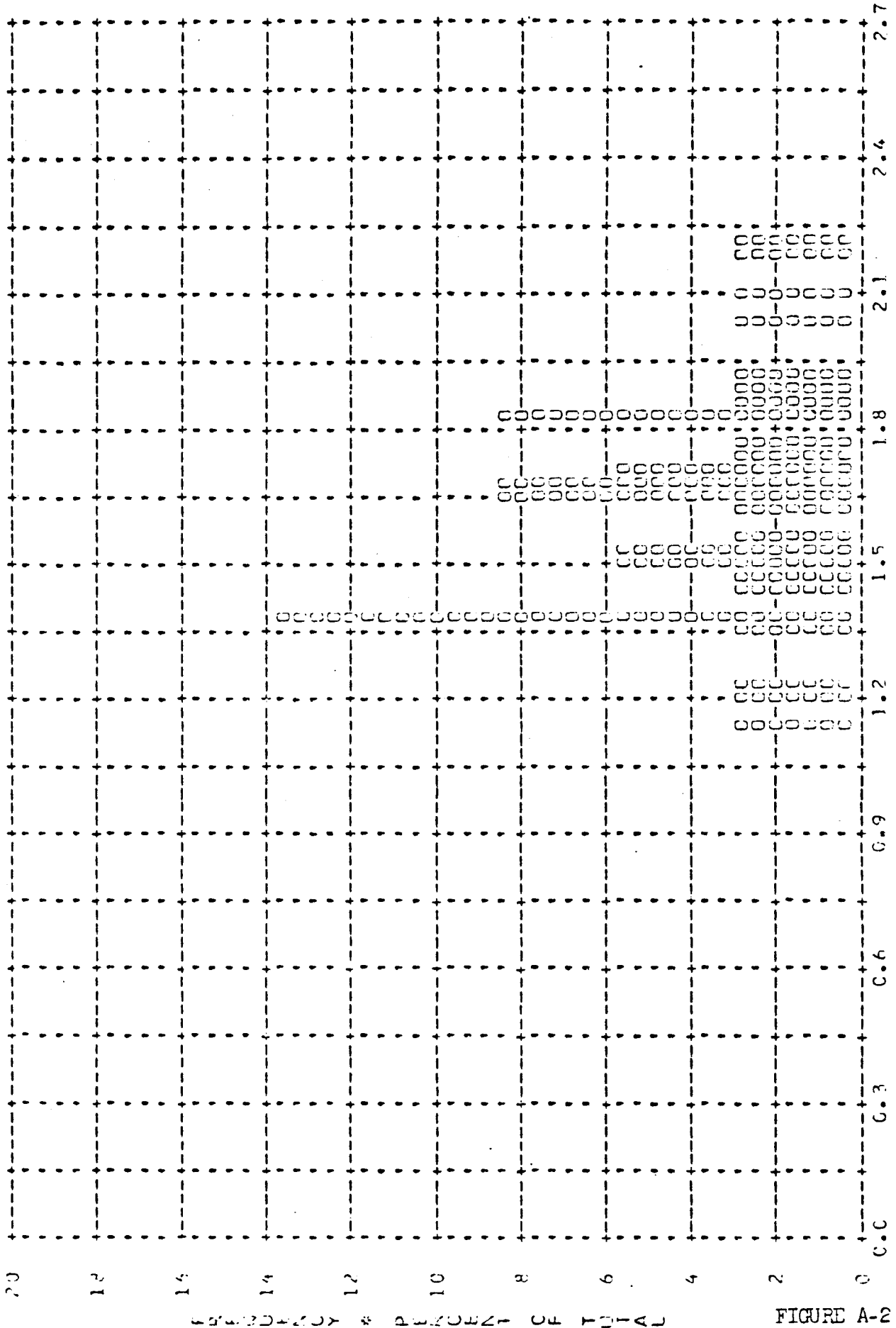


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ C

~~SECRET~~

MISSION \* 1044-1 \* INSIP \* AFI \* 1/16/68 PLCT OF D MAX \* CLOUD \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 2.07 \* MEDIAN \* 2.12 \* STD DEV \* 0.24 \* RANGE \* 1.44 TO 2.46 WITH 22 SAMPLES

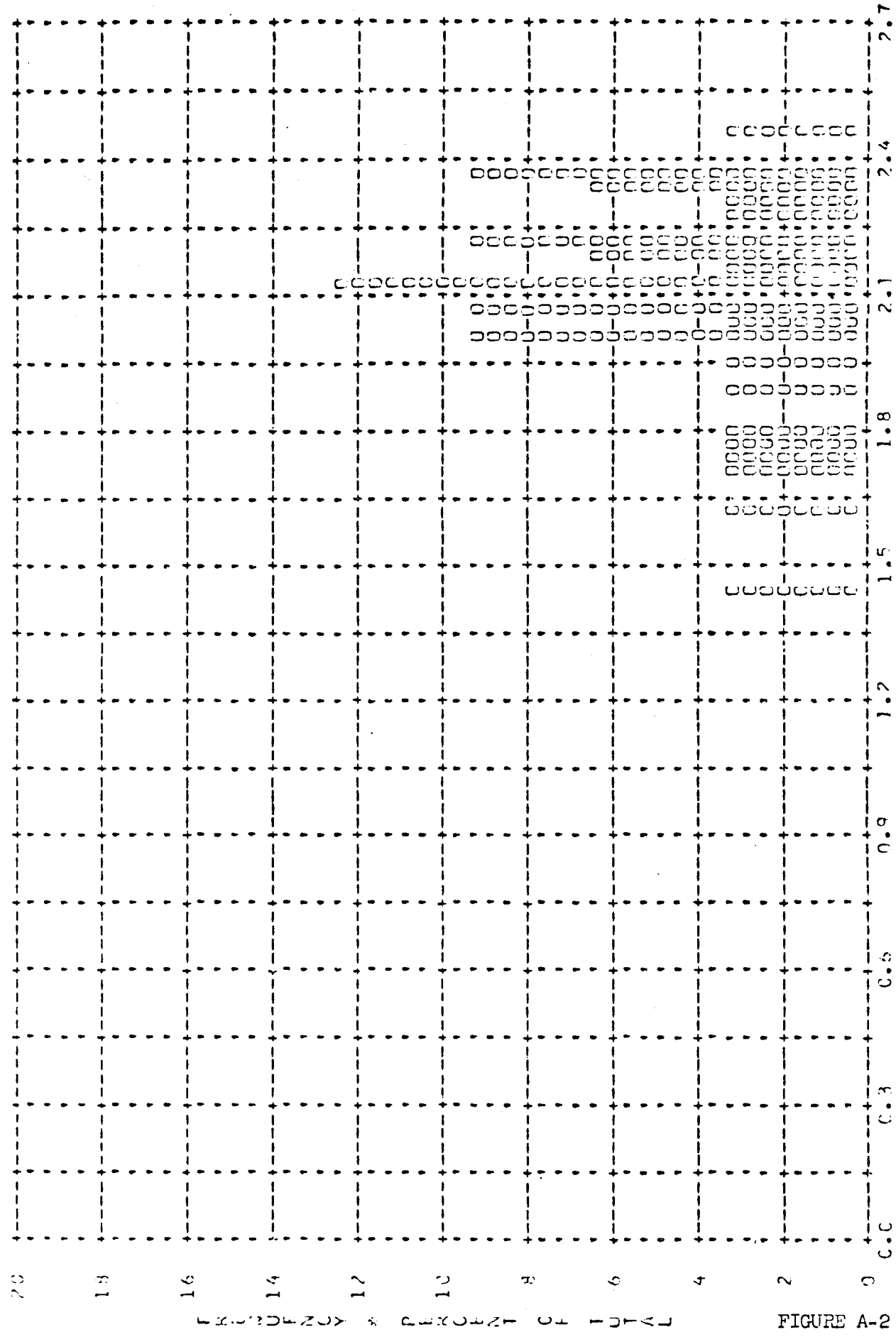


FIGURE A-2

~~TOP SECRET~~

~~TOP SECRET~~ C

MISSION \* 1044-1 \* TASK \* AFT \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* FULL  
WITH MEAN \* 0.55 \* MEDIAN \* 0.50 \* STD DEV \* 0.22 \* RANGE \* 0.27 TO 1.42 WITH 126 SAMPLES

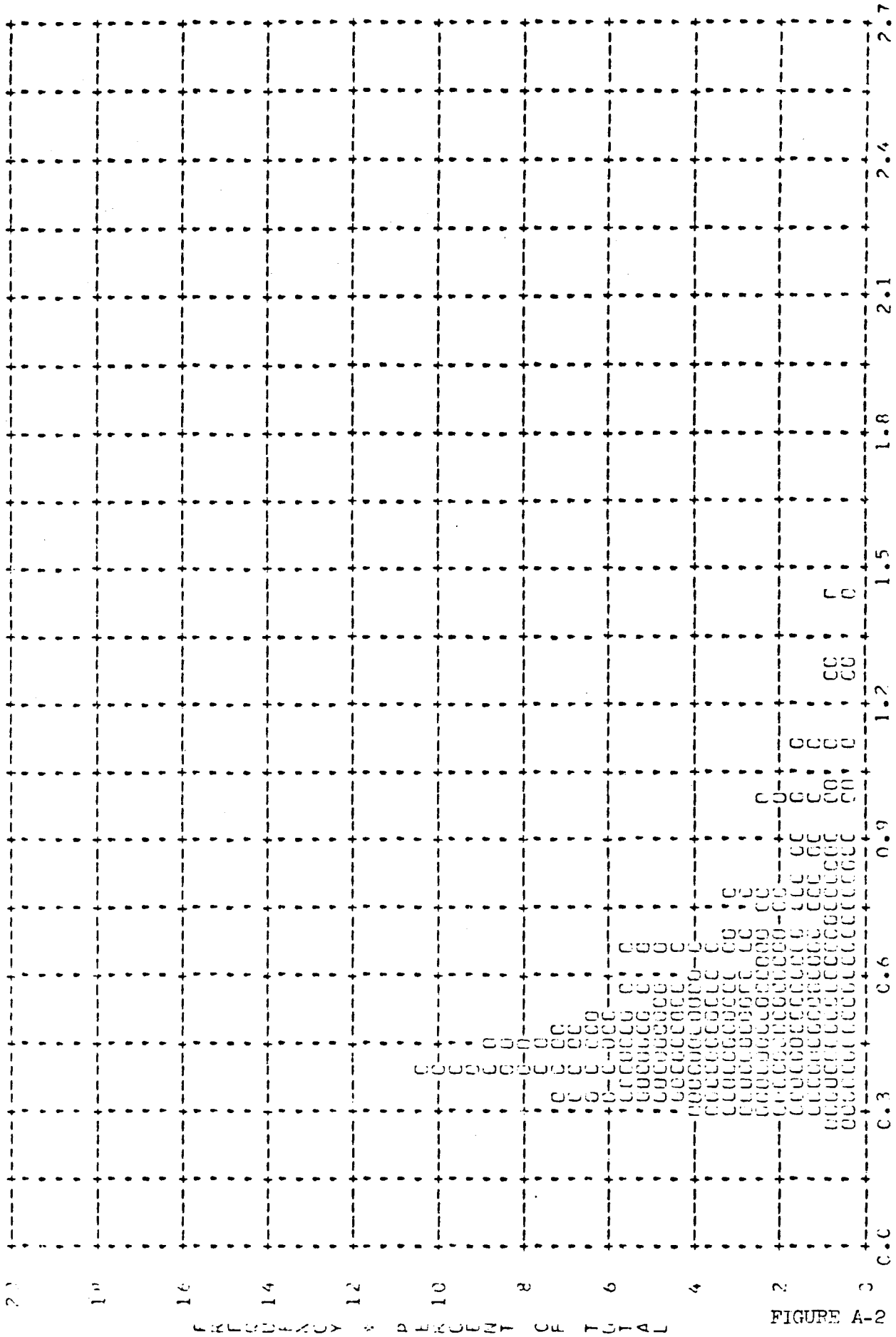


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ C

~~TOP SECRET~~ C

MISSION \* 1044-1 \* INSTP \* APT \* 1/16/63 PLCT OF D MAX \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 1.55 \* MEDIAN \* 1.54 \* STD DEV \* 0.36 \* RANGE \* 0.76 TO 2.41 WITH 126 SAMPLES

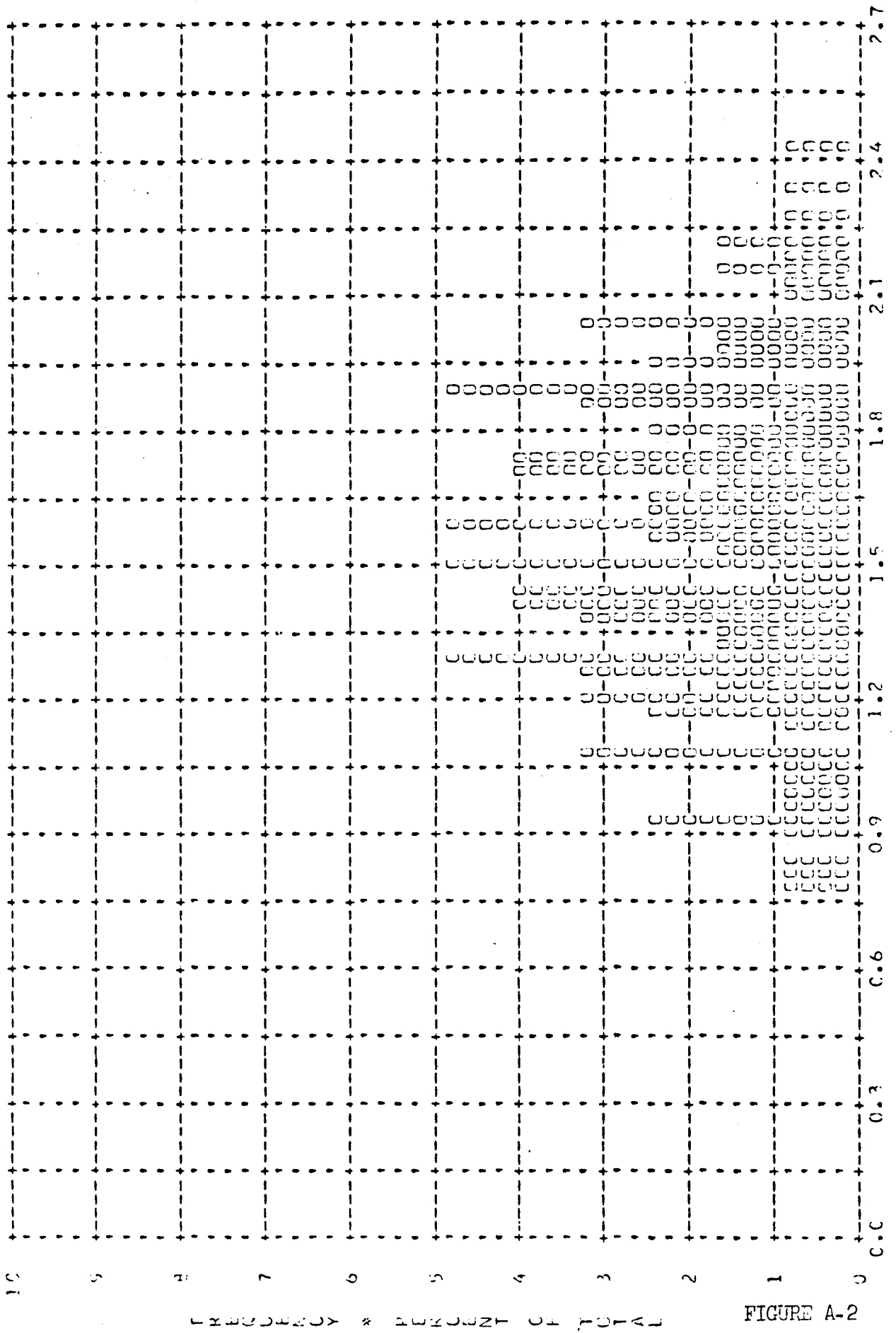


FIGURE A-2

\* DENSITY

~~TOP SECRET~~ C

MISSILE \* 1044-1 \* INSTR \* AFT \* 1/16/68 PLOT CF D MAX \* CLOUD \* PROCESSING \* FULL  
WRITE PEAK \* 2.0P \* MEDIAN \* 2.13 \* STD DEV \* 0.37 \* RANGE \* 0.22 TO 2.47 WITH 104 SAMPLES

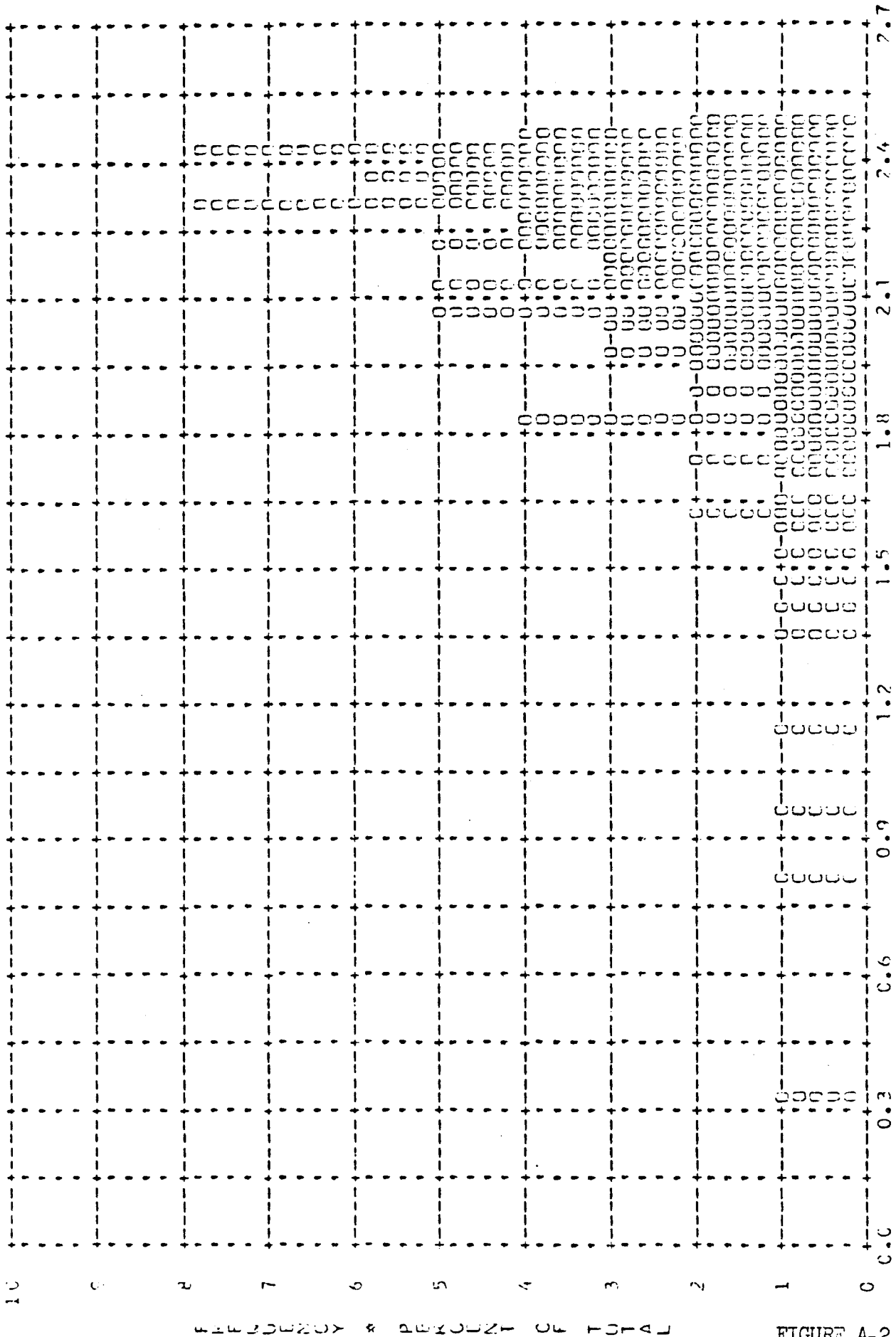


FIGURE A-2

~~TOP SECRET~~ C

MISSION # 1044-1 \* INSTR # AFT \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* ALL LEVELS  
ARIP MEAN \* 0.55 \* MEAN \* 0.53 \* STD DEV \* 0.27 TO 1.45 WITH 163 SAMPLES

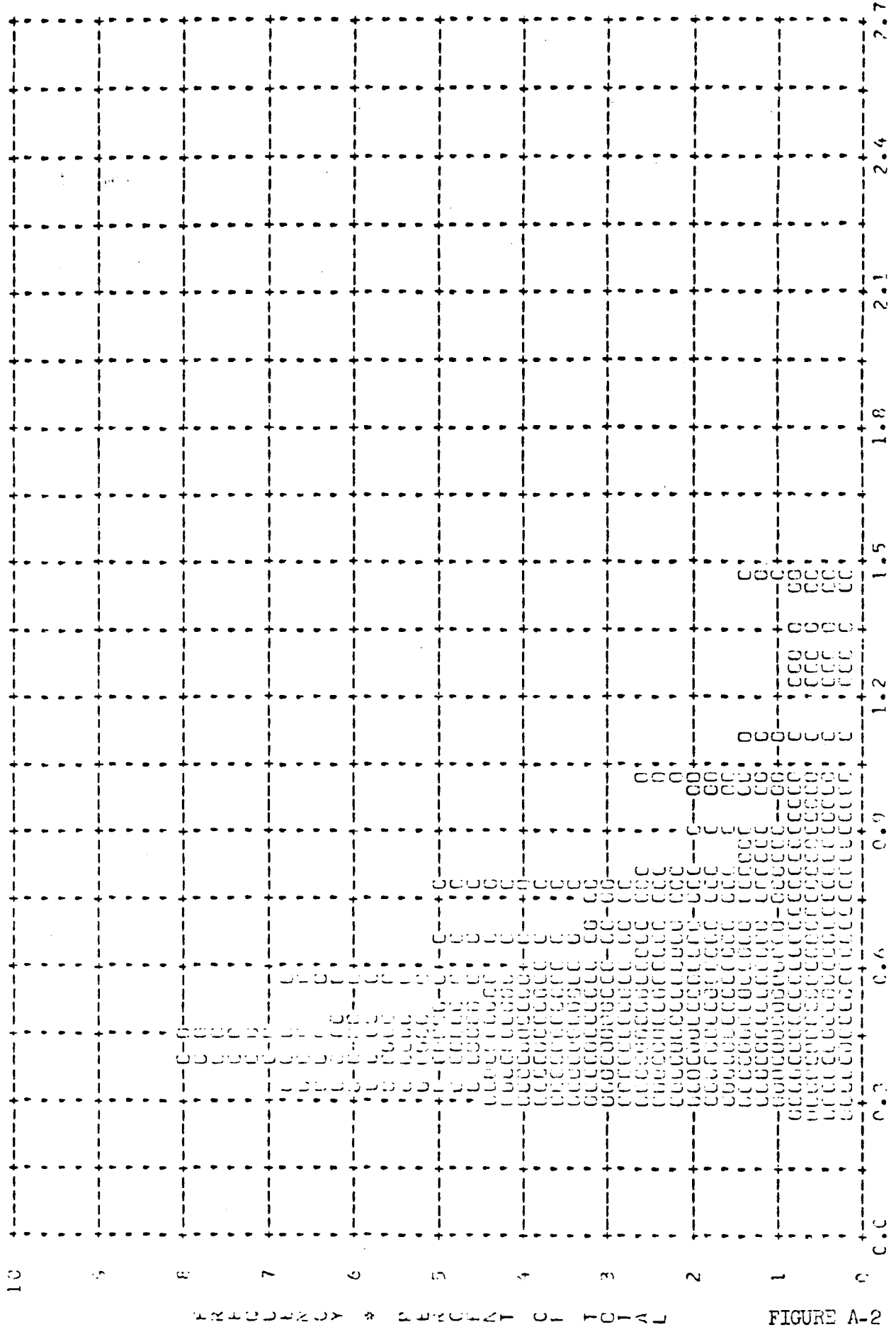


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ C



TOP SECRET

MISSION # 1046-1 \* INSTR # AFI \* 1/16/68 \* PLOT OF 0 MAX \* TERRAIN \* PROCESSING \* ALL LEVELS  
 ARITH MEAN \* 1.57 \* MEDIAN \* 0.74 \* RANGE \* 0.76 TO 2.41 WITH 163 SAMPLES

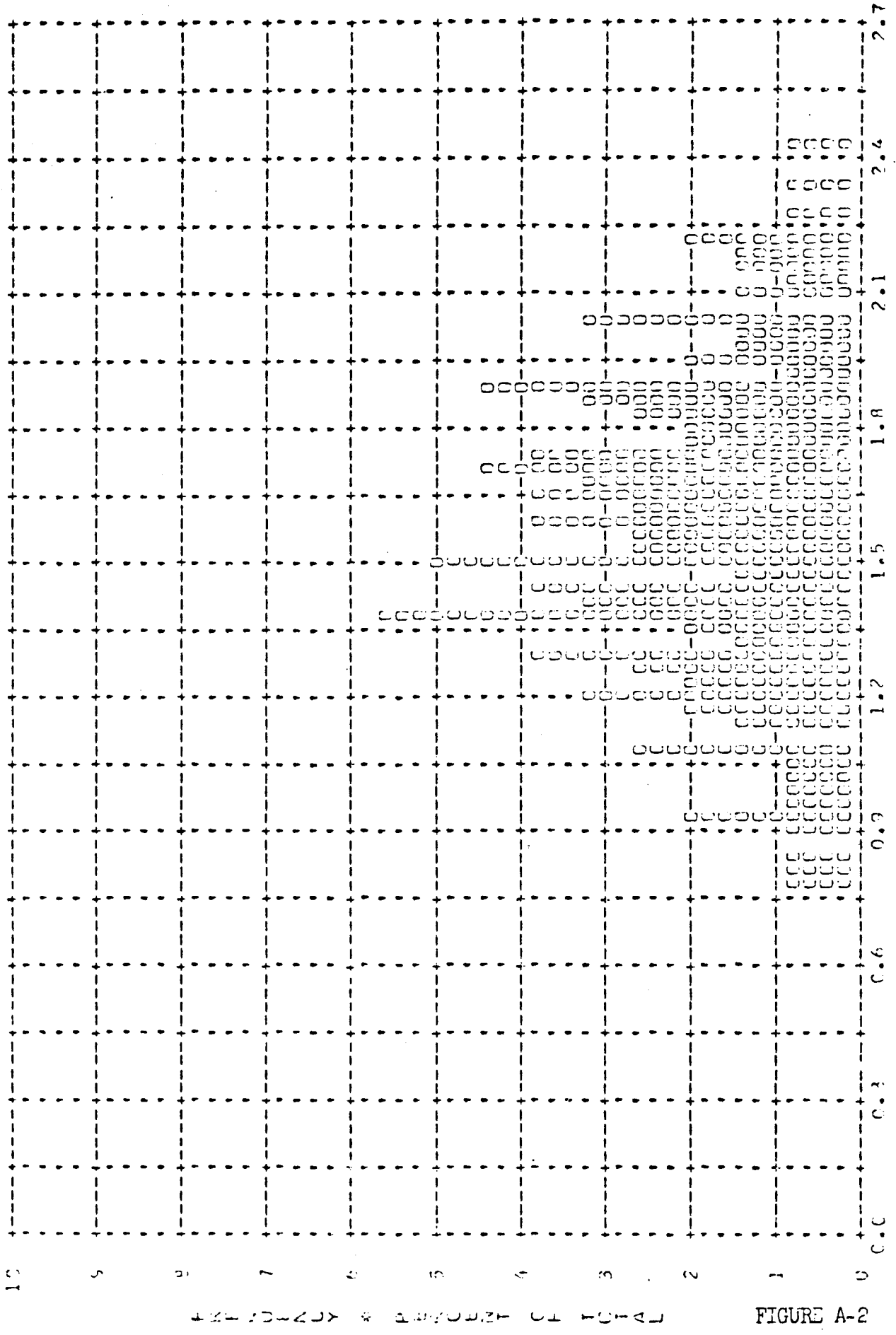


FIGURE A-2

TOP SECRET

~~TOP SECRET~~ CA

MISSION \* 10-0-1 \* INSTR \* AFT \* 1/16/68 FLCT CF D MAX \* CLOUD \* PROCESSING \* ALL LEVELS  
AFTL \* MEAS \* 4.00 \* MEAS \* 2.15 \* STD DEV \* 0.35 \* RANGE \* 0.72 TO 2.47 WITH 137 SAMPLES

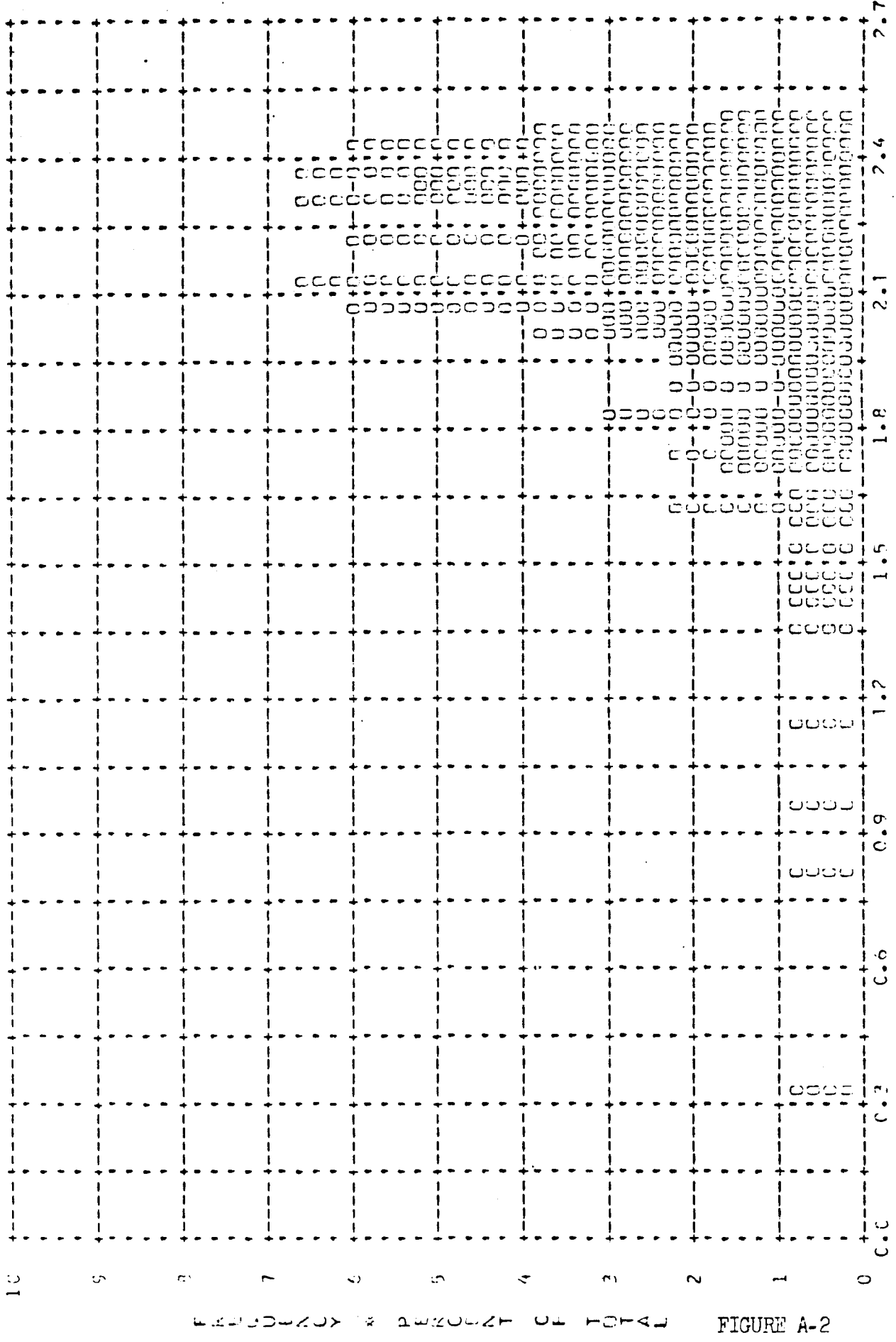


FIGURE A-2

~~TOP SECRET~~ CA

~~TOP SECRET~~ C

MISSION \* AD49-1 \* INSTP \* REF \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* DUAL GAMMA  
GRITH MEAN \* 0.42 \* MEDIAN \* 0.41 \* STD DEV \* 0.20 \* RANGE \* 0.26 TO 1.05 WITH 68 SAMPLES

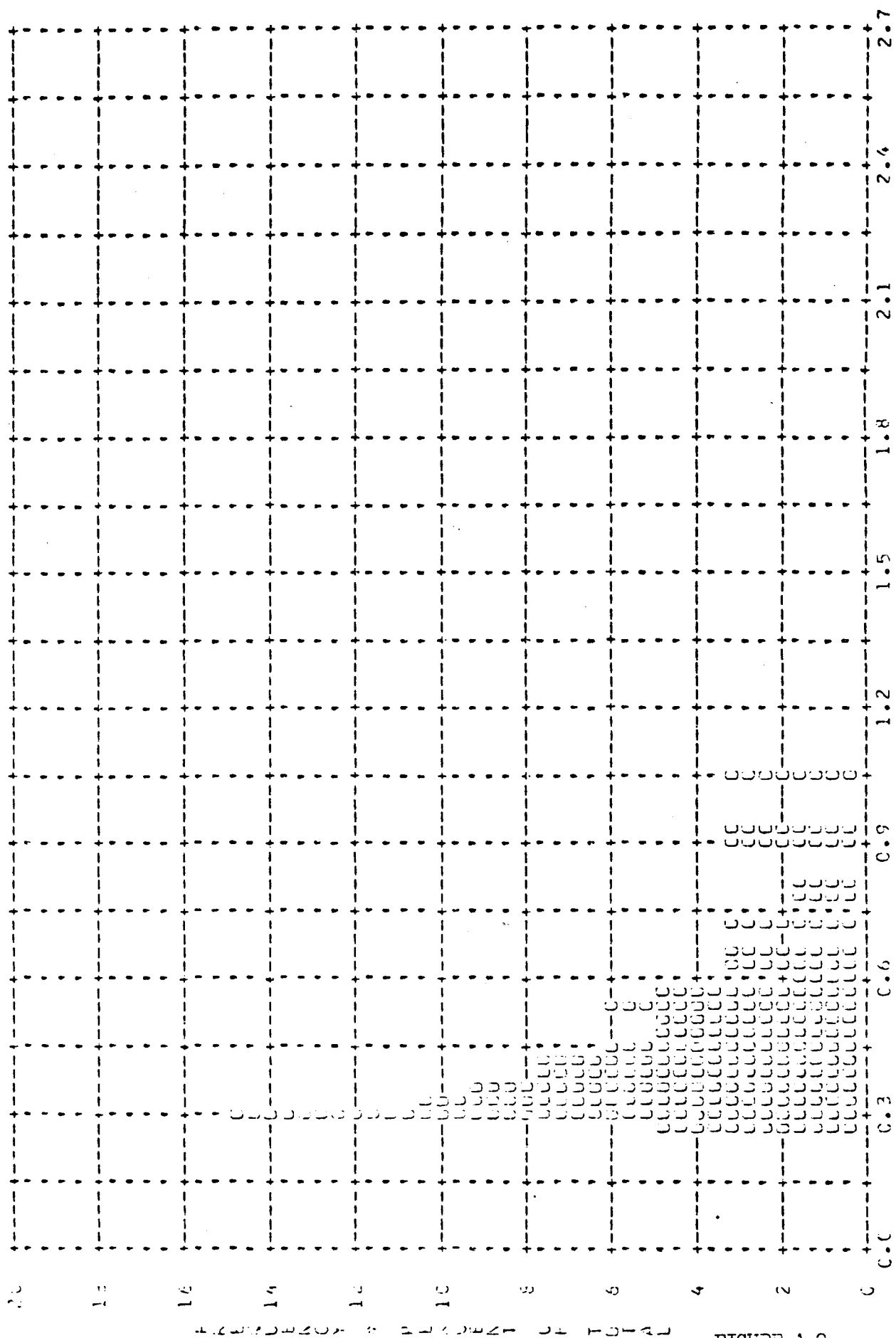


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ C

~~TOP SECRET~~ CA

MISSION \* 1044-1 \* INSTR \* AFT \* 1/16/68 PLCT CF D MAY \* TERRAIN \* PROCESSING \* DUAL GAMMA  
ARITH MEAN \* 1.52 \* REGION \* 1.42 \* STD DEV \* 0.35 \* RANGE \* 0.42 TO 1.75 WITH 68 SAMPLES

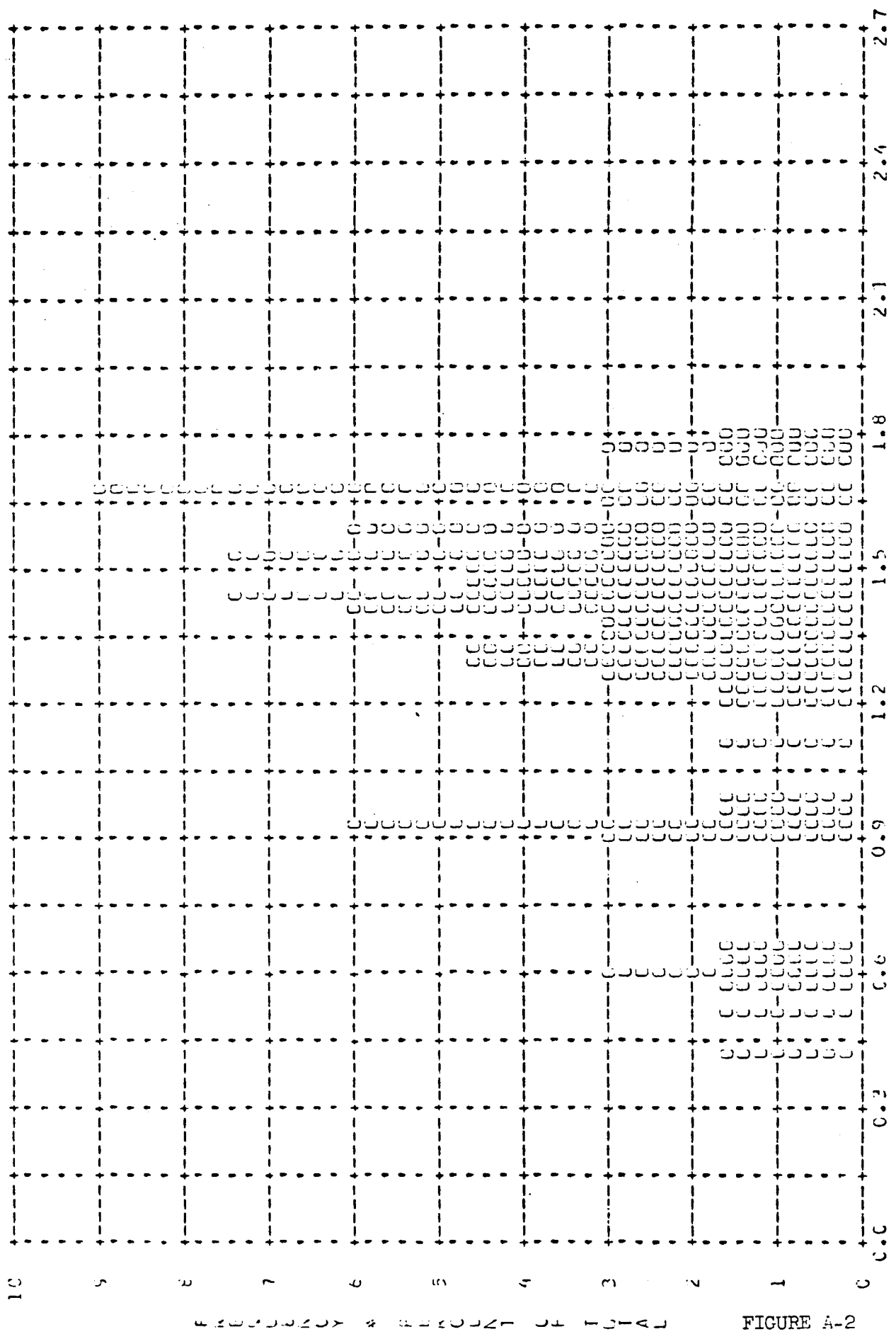


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ C

~~TOP SECRET~~ U

MISSION \* 1046-1 \* INSTR \* AFT \* 1/16/68 FLCT CF D MAX \* CLCUD \* PROCESSING \* DUAL GAMMA  
ARITH MEAN \* 1.01 \* MEDIAN \* 1.71 \* STD DEV \* 0.25 \* RANGE \* 0.7C TO 1.8B WITH 58 SAMPLES

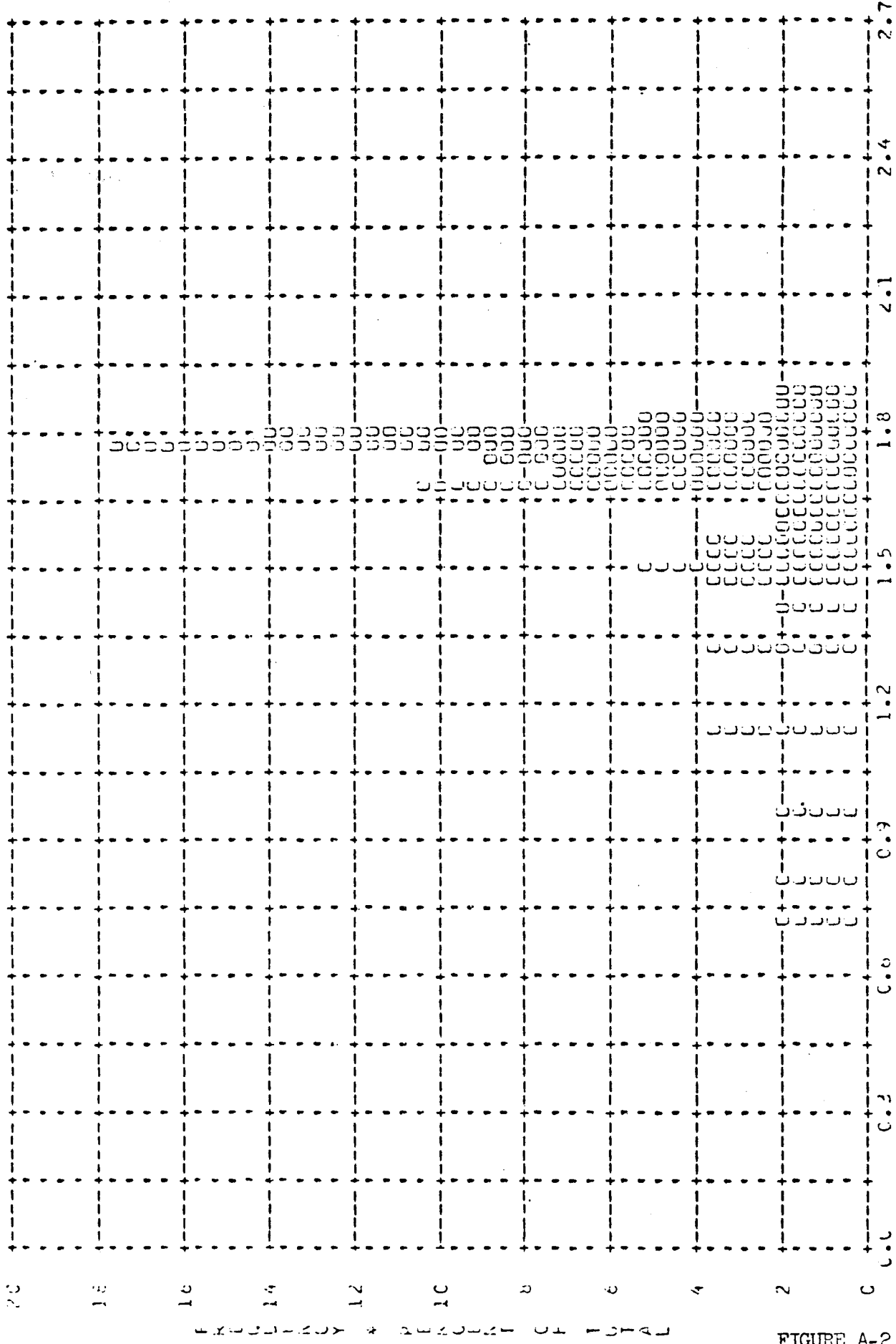


FIGURE A-2

\* DENSITY \*

~~TOP SECRET~~ U

~~TOP SECRET~~

MISSION \* 1044-2 \* INSTR \* FWD \* 1/16/68 \* PLOT OF D MIN \* TERRAIN \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 0.26 \* MEDIAN \* 0.94 \* STD DEV \* 0.16 \* RANGE \* 0.55 TO 1.07 WITH 12 SAMPLES

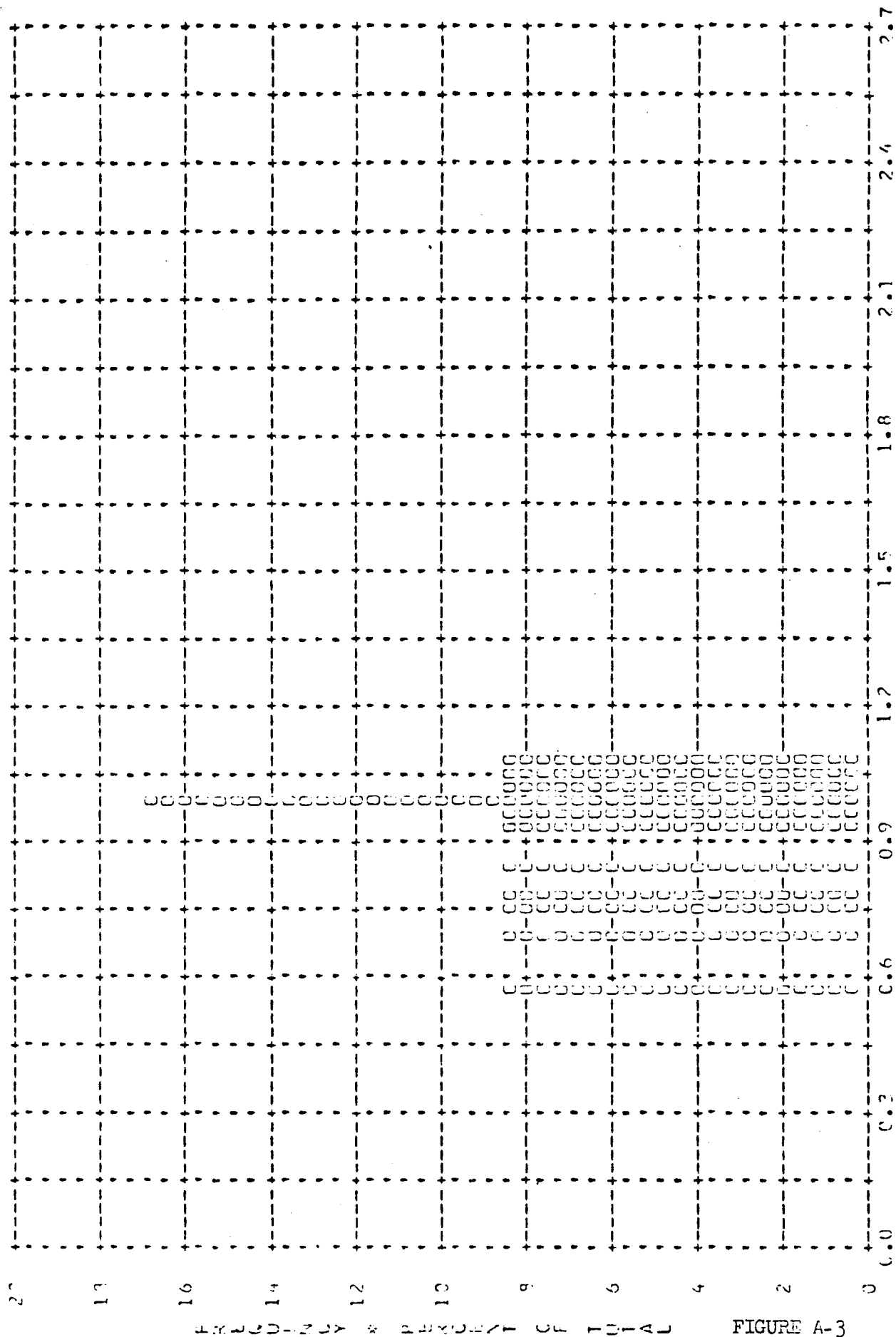


FIGURE A-3

~~TOP SECRET~~

~~TOP SECRET~~

MISSION # 1044-2 \* INSIP \* FWD \* 1/16/68 \* PLCT CF D MAX \* TERRAIN \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 1.77 \* MEDIAN \* 1.77 \* STD DEV \* 0.19 \* RANGE \* 1.51 TO 2.19 WITH 12 SAMPLES

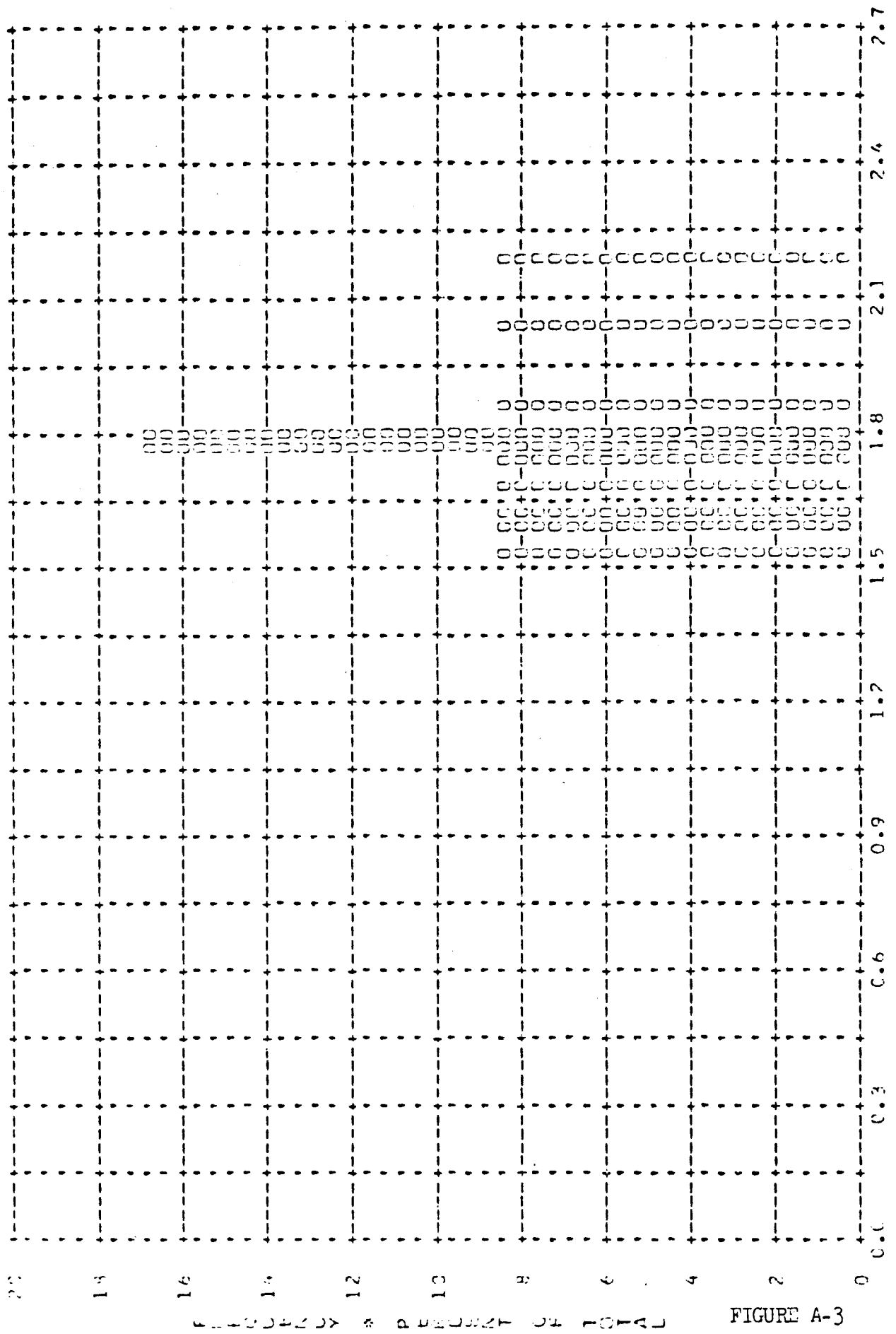


FIGURE A-3

\* DENSITY \*

~~TOP SECRET~~

[REDACTED]

MISSION # 1044-2 \* INSIR # FWD \* 1/16/68 PLOT OF P MAX # CLOUD \* PROCESSING \* INTERMEDIATE  
APRIL MEAN # 2.08 \* MEDIAN # 2.23 \* STD DEV # 0.26 \* RANGE # 1.65 TO 2.37 WITH 8 SAMPLES

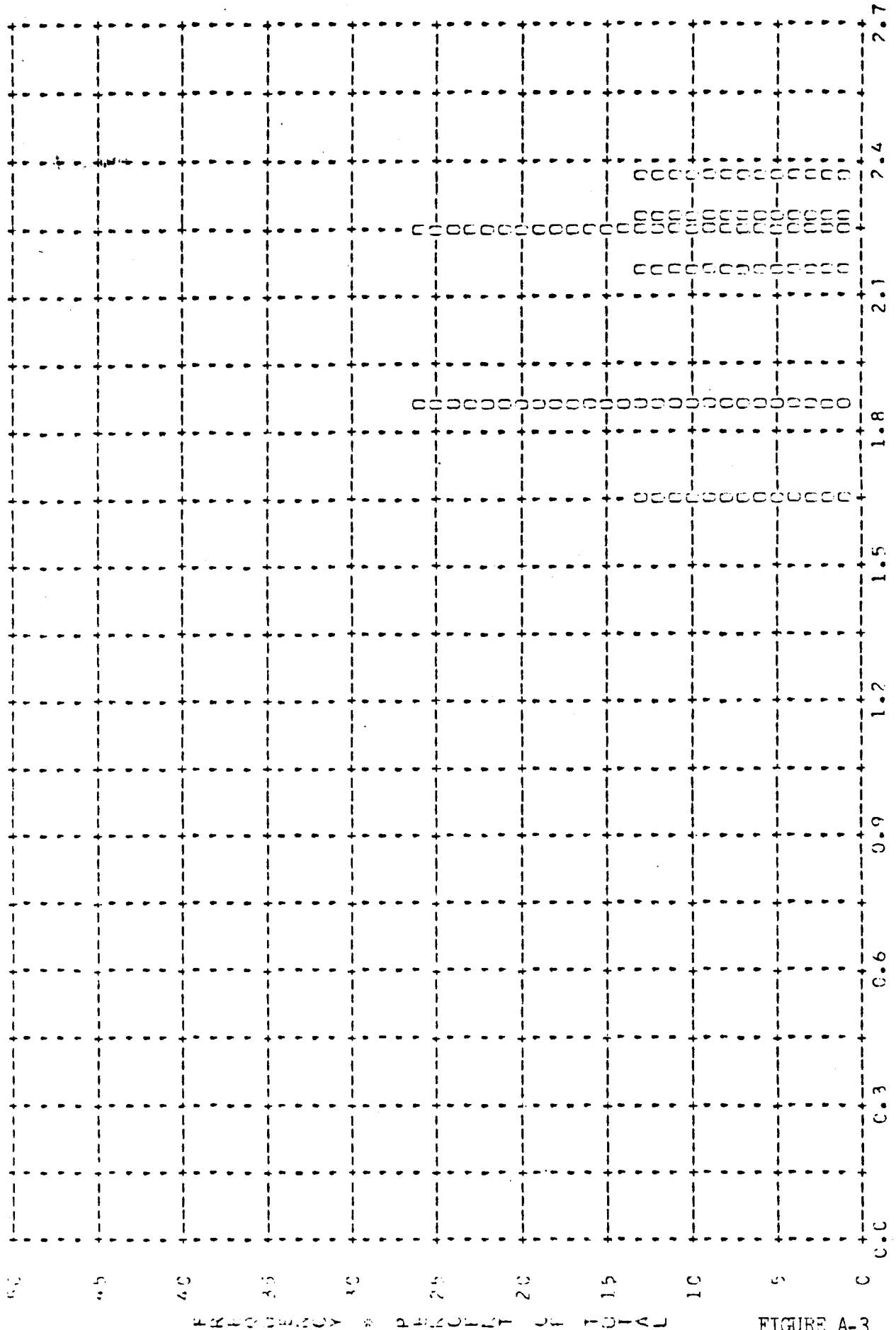


FIGURE A-3

[REDACTED]



TOP SECRET

MISSION # 1044-2 \* INSTR # EMP \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 0.50 \* MEDIAN \* 0.47 \* STD DEV \* 0.21 \* RANGE \* 0.29 TO 1.32 WITH 168 SAMPLES

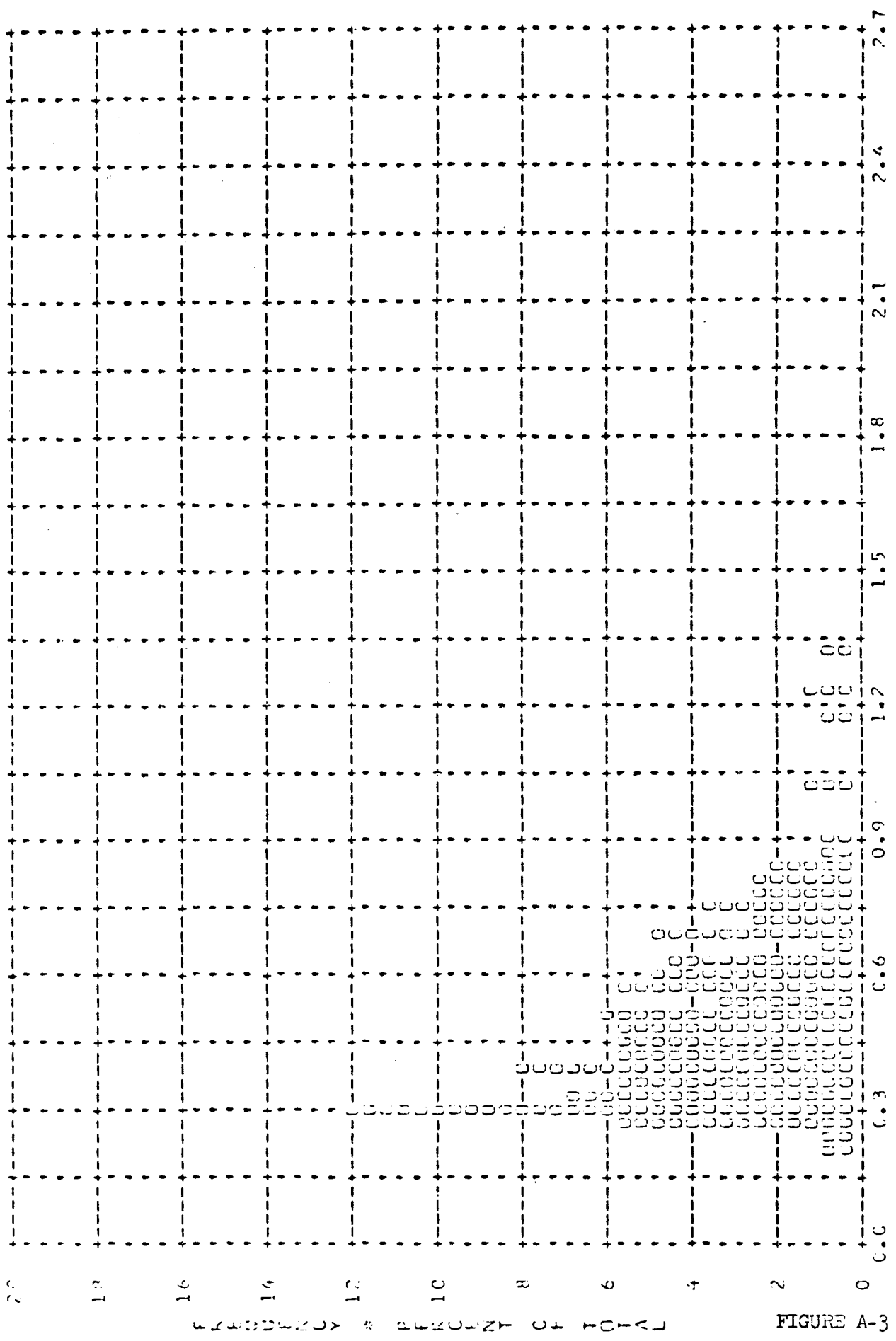


FIGURE A-3

\* DENSITY \*

TOP SECRET

SECRET

MISSION # 1044-2 \* INSTR # FWD \* 1/16/68 PLCT OF C MAX \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN # 1.52 \* MEDIAN # 1.53 \* STD DEV # 0.42 \* RANGE # 0.40 TO 2.37 WITH 168 SAMPLES

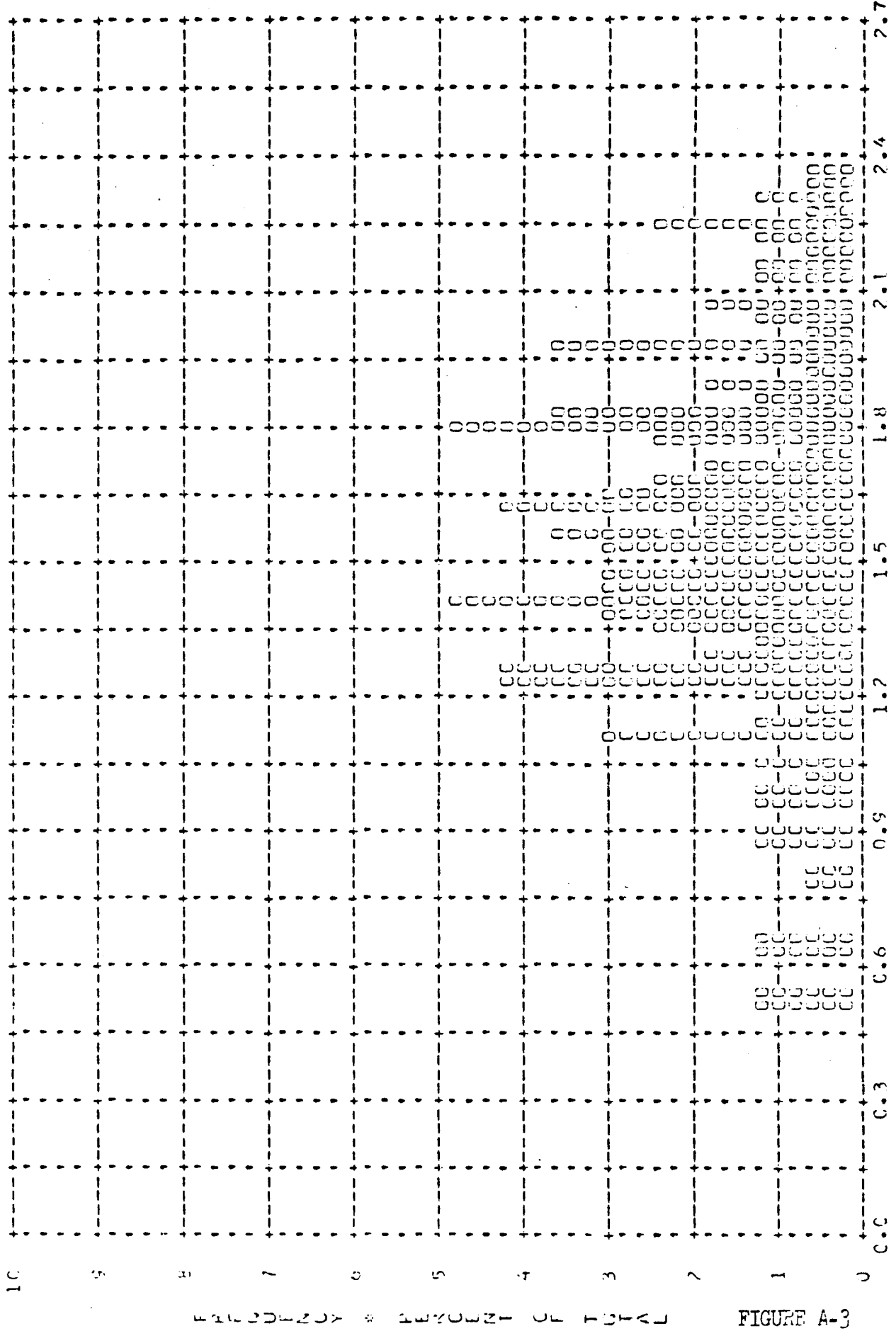


FIGURE A-3

\* DENSITY \*

SECRET

MISSION # 1044-2 \* INSTR # FWD \* 1/16/68 PLOT OF D VAX \* CLCUD \* PROCESSING \* FULL  
 ARITH MEAN \* 2.00 \* MEDIAN \* 2.24 \* STD DEV \* 0.41 \* RANGE \* 0.76 TO 2.49 WITH 162 SAMPLES

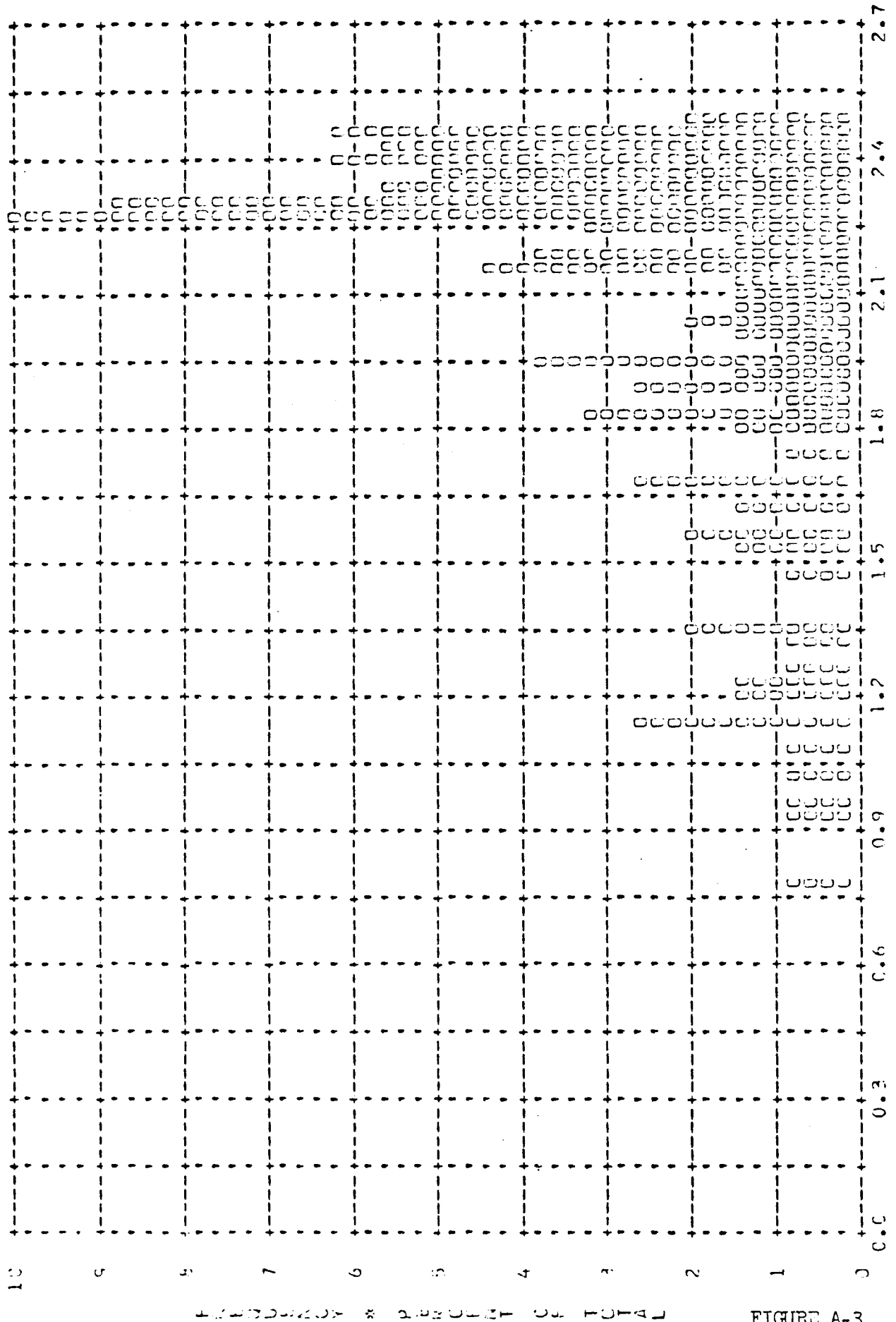


FIGURE A-3

~~TOP SECRET~~

MISSION \* 1044-2 \* INSTR \* F42 \* 1/16/68 PLCT CF D MIN \* TERPAIN \* PROCESSING \* ALL LFVFLS  
ARITH MEAN \* 0.53 \* MEDIAN \* 0.49 \* STD DEV \* 0.23 \* RANGE \* 0.20 TO 1.32 WITH 180 SAMPLES

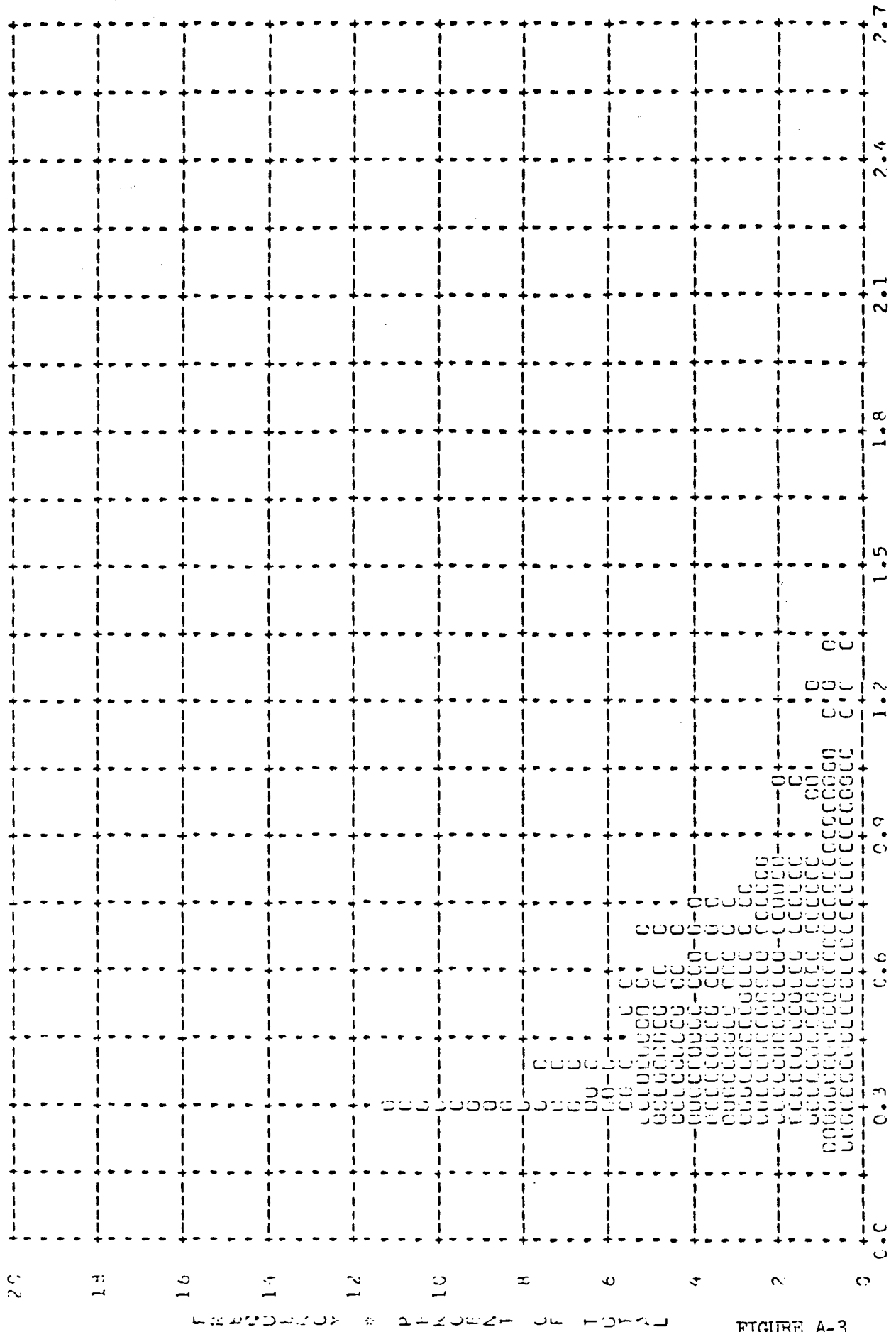


FIGURE A-3

~~TOP SECRET~~

\* DENSITY \*

TOP SECRET C

MISSION # 1044-2 \* INSTE \* FWD \* 1/16/58 PLCT OF D MAX \* TERRAIN \* PROCESSING # ALL LGVLS  
ARITH \* FWD \* 1.56 \* FLUIDIAN \* 1.55 \* STD DEV \* 0.41 \* RANGE \* 0.69 TO 2.37 WITH 180 SAMPLES

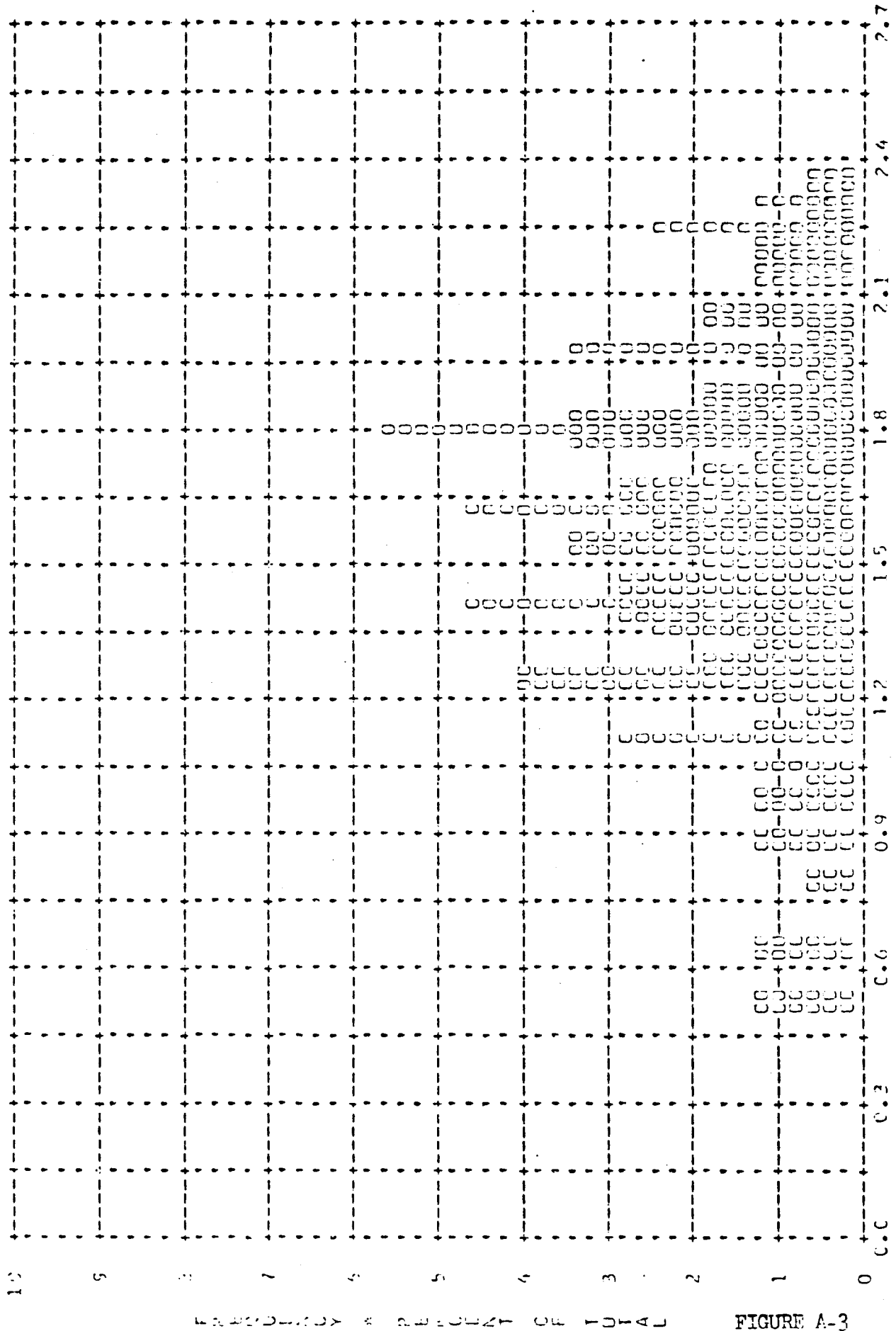


FIGURE A-3

\* DENSITY \*

TOP SECRET C

~~TOP SECRET~~ C

MISSION # 1044-2 \* INSIR # F4D \* 1/16/68 PLCT CF D MAY \* CLCUD \* PROCESSING \* ALL LFVFLS  
ARITH MEAN # 2.06 \* MEDIAN # 2.23 \* STD DEV # 0.40 \* RANGE # 0.76 TO 2.49 WITH 170 SAMPLES

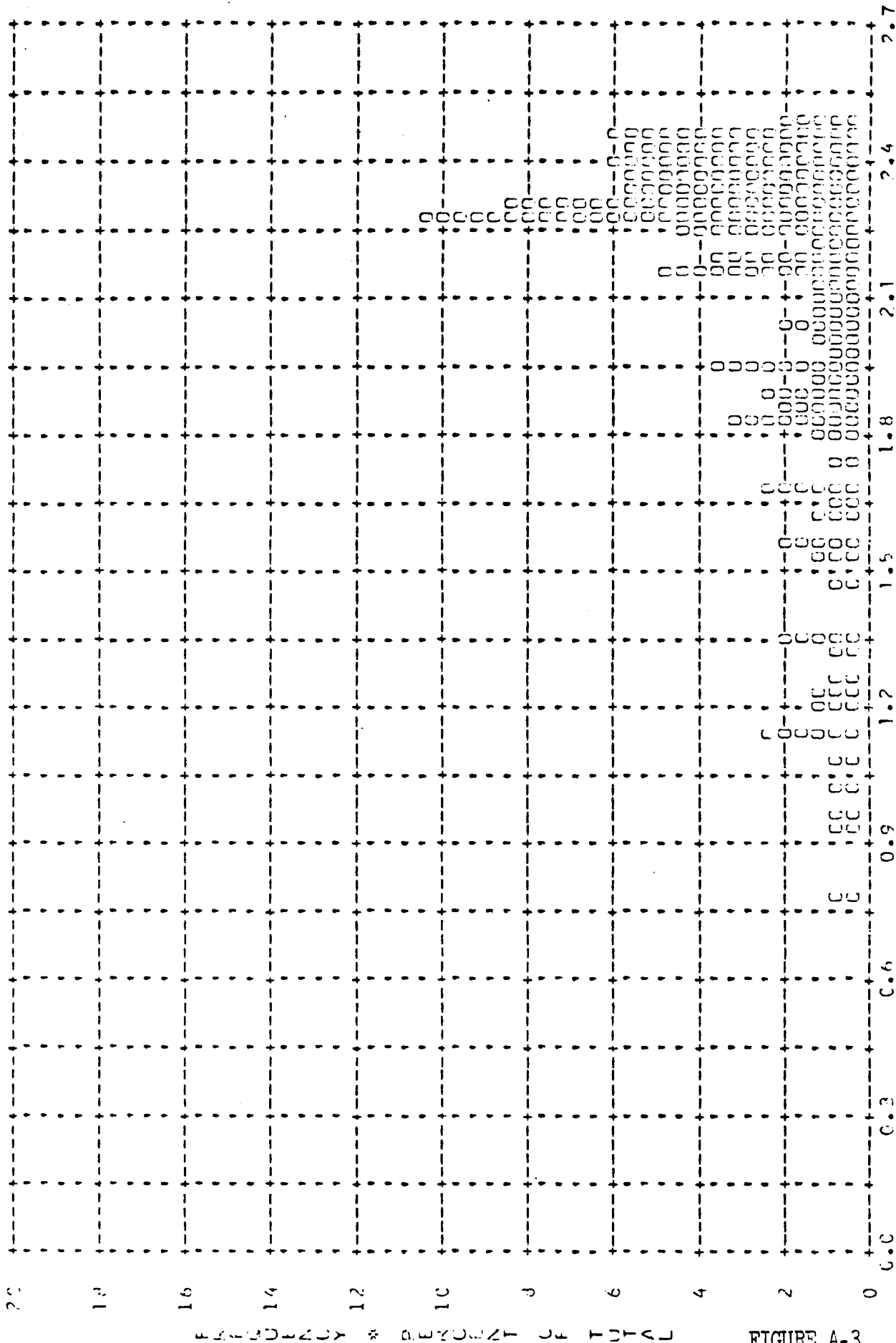


FIGURE A-3

\* DENSITY \*

~~TOP SECRET~~ C

TOP SECRET

MISSION # 1044-2 \* INSTL # FAD \* 1/16/68 PLCT CF 0 4IN \* TERRAIN \* PROCESSING \* DUAL GAUSS  
 WITH MEAN \* 0.67 \* MEDIAN \* 0.38 \* STD DEV \* 0.21 \* RANGE \* 0.26 TO 1.11 WITH 73 SAMPLES

PERCENTILE	0	0.3	0.6	0.9	1.2	1.5	1.8	2.1	2.4	2.7
20										
18										
16										
14										
12		00								
10		0000								
8		00000								
6		000000	00							
4		0000000	0000							
2		00000000	000000	00						
0	0	000000000	0000000000	00000000000	000000000000	0000000000000	00000000000000	000000000000000	0000000000000000	00000000000000000

FIGURE A-3

\* DENSITY \*

TOP SECRET

TOP SECRET

MISSION \* 1044-2 \* INSTR \* FWD \* 1/16/68 PLCT CF D MAX \* TERRAIN \* PROCESSING \* DUAL GAMMA  
 ARITH MEAN \* 1.32 \* MEDIAN \* 1.37 \* STD DEV \* 0.32 \* RANGE \* 0.40 TO 1.81 WITH 73 SAMPLES

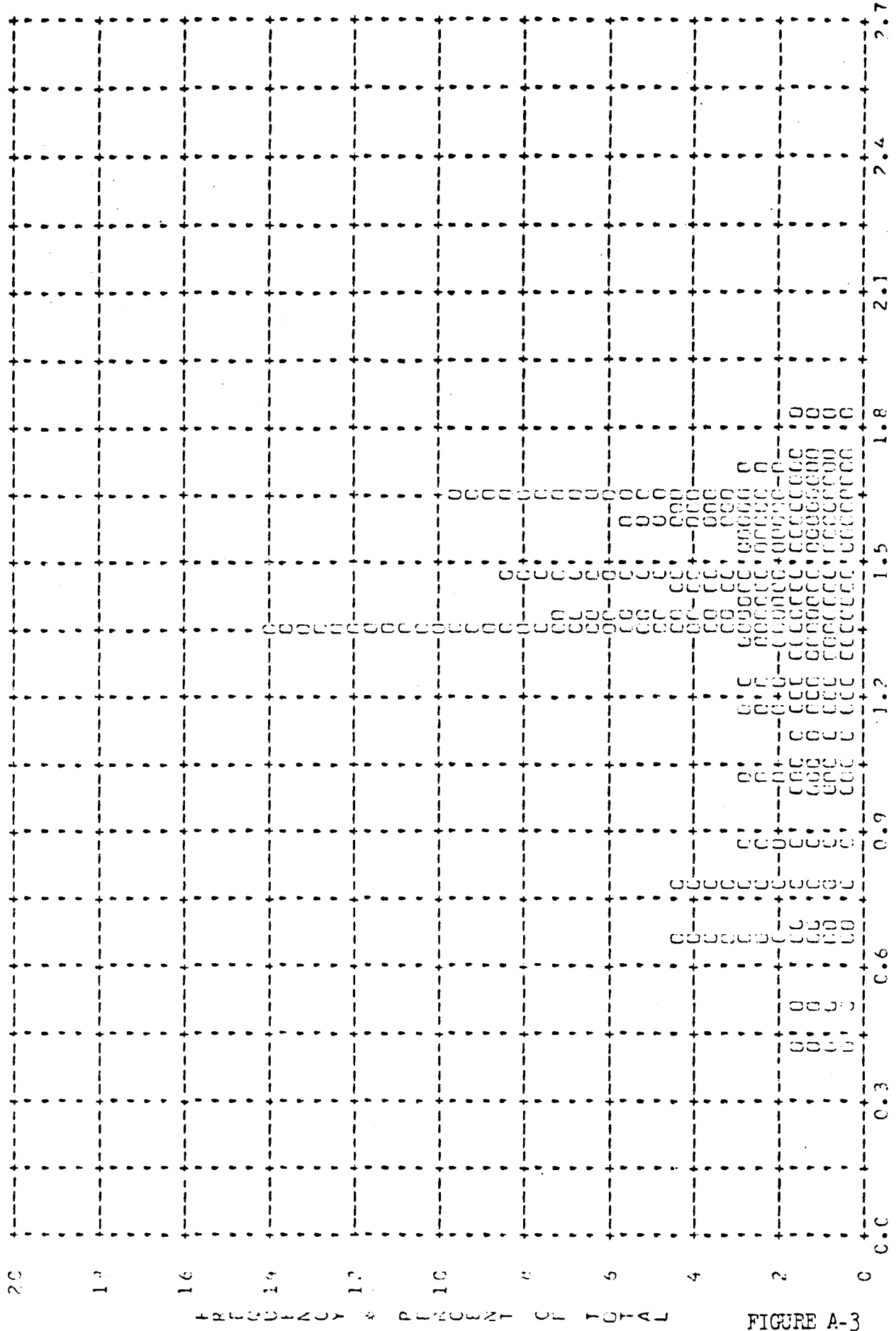


FIGURE A-3

TOP SECRET



TOP SECRET

MISSION # 1044-2 \* TASSIR # FWD \* 1/16/68 PICT CF D MAX \* CLOUD \* PROCESSING \* DUAL GAMMA  
ARITH MEAN # 1.57 \* MEDIAN # 1.64 \* STD DEV # 0.22 \* RANGE # 0.82 TO 1.86 WITH 63 SAMPLES

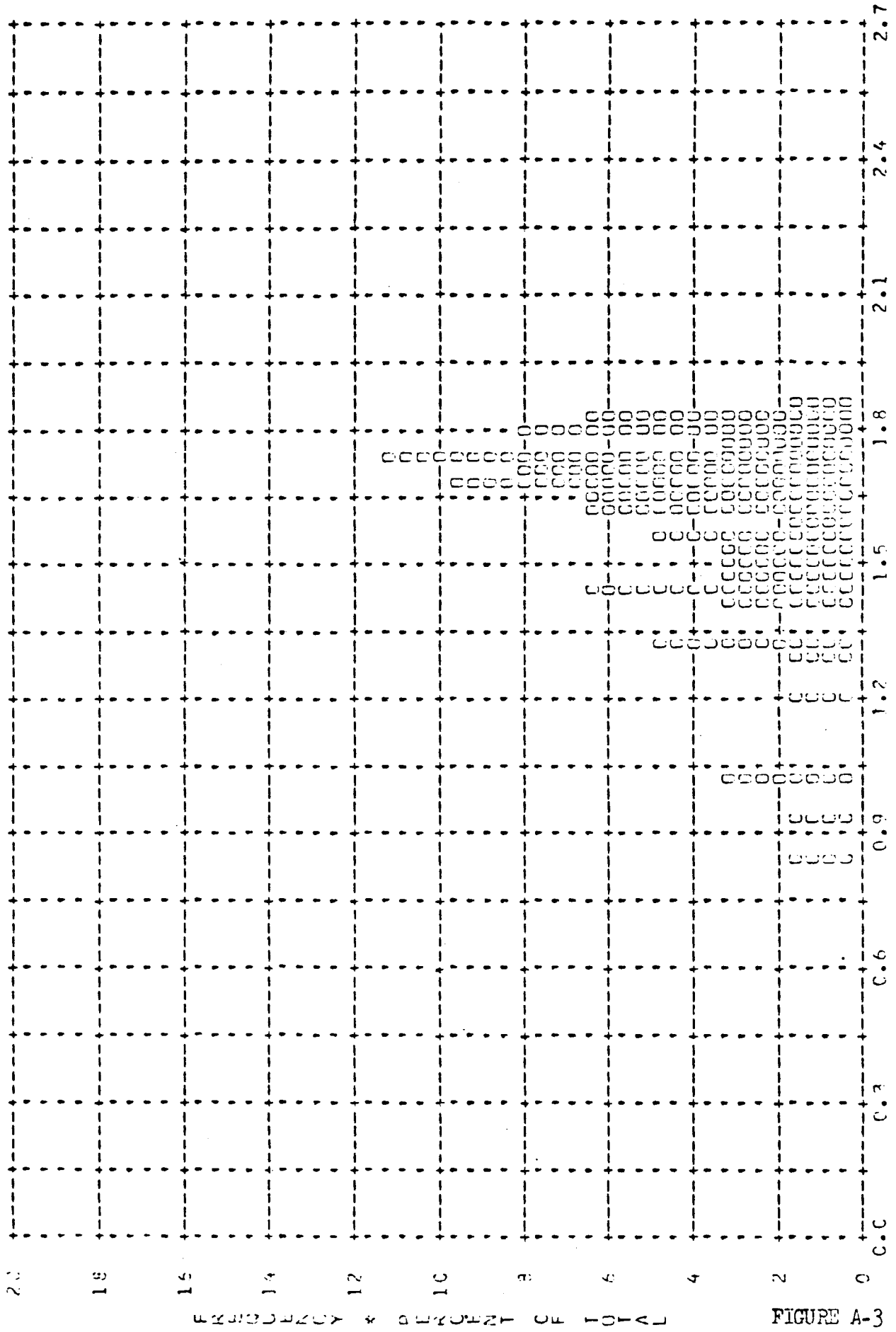


FIGURE A-3

\* DEASITY \*

TOP SECRET

TOP SECRET

MISSION \* 1044-2 \* INSTR \* AFT \* 1/16/68 PLCT CF D MIN \* TERRAIN \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 0.58 \* MEDIAN \* 0.54 \* STD DEV \* 0.26 \* RANGE \* 0.22 TO 1.31 WITH 46 SAMPLES

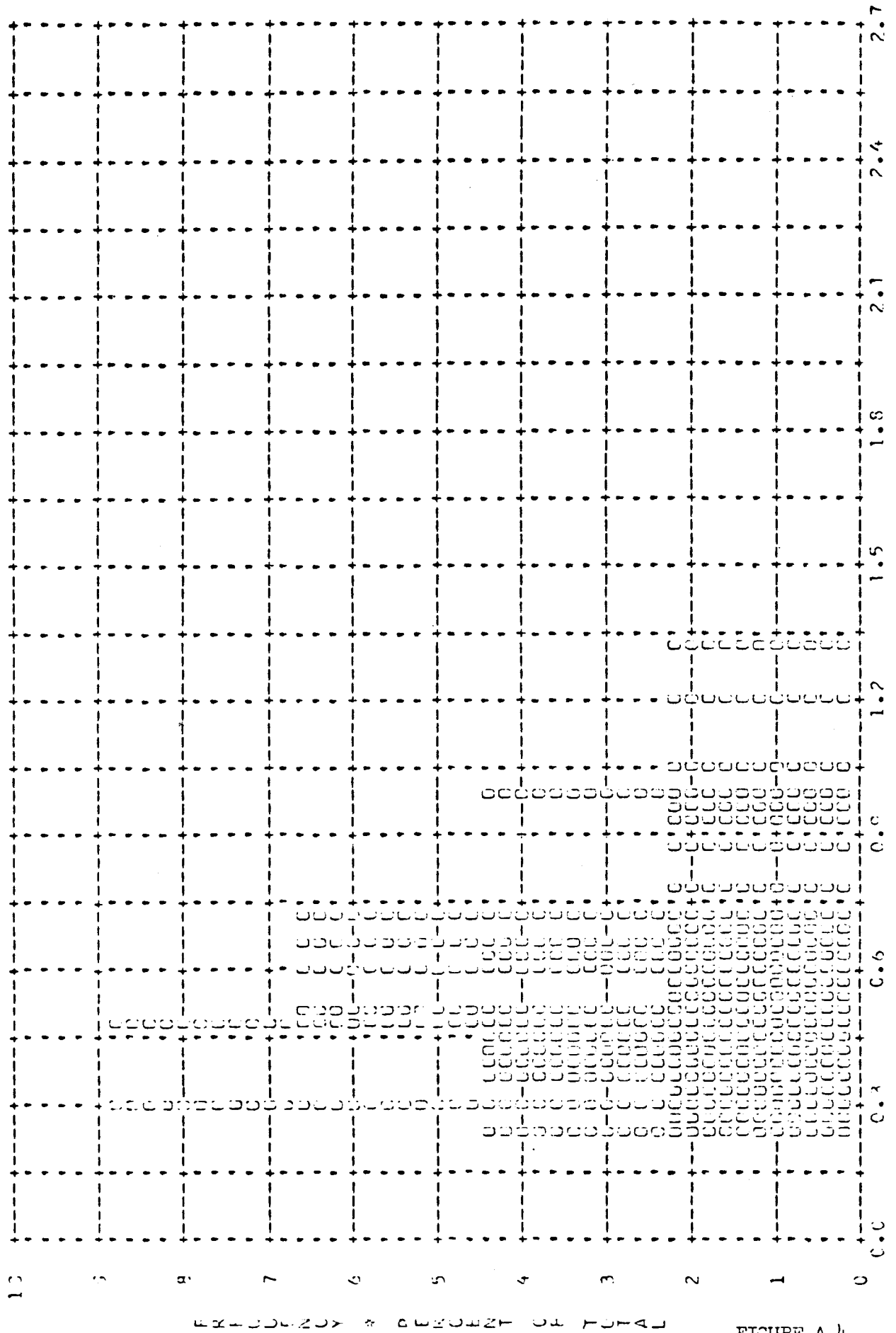


FIGURE A-4

\* DENSITY #

TOP SECRET

~~TOP SECRET~~ C

MISSION \* IG44-2 \* INSTR \* AFT \* 1/16/69 PLCT CF D MAX \* TERRAIN \* PROCESSING \* INTERMEDIATE  
ARITH MEAN \* 1.45 \* MEDIAN \* 1.43 \* STD DEV \* 0.36 \* RANGE \* 0.52 TO 2.07 WITH 46 SAMPLES

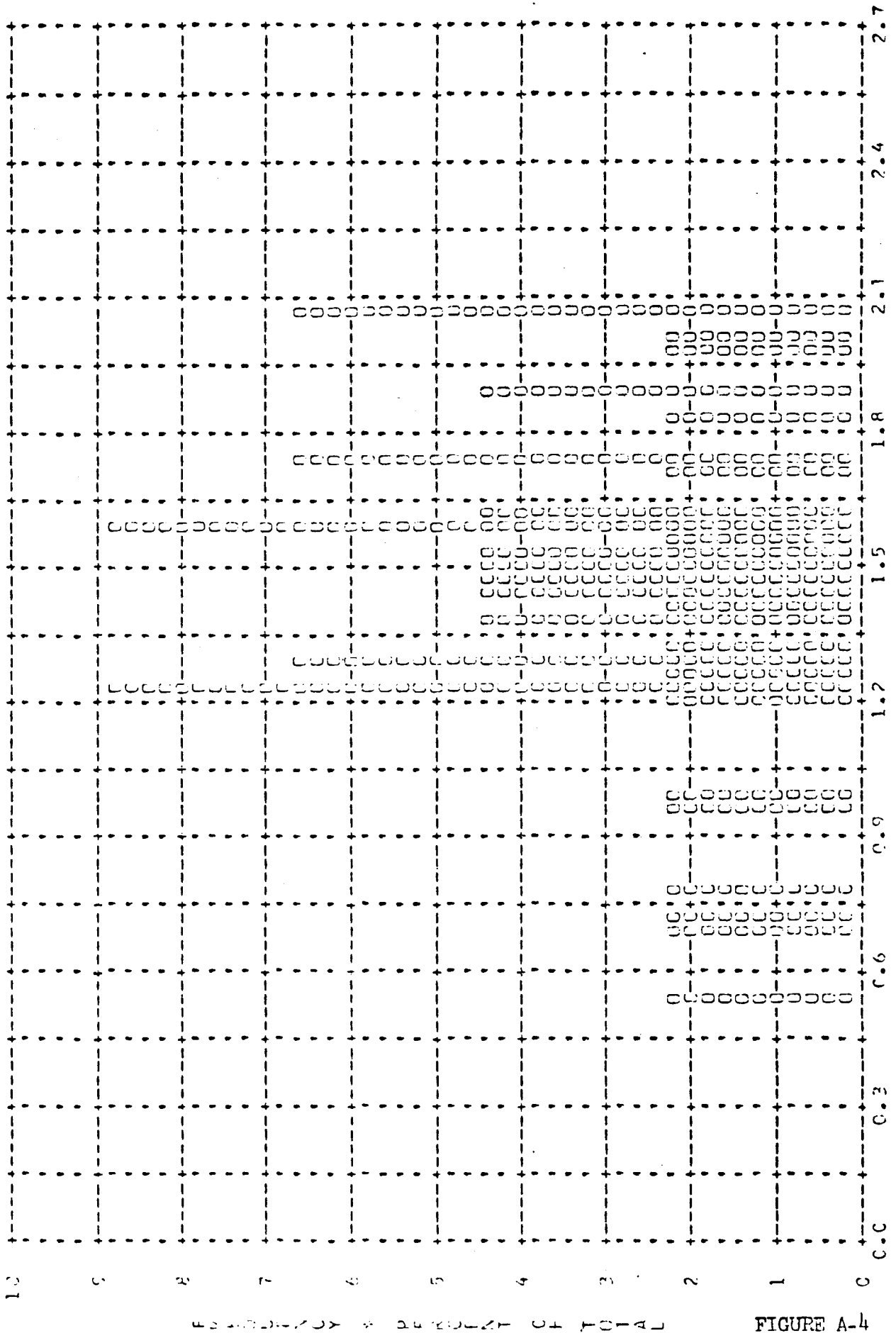


FIGURE A-4

~~TOP SECRET~~ C

TOP SECRET C

MISSION # 1044-2 \* INSIP # AFT # 1/16/63 FLOT OF D MAX # CLCUD \* PROCESSING \* INTERMEDIATE  
ARITH MEAN # 2.10 \* MEDIAN # 2.22 \* STD DEV # 0.47 \* RANGE # 0.44 TO 2.54 WITH 35 SAMPLES

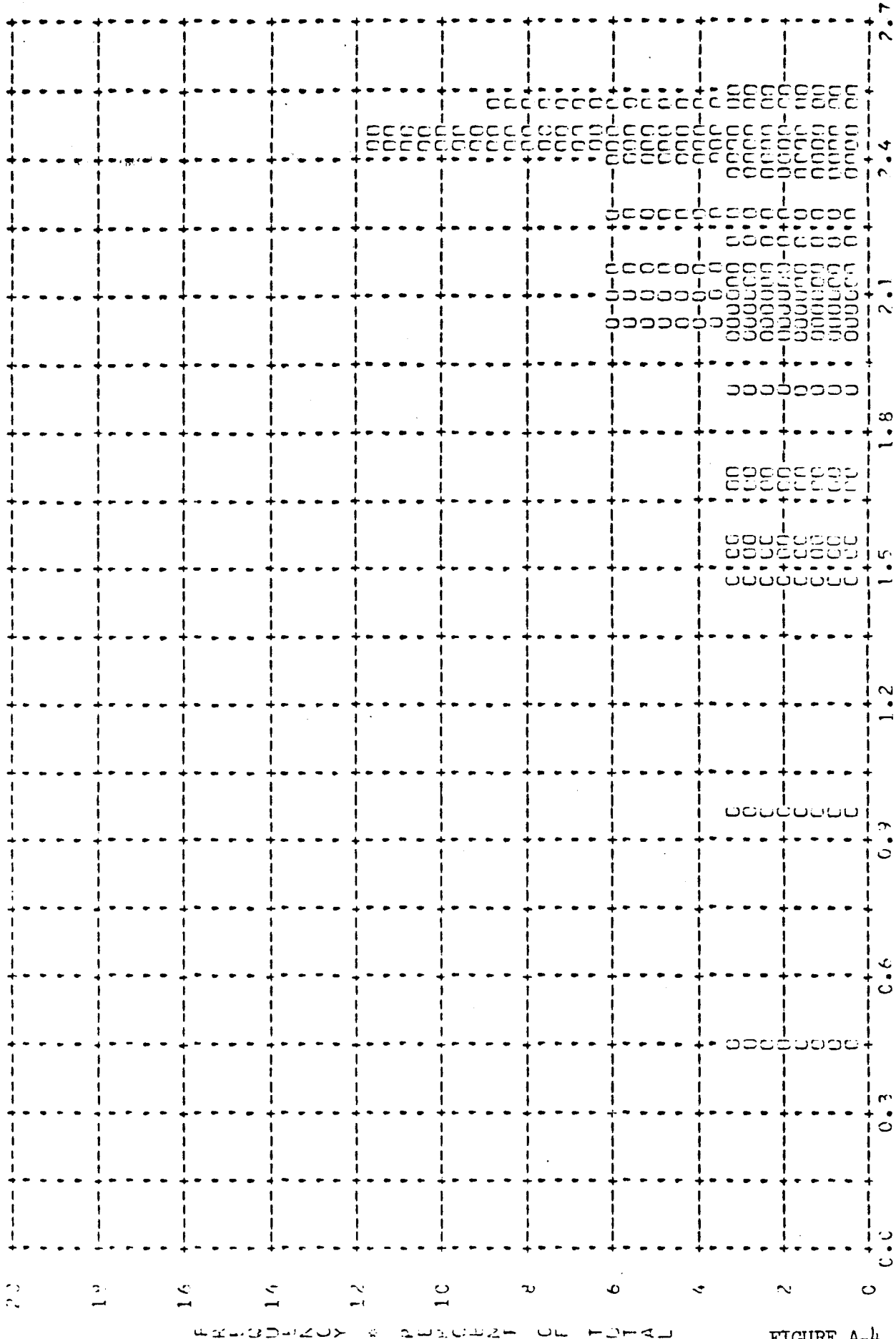


FIGURE A-4

\* DENSITY

TOP SECRET C

SECRET

MISSION # 1044-C \* INSTR # AFI \* 1/16/68 PLOT OF D MIN \* TERRAIN \* PROCESSING \* FULL  
WIDTH # 1/4 \* MEDIAN \* 0.46 \* STD DEV \* 0.18 \* RANGE \* 0.25 TO 1.08 WITH 116 SAMPLES

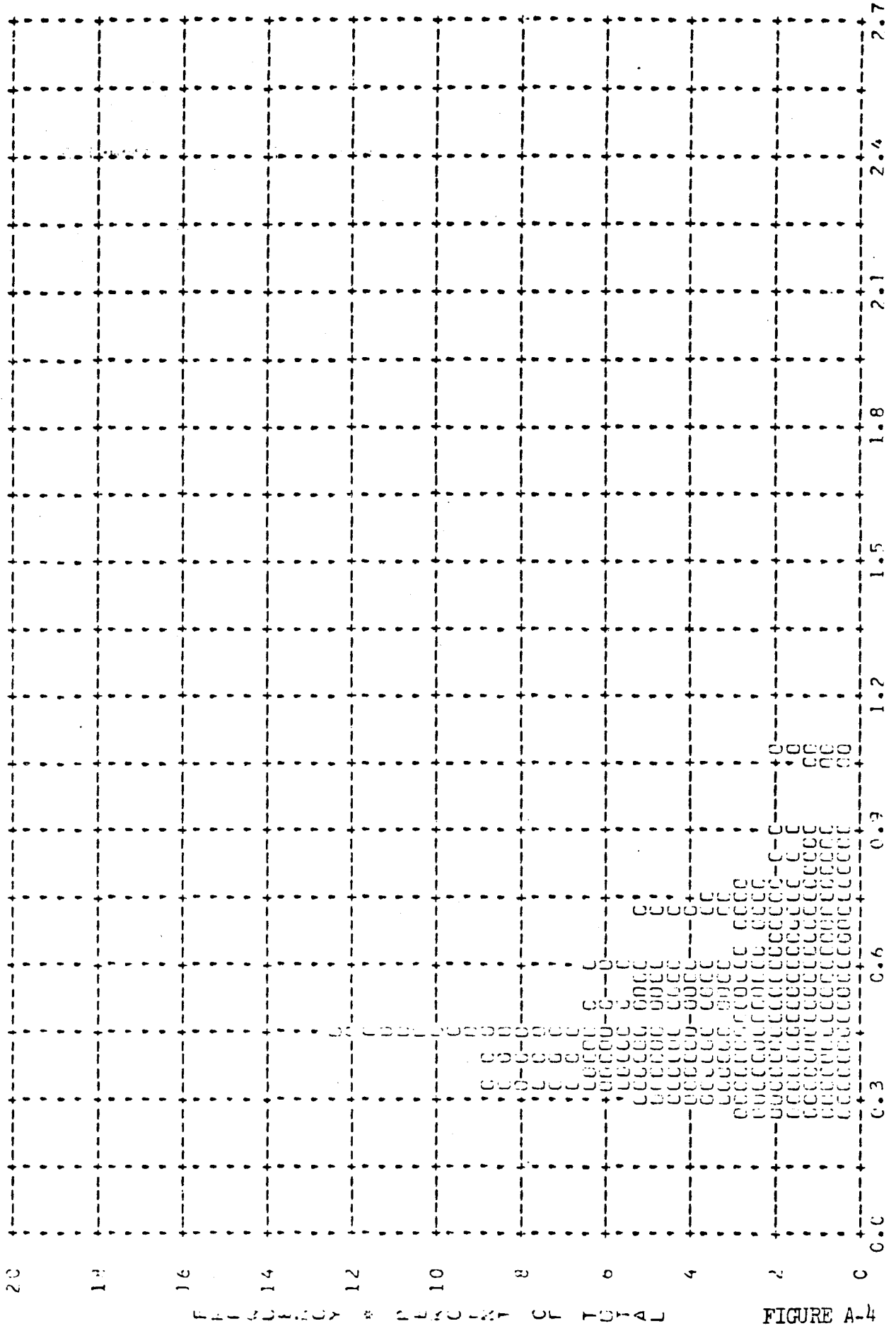


FIGURE A-4

# DENSITY #

SECRET

MISSION \* 1044-2 \* INSTP \* AFT \* 1/16/58 PLCT OF D MAX \* TERRAIN \* PROCESSING \* FULL  
ARITH MEAN \* 1.59 \* MEDIAN \* 1.61 \* STD DEV \* 0.44 \* RANGE \* 0.50 TO 2.48 WITH 116 SAMPLES

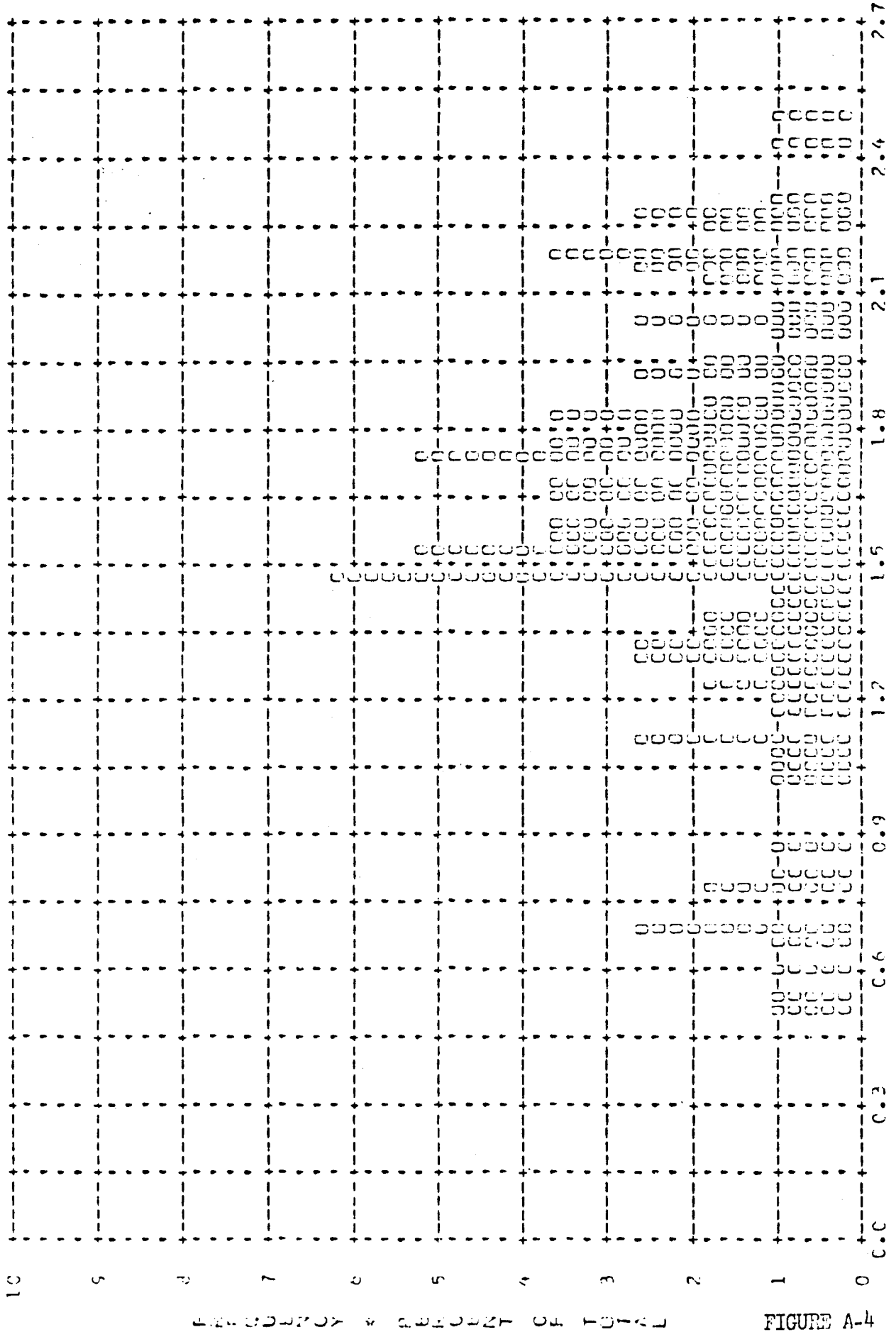


FIGURE A-4

~~TOP SECRET C~~

MISSION # 1044-2 \* INSTR # AFT \* 1/16/68 PLCT CF D MAX # CLOUD \* PROCESSING # FULL  
ARITH MEAN # 2.04 \* MEDIAN # 2.12 \* STD DEV # 0.42 \* RANGE # 0.91 TO 2.55 WITH 93 SAMPLES

	1.0	1.1	1.2	1.3	1.4	1.5	1.6	1.7	1.8	1.9	2.0	2.1	2.2	2.3	2.4	2.5	2.6	2.7
10																		
9																		
8																		
7																		
6																		
5																		
4																		
3																		
2																		
1																		
0																		

FREQUENCY \* PERCENT OF TOTAL

FIGURE A-4

\* DENSITY #

~~TOP SECRET C~~

~~TOP SECRET~~ U

MISSION # 1044-2 \* INSTR \* AFT \* 1/16/69 PLCT OF D MIN \* TERAIN \* PROCESSING \* ALL LEVELS  
ARITH MEAN \* 0.53 \* MEDIAN \* 0.49 \* STD DEV \* 0.21 \* RANGE \* 0.22 TO 1.31 WITH 162 SAMPLES

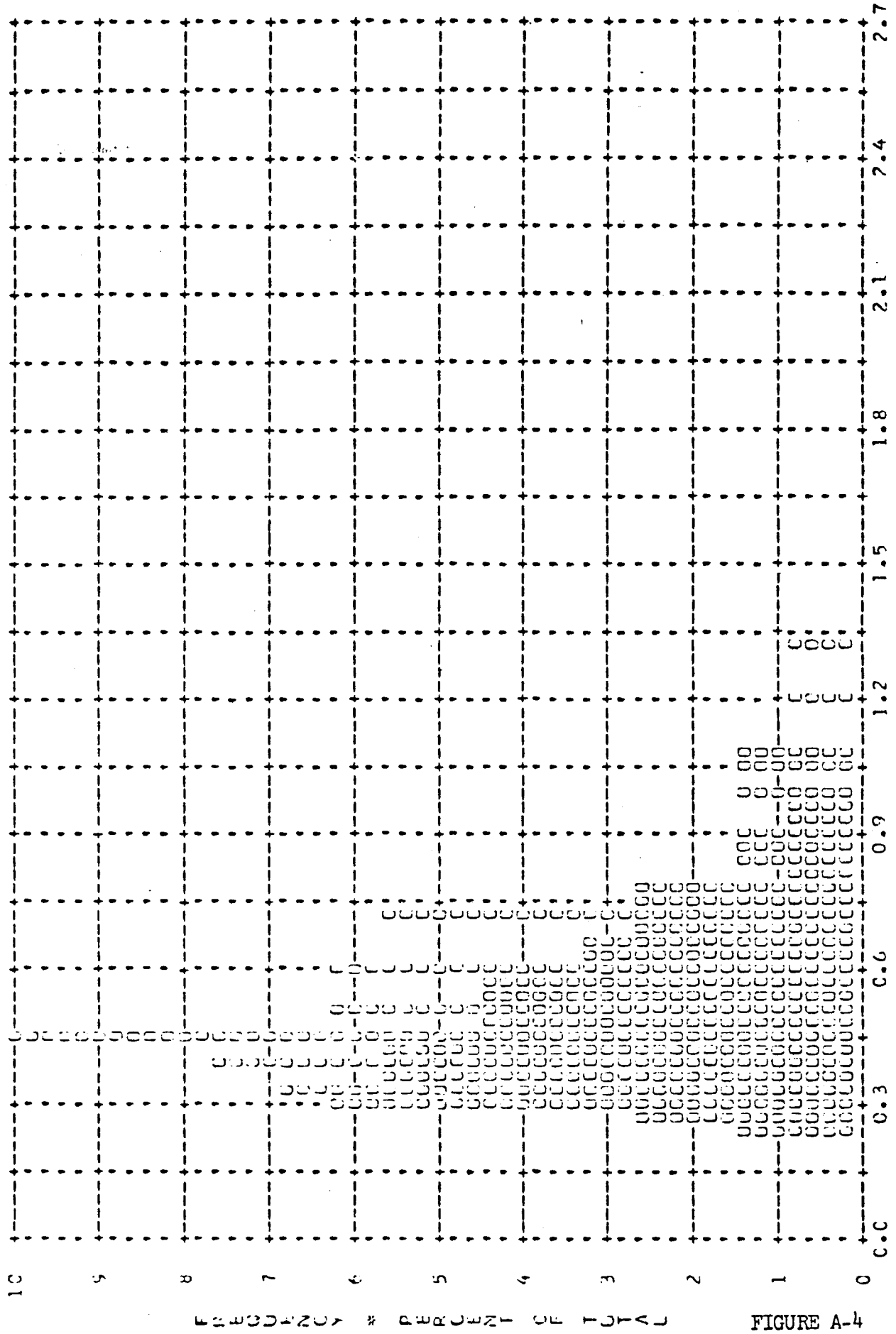


FIGURE A-4

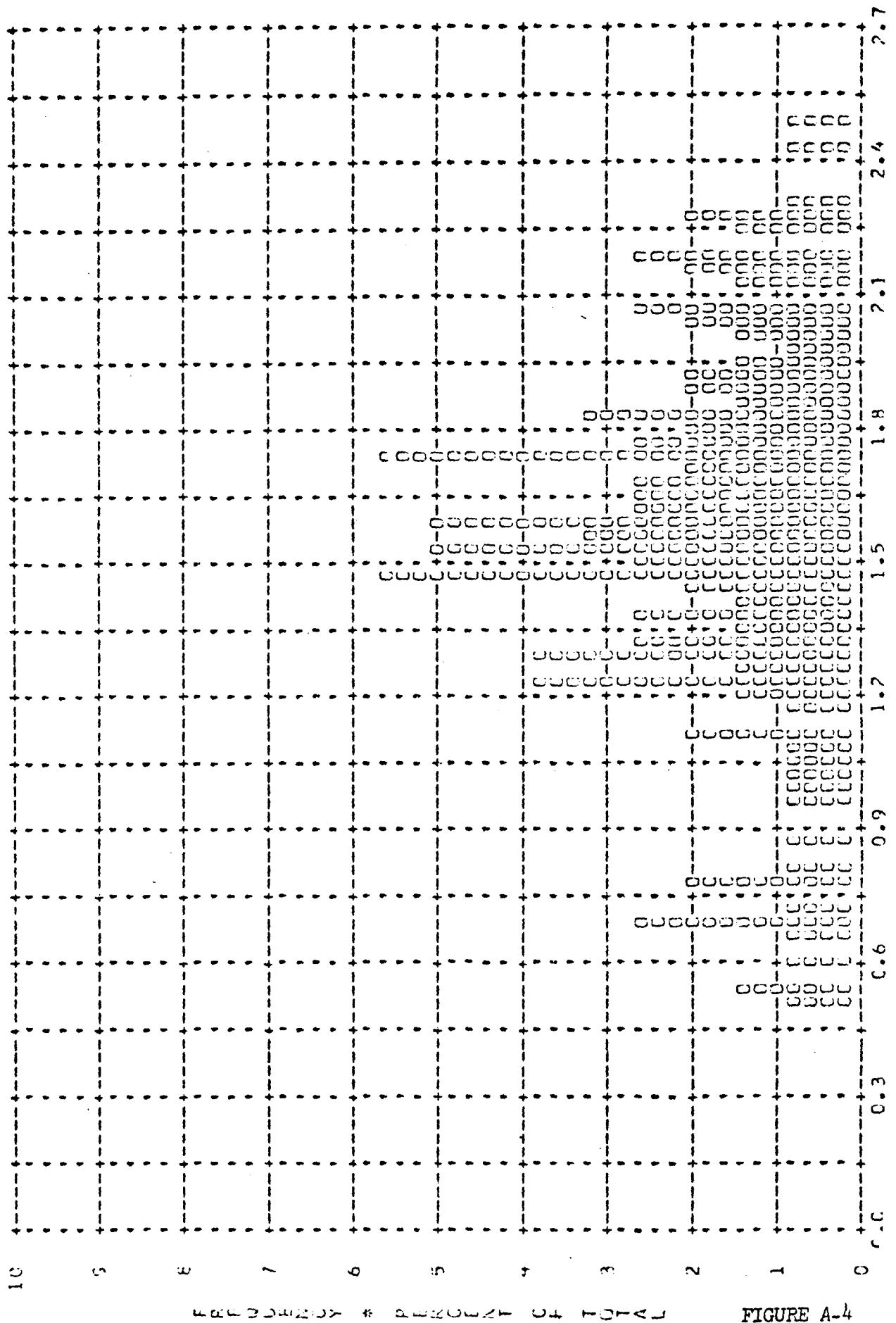
~~TOP SECRET~~ C

\* DENSITY \*



TOP SECRET

MISSION # 1044-2 \* INSTR # AFT \* 1/16/68 PLCT OF D MAY \* TERRAIN \* PROCESSING \* ALL LEVELS  
WITH MEAN \* 1.55 \* MEDIAN \* 1.57 \* STD DEV \* 0.42 \* RANGE \* 0.50 TO 2.48 WITH 162 SAMPLES



\* DENSITY \*

FIGURE A-4

~~XXXXXXXXXX~~ C/

MISSION # 1044-2 \* INSTR # AFT \* 1/16/69 PLCT CF C MAX \* CLUD \* PRUCSSNG \* ALL LEVELS  
 ARITH MEAN \* 2.05 \* MEDIAN \* 2.16 \* STD DEV \* 0.43 \* RANGF \* 0.44 TO 2.55 WITH 128 SAMPLES

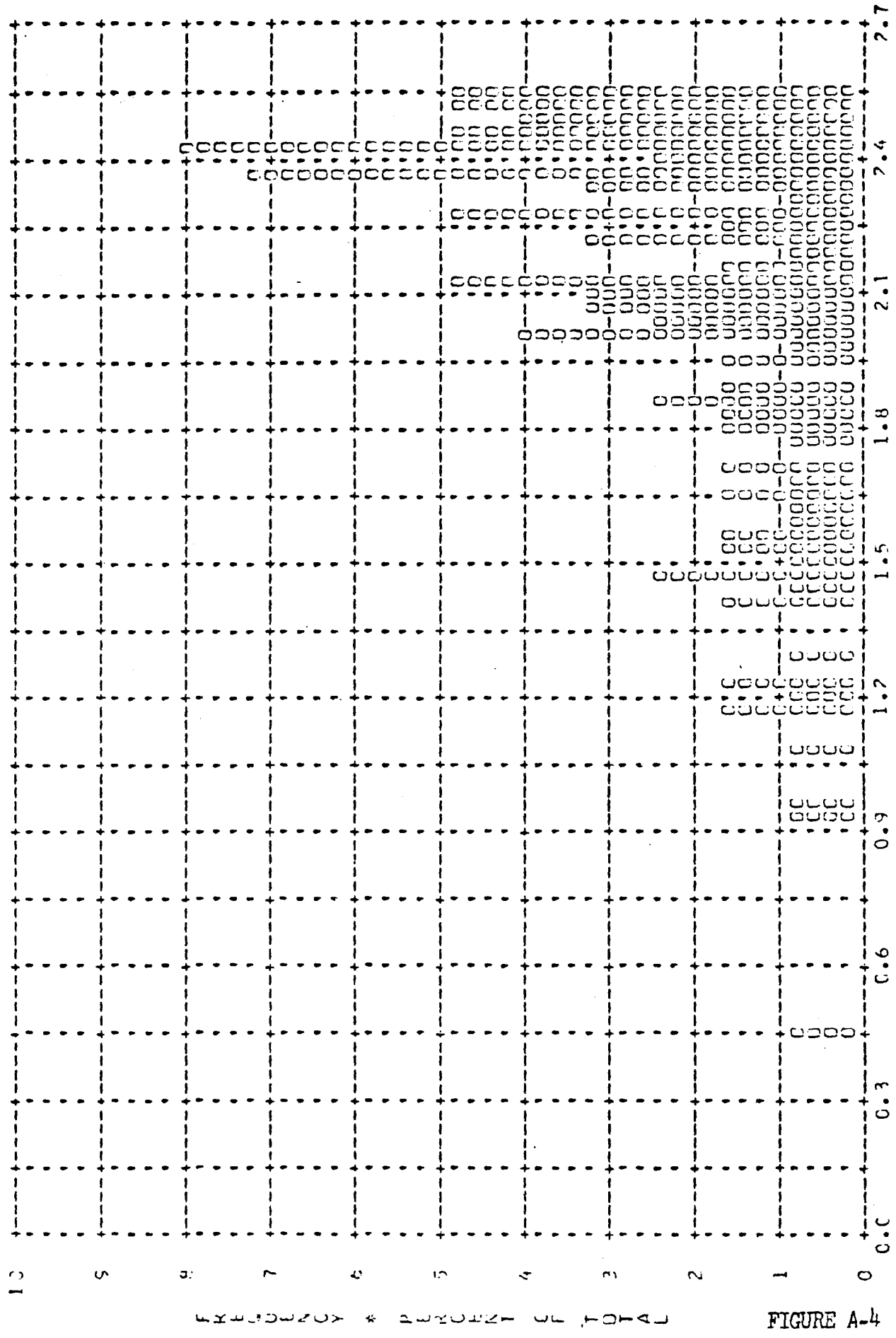
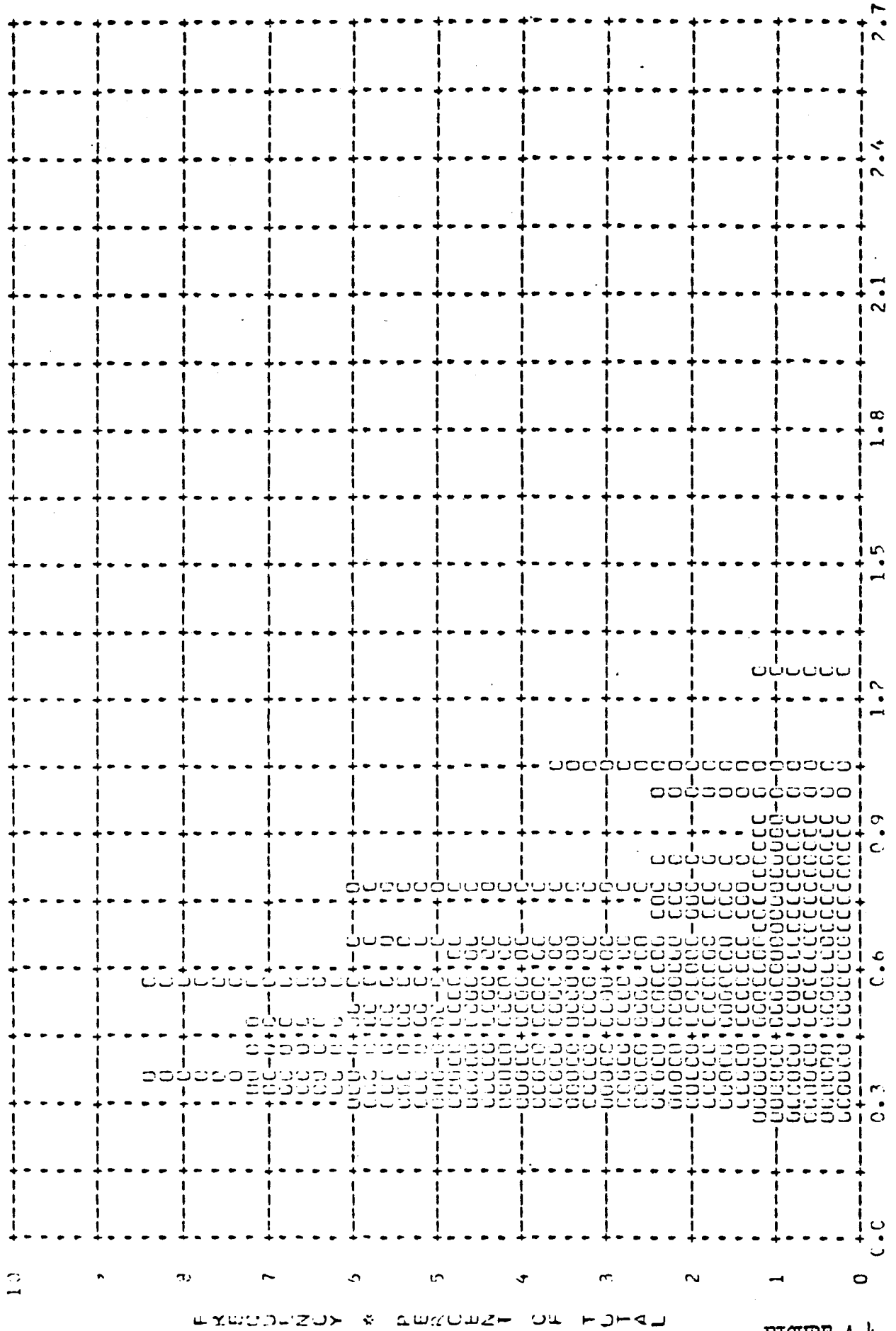


FIGURE A-4

TOP SECRET

MISSION # 1044-2 \* INSTR # AFT # 1/16/68 PLCT CF D MIN \* TERRAIN \* PROCESSING \* DUAL GAMMA  
ARITH MEAN # 0.56 \* MEDIAN # 0.53 \* STD DEV # 0.22 \* RANGE # 0.27 TO 1.25 WITH 84 SAMPLES



\* DENSITY #

TOP SECRET

FIGURE A-4

[REDACTED]

MISSION \* 1044-2 \* INSTR \* AFT \* 1/16/68 PLCT CF D MAX \* TERRAIN \* PROCESSING \* DUAL GAMMA  
ARITH MEAN \* 1.36 \* MEDIAN \* 1.41 \* STD DEV \* 0.29 \* RANGE \* 0.54 TO 1.83 WITH 84 SAMPLES

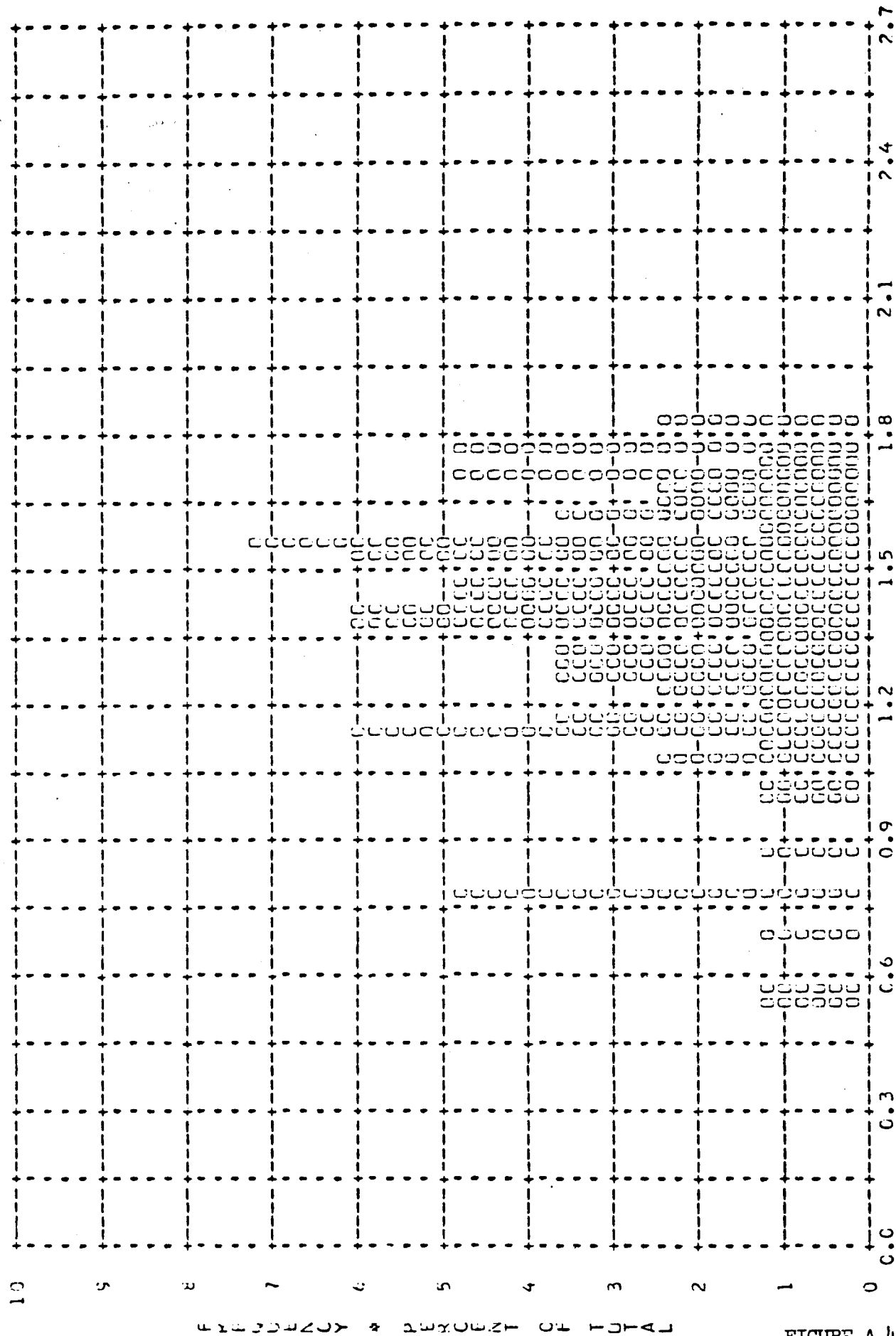


FIGURE A-4

\* DENSITY \*

[REDACTED]

~~TOP SECRET~~ C

~~TOP SECRET~~ C

MISSION \* 1044-2 \* INSTR \* AFT \* 1/16/68 FLCT CF D MAX \* CLOUD \* PROCESSING \* DUAL GAMMA  
ARITH MEAN \* 1.66 \* MEDIAN \* 1.74 \* STD DEV \* 0.22 \* RANGE \* 0.73 TO 1.88 WITH 93 SAMPLES

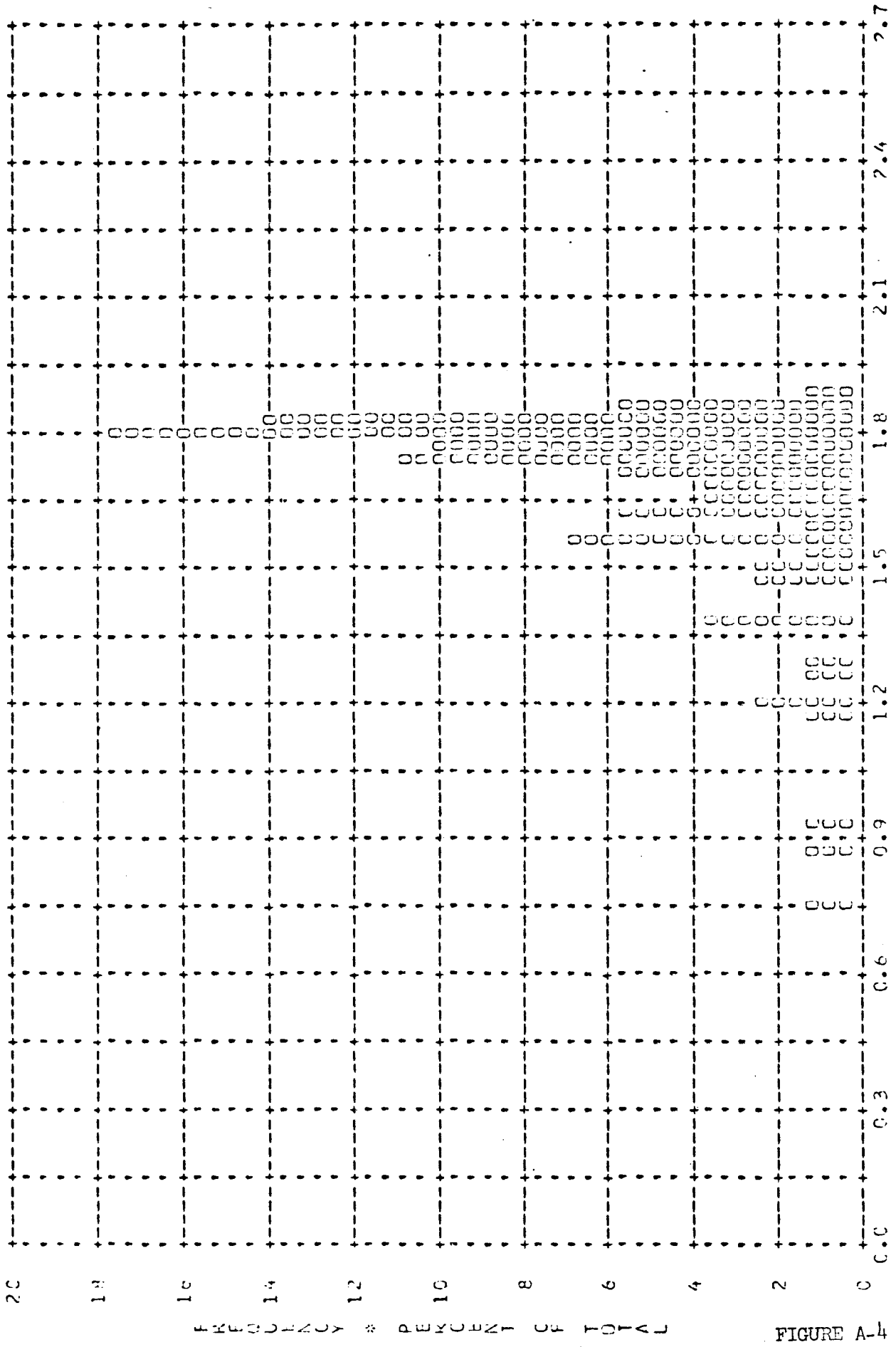


FIGURE A-4

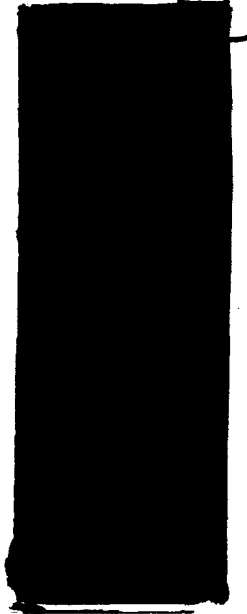
~~TOP SECRET~~ C

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